



**NILE ACADEMY**

**FACULTY OF ENGINEERING**

**CIVIL ENGINEERING DEPARTEMENT**

# **REINFORCED CONCRETE PROJECT**

**( مشروع تصميم منشآت خرسانية ) ( 2023-2024 )**

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**Before we start to write this book, we should express our deepest feelings of thanking and appreciation to our supervisors who helped us throughout this year to complete this project, there are no words that can describe their effort and time to help us in this project; so we thank:**

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**At the end, we would like to thank our biggest supportive team - our parents - for their support through this year**

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## Project Definition

Structural design is an essential component of civil engineering that plays a pivotal role in the construction and maintenance of built environments. It involves the application of scientific principles to ensure that structures are capable of bearing anticipated loadings without failure during their intended lifespan. This discipline is crucial for the safety and functionality of various structures, including buildings, bridges, tunnels, and dams.

The importance of structural design lies in its ability to create frameworks that are not only strong and stable but also cost-effective and sustainable. By carefully selecting materials and designing efficient load paths, structural engineers minimize the risk of structural failure, which can lead to catastrophic consequences such as loss of life, economic disruption, and environmental damage.

Moreover, structural design is integral to achieving architectural vision and aesthetic goals. It enables architects to realize complex shapes and forms while ensuring that these structures are practical and safe. This collaboration between form and function is what allows for innovation in the field of construction.

In addition to these considerations, structural design also addresses environmental concerns. It seeks to reduce the carbon footprint of construction projects by optimizing material usage and promoting the use of renewable resources and energy-efficient designs. This approach not only conserves natural resources but also ensures that structures contribute positively to their surroundings.

Furthermore, in regions prone to natural disasters such as earthquakes, hurricanes, or floods, structural design becomes even more critical. Engineers must account for these additional stresses and design structures that can withstand such extreme conditions. This resilience is key to protecting communities and ensuring that infrastructure remains functional in the aftermath of a disaster.

In summary, structural design is a multifaceted discipline that underpins the safety, sustainability, and success of construction projects. Its importance cannot be overstated, as it directly impacts the well-being of individuals and the collective progress of society.

The field of structural design is a cornerstone of civil engineering, where the primary motivation lies in creating structures that are safe, efficient, and sustainable. The aim of this project is to explore advanced methodologies in structural design that can be applied to modern construction challenges. The contributions of this project include innovative design solutions that enhance structural integrity and cost-effectiveness

### **Economic Feasibility Study and Engineering Standards**

Structural design is deeply intertwined with environmental considerations and economic benefits. An economic feasibility study is crucial to evaluate the viability of design choices, ensuring that projects are not only technically sound but also financially sustainable. Adherence to engineering standards guarantees that structures meet the required safety and performance criteria while being mindful of their environmental impact.

#### **Study Objectives**

The study objectives for concrete structural design are centered around understanding and optimizing the use of concrete, a versatile and widely used construction material. These objectives include:

1. **Material Properties:** To investigate the physical and mechanical properties of concrete, including its strength, durability, and behavior under various environmental conditions.
2. **Design Techniques:** To develop efficient design methodologies that maximize the load-bearing capacity of concrete structures while minimizing material usage and cost.
3. **Sustainability:** To explore sustainable practices in concrete production and usage, aiming to reduce the environmental impact of construction activities
- Innovation:** To integrate new technologies such as high-performance concrete, fiber reinforcement, and smart sensors to enhance the functionality and resilience of concrete structures.
4. **Safety Standards:** To ensure that concrete structural designs comply with international safety standards and can withstand natural disasters like earthquakes and floods.
5. **Economic Analysis:** To perform cost-benefit analyses to determine the most economically feasible solutions for concrete structural design projects. These objectives

guide the continuous improvement and innovation within the field of concrete structural design.

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5. **Safety Standards:** To ensure that concrete structural designs comply with international safety standards and can withstand natural disasters like earthquakes and floods.
6. **Economic Analysis:** To perform cost-benefit analyses to determine the most economically feasible solutions for concrete structural design projects.

These objectives guide the continuous improvement and innovation within the field of concrete structural design.

## **Concrete Challenges: Appeal and Solutions**

Reinforced concrete is a composite material that combines the high compressive strength of concrete with the high tensile strength of steel reinforcement. While it is one of the most popular materials used in construction due to its versatility, durability, and cost-effectiveness, it faces several challenges:

1. **Corrosion:** Steel reinforcement is susceptible to corrosion, especially in aggressive environments, leading to structural weakness.

2. **Cracking:** Concrete can develop cracks over time due to thermal expansion, shrinkage, or excessive loads.
3. **Sustainability:** The production of cement, a key ingredient in concrete, contributes significantly to carbon emissions.

The appeal of reinforced concrete lies in its adaptability to various shapes and forms, its inherent fire resistance, and its ability to be made with locally available materials.

Solutions to these challenges include:

1. **Corrosion-resistant Alloys:** Using stainless steel or other corrosion-resistant alloys for reinforcement.
  2. **Advanced Admixtures:** Incorporating admixtures that improve the durability and reduce the permeability of concrete.
  3. **Green Cement:** Developing alternative cementitious materials that reduce the carbon footprint of concrete production.

By addressing these challenges with innovative solutions, the long-term performance and sustainability of reinforced concrete can be significantly enhanced

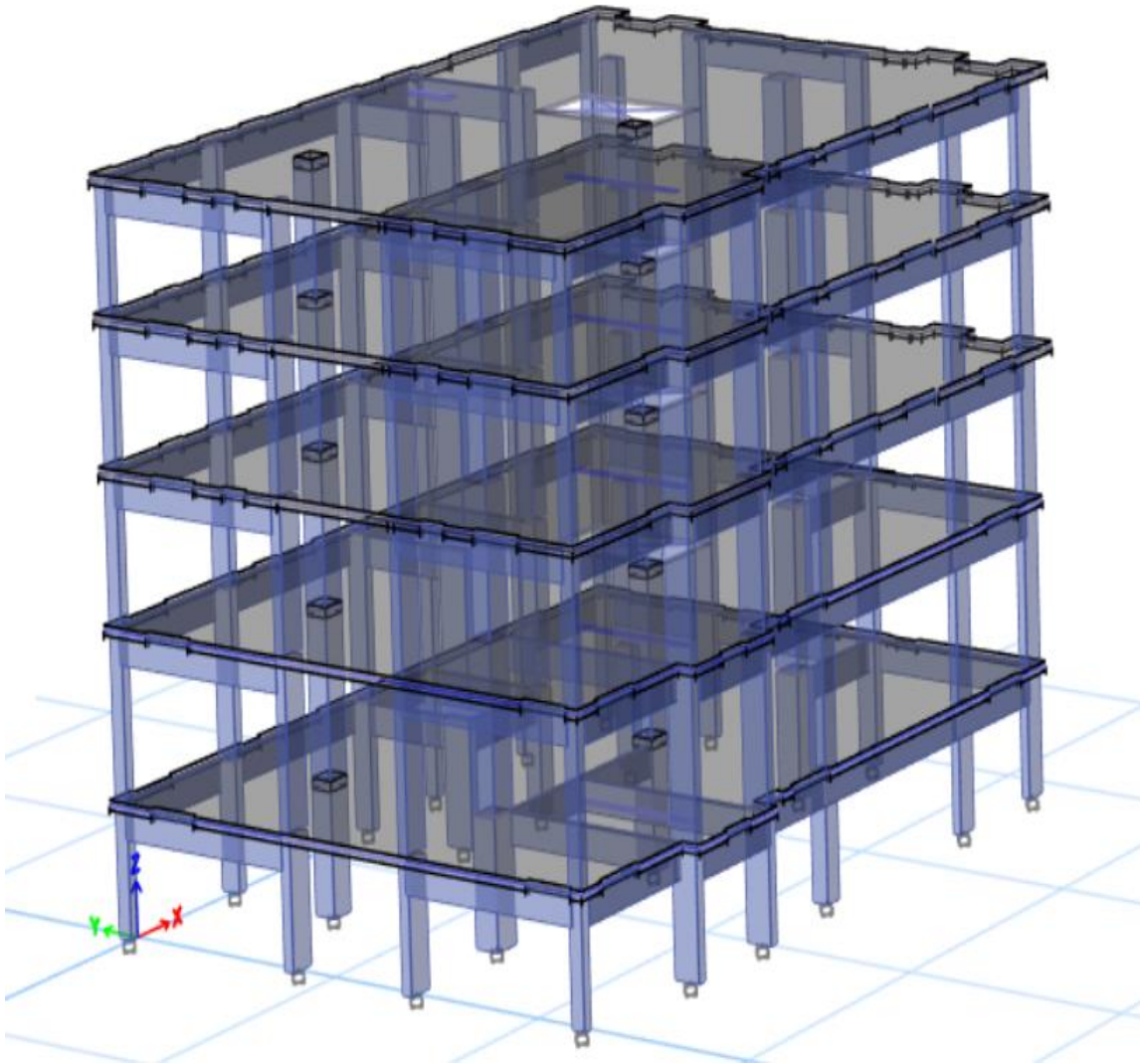
### **The environmental challenges of concrete, its appeal, and how the designer can overcome them**

1. **Efficient Design:** By optimizing the structural layout, engineers can reduce the amount of concrete required, leading to lower material costs.
2. **Durability:** Designing for longevity reduces the need for repairs and replacements, saving money over the structure's lifespan.
3. **Recycled Materials:** Using recycled aggregates in concrete mixtures can lower costs and reduce waste.
4. **Thermal Mass:** Concrete's thermal mass can improve energy efficiency in buildings, cutting heating and cooling expenses.
5. **Green Concrete:** Incorporating sustainable materials like fly ash or slag in concrete reduces the environmental impact and can be more economical than traditional cement.

By focusing on these aspects, concrete structural design not only becomes cost-effective but also environmentally friendly.



## *Unit (1) Villa*



## 1.1 INTRODUCTION

### 1.1.1 Villa Consists of:

- Basement of (2.90) m height
- Ground Floor of (3.80) m height
- First Floor (3.40) m height
- Second Floor (3.30) m height
- Third Floor of (3.00) m height

### 1.1.2 Material Properties Used:

- $F_{cu}=350 \text{ kg/cm}^2$
- $F_{y(\text{main steel})}=3600 \text{ kg/cm}^2$
- $F_{y(\text{stirrups})}=2400 \text{ kg/cm}^2$
- Weight of used brick =  $1400 \text{ kg/m}^3$
- Bearing Capacity of Soil =  $1.0 \text{ kg/m}^2$

### 1.1.3 Cover Thickness

- Slabs Cover = 2 cm
- Beams Cover = 2 cm
- Columns Cover = 2.5 cm
- Foundations Cover = 7 cm
- Stairs Cover = 2 cm
- Ramp Cover = 2 cm
- Semell Cover = 3 cm

### 1.1.4 Loads Used:

- L.L= According to every Floor
- Cover = 0.15 ton
  - رمل تسوية بسمك 5 سم  $0.05 * 1.5$
  - مونة أسمنتية بسمك 1 سم  $0.01 * 2.1$
  - بلاط سيراميك بسمك 2 سم  $0.02 * 2.1$
  - محارة أسفل البلاط بسمك 2 سم  $0.02 * 2.1$
- Wall = According to every Floor
- D.L = Own weight + Covering Material + Wall Load

**1.1.5 Design Method:**

- Ultimate limit state design

**1.1.6 Computer Programs Used in Analysis :**

- (Etabs + Safe + SAP2000 + Excel)

**1.1.7 Design Code:**

**Egyptian code of practice 2020**

## 1.2 DESIGN OF SLABS:

### 1.2.1 Basement Slab: (Flat Slab System)

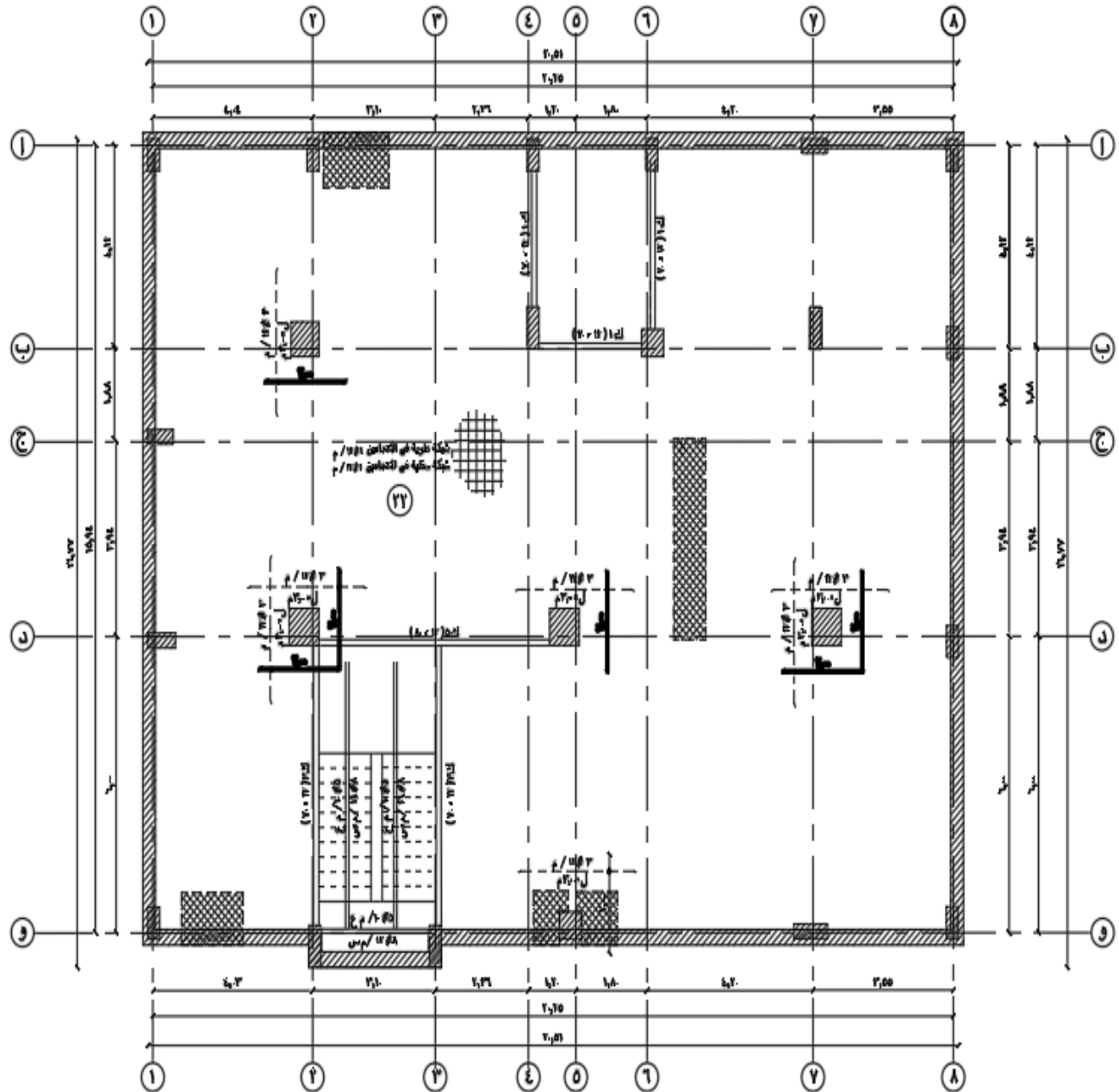


Figure 1.1 Statical System of Basement Roof

- ❖ Slab Thickness = 22 cm
- ❖ Own weight =  $0.22 \times 2.5 = 0.55 \text{ t/m}^2$ .
- ❖ Covering =  $150 \text{ kg/m}^2 = 0.15 \text{ t/m}^2$ .
- ❖ Live load =  $250 \text{ kg/m}^2 = 0.25 \text{ t/m}^2$
- ❖ Wall load =  $250 \text{ kg/m}^2 = 0.25 \text{ t/m}^2$ .

**Solving This flat slab By Using CSI Safe program:**

- $D.L = O.W + W_{wall} + \text{Covering material}$   
 $= 0.55 + 0.25 + 0.15 = 0.95 \text{ t/m}^2$
- $L.L = 250 \text{ kg/cm}^2 = 0.25 \text{ t/m}^2$
- $W_u = 1.4 D.L + 1.6 L.L = 1.4 (0.95) + 1.6 (0.25) = 1.73 \text{ t/m}^2$

**For ultimate design:-**

- $A_s = A_s = \left[ \frac{M_u}{F_y * j * d} \right]$
- 
- $M_u = A_s * F_y * j * d = 6 * \left( \frac{\pi * (1.2)^2}{4} \right) * 3600 * 0.826 * 18 * (10)^{-5}$
- $M(r) = 3.63 \text{ t.m} \Rightarrow \text{Use } 6 \text{ } \phi 12 / \text{m in each Direction}$
- Additional RFT (3  $\phi 12 / \text{m}$ ) & 6  $\phi 12 / \text{m}$  upper and lower

**In X-Direction: (Upper and Lower)**

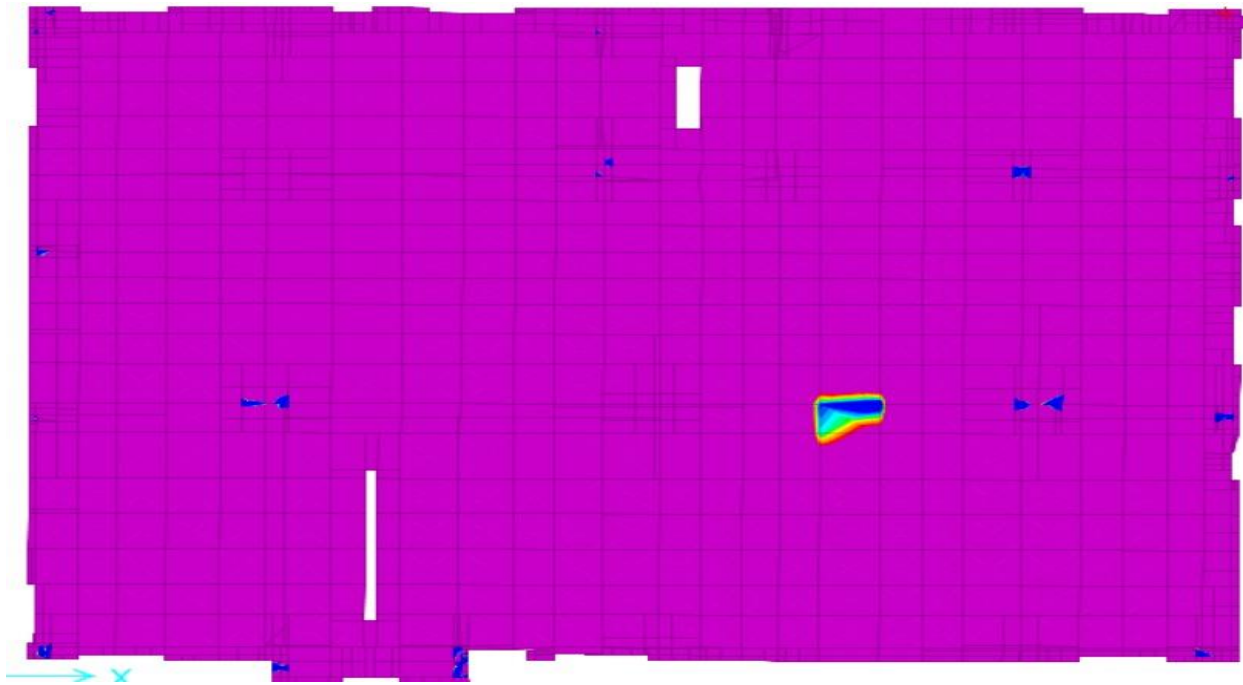


Figure 1.2 Additional Reinforcement in X-Direction (Upper and Lower)

**In Y-Direction (Upper and Lower):**

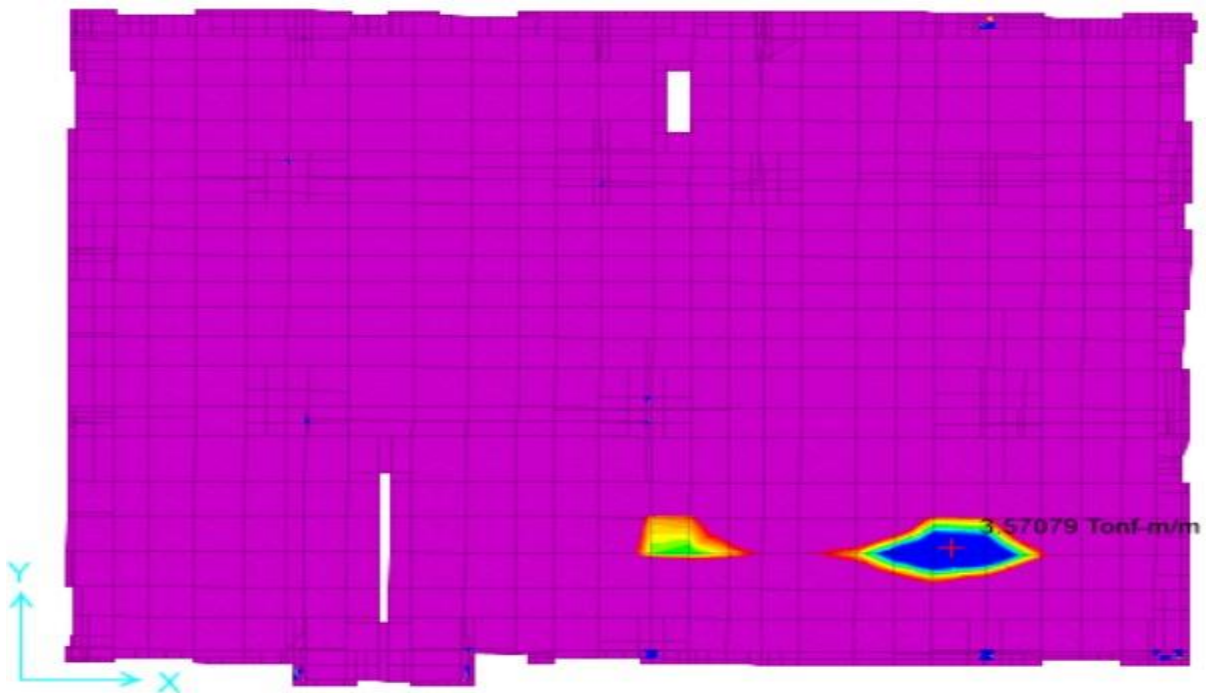


Figure 1.3 Additional Reinforcement in Y-Direction (Upper and Lower)

### 1.2.1.1 Check for Long Term Deflection:

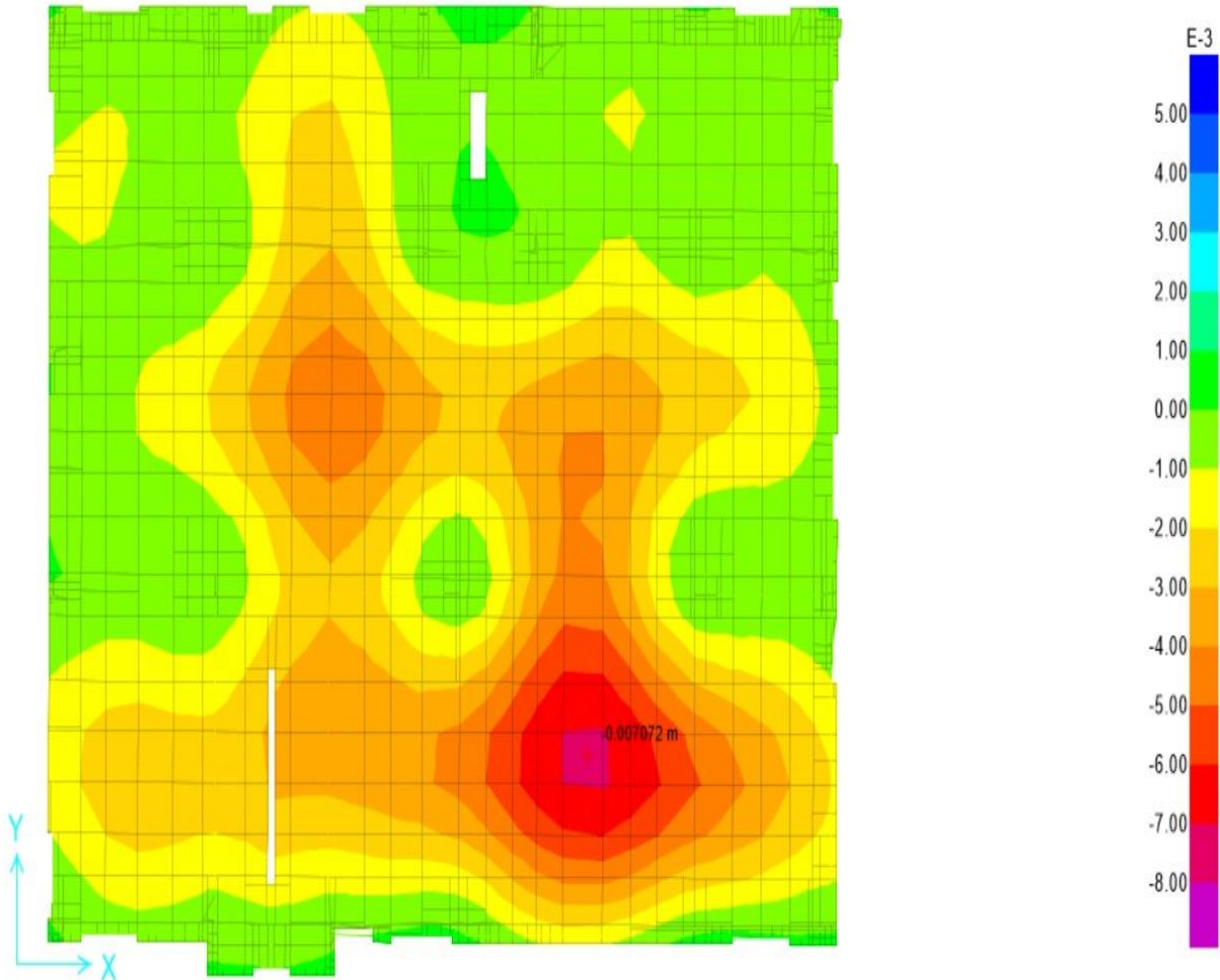


Figure 1.4 Long Term Deflection

- From Code Check =  $L/250$
- Span for Check = 6.00 m
- Allowable Deflection =  $6/250 = 0.024$  m
- Maximum Deflection = 0.007072 m



## 1.2.2 Ground Slab (Flat Slab System)

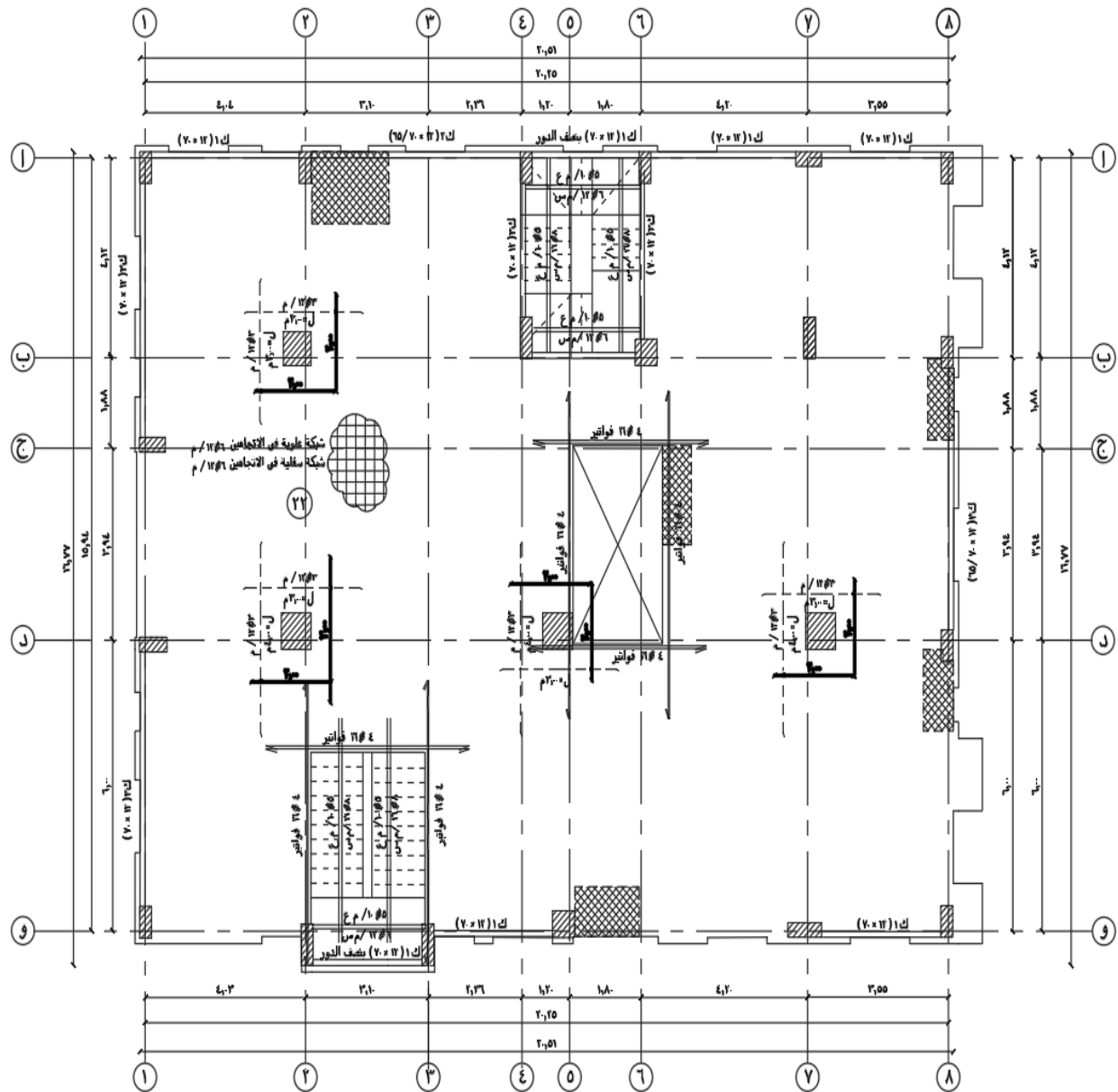


Figure 1.5 Static System of Ground Roof

- ❖ Slab Thickness = 22 cm
- ❖ Own weight =  $0.22 \times 2.5 = 0.55 \text{ t/m}^2$ .
- ❖ Covering =  $150 \text{ kg/m}^2 = 0.150 \text{ t/m}^2$ .
- ❖ Live load =  $250 \text{ kg/m}^2 = 0.250 \text{ t/m}^2$
- ❖ Wall load =  $250 \text{ kg/m}^2 = 0.250 \text{ t/m}^2$ .



### Solving This flat slab By Using CSI Safe program:

- $D.L = O.W + W_{wall} + \text{Covering material}$   
 $= 0.55 + 0.25 + 0.15 = 0.95 \text{ t/m}^2$
- $L.L = 250 \text{ kg/cm}^2 = 0.25 \text{ t/m}^2$
- $W_u = 1.4 D.L + 1.6 L.L = 1.4 (0.95) + 1.6 (0.25) = 1.73 \text{ t/m}^2$

### For ultimate design:-

- $A_s = \left[ \frac{M_u}{F_y * j * d} \right]$
- $M_u = A_s * F_y * j * d = 6 * \left( \frac{\pi * (1.2)^2}{4} \right) * 3600 * 0.826 * 18 * (10)^{-5}$
- $M(r) = 3.63 \text{ t.m} \Rightarrow \text{Use } 6 \text{ } \phi 12 / \text{m in each Direction}$
- Additional RFT (3  $\phi 12 / \text{m}$ ) & ( 6  $\phi 12 / \text{m}$ ) upper and lower

**In X-Direction: (Upper and Lower)**

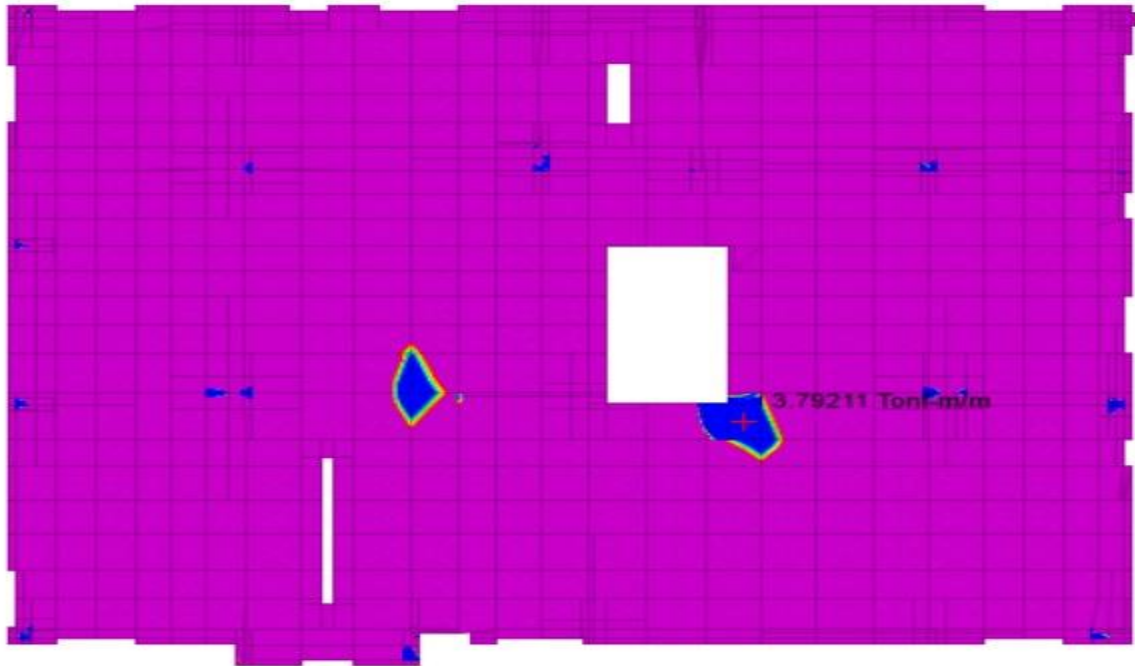


Figure 1.6 Additional Reinforcement in X-Direction (Upper and Lower)

**In Y-Direction: (Upper and Lower)**

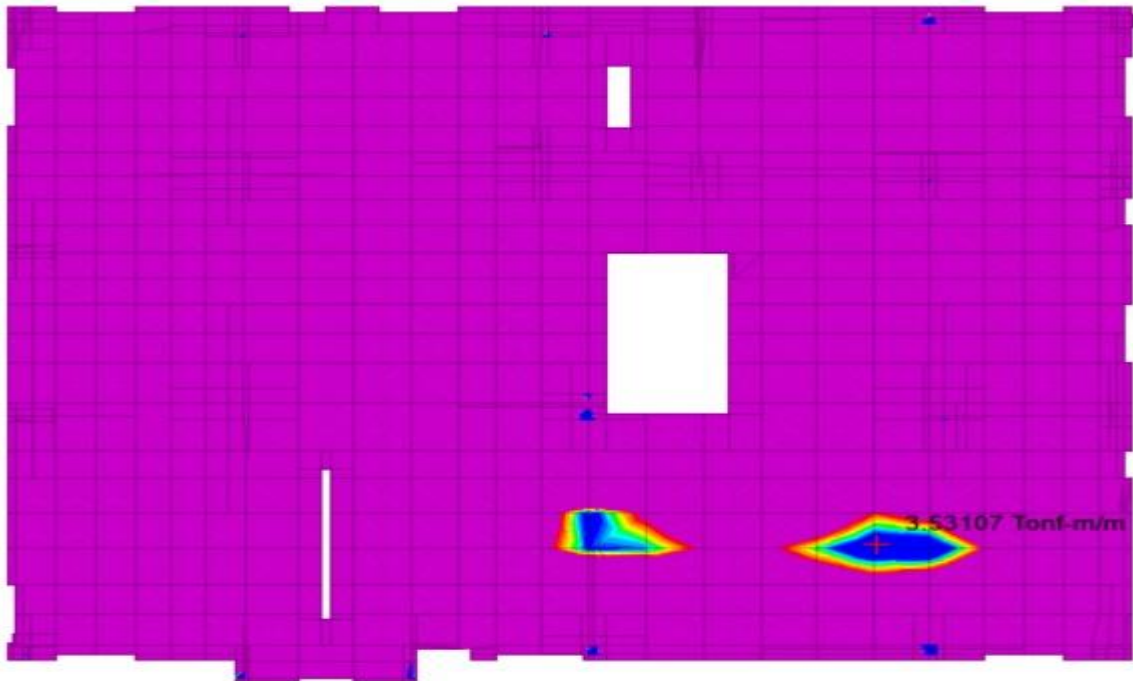


Figure 1.7 Additional Reinforcement in Y-Direction (Upper and Lower)

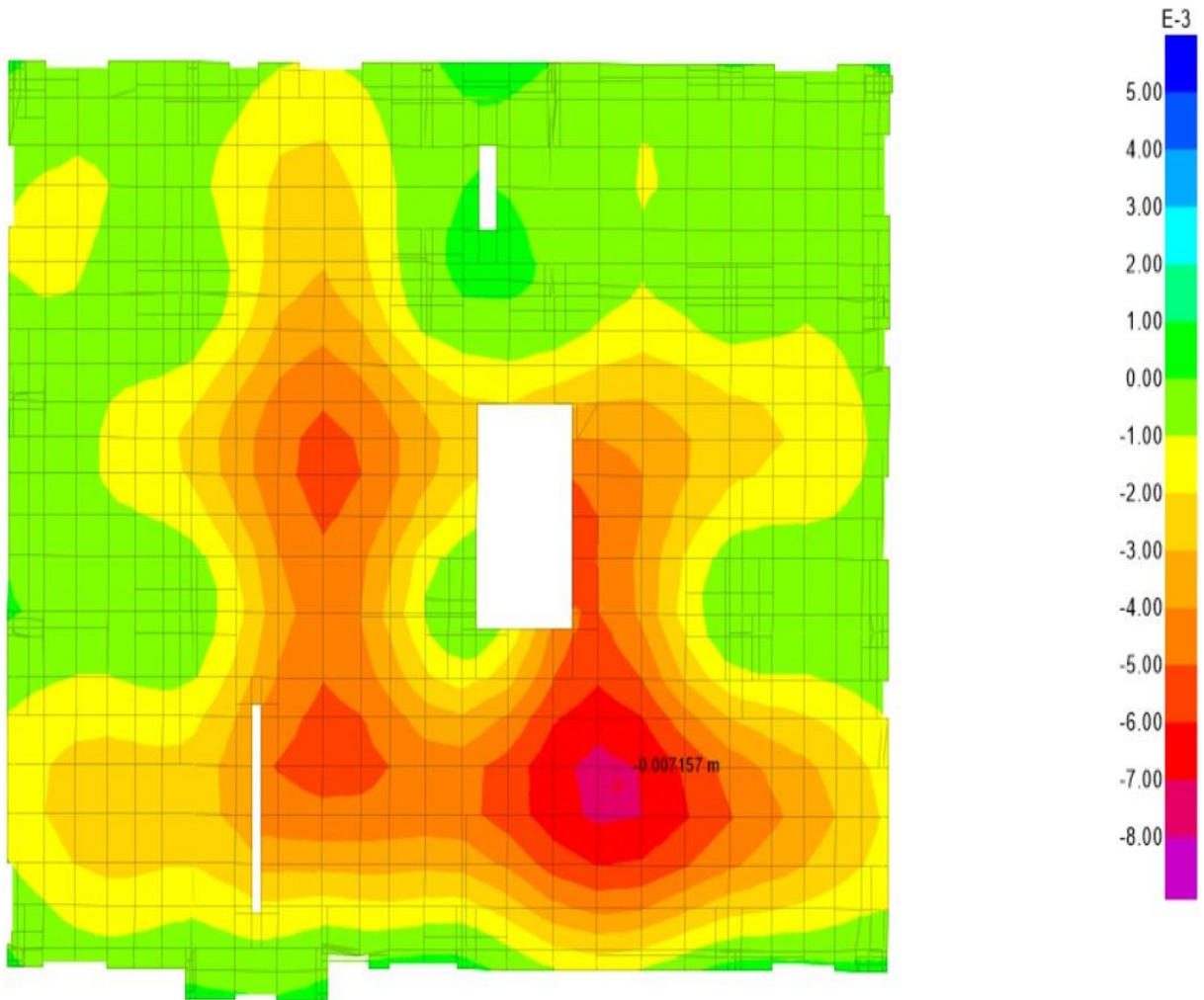
**1.2.2.1 Check for Long Term Deflection:**

Figure 1.8 Long Term Deflection

- From Code Check =  $L/250$
- Span for Check = 6.00 m
- Allowable Deflection =  $6/250 = 0.024$  m
- Maximum Deflection = 0.007157 m

### 1.2.3 First Slab (Flat Slab System)

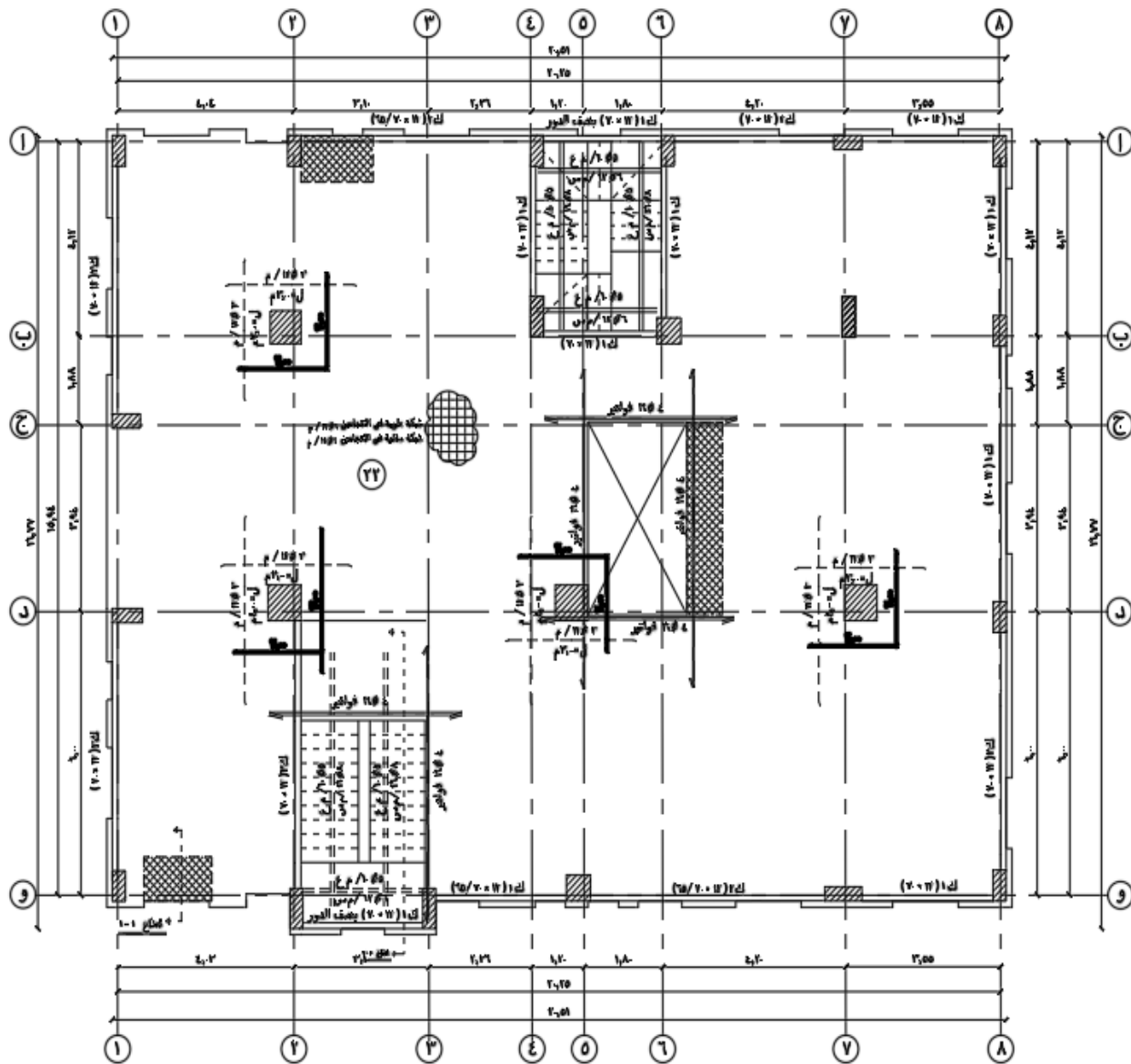


Figure 1.9 Static System of First Roof

- ❖ Slab Thickness = 22 cm
- ❖ Own weight =  $0.22 \times 2.5 = 0.55 \text{ t/m}^2$ .
- ❖ Covering =  $150 \text{ kg/m}^2 = 0.15 \text{ t/m}^2$ .
- ❖ Live load =  $250 \text{ kg/m}^2 = 0.25 \text{ t/m}^2$ .
- ❖ Wall load =  $250 \text{ kg/m}^2 = 0.25 \text{ t/m}^2$ .

**Solving This flat slab By Using CSI Safe program:**

- $D.L = O.W + W_{wall} + \text{Covering material}$   
 $= 0.55 + 0.25 + 0.15 = 0.95 \text{ t/m}^2$
- $L.L = 250 \text{ kg/cm}^2 = 0.25 \text{ t/m}^2$
- $W_u = 1.4 D.L + 1.6 L.L = 1.4 (0.95) + 1.6 (0.25) = 1.73 \text{ t/m}^2$

**For ultimate design:-**

- $As = \left[ \frac{Mu}{F_y * J * d} \right]$
- $M_u = As * F_y * J * d = 6 * \left( \frac{\pi * (1.2)^2}{4} \right) * 3600 * 0.826 * 18 * (10)^{-5}$
- $M(r) = 3.63 \text{ t.m} \Rightarrow \text{Use } 6 \text{ } \phi 12 / \text{m in each Direction}$
- Additional RFT (3  $\phi 12 / \text{m}$ ) & ( 6  $\phi 12 / \text{m}$ ) upper and lower

**In X-Direction: (upper and Lower)**

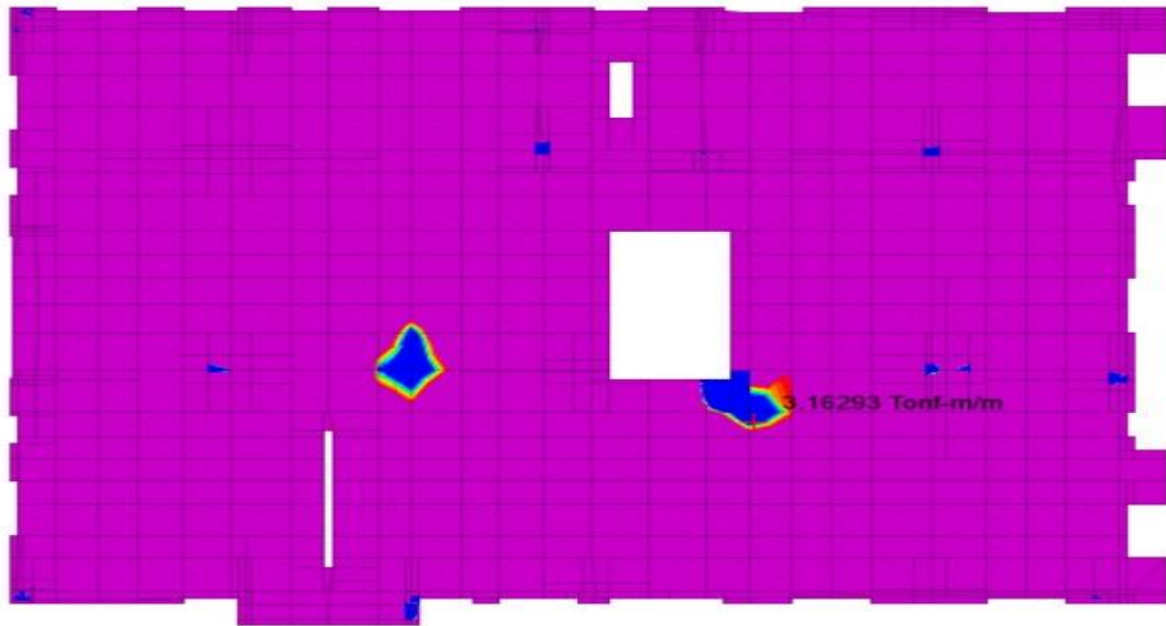


Figure 1.10 Additional Reinforcement in X-Direction (upper and Lower)

**In Y-Direction: (upper and Lower)**

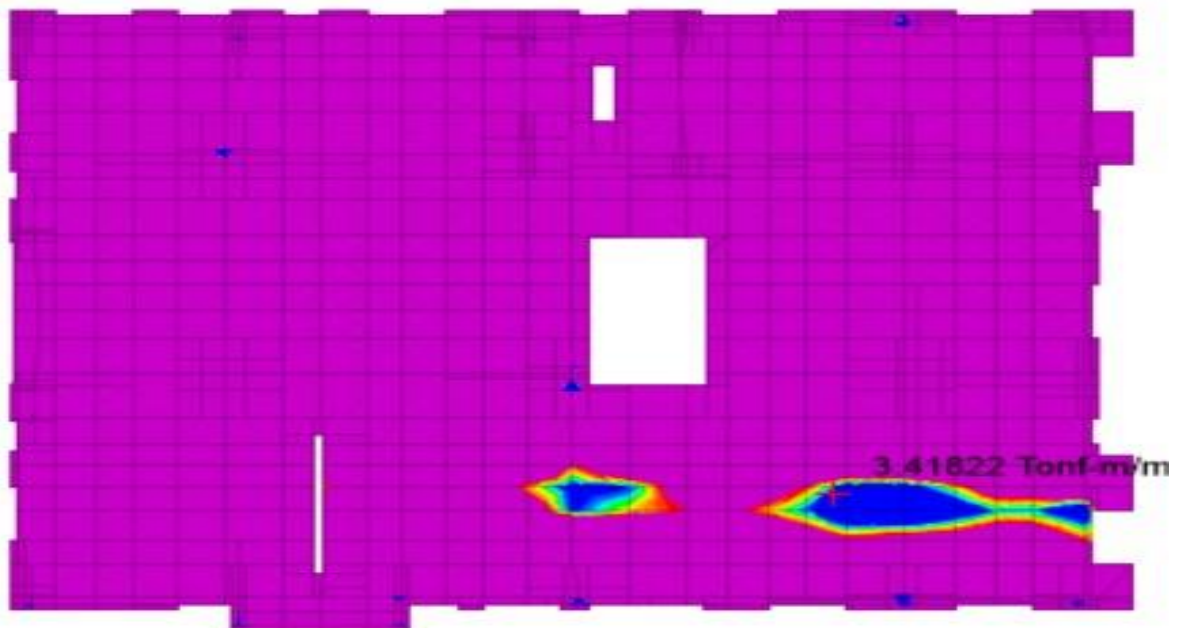


Figure 1.11 Additional Reinforcement in Y-Direction (upper and Lower)



### 1.2.3.1 Check for Long Term Deflection:

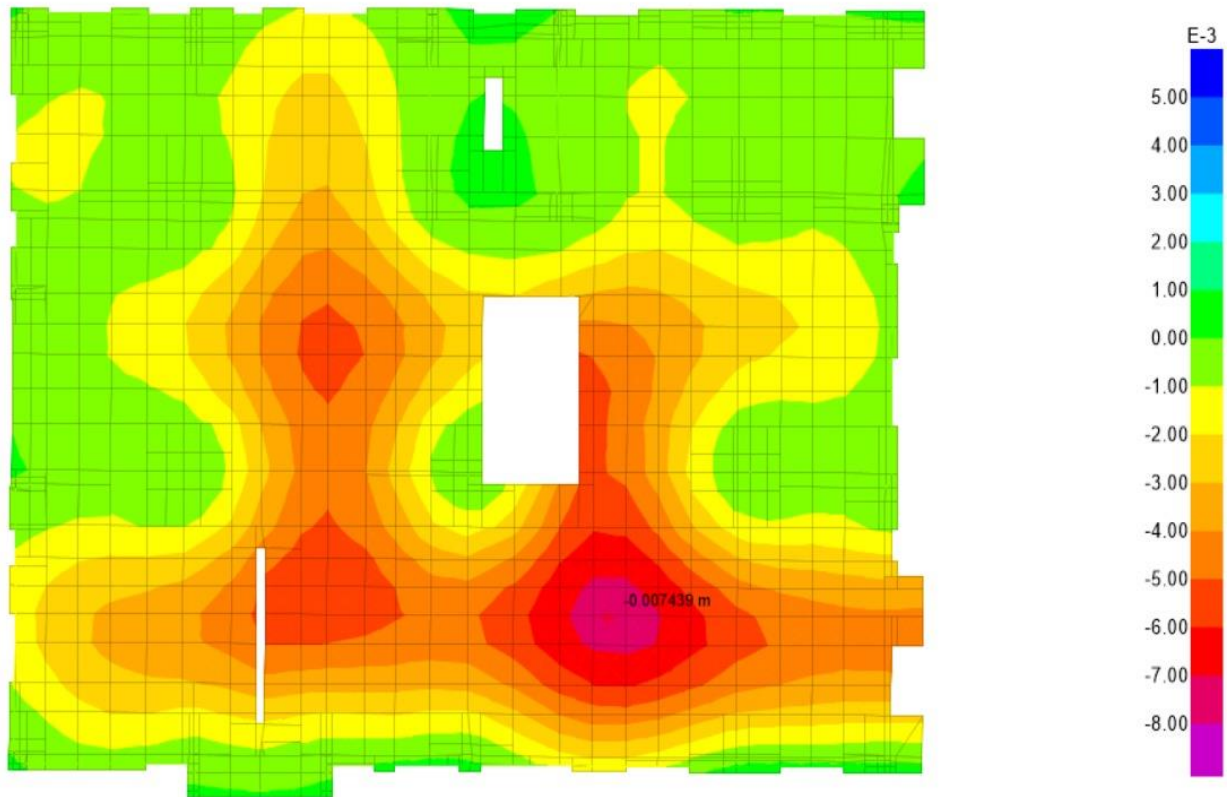


Figure 1.12 Long Term Deflection

- From Code Check =  $L/250$
- Span for Check = 6.00 m
- Allowable Deflection =  $6 / 250 = 0.024$  m
- Maximum Deflection = 0.007439 m

## 1.2.4 Check of Punching Shear: (Basement Roof)

### 1.2.4.1 Corner Column (C<sub>1</sub>=30\*65) on (1 -i) Axis

- ❖ Slab Thickness = 22 cm
- ❖ Own weight =  $0.22 * 2.5 = 0.55 \text{ t/m}^2$ .
- ❖ F.Covering =  $150 \text{ kg/m}^2 = 0.15 \text{ t/m}^2$ .
- ❖ Live load =  $250 \text{ kg/m}^2 = 0.25 \text{ t/m}^2$
- ❖ Wall load =  $250 \text{ kg/m}^2 = 0.25 \text{ t/m}^2$ .

- D.L = O.W + W<sub>wall</sub> + Covering material

- =  $0.55 + 0.25 + 0.15 = 0.95 \text{ t/m}^2$

- L.L =  $250 \text{ kg/cm}^2 = 0.25 \text{ t/m}^2$

- $W_u = 1.4 D.L + 1.6 L.L = 1.4(0.95) + 1.6(0.25) = 1.73 \text{ t/m}^2 = 17.3 \text{ KN/m}^2$

- $d = t_s - 20 \text{ mm} = 220 - 20 = 200 \text{ mm} = 0.200 \text{ m}$

- $b_o = 400 + 750 = 1150 \text{ mm}$

- $Q_{up} = W_u (L_1 * L_2 - A_p) = 17.3 (2.06 * 2.02 - 0.40 * 0.75) = 66.79 \text{ KN/m}^2$

- $q_{up} = \frac{Q_{up}}{b_o * d} * \beta = \frac{66.79 * 1000}{1150 * 200} * 1.5 = 0.33 \text{ N/mm}^2$

- $q_{cup} = \text{the least of:-}$

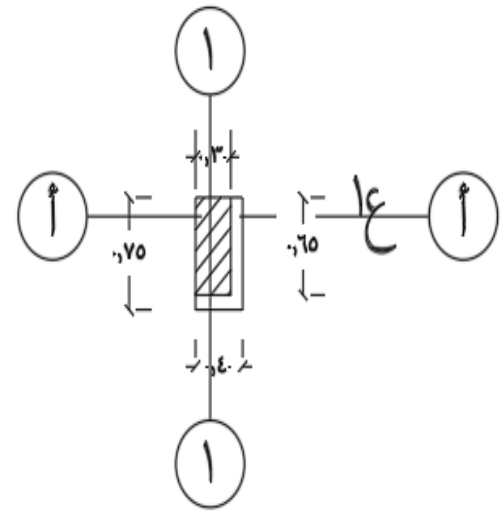
- $1.7 \text{ N/mm}^2$

- $0.316 \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.316 \sqrt{\frac{35}{1.5}} = 1.53 \text{ N/mm}^2$

- $0.316 \left( \frac{a}{b} + 0.5 \right) \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.316 \left( \frac{300}{650} + 0.5 \right) \sqrt{\frac{35}{1.5}} = 1.46 \text{ N/mm}^2$

- $0.8 \left( \frac{\alpha * d}{b_o} + 0.5 \right) \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.8 \left( \frac{2 * 200}{1150} + 0.2 \right) \sqrt{\frac{35}{1.5}} = 2.11 \text{ N/mm}^2$

- $q_{up} = 0.33 \text{ N/mm}^2 \leq q_{cup} = 1.46 \text{ N/mm}^2$



OK , safe punching



### 1.2.4.2 Edge Column (C2 = 30\*85) on (7-9) Axis:

- ❖ Slab Thickness = 22 cm
- ❖ Own weight =  $0.22 * 2.5 = 0.55 \text{ t/m}^2$ .
- ❖ F.Covering =  $150 \text{ kg/m}^2 = 0.15 \text{ t/m}^2$ .
- ❖ Live load =  $250 \text{ kg/m}^2 = 0.25 \text{ t/m}^2$
- ❖ Wall load =  $250 \text{ kg/m}^2 = 0.25 \text{ t/m}^2$ .

$$\begin{aligned} \text{D.L} &= \text{O.W} + \text{W}_{\text{wall}} + \text{Covering material} \\ &= 0.55 + 0.25 + 0.15 = 0.95 \text{ t/m}^2 \end{aligned}$$

$$\text{L.L} = 250 \text{ kg/cm}^2 = 0.25 \text{ t/m}^2$$

$$\begin{aligned} W_u &= 1.4 \text{ D.L} + 1.6 \text{ L.L} = 1.4(0.95) + 1.6(0.25) = \\ &1.73 \text{ t/m}^2 = 17.3 \text{ Kn/m}^2 \end{aligned}$$

$$d = t_s - 20 \text{ mm} = 220 - 20 = 200 \text{ mm} = 0.200 \text{ m}$$

$$b_o = 1005 + 400 + 400 = 1805 \text{ mm}$$

$$Q_{up} = W_u (L_1 * L_2 - A_p) = 17.3 (3.9 * 3 - 1.05 * 0.400) = 195.144 \text{ KN/m}^2$$

$$q_{up} = \frac{Q_{up}}{b_o * d} * \beta = \frac{195.144 * 1000}{1805 * 200} * 1.3 = 0.70 \text{ N/mm}^2$$

- $q_{cup} = \text{the least of:-}$

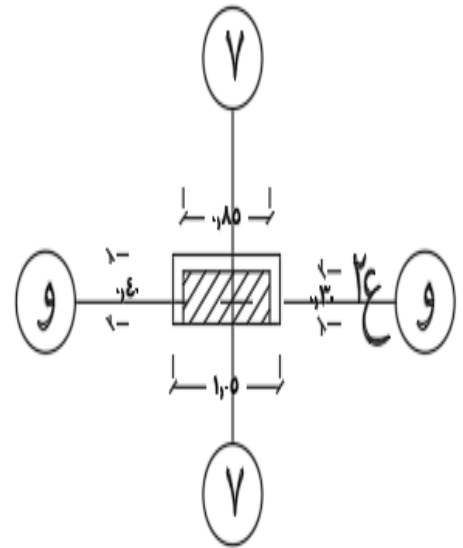
$$\circ 1.70 \text{ N/mm}^2$$

$$\circ 0.316 \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.316 \sqrt{\frac{35}{1.5}} = 1.53 \text{ N/mm}^2$$

$$\circ 0.316 \left( \frac{a}{b} + 0.5 \right) \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.316 \left( \frac{300}{850} + 0.5 \right) \sqrt{\frac{25}{1.5}} = 1.30 \text{ N/mm}^2$$

$$\circ 0.8 \left( \frac{\alpha * d}{b_o} + 0.5 \right) \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.8 \left( \frac{3 * 200}{1805} + 0.2 \right) \sqrt{\frac{35}{1.5}} = 2.06 \text{ N/mm}^2$$

$$\bullet q_{up} = 0.70 \text{ N/mm}^2 \leq q_{cup} = 1.30 \text{ N/mm}^2$$



OK , safe punching

## 1.2.4.3 Interior Column (C3 = 75\*75) on (5 - 3) Axis:

- ❖ Slab Thickness = 22 cm
- ❖ Own weight =  $0.22 * 2.5 = 0.55 \text{ t/m}^2$ .
- ❖ Covering =  $150 \text{ kg/m}^2 = 0.15 \text{ t/m}^2$ .
- ❖ Live load =  $250 \text{ kg/m}^2 = 0.25 \text{ t/m}^2$
- ❖ Wall load =  $250 \text{ kg/m}^2 = 0.25 \text{ t/m}^2$ .

$$- \text{ D.L} = \text{O.W} + \text{W}_{\text{wall}} + \text{F.Covering material}$$

$$= 0.55 + 0.25 + 0.15 = 0.95 \text{ t/m}^2$$

$$- \text{ L.L} = 250 \text{ kg/cm}^2 = 0.25 \text{ t/m}^2$$

$$\bullet \text{ } W_u = 1.4 \text{ D.L} + 1.6 \text{ L.L} = 1.4(0.95) + 1.6(0.25) = 1.73 \text{ t/m}^2 = 17.3 \text{ Kn/m}^2$$

$$\bullet \text{ } dp = t_s - 20 \text{ mm} = 220 - 20 = 200 \text{ mm} = 0.2 \text{ m}$$

$$\bullet \text{ } b_o = 2(a + b) = 2 * (750 + 750) = 3000 \text{ mm}$$

$$\bullet \text{ } Q_{up} = W_u (L_1 * L_2 - A_p) = 17.3 (6.3 * 5.91 - 0.75 * 0.75) = 634.4 \text{ Kn/m}^2$$

$$\bullet \text{ } q_{up} = \frac{Q_{up}}{b_o * d} * \beta = \frac{634.4 * 1000}{3000 * 200} * 1.15 = 1.21 \text{ N/mm}^2$$

$$\bullet \text{ } q_{cup} = \text{the least of:-}$$

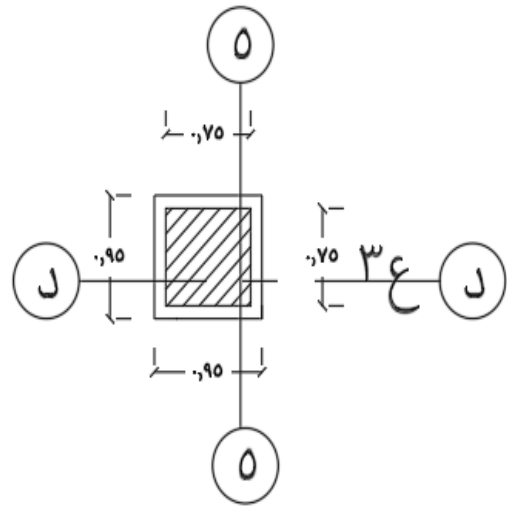
$$\circ \text{ } 1.70 \text{ N/mm}^2$$

$$\circ \text{ } 0.316 \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.316 \sqrt{\frac{35}{1.5}} = 1.53 \text{ N/mm}^2$$

$$\circ \text{ } 0.316 \left( \frac{a}{b} + 0.5 \right) \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.316 \left( \frac{75}{75} + 0.5 \right) \sqrt{\frac{35}{1.5}} = 2.29 \text{ N/mm}^2$$

$$\circ \text{ } 0.8 \left( \frac{\alpha * d}{b_o} + 0.5 \right) \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.8 \left( \frac{4 * 180}{3000} + 0.2 \right) \sqrt{\frac{35}{1.5}} = 1.80 \text{ N/mm}^2$$

$$\bullet \text{ } q_{up} = 1.21 \text{ N/mm}^2 \leq q_{cup} = 1.53 \text{ N/mm}^2$$



OK safe punching

### 1.3 Design of Stairs ( TWO Flight Stair Axis ' د - و ' )

#### 1.3.1 Manual solution

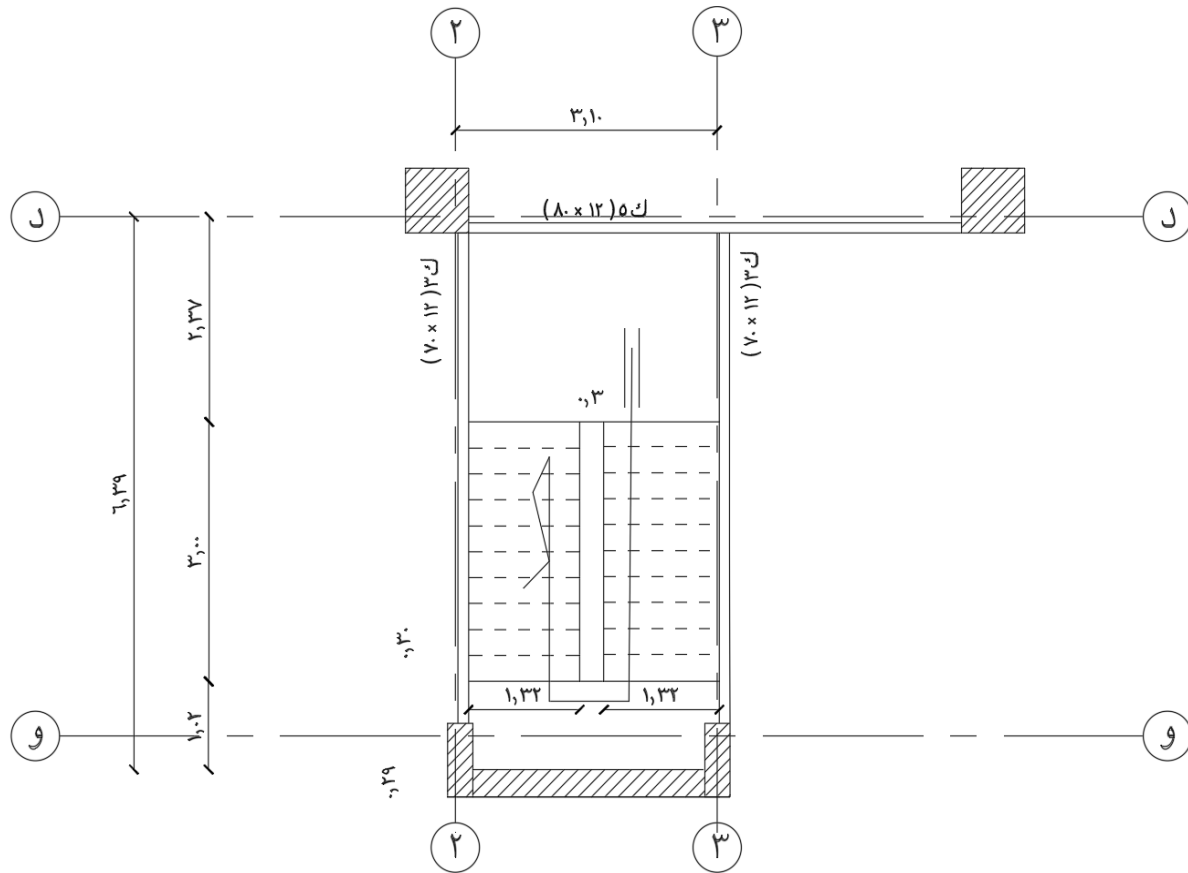


Figure 1.13 Stair Cross Section

#### 1) Dimensions:

- $ts = \frac{Span}{24:30} = \frac{6.39\text{ m}}{24:30} = 0.22\text{ m}$
- $H_{\text{Story}} = 2.90\text{ m}$
- $Rise = 0.15\text{ m}$
- $Going = 0.30\text{ m}$
- $\theta = \tan^{-1}\left(\frac{0.15}{0.30}\right) = 26.56^\circ$
- $t^* = \frac{ts}{\cos\theta} = \frac{22}{\cos(26.56)} = 24.59\text{ cm}$
- $t_{av} = t^* + \frac{Rise}{2} = 24.59 + 7.5 = 32\text{ cm}$

**2 ) Loads:**

- $W_{Su} = 1.4 \text{ D.L} + 1.6 \text{ L.L}$   
 $= 1.4 ( 25 \times 0.32 + 1.0 ) + 1.6 ( 2.5 )$   
 $= 16.6 \text{ KN/m}^2$
- $W_{u \text{ landing}} = 1.4 \text{ D.L} + 1.6 \text{ L.L}$   
 $= 1.4 ( 25 \times 0.22 + 1.5 ) + 1.6 ( 2.5 )$   
 $= 13.8 \text{ KN/m}^2$

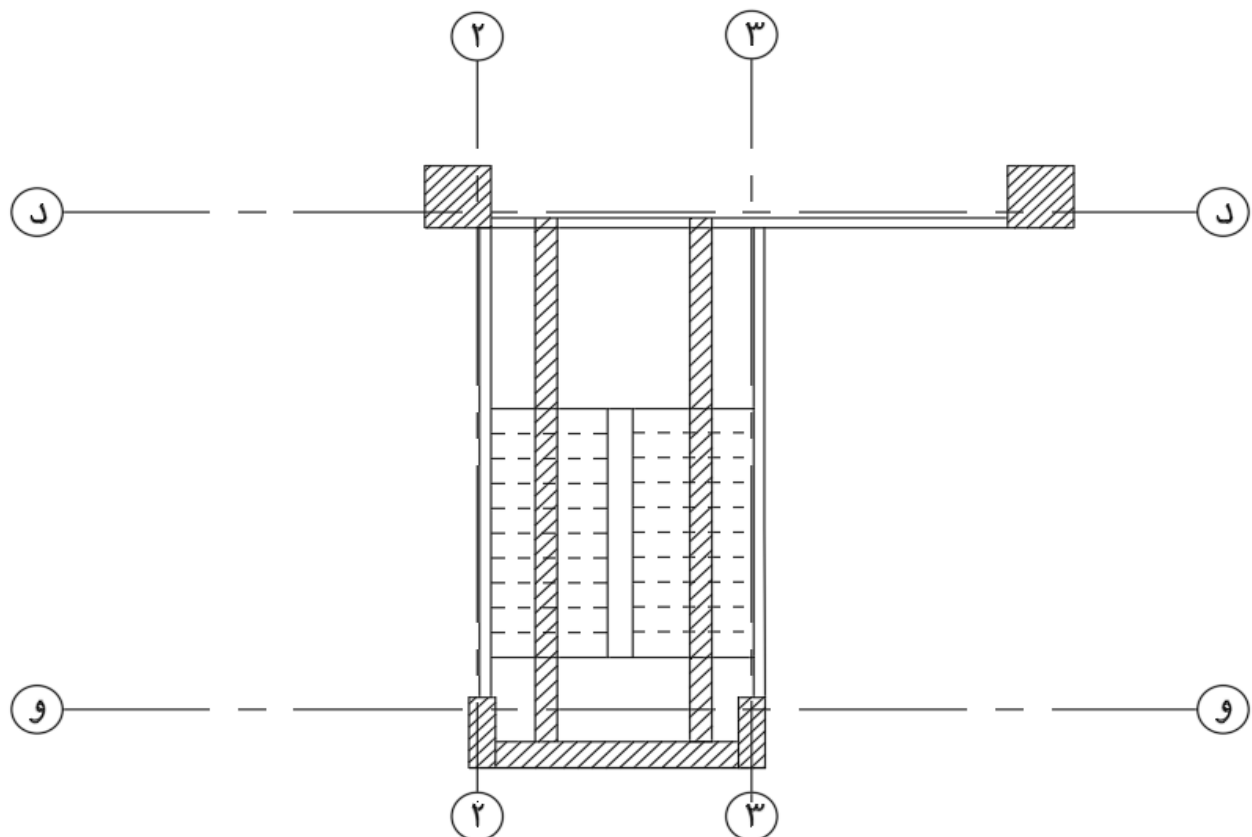
**3) For Strips**  
show in figure

Figure 1.14 Strips Of Stair

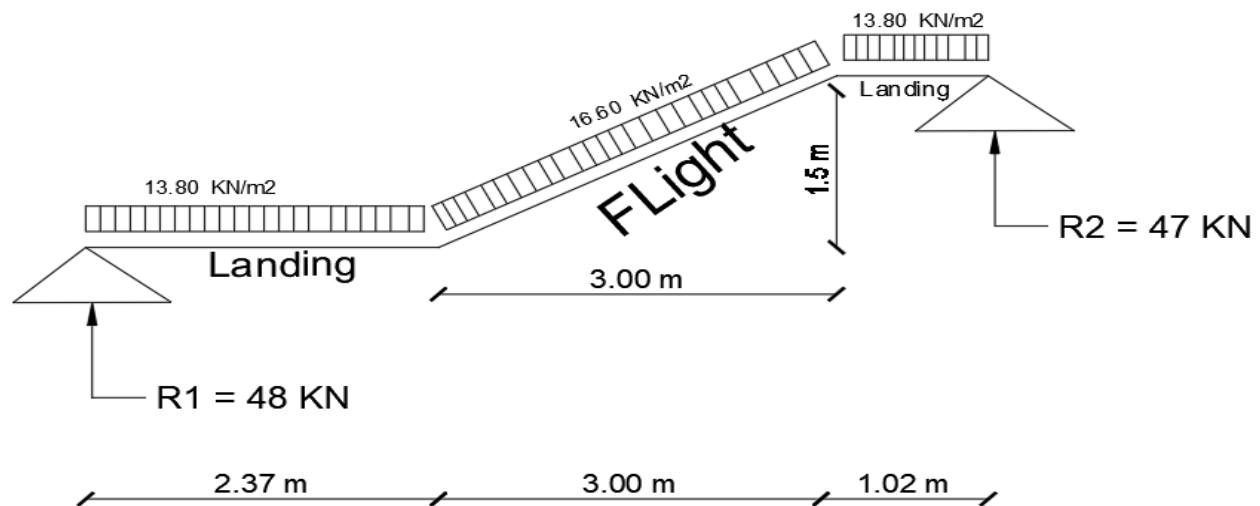
**For Strip :-**

$$R1 (6.39) = (13.8 * 2.37) \left( \frac{2.37}{2} + 3 + 1.02 \right) + (16.6 * 3) \left( \frac{3}{2} + 1.02 \right) + (13.8 * 1.02) \left( \frac{1.02}{2} \right)$$

$$= 302.909$$

$$\therefore R1 = \frac{302.909}{6.39} = 48 \text{ KN}$$

$$\therefore R2 = (13.8 * 2.37) + (16.6 * 3) + (13.8 * 1.02) - 48 = 47 \text{ KN}$$



**Moment :-**

**. M1**

$$R1(2.37) - 13.8 * 2.37 * \frac{2.37}{2} = 48(2.37) - 13.8 * 2.37 * \frac{2.37}{2} = 75 \text{ KN.m}$$

$$\therefore M1 = 75 \text{ KN.m}$$

**. M2**

$$R2(4.02) - 13.8 * 1.02 \left( \frac{1.02}{2} + 3 \right) - 16.6 * 3 * \frac{3}{2}$$

$$= 47(4.02) - 13.8 * 1.02 \left( \frac{1.02}{2} + 3 \right) - 16.6 * 3 * \frac{3}{2} = 64.8 \text{ KN.m}$$

$$\therefore M2 = 64.8 \text{ KN.m}$$

**. M3**

$$R1(5.37) - 13.8 * 2.37 \left( \frac{2.37}{2} + 3 \right) - 16.6 * 3 * \frac{3}{2} \\ = 48(5.37) - 13.8 * 2.37 \left( \frac{2.37}{2} + 3 \right) - 16.6 * 3 * \frac{3}{2} = 46.18 \text{ KN.m} \\ \therefore M3 = 46.18 \text{ KN.m}$$

**. M4**

$$R2(1.02) - 13.8 * 1.02 * \frac{1.02}{2} = 47(1.02) - 13.8 * 1.02 * \frac{1.02}{2} = 40.76 \text{ KN.m} \\ \therefore M4 = 40.76 \text{ KN.m}$$

**$\therefore$  Moment of flight = 55.5 KN.m**

**Design of section**

**1- M = 55.5 KN.m**

$$R_1 = \frac{M_u}{F_{cu} b d^2} = \frac{55.5 * 10^6}{35 * 1000 * 200^2} = 0.04 < R_{x \max} = 0.129$$

$$u = 0.05$$

$$\therefore A_s = u \frac{F_{cu}}{F_y} b d = 0.05 * \frac{35}{360} * 1000 * 200 = 972 \text{ mm}^2 = 9.72 \text{ cm}^2 \\ \text{Use } A_s \Rightarrow 6 \text{ } \phi 16 / \text{m} \Rightarrow A_{\text{sact}} = 1205.76 \text{ mm}^2$$

**2- M1 = 75 KN.m**

$$R_1 = \frac{M_u}{F_{cu} b d^2} = \frac{75 * 10^6}{35 * 1000 * 200^2} = 0.05 < R_{x \max} = 0.129$$

$$u = 0.06$$

$$\therefore A_s = u \frac{F_{cu}}{F_y} b d = 0.06 * \frac{35}{360} * 1000 * 200 = 1167 \text{ mm}^2 = 11.67 \text{ cm}^2$$

$$\text{Use } A_s \Rightarrow 6 \text{ } \phi 16 / \text{m} \Rightarrow A_{\text{sact}} = 1205.76 \text{ mm}^2$$

## Draw of moment and section and details of RFT

### Moment

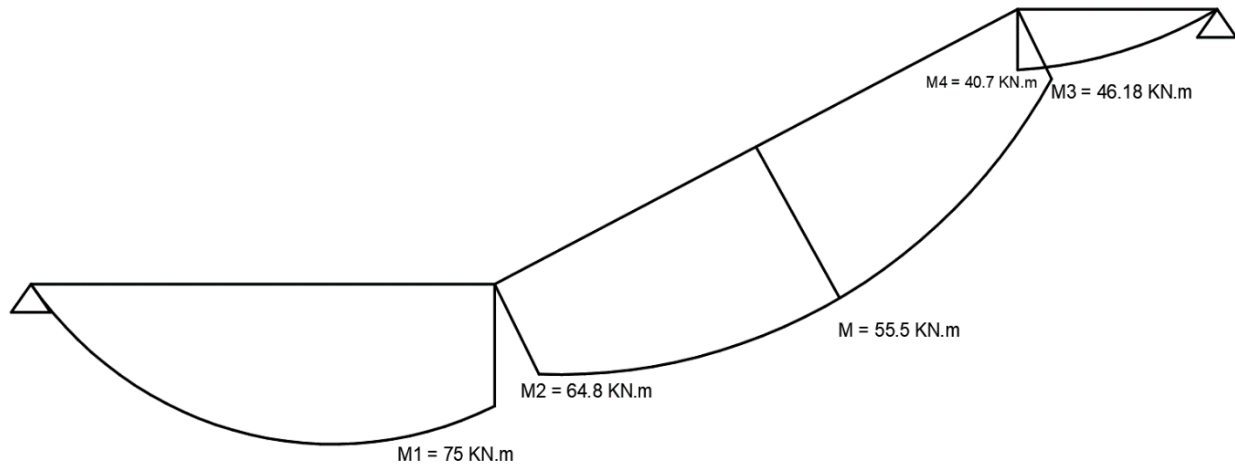


Figure 1.15 BMD

### Section and details of RFT

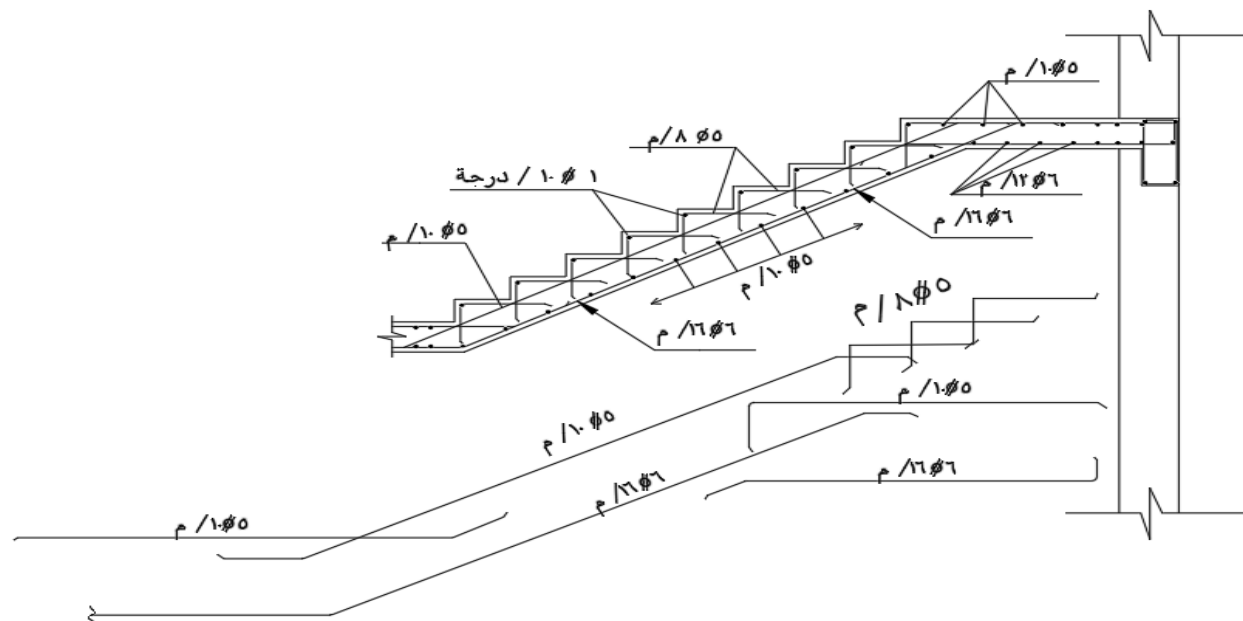


Figure 1.16 Section and details of RFT

### 1.3.2 Using Sap Program and Excel

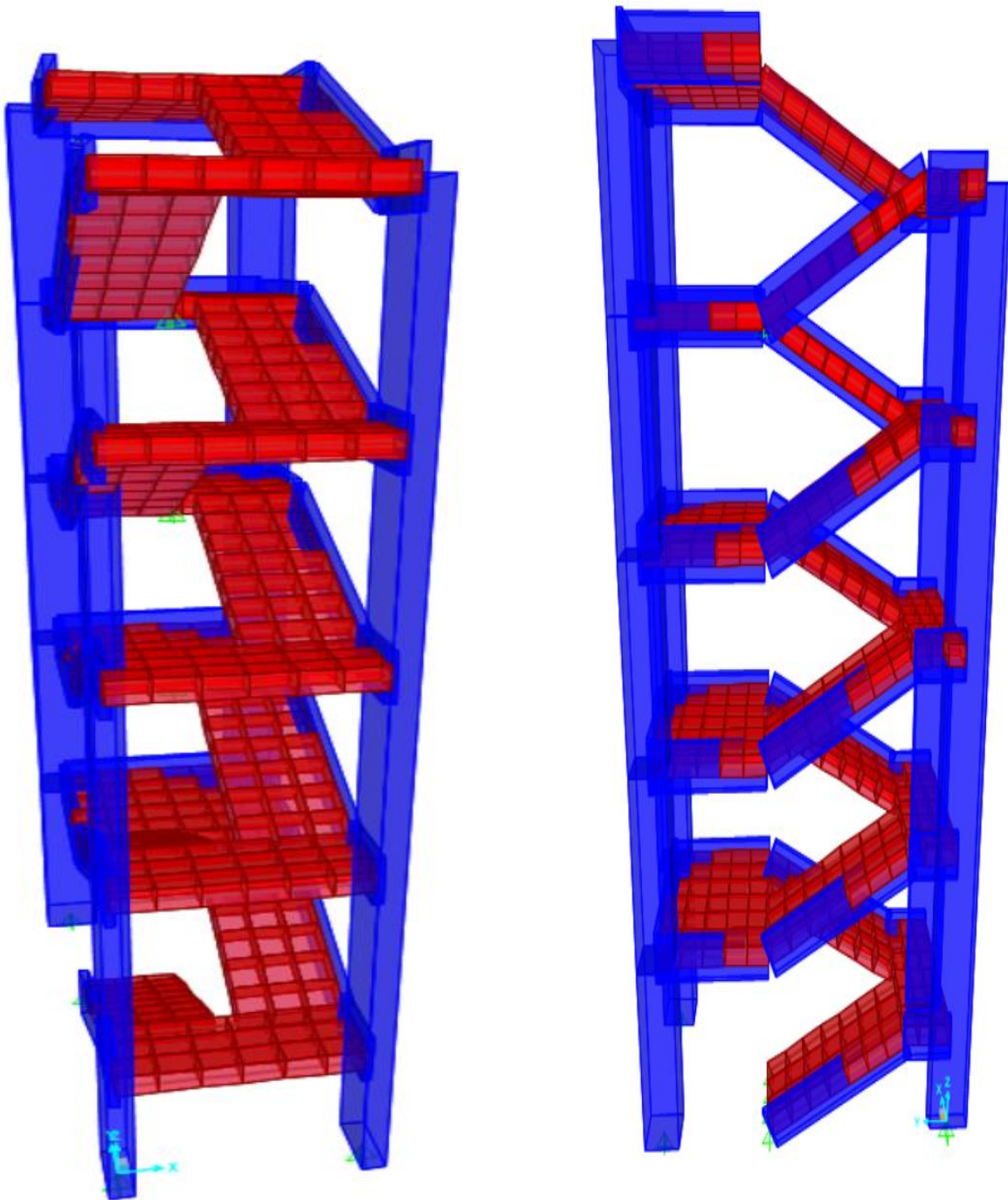


Figure 1.17 Stair 3D



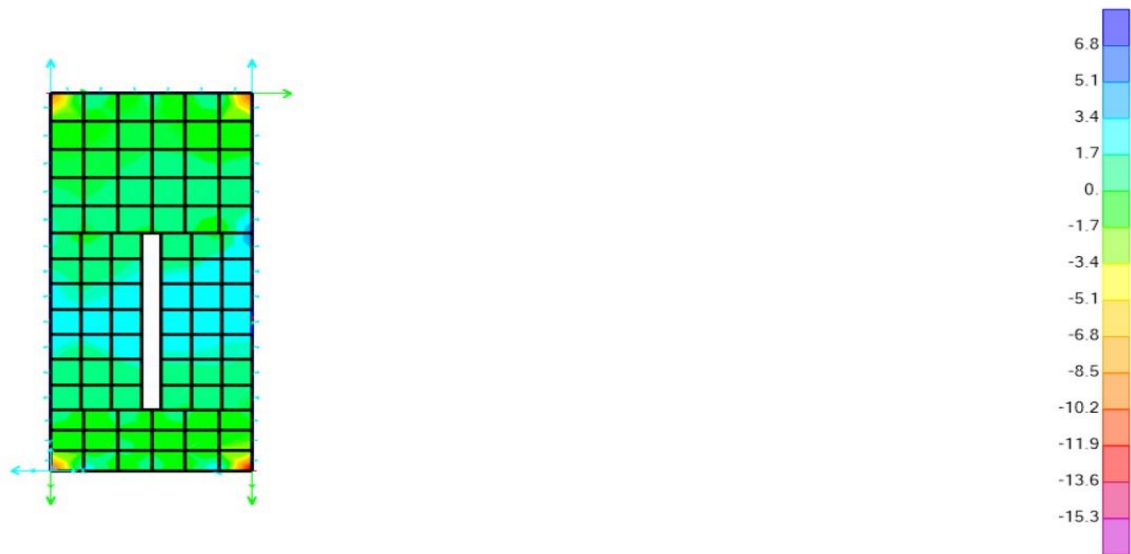


Figure 1.18 Moment axis xy

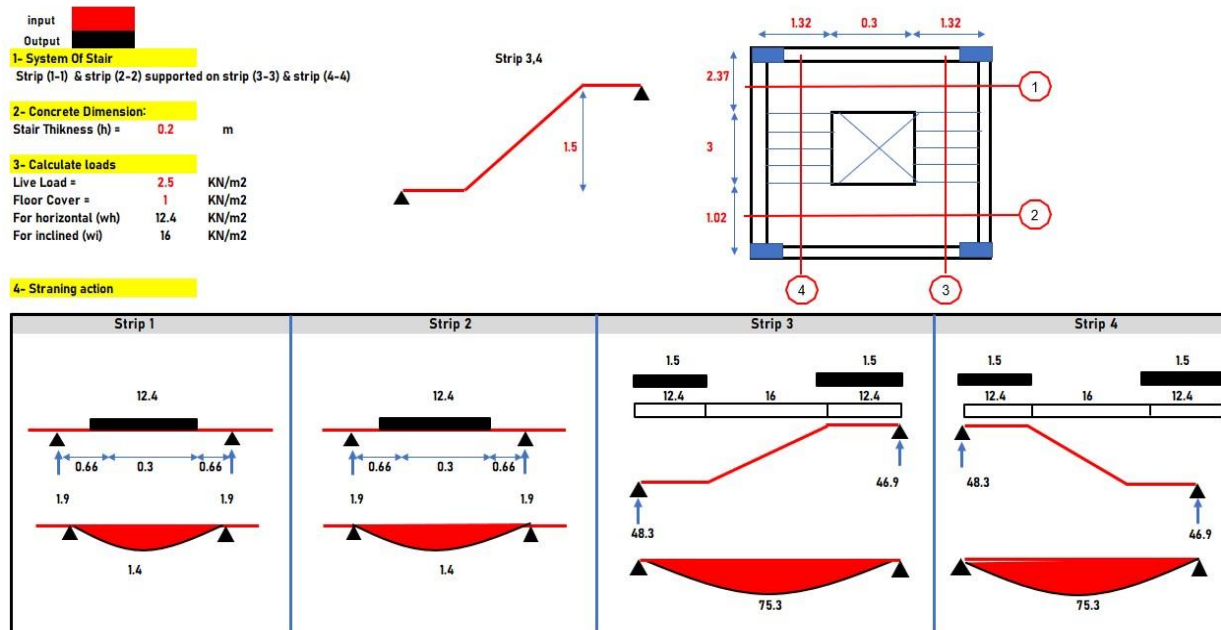


Figure 1.19 Sheet Excel of moment and BMD

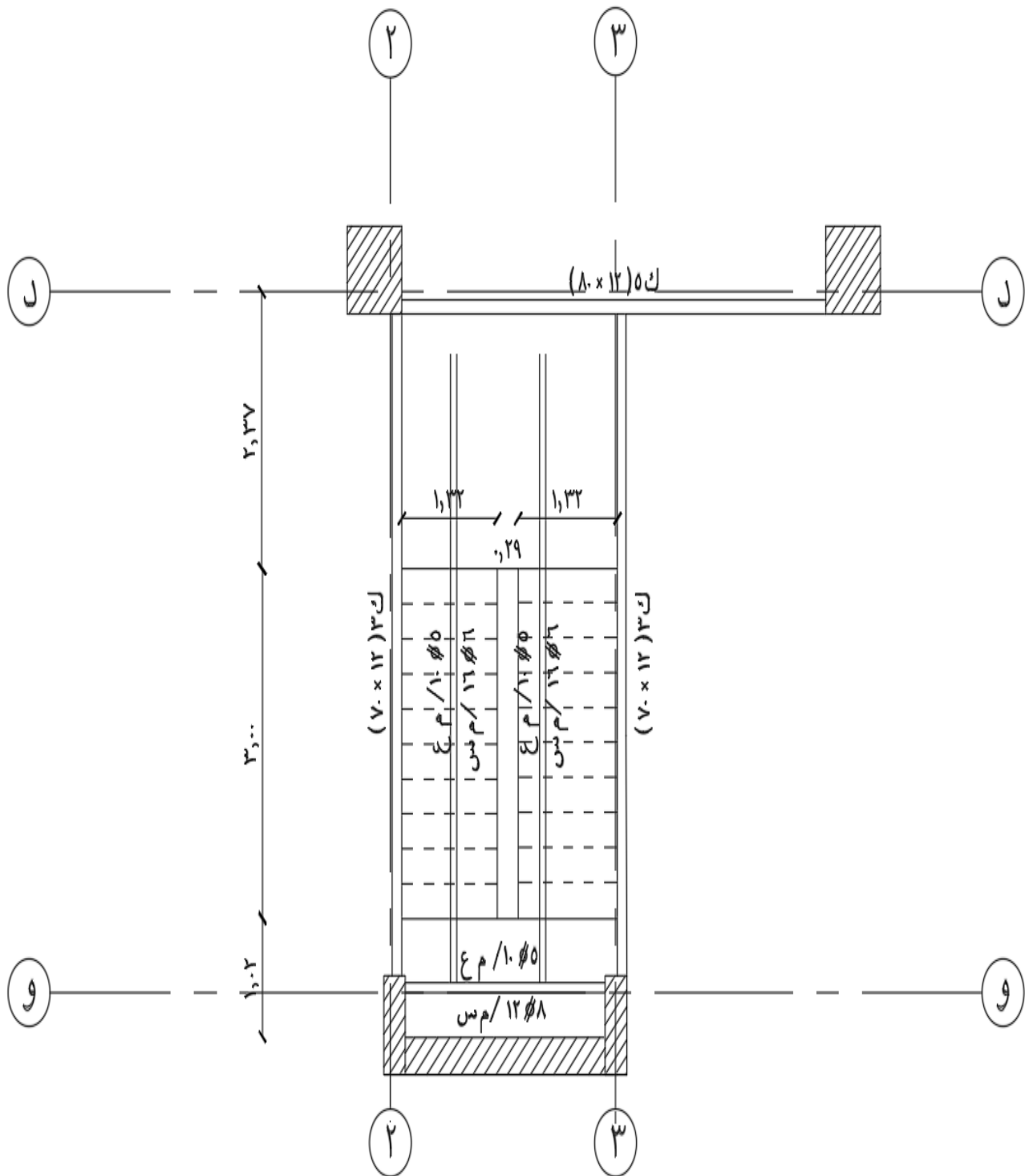


Figure 1.20 Stair Reinforcement in plan

### 1.3 Design of Stairs ( Three Flight Stair Axis / ب - أ )

#### 1.3.3 Manual solution

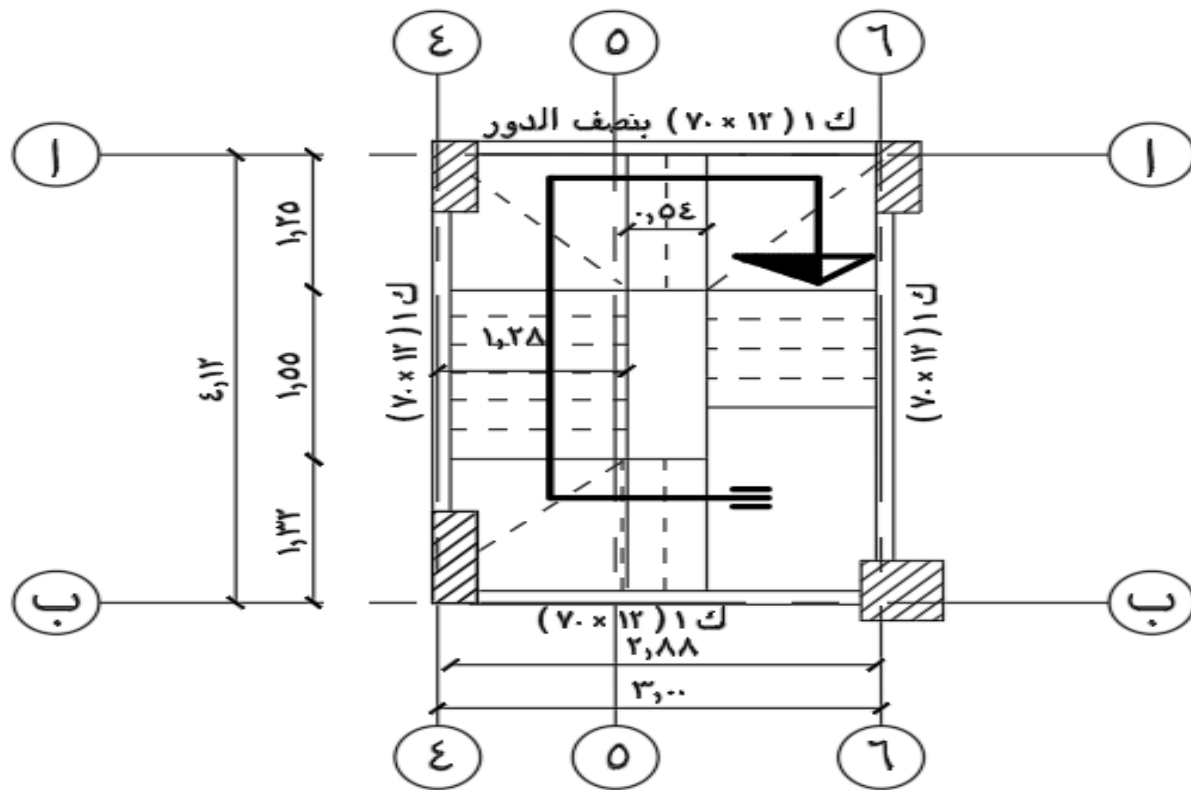


Figure 1.21 Stair Cross Section

#### 1) Dimensions:

$$ts = \frac{Span}{24:30} = \frac{4.12m}{24:30} = 0.16 m$$

$$H_{Story} = 3.80 m$$

$$Rise = 0.15 m$$

$$Going = 0.30 m$$

$$\theta = \tan^{-1}\left(\frac{0.15}{0.30}\right) = 26.56^\circ$$

$$t^* = \frac{ts}{\cos\theta} = \frac{16}{\cos(26.56)} = 17.88 cm$$

$$t_{av} = t^* + \frac{Rise}{2} = 17.88 + \frac{15}{2} = 25.38 cm$$

## 2 ) Loads:

- $W_{su} = 1.4 \text{ D.L} + 1.6 \text{ L.L}$ 

$$= 1.4 ( 25 \times 0.2538 + 1.0 ) + 1.6 ( 2.5 )$$

$$= 14.283 \text{ KN/m}^2$$
- $W_{u \text{ landing}} = 1.4 \text{ D.L} + 1.6 \text{ L.L}$ 

$$= 1.4 ( 25 \times 0.16 + 1.5 ) + 1.6 ( 2.5 )$$

$$= 11.7 \text{ KN/m}^2$$

## 3) For Strips :-

Shown In The figure

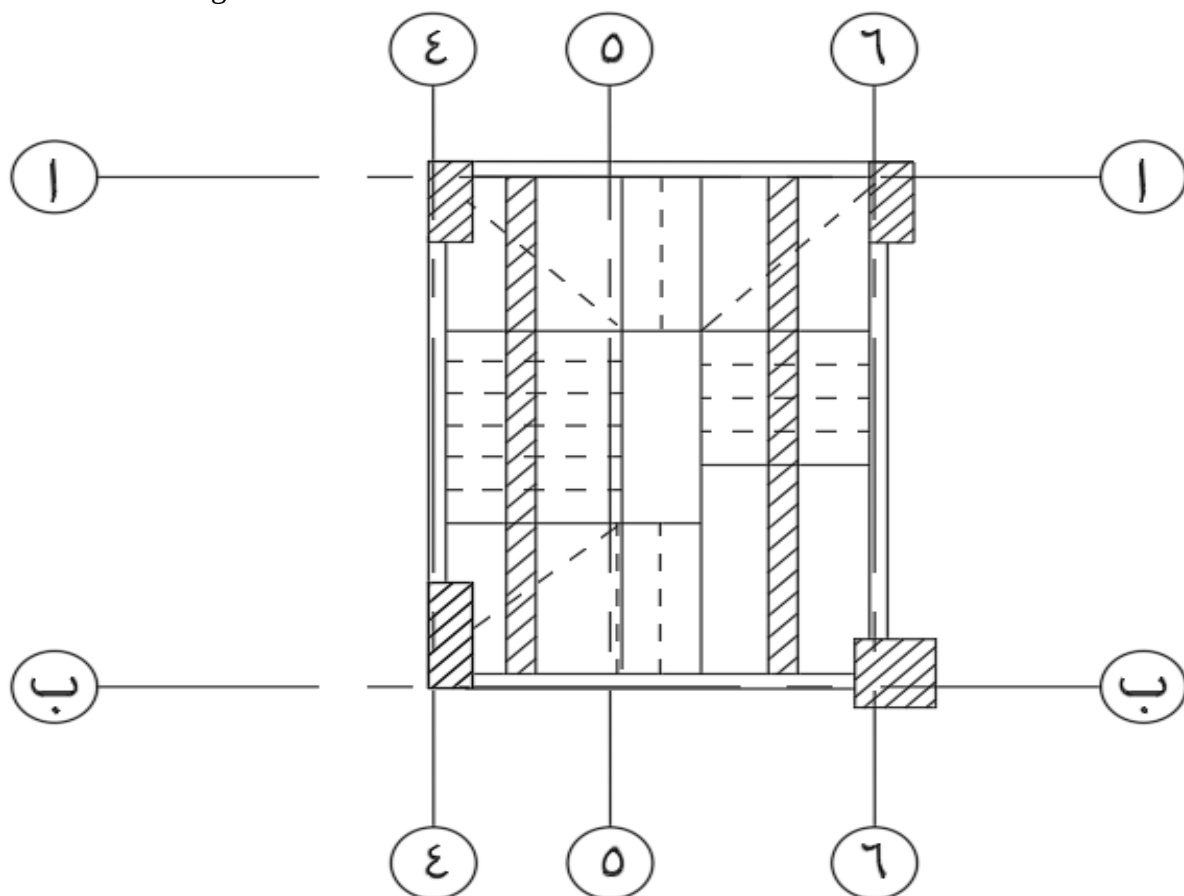


Figure 1.22 Strips Of Stair

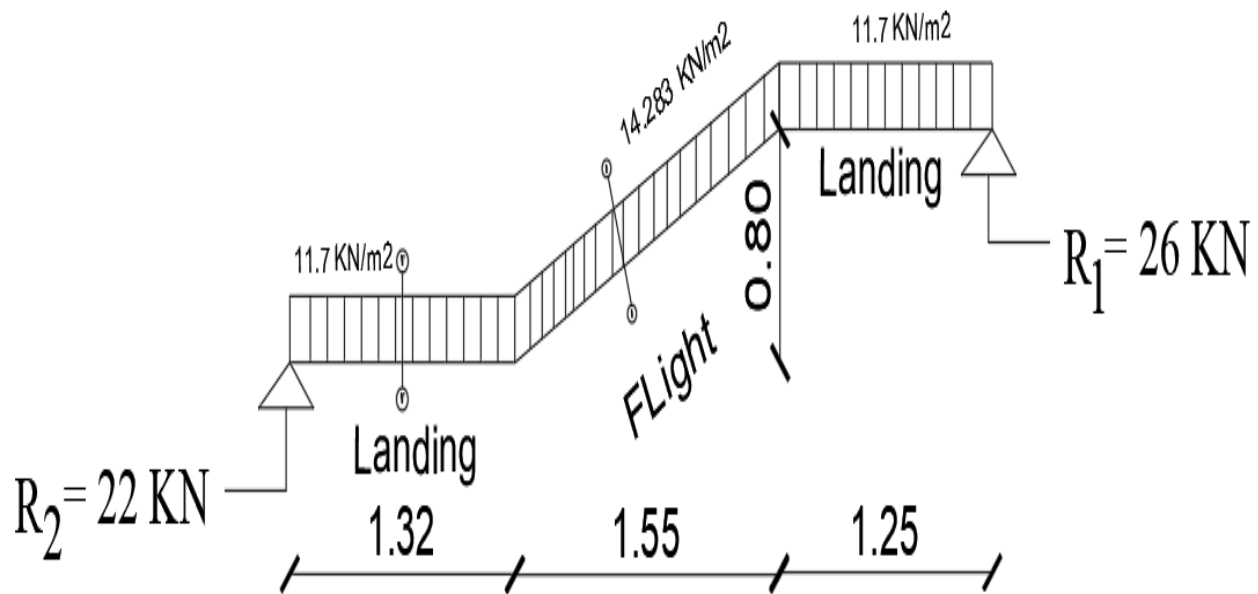
## For Strip :-

- $$R_1 * 4.12 = (11.7 * 1.32) \left( \frac{1.32}{2} + 1.55 + 1.25 \right) + (14.283 * 1.55) * \left( \frac{1.55}{2} + 1.32 \right) + (11.7 * 1.25) \left( \frac{1.25}{2} \right)$$

$$\therefore R_1 = 13.73 \text{ KN}$$

- $$R_2 = (11.7 * 1.32) + (14.283 * 1.55) + (11.7 * 1.25) - 26$$

$$\therefore R_2 = 22 \text{ KN}$$



## Moment :-

### . M1

$$R_2 (1.32) - 11.7 * 1.32 * \frac{1.32}{2} = 22 (1.32) - 11.7 * 1.32 * \frac{1.32}{2} = 18.8 \text{ KN.m}$$

$$\therefore M1 = 18.8 \text{ KN.m}$$

### . M2

$$R_1 (1.25 + 1.55) - 11.7 * 1.25 \left( \frac{1.25}{2} + 1.55 \right) - 14.283 * 1.55 * \frac{1.55}{2} = 26 (2.8) - 11.7 * 1.25 \left( \frac{1.25}{2} + 1.55 \right) - 14.283 * 1.55 * \frac{1.55}{2} = 23.8 \text{ KN.m}$$

$$\therefore M2 = 23.8 \text{ KN.m}$$

. M3

$$R2(1.32 + 1.55) - 11.7 * 1.32 \left( \frac{1.32}{2} + 1.55 \right) - 14.283 * 1.55 * \frac{1.55}{2}$$

$$= 22(2.87) - 11.7 * 1.32 \left( \frac{1.32}{2} + 1.55 \right) - 14.283 * 1.55 * \frac{1.55}{2} = 11.85 \text{ KN.m}$$

$\therefore M3 = 11.85 \text{ KN.m}$

. M4

$$R1(1.25) - 11.7 * 1.25 * \frac{1.25}{2} = 26(1.25) - 11.7 * 1.25 * \frac{1.25}{2} = 23.35 \text{ KN.m}$$

$\therefore M4 = 23.35 \text{ KN.m}$

$\therefore$  **Moment of flight = 23.57 KN.m**

#### 4) . Design of Section

1- **M = 23.57 KN.m**

$$R_1 = \frac{M_u}{F_{cu} b d^2} = \frac{23.57 * 10^6}{35 * 1000 * 200^2} = 0.02 < R_{x \max} = 0.129$$

$$w = 0.03$$

$$A_s = w \frac{F_{cu}}{F_y} b d = 0.03 * \frac{35}{360} * 1000 * 200 = 583.33 \text{ mm}^2 = 5.833 \text{ cm}^2$$

$$\text{Use } A_s \Rightarrow 6 \text{ } \phi 16 / \text{m} \Rightarrow A_{\text{sact}} = 1205.76 \text{ mm}^2$$

2- **M1 = 18.8 KN.m**

$$R_1 = \frac{M_u}{F_{cu} b d^2} = \frac{18.8 * 10^6}{35 * 1000 * 200^2} = 0.02 < R_{x \max} = 0.129$$

$$w = 0.03$$

$$\therefore A_s = w \frac{F_{cu}}{F_y} b d = 0.03 * \frac{35}{360} * 1000 * 200 = 583.33 \text{ mm}^2 = 5.833 \text{ cm}^2$$

$$\text{Use } A_s \Rightarrow 6 \text{ } \phi 16 / \text{m} \Rightarrow A_{\text{sact}} = 1205.76 \text{ mm}^2$$

### Draw of moment and section and details of RFT

## Moment

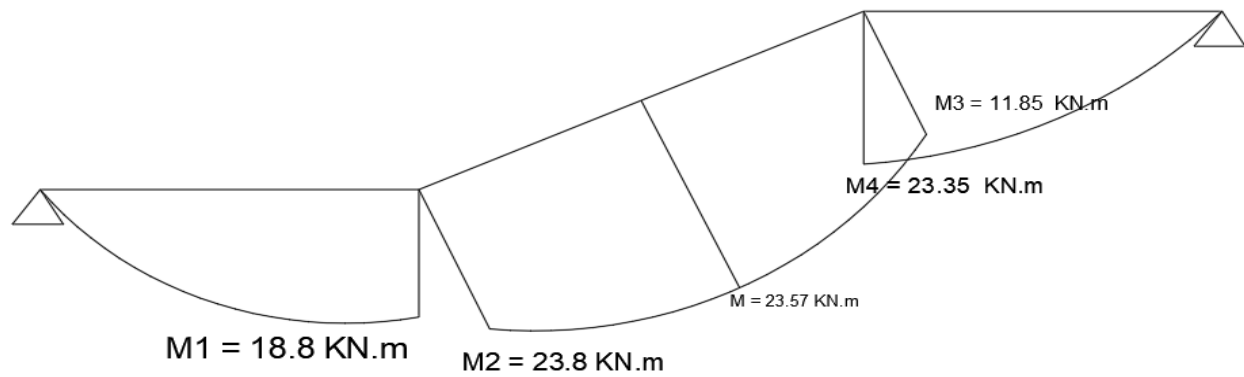


Figure 1.23 BMD

### Section and details of RFT

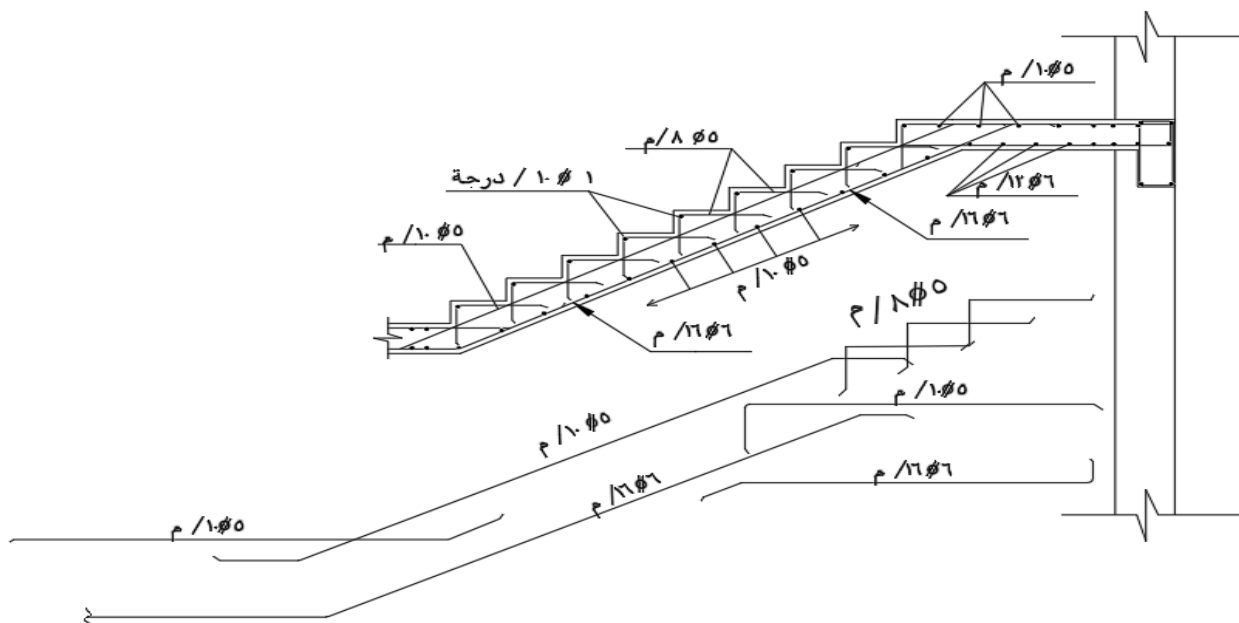


Figure 1.24 Section and details of RFT

### 1.3.4 Using Sap Program

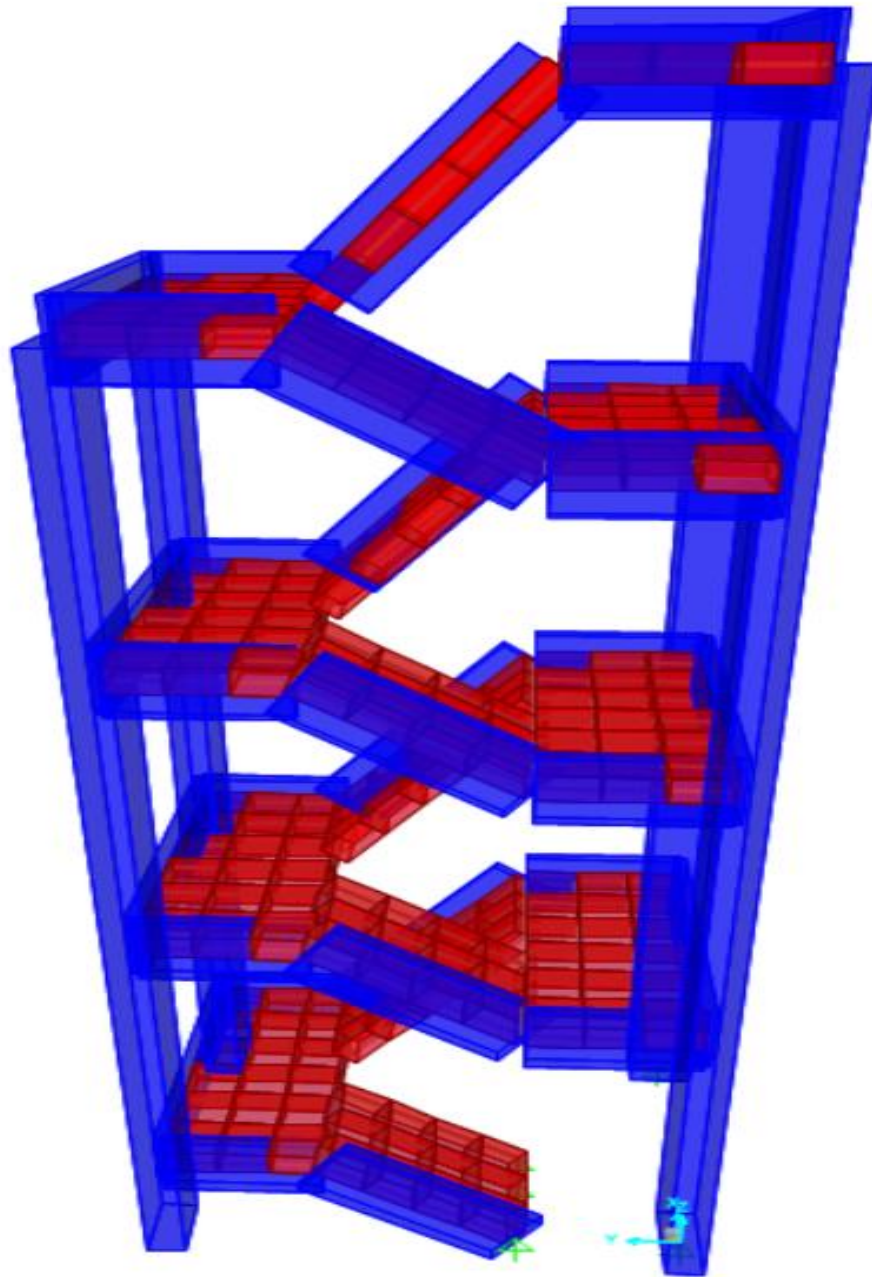


Figure 1.25 3D Stair



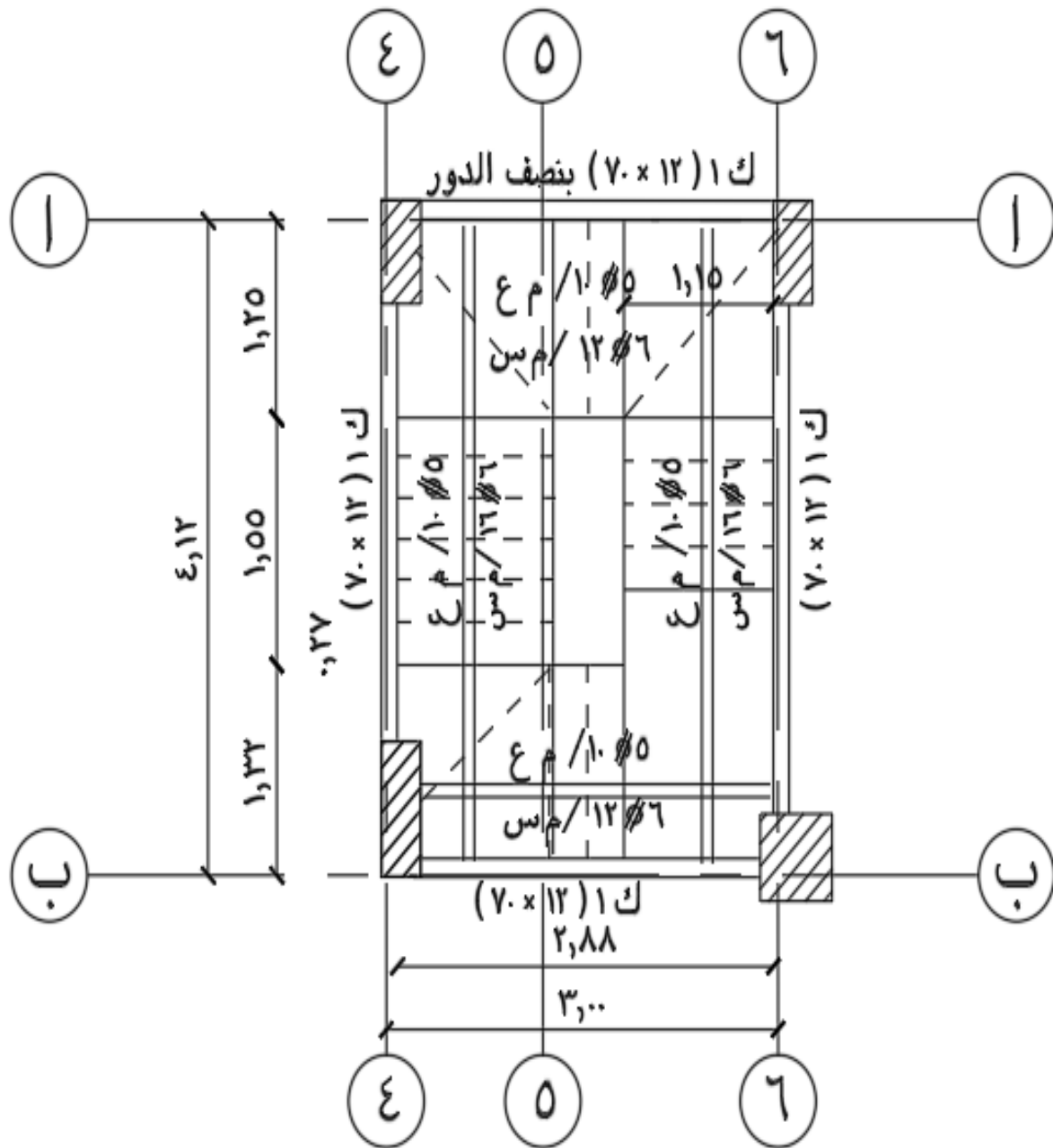


Figure 1.26 Stair Reinforcement in plan

## 1.4 Design of Beams

### Concrete dimensions

- ❖ Assume  $b = 12 \text{ cm}$
- ❖  $d = \frac{L}{12} = 80 \text{ cm}$ ,  $d = \frac{L}{12} = 70 \text{ cm}$
- ❖ **Loads on beams**
  - ❖ Own weight
  - ❖ Load from slab
  - ❖ Load from wall

$$\text{❖ } A_s = \frac{M_u}{F_y * j * d}$$

### 1.4.1 DESIGN OF Beam

#### Moment and shear

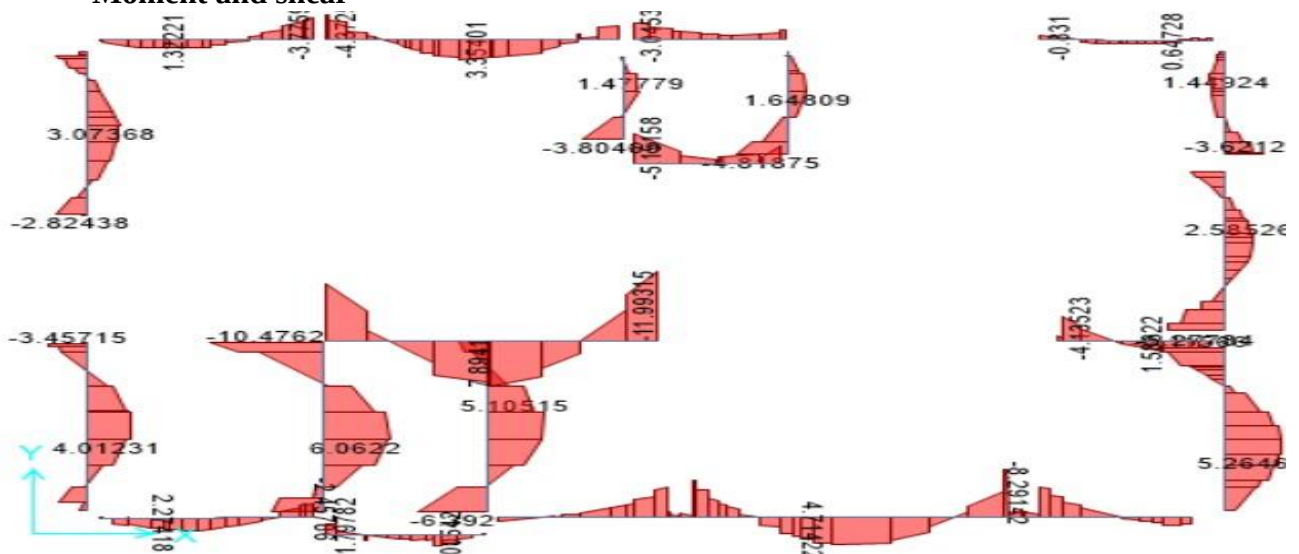


Figure 1.27 Moment Basement Slab

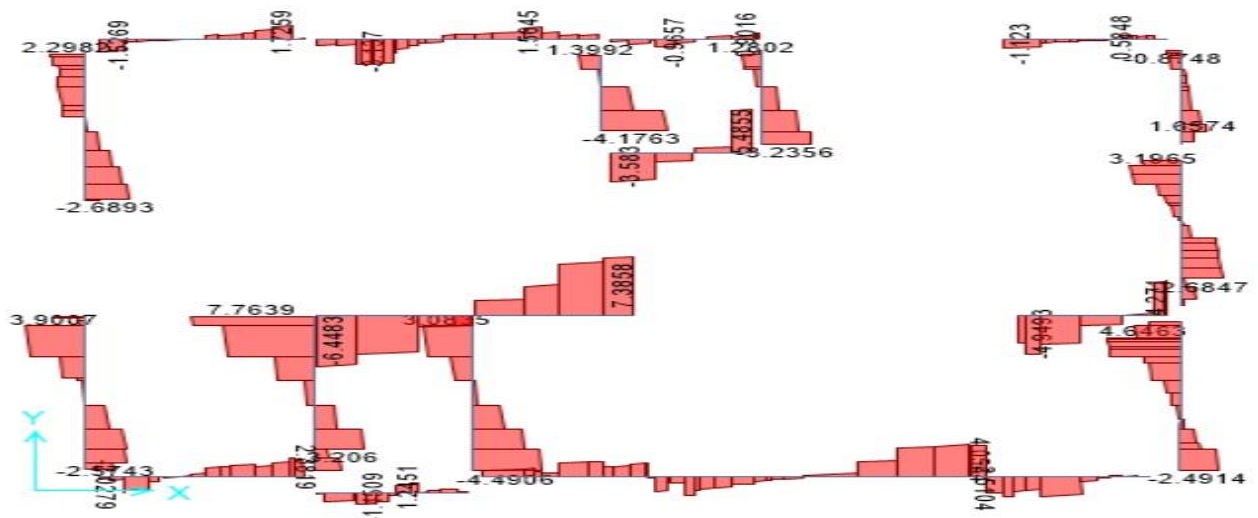


Figure 1.28 Shear Basement Slab

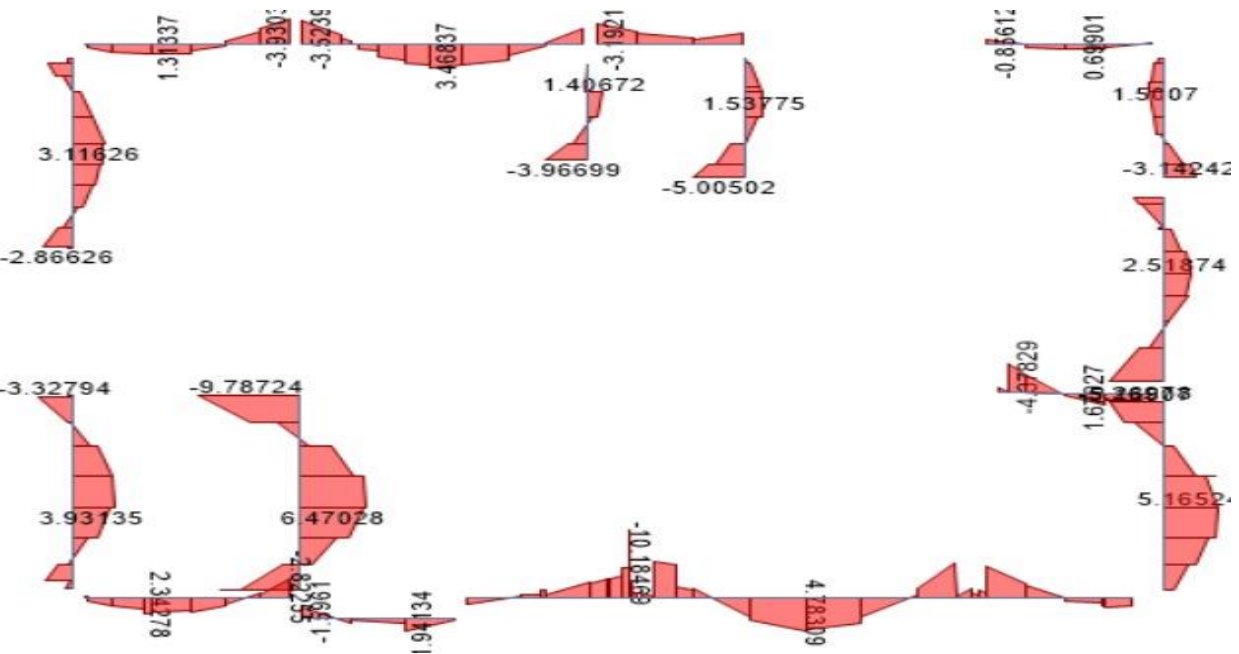


Figure 1.29 Moment Ground Slab

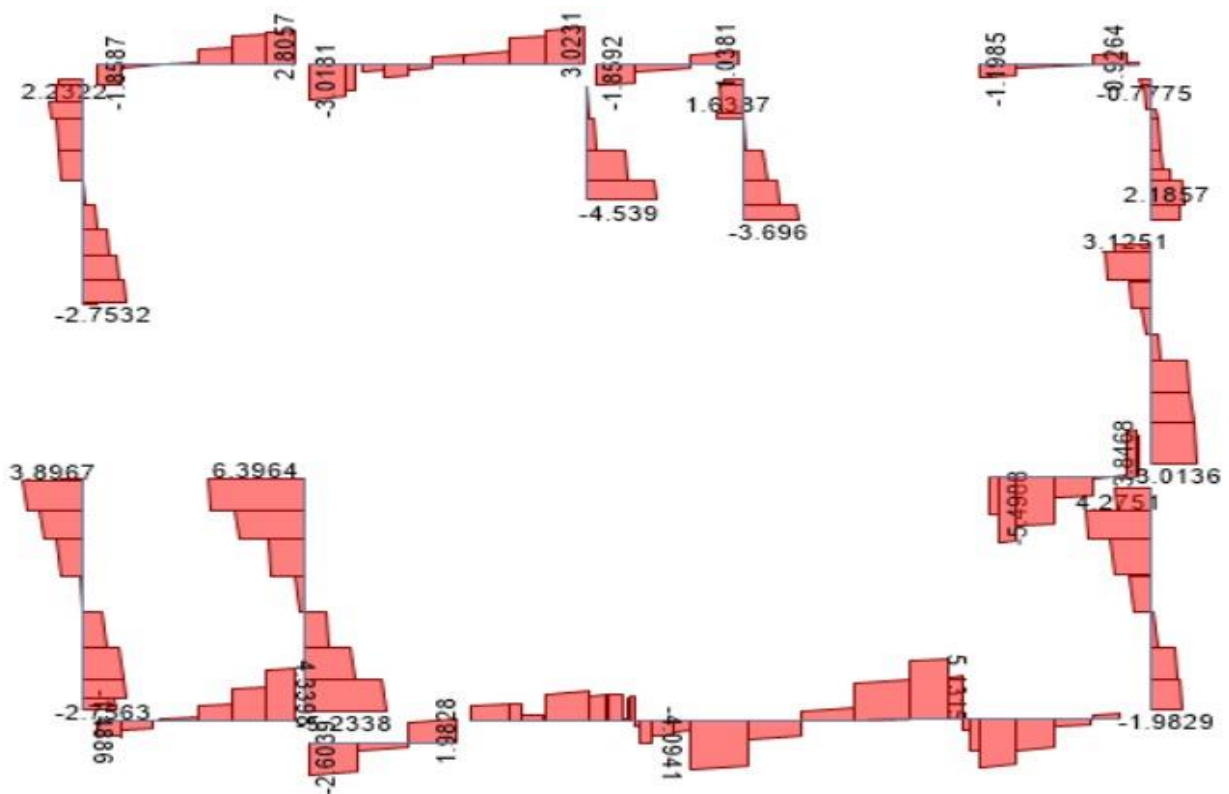


Figure 1.30 Shear Beam Ground Slab

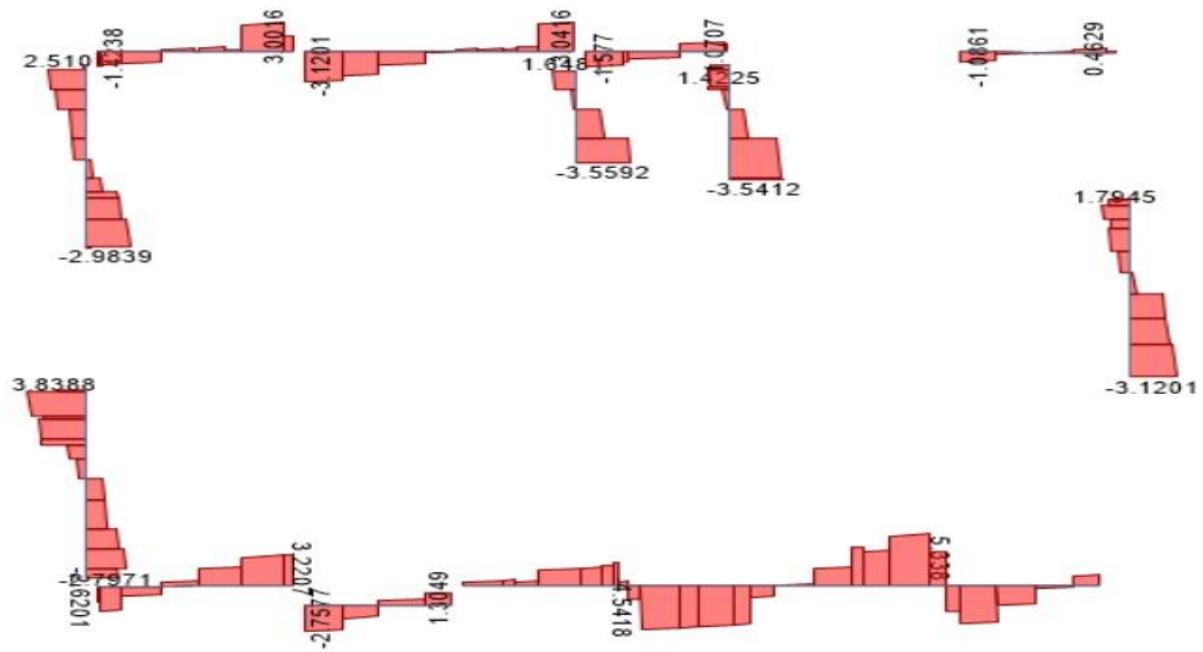


Figure 1.31 Moment Beam First Slab

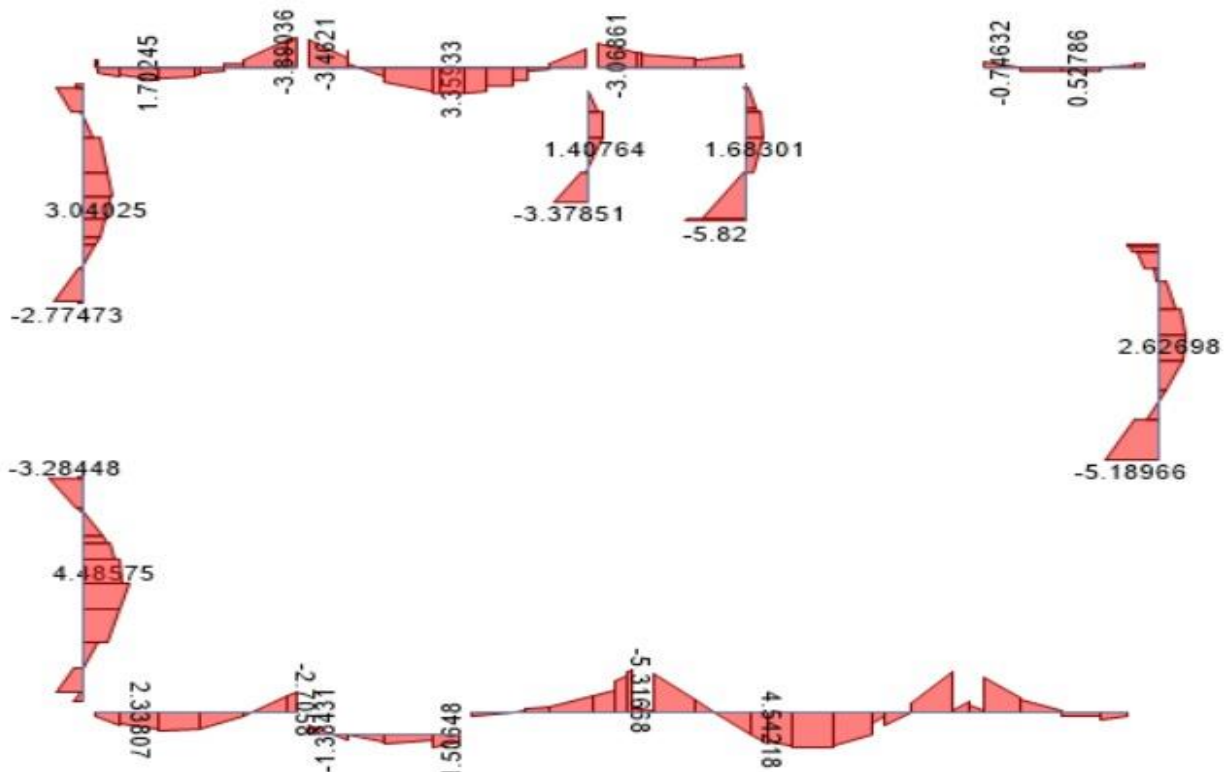


Figure 1.32 Shear Beam First Slab

## جدول تسليح الكمرات :

| ملاحظات | كانات / م | تسليح علوى  |             | تسليح سفلى | نموذج |
|---------|-----------|-------------|-------------|------------|-------|
|         |           | منتصف البحر | فوق الركيزه | عدل        |       |
|         | ٨ Ø ٥     | -----       | ١٢ Ø ٤      | ١٢ Ø ٢     | ك ١   |
|         | ٨ Ø ٥     | -----       | ١٢ Ø ٤      | ١٢ Ø ٤     | ك ٢   |
|         | ٨ Ø ٥     | -----       | ١٦ Ø ٢      | ١٦ Ø ٢     | ك ٣   |
|         | ٨ Ø ٥     | -----       | ١٦ Ø ٤      | ١٦ Ø ٤     | ك ٤   |
|         | ٨ Ø ٦     | -----       | ١٦ Ø ٦      | ١٦ Ø ٦     | ك ٥   |

Figure 1.33 Table RFT beam

## 1.4.2 Check of Shear on Beam Section:

Max shear on Basement Slab  $Q_{max} = 7.76 \text{ t}$ 

- $q_{cu}(uncracked) = 0.16 \sqrt{\frac{F_{cu}}{\gamma_c}} = 0.16 \sqrt{\frac{35}{1.5}} = 0.77 \text{ N/mm}^2$
  - $q_{max} = 0.7 \sqrt{\frac{F_{cu}}{\gamma_c}} = 0.7 \sqrt{\frac{35}{1.5}} = 3.38 \text{ N/mm}^2$
  - $q_u = \frac{Q_{max}}{b*d} = \frac{7.76*10^4}{250*780} = 0.4 \text{ N/mm}^2$
- $q_{cu} > q_u$  , Use Stirrups 6 Ø 8 / m as minimum

Max shear on Ground Slab  $Q_{max} = 6.4 \text{ t}$ 

- $q_{cu}(uncracked) = 0.16 \sqrt{\frac{F_{cu}}{\gamma_c}} = 0.16 \sqrt{\frac{35}{1.5}} = 0.77 \text{ N/mm}^2$
  - $q_{max} = 0.7 \sqrt{\frac{F_{cu}}{\gamma_c}} = 0.7 \sqrt{\frac{35}{1.5}} = 3.38 \text{ N/mm}^2$
  - $q_u = \frac{Q_{max}}{b*d} = \frac{6.4*10^4}{250*680} = 0.46 \text{ N/mm}^2$
- $q_{cu} > q_u$  , Use Stirrups 5 Ø 8 / m as minimum

Max shear on First Slab  $Q_{max} = 5.3 \text{ t}$ 

- $q_{cu}(uncracked) = 0.16 \sqrt{\frac{F_{cu}}{\gamma_c}} = 0.16 \sqrt{\frac{35}{1.5}} = 0.77 \text{ N/mm}^2$
- $q_{max} = 0.7 \sqrt{\frac{F_{cu}}{\gamma_c}} = 0.7 \sqrt{\frac{35}{1.5}} = 3.38 \text{ N/mm}^2$
- $q_u = \frac{Q_{max}}{b*d} = \frac{5.3*10^4}{250*680} = 0.31 \text{ N/mm}^2$

 $q_{cu} > q_u$  , Use Stirrups 5 Ø 8 / m as minimum

## 1.5 Design of Columns

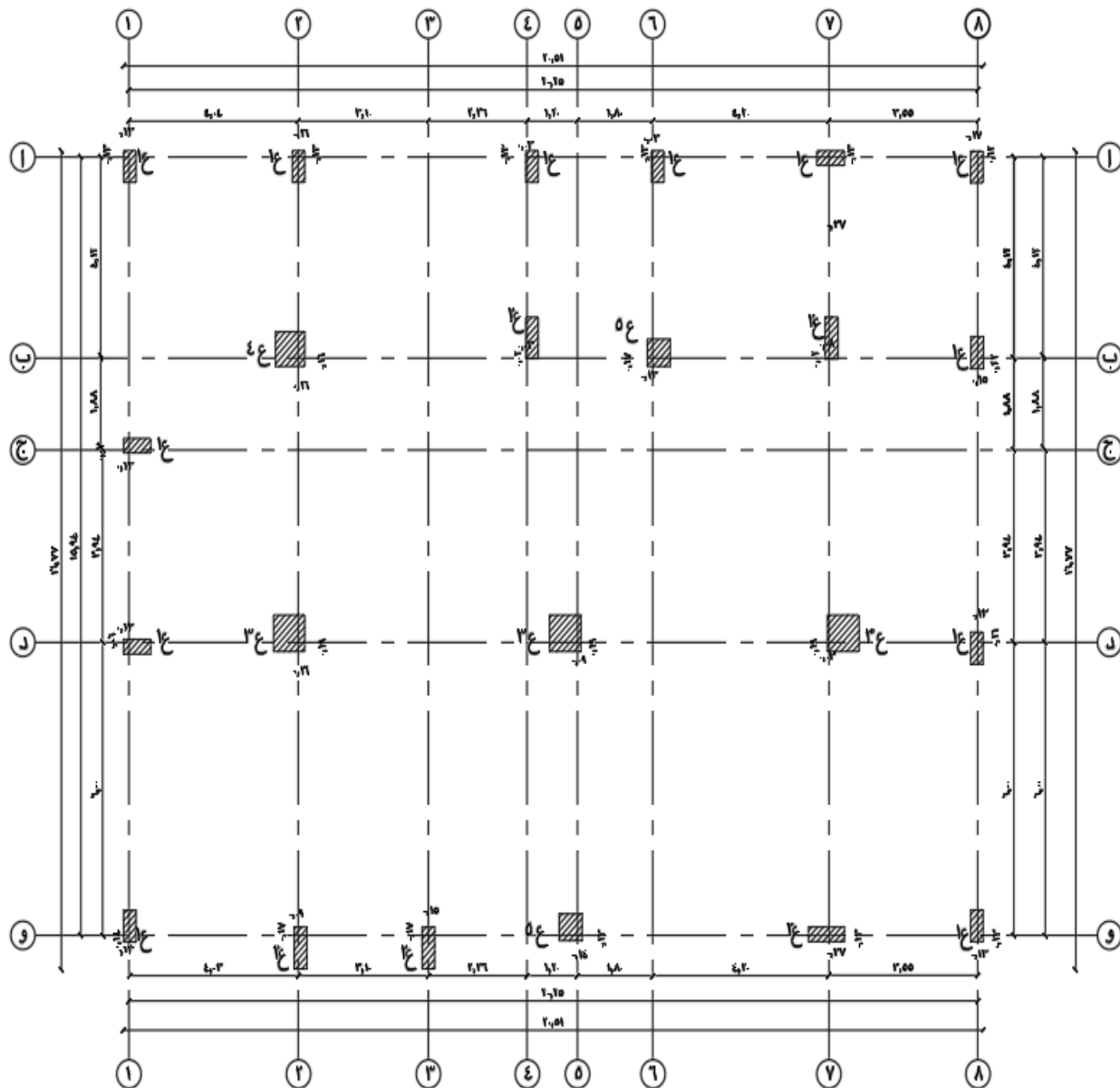


Figure 1.34 Columns Axis

### 1.5.1 Design of Column Section (subjected to axial compression force)

❖ For 1ξ on axis (8 - د)

#### ➤ Concrete Dimension

Use the largest Of

- $\frac{L}{20} = \frac{6}{20} = 0.3 \text{ m} = 30 \text{ cm}$
- $\frac{h}{15} = \frac{3}{15} = 0.20 \text{ m} = 20 \text{ cm}$
- 30 cm

Use b = 30 cm

- $P_U = 150 \text{ ton}$  (From Etabs Program)
- $P_{u \text{ actual}} = 150 \text{ ton}$
- $P_{u \text{ actual}} = 0.35 (A_c - A_s) F_{cu} + 0.67 A_s F_y \rightarrow \text{assume } A_s = 0.01 A_c$
- $150 * 10^4 = 0.35 * (0.99 A_c) * 35 + 0.67 * 360 * 0.01 A_c$
- $A_c = 103167.23 \text{ mm}^2$
- $t = \frac{A_c}{b} = \frac{103167.23}{300} = 343.89 \text{ mm} \sim 650 \text{ mm}$
- $A_{\text{cact}} = 30 * 65 = 1950 \text{ mm}^2$

## ❖ Check of Buckling (braced column)

- in short direction

$$\lambda = \frac{k*H_o}{b} = \frac{.8*3}{0.3} = 8 < 15 \Rightarrow$$

Short Column

- in Long direction

$$\lambda = \frac{k*H_o}{t} = \frac{0.8*3}{.650} = 3.69 < 15 \Rightarrow \text{Short Column}$$

- $A_s = 0.01 * 30 * 65 = 19.5 \text{ cm}^2$

Use 10  $\phi$  16

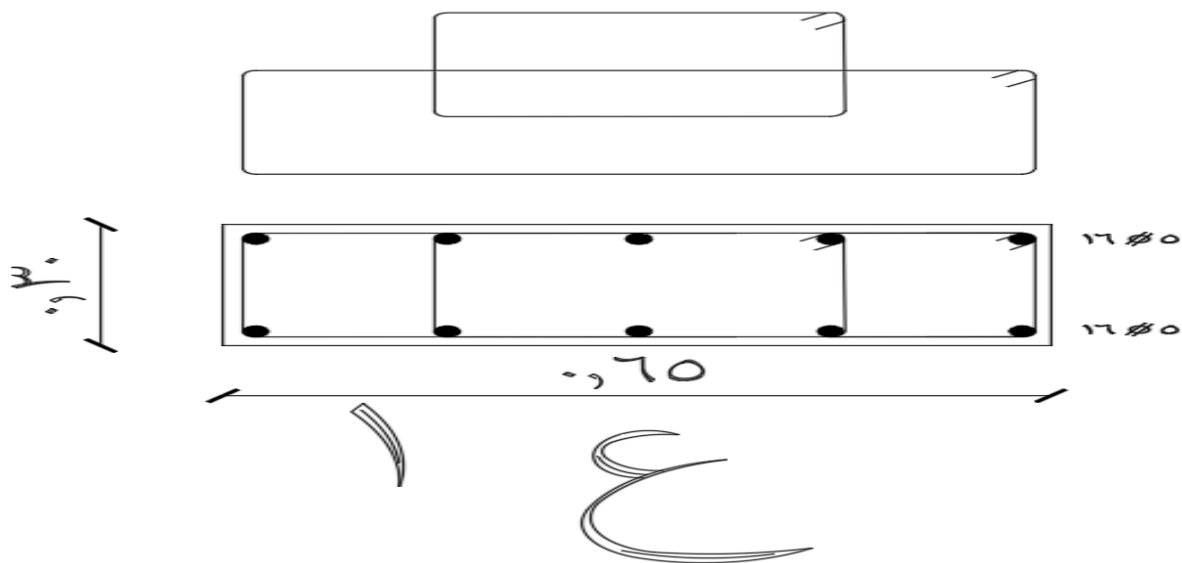


Figure 1.35 Column Cross Section

## 1.5.2 Table of Columns

| Joint | $P_{u \text{ etabs}}$<br>(T) | $P_{u \text{ act}}$ (T) | $\mu = A_s/A_c$ | b (cm) | t (cm) | §     | Sample |
|-------|------------------------------|-------------------------|-----------------|--------|--------|-------|--------|
| 1     | 150.01                       | 150                     | 1.00%           | 30     | 65     | 10§16 | C1     |
| 2     | 135.355                      | 135                     | 1.00%           | 30     | 65     | 10§16 | C1     |
| 3     | 135.355                      | 135                     | 1.00%           | 30     | 65     | 10§16 | C1     |
| 4     | 134.09                       | 134                     | 1.00%           | 30     | 65     | 10§16 | C1     |
| 5     | 144.21                       | 144                     | 1.00%           | 30     | 65     | 10§16 | C1     |
| 6     | 129.03                       | 129                     | 1.00%           | 30     | 65     | 10§16 | C1     |
| 7     | 112.585                      | 113                     | 1.00%           | 30     | 65     | 10§16 | C1     |
| 8     | 108.79                       | 109                     | 1.00%           | 30     | 65     | 10§16 | C1     |
| 9     | 98.67                        | 99                      | 1.00%           | 30     | 65     | 10§16 | C1     |
| 10    | 98.67                        | 99                      | 1.00%           | 30     | 65     | 10§16 | C1     |
| 11    | 92.345                       | 92                      | 1.00%           | 30     | 65     | 10§16 | C1     |
| 12    | 56.925                       | 57                      | 1.00%           | 30     | 65     | 10§16 | C1     |
| 13    | 228.8                        | 229                     | 1.00%           | 30     | 85     | 12§16 | C2     |
| 14    | 214.5                        | 215                     | 1.00%           | 30     | 85     | 12§16 | C2     |
| 15    | 187.22                       | 187                     | 1.00%           | 30     | 85     | 12§16 | C2     |
| 16    | 165.715                      | 166                     | 1.00%           | 30     | 85     | 12§16 | C2     |
| 17    | 120.175                      | 120                     | 1.00%           | 30     | 85     | 12§16 | C2     |
| 18    | 440.22                       | 440                     | 1.00%           | 75     | 75     | 28§16 | C3     |
| 19    | 400                          | 440                     | 1.00%           | 75     | 75     | 28§16 | C3     |
| 20    | 431.365                      | 431                     | 1.00%           | 75     | 75     | 28§16 | C3     |
| 21    | 333.96                       | 334                     | 1.00%           | 70     | 70     | 22§16 | C4     |
| 22    | 222.64                       | 223                     | 1.00%           | 55     | 55     | 20§16 | C5     |
| 23    | 211.255                      | 211                     | 1.00%           | 55     | 55     | 20§16 | C5     |

Table 1.36 Columns Load And Section (Ultimate)



## 1.6 Design of Foundation

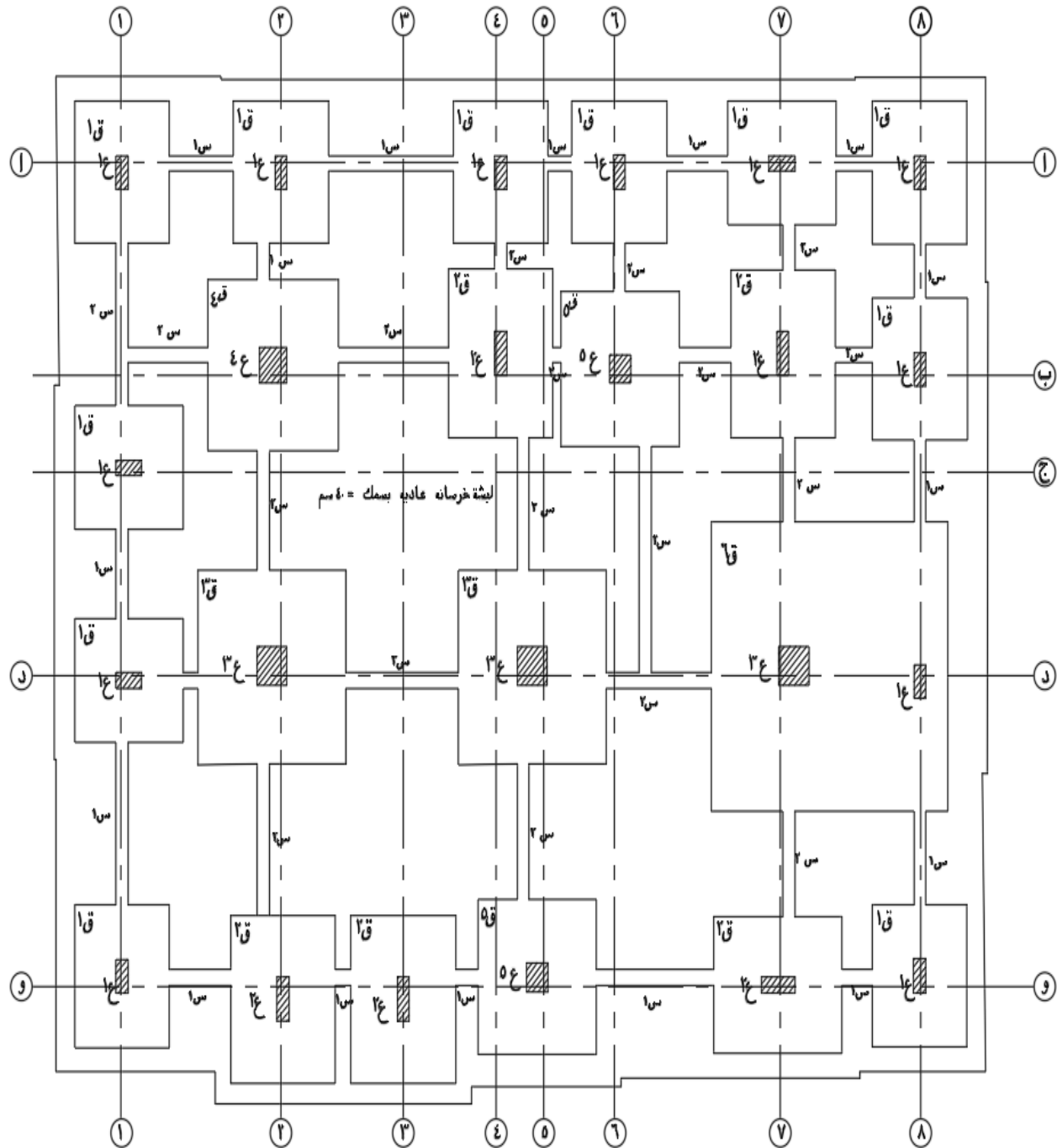


Figure 1.37 Foundations

- 1) Bearing Capacity for Soil =  $1 \text{ t/m}^2 = 10 \text{ kg/cm}^2$
- 2) Tensile Steel Stress  $F_y = 360$
- 3) Compressive Concrete Stress  $F_{cu} = 35$

### 1.6.1 For (1ق)

#### 1.1 Input Data:

- Column Working Load = 106.71 ton
- Column Dimension a = 30 cm  
b = 65 cm
- Plain Concrete Depth = 0.4 m
- Plain Concrete Extension = 0.40 m

#### 1.2 Concrete Dimension :

Figure 1.64 Section of footing (elevation)

- $B.C = \frac{106.71}{A_{pc}}$
- $10 = \frac{106.7}{A_{pc}}$
- $A_{pc} = 10.671 \text{ m}^2 = L_{pc} * B_{pc}$   
 $= (0.65 + 2c) * (0.30 + 2c)$

$$C = 1.441 \text{ m}$$

- $L_{pc} = 0.65 + 2 * 1.441 = 3.54 \text{ m}$
- $B_{pc} = 0.30 + 2 * 1.441 = 3.18 \text{ m}$

$$\therefore L_{RC} = L_{pc} - 2t_{pc} = 3.54 - 2 * 0.40 = 2.75 \text{ m}$$

$$B_{RC} = B_{pc} - 2t_{pc} = 3.18 - 2 * 0.40 = 2.4 \text{ m}$$

#### 1.3 Check of Stress:

- $q_{act} = \frac{106.71}{A_{pc}}$
- $q_{act} = \frac{106.71}{3.54 * 3.18} = 9.47 \text{ t/m}^2 < B.C = 10 \text{ t/m}^2, \text{ Ok}$

$$q_{act} = \frac{P_U}{A_{RC}} = \frac{106.71}{2.75 * 2.4} = 161.68 \text{ KN}$$

$$Z = \frac{2.75 - 0.65}{2} = 1$$

$$M_U = \frac{q_{act} * Z^2}{2} = \frac{161.68 * 1^2}{2} = 85 \text{ KN.m}$$

$$d = c1 \sqrt{\frac{M_U}{F_{CU} * b}} = C1 \sqrt{\frac{85 * 10^6}{35 * 1000}} = 246.40 \Rightarrow d = 600 \Rightarrow t = 700$$

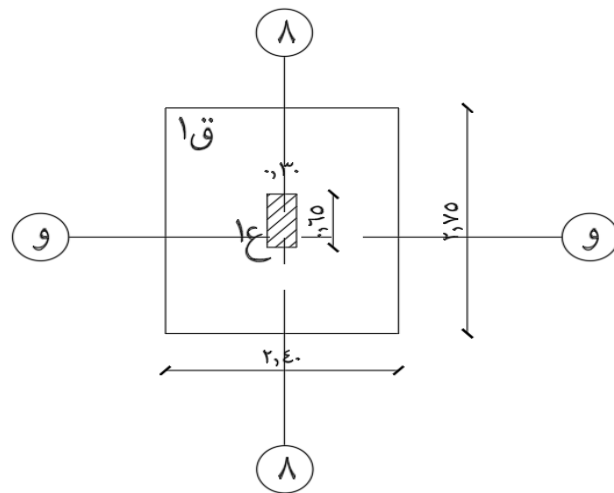


Figure 1.38. Sections of footing

$$Q_p = P_{co} - q_{act} * (a+d) * (b+d)$$

- $Q_p = 1067 - 161.68 * (0.3 + 0.6) * (0.64 + 0.6) = 887 \text{ KN}$
- $q_p = \frac{Q_p}{2((a+d)+(a+d))*d}$
- $q_p = \frac{887 * 10^3}{2((300 + 600) + (650 + 600)) * 600} = 0.35 \text{ N/mm}^2$

$$q_{p(allow)} = 1 - 1.7$$

$$2- 0.316 * (.5 + \frac{a}{b}) * \sqrt{\frac{F_{cu}}{\gamma_c}} = 0.316 * (0.5 + \frac{0.3}{0.65}) * \sqrt{\frac{35}{1.5}} = 1.46 \text{ N/mm}^2$$

$$3- 0.316 * \sqrt{\frac{F_{cu}}{\gamma_c}} = 0.316 * \sqrt{\frac{35}{1.5}} = 1.53 \text{ N/mm}^2$$

$$4- 0.8 * (\frac{\alpha * d}{2(a+d+b+d)} + .2) * \sqrt{\frac{F_{cu}}{\gamma_c}} =$$

$$= 0.8 * (\frac{4 * 600}{2(300 + 600 + 650 + 600)} + 0.2) * \sqrt{\frac{35}{1.5}} = 2.93 \text{ N/mm}^2$$

$$q_p < q_{p(allow)} \Rightarrow \text{OK, safe}$$

#### 1.4 Check of Shear

$$Q_{sh1} = q_{act} * c = 161.68 * 1.441 = 232.98 \text{ ton}$$

$$q_{sh} = \frac{Q_{sh}}{B * d} = \frac{232.98 * 10^3}{2400 * 600} = 0.161 \text{ KN / mm}^2$$

$$q_{sh(allow)} = 0.16 * \sqrt{\frac{F_{cu}}{\gamma_c}} = 0.16 * \sqrt{\frac{35}{1.5}} = 0.773 \text{ KN / mm}^2$$

$$q_{sh} < q_{sh(allow)} \text{ OK .Safe}$$

### 1.5 RFT

$$A_s = \frac{M_U}{F_Y * j * d} = \frac{85 * 10^6}{360 * 0.826 * 600} = 476.41 \text{ mm}^2$$

$$A_s = 9 \text{ } \phi 12 / \text{m}$$

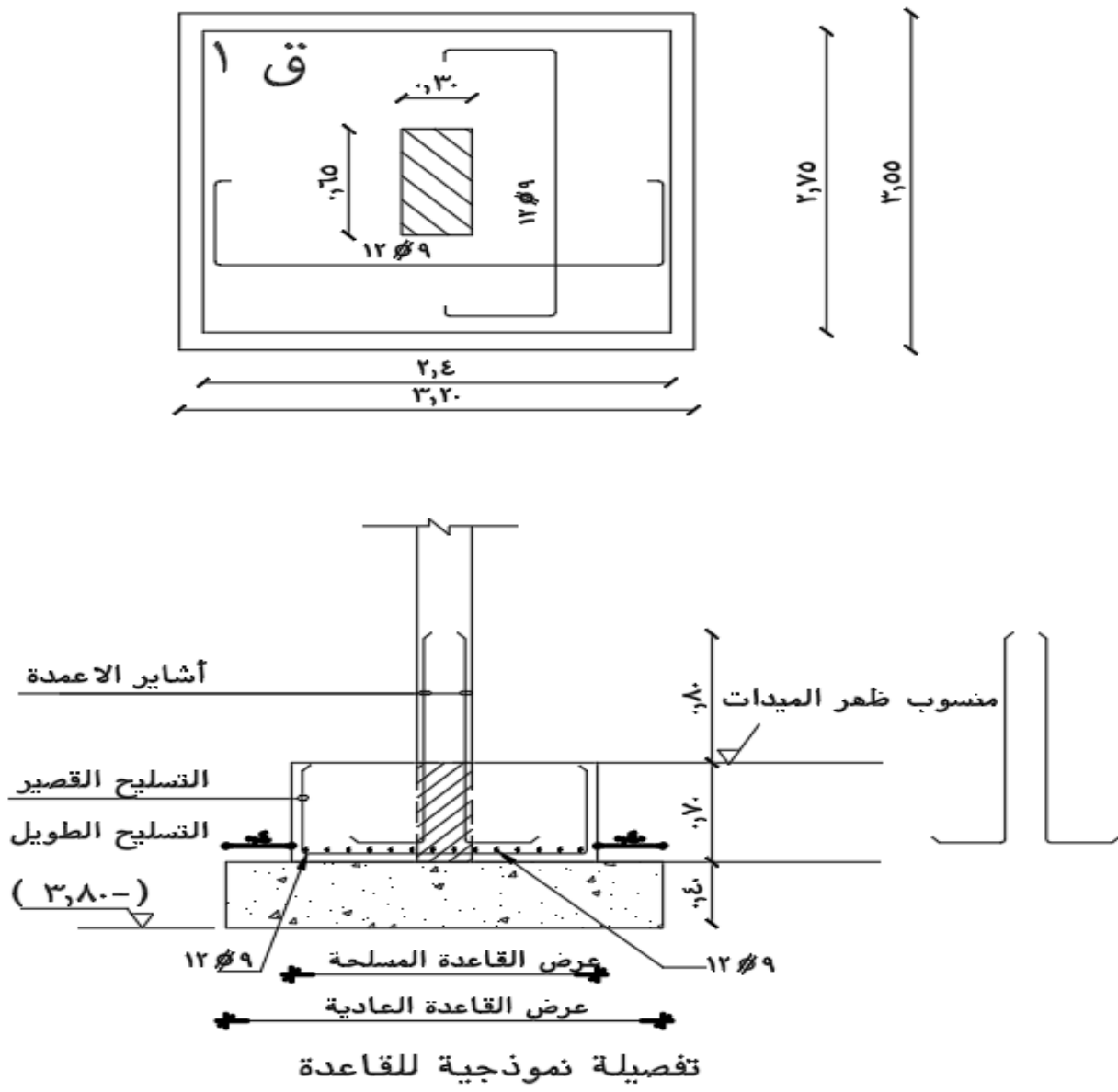


Figure 1.39. Sections of footing and details' of RFT

## 1.6.2 For ( ق 6 )

### 2.1 Input Data:

- Column 1 Working Load  $P_1 = 118$  ton  
Column 3 Working Load  $P_2 = 316$  ton
- Column Dimension  $a_1 = 30$  cm  
 $B_1 = 65$  cm  
Column Dimension  $a_2 = 75$  cm  
 $B_2 = 75$  cm
- Plain Concrete Depth = 0.4 m
- Plain Concrete Extension = 0.40 m

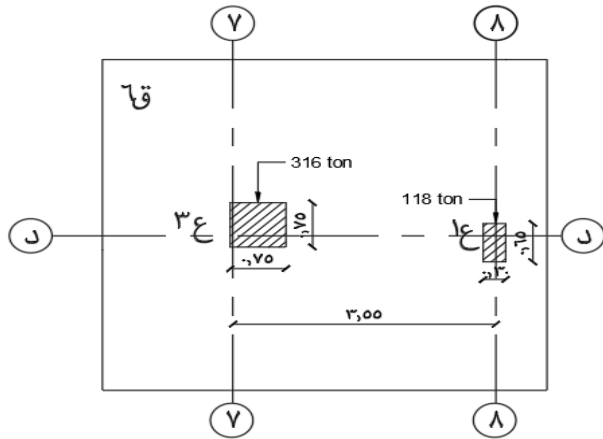


Figure 1.40. Sections of footing

### SOLUTION :-

#### 2.2 - Get R Location

$$R = P_1 + P_2 = 118 + 316 = 434 \text{ ton}$$

$$em @ c1 = 0.0$$

$$R * x = P_2 * s$$

$$434 * x = 316 * 3.55$$

$$\therefore X = 2.58 \text{ m}$$

#### 2.3 - Concrete Dimension

$$B.C = P_w / A_{pc}$$

$$10 = \frac{470}{A_{pc}}$$

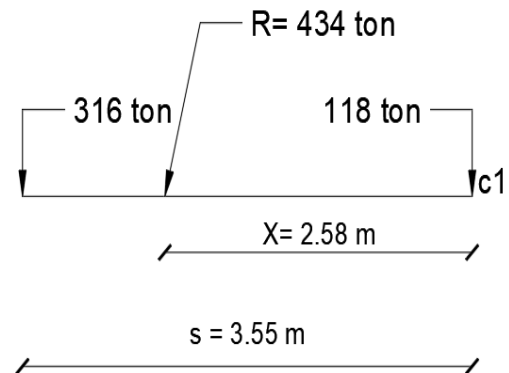
$$A_{pc} = 47 \text{ m}^2$$

$$L_{RC} = 2 \left( X + \frac{a_1}{2} \right) = 2 \left( 2.58 + \frac{0.09}{2} + 0.5 \right) = 6.2 \text{ m}$$

$$\therefore B_{RC} = 5.45 \text{ m}$$

$$\therefore B_{PC} = 5.45 + 2t_{PC} = 5.45 + 2 * 0.40 = 6.25 \text{ m}$$

$$L_{PC} = 6.2 + 2 * 0.40 = 7 \text{ m}$$



#### 2.4 - Check of Stress

- $q_{act} = \frac{434}{A_{pc}}$
- $q_{act} = \frac{434}{6.25 * 7} = 9.98 \text{ t/m}^2 < B.C = 10 \text{ t/m}^2$  , Ok safe

## 2.5 Design of Longitudinal direction

Calculate of actual uniform load on Rc Footing ( U.L ) as a beam :-

$$P2_{UL} = 316 * 1.5 = 474 \text{ ton}$$

$$P1_{UL} = 118 * 1.5 = 177 \text{ ton}$$

$$Pt_{UL} = 434 * 1.5 = 651 \text{ ton}$$

$$W_{UL} = \frac{Pt_{UL}}{L_{RC}} = \frac{651}{6.2} = 105 \text{ ton / m}$$

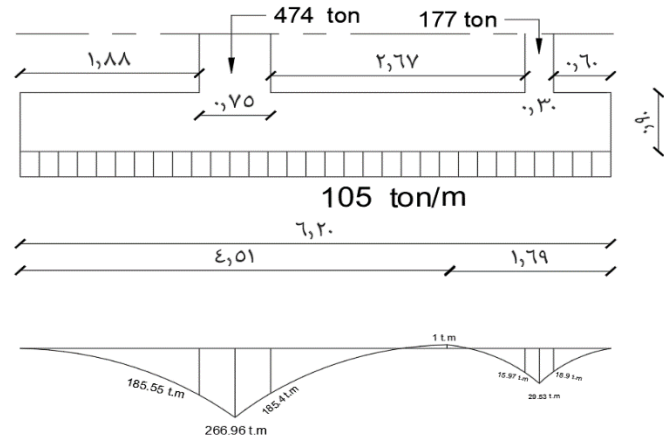


Figure 1.41 BMD

point of zero shear and drawing ( BMD ) :-

$$\begin{aligned} \text{zero shear : } 105 * y - 474 &= 0.0 \\ y &= 4.51 \text{ m} \end{aligned}$$

$$M1 = \frac{w * L * L}{2} = \frac{105 * 1.88 * 1.88}{2} = 185.55 \text{ t.m}$$

$$M2 = \frac{w * L * L}{2} = \frac{105 * 2.25 * 2.25}{2} = 266.96 \text{ t.m}$$

$$M3 = 105 * \frac{L * L}{2} - P2 * \frac{0.75}{2} = 105 * \frac{2.63 * 2.63}{2} - 474 * 0.375 = 185.4 \text{ t.m}$$

$$M_{\text{zero shear}} = ( 105 * \frac{Y * Y}{2} ) - ( P2 (0.375 + 1.83) ) = ( 105 * \frac{4.51 * 4.51}{2} ) - ( 474 * 2.255 ) = -1.014 \text{ t.m}$$

$$M4 = \frac{w * L * L}{2} - P1 * 0.15 = \frac{105 * 0.90 * 0.90}{2} - 177 * 0.15 = 15.975 \text{ t.m}$$

$$M5 = \frac{w * L * L}{2} = \frac{105 * 0.75 * 0.75}{2} = 29.53 \text{ t.m}$$

$$M6 = \frac{w * L * L}{2} = \frac{105 * 0.60 * 0.60}{2} = 18.9 \text{ t.m}$$

Use **M max = 266.96 t.m = 2669.6 KN.m**

$$d = K1 \sqrt{\frac{M_u}{B}} = 0.33 * \sqrt{\frac{266.96 * 10^5}{545}} = 73.05 \text{ cm} \Rightarrow d = 80 \text{ cm} \Rightarrow t = 90 \text{ cm}$$

## 2.6 - Check of Shear :-

$$Q_1 = 105 * 1.88 = 197.4 \text{ ton}$$

$$Q_2 = 105 ( 1.88 + 0.75 ) - 474 = -197.85 \text{ ton}$$

$$Q_3 = 105 ( 1.88 + 0.75 + 2.67 ) - 474 = 82.5 \text{ ton}$$

$$Q_4 = 105 ( 1.88 + 0.75 + 0.30 + 2.67 ) - 474 - 177 = -63 \text{ ton}$$

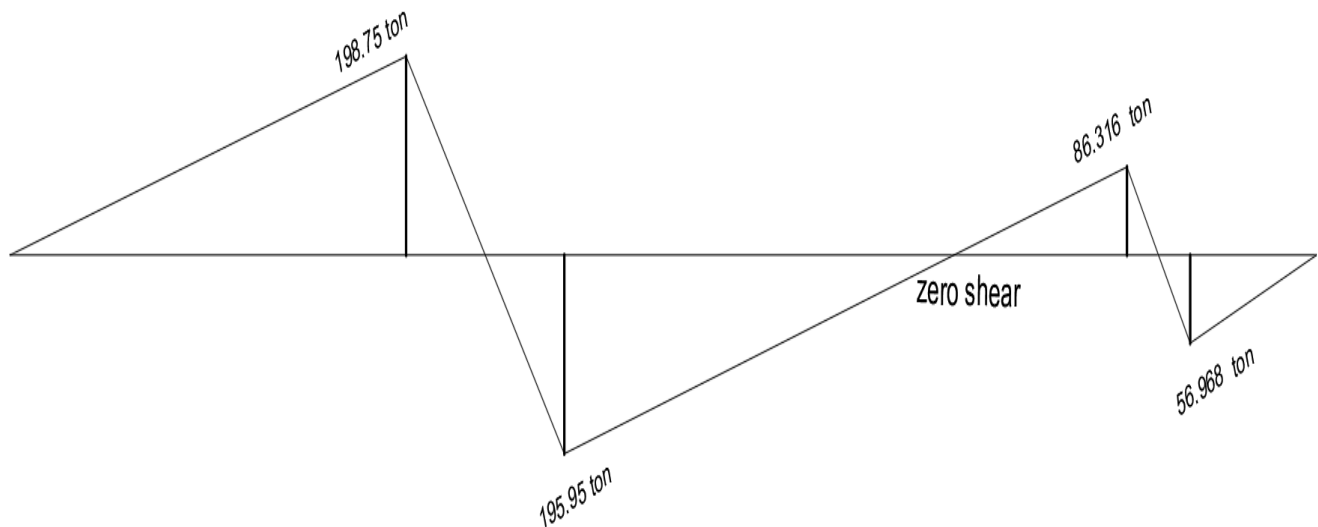


Figure 1.42. SFD

**USE Q Max = 197.85 ton**

**OR**

$$Q = 197.85 - W_{UL} * \frac{d}{2} = 197.85 - 105 * \frac{0.80}{2} = 155.85 \text{ ton} = 1558.5 \text{ KN}$$

$$q_{sh} = \frac{Q_{sh}}{B*d} = \frac{1558.4 * 10^3}{5450 * 800} = 0.35 \text{ KN / mm}^2$$

$$q_{sh(allow)} = 0.16 * \sqrt{\frac{F_{CU}}{\gamma_c}} = 0.16 * \sqrt{\frac{35}{1.5}} = 0.773 \text{ KN / mm}^2$$

$q_{sh} < q_{sh(allow)}$  OK .Safe

**OR**

$$q_{sh} = \frac{Q*1000}{B*d} = \frac{155.85*1000}{545*80} = 3.58 \text{ Kg/cm}^2 < 4.5 \text{ Kg/cm}^2 , \text{ ok safe}$$

## 2.7 - Check of punching :-

$$q_{act} = \frac{PUL}{B L} = \frac{6551}{6.2 \times 5.45} = 19.26 \text{ t / m}^2 = 192.6 \text{ KN/m}^2$$

### Column 1

$$P_u = 474 \text{ ton} = 4740 \text{ KN}$$

$$Q_p = P_{co} - q_{act} * (a+d) * (b+d)$$

- $Q_p = 4740 - 192.9 * (0.75 + 0.8) * (0.75 + 0.8) = 4276.5 \text{ KN}$
- $q_p = \frac{Q_p}{2((a+d)+(a+d))*d}$

$$q_p = \frac{4276.5 * 10^3}{2((750 + 800) + (750 + 800)) * 800} = 0.86 \text{ N/mm}^2$$

$$q_{p(allow)} = 1 - 1.7 \text{ N/mm}^2$$

$$2- 0.316 * (.5 + \frac{a}{b}) * \sqrt{\frac{F_{cu}}{\gamma_c}} = 0.316 * (0.5 + \frac{0.3}{0.65}) * \sqrt{\frac{35}{1.5}} = 1.46 \text{ N/mm}^2$$

$$3- 0.316 * \sqrt{\frac{F_{cu}}{\gamma_c}} = 0.316 * \sqrt{\frac{35}{1.5}} = 1.53 \text{ N/mm}^2$$

$$4- 0.8 * (\frac{\alpha * d}{2(a+d+b+d)} + .2) * \sqrt{\frac{F_{cu}}{\gamma_c}} =$$

$$= 0.8 * (\frac{4 * 800}{2(300 + 800 + 650 + 800)} + 0.2) * \sqrt{\frac{35}{1.5}} = 3.2 \text{ N/mm}^2$$

$$q_p < q_{p(allow)} \Rightarrow \text{OK, safe}$$

### Column 2

$$P_u = 177 \text{ ton} = 1770 \text{ KN}$$

$$Q_p = P_{co} - q_{act} * (a+d) * (b+d)$$

- $Q_p = 1770 - 192.9 * (0.30 + 0.8) * (0.65 + 0.8) = 1462.32 \text{ KN}$
- $q_p = \frac{Q_p}{2((a+d)+(a+d))*d}$
- $q_p = \frac{1462.32 * 10^3}{2((300 + 800) + (650 + 800)) * 800} = 0.358 \text{ N/mm}^2$



$$q_{p(allow)} = 1- 1.7 \text{ N/mm}^2$$

$$2- 0.316 * (0.5 + \frac{a}{b}) * \sqrt{\frac{F_{CU}}{\gamma_c}} = 0.316 * (0.5 + \frac{0.75}{0.75}) * \sqrt{\frac{35}{1.5}} = 2.289 \text{ N/mm}^2$$

$$3- 0.316 * \sqrt{\frac{F_{CU}}{\gamma_c}} = 0.316 * \sqrt{\frac{35}{1.5}} = 1.53 \text{ N/mm}^2$$

$$4- 0.8 * (\frac{\alpha * d}{2(a+d+b+d)} + 0.2) * \sqrt{\frac{F_{CU}}{\gamma_c}} =$$

$$= 0.8 * (\frac{4 * 800}{2(750 + 800 + 750 + 800)} + 0.2) * \sqrt{\frac{35}{1.5}} = 2.76 \text{ N/mm}^2 , q_p < q_{p(allow)} \text{ OK, safe}$$

## 2.8 – RFT

$$A_s = \frac{M_U}{F_y * J * d} = \frac{266.96 * 10^6}{360 * 0.826 * 800} = 1122.20 \text{ mm}^2 , 8 \text{ } \phi 16 / \text{m}$$

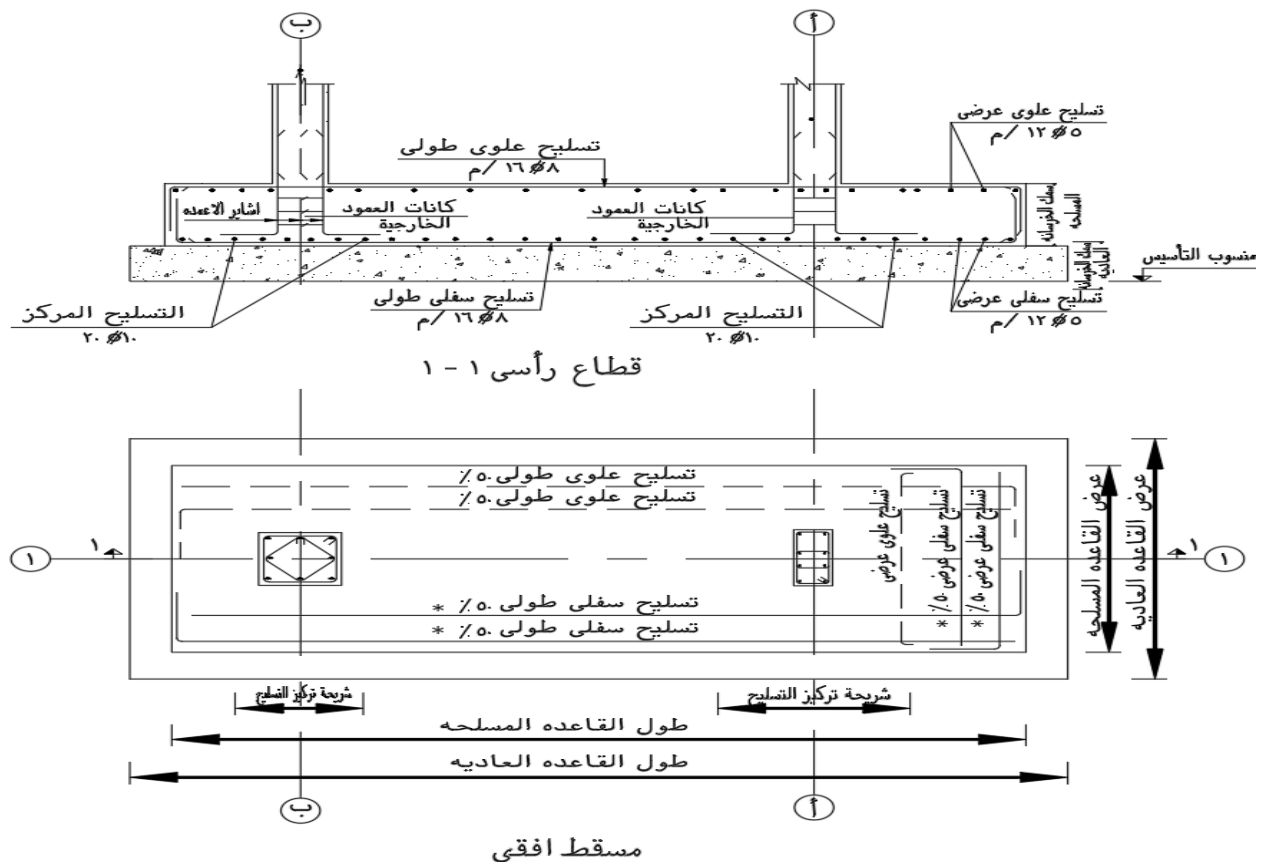


Figure 1.43. Sections of footing and details' of RFT

## 1.6.3 Table foundation

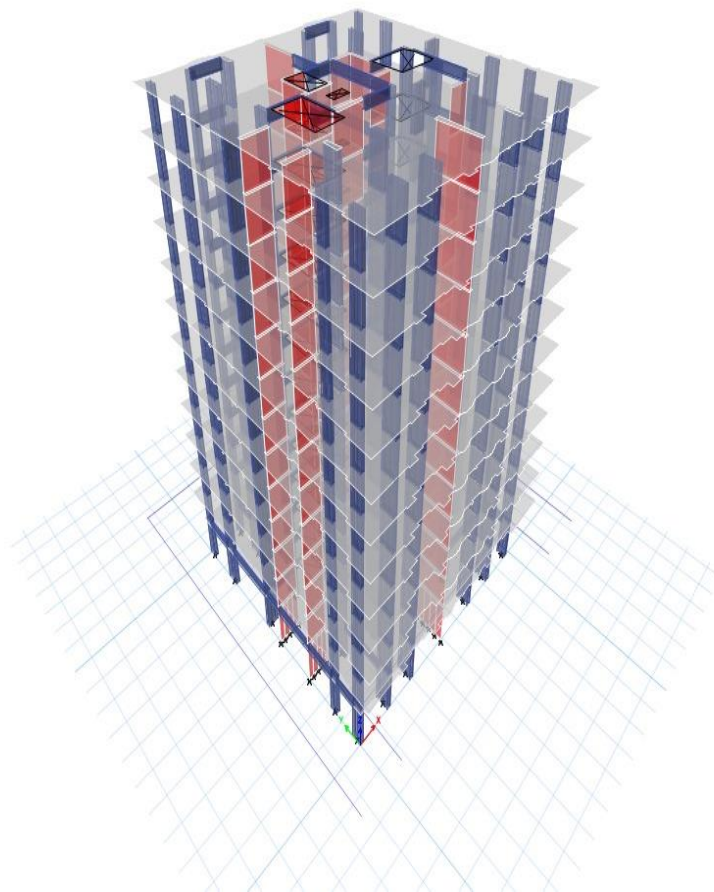
جدول القواعد على اجهاد لا يقل عن ١,٠٠ كجم/سم<sup>٢</sup>

| النموذج | الابعاد                   |  | الابعاد           |  | التسليح السفلي |            | التسليح العلوي |            |
|---------|---------------------------|--|-------------------|--|----------------|------------|----------------|------------|
|         | عادية                     |  | مسلحة             |  | قصير           | طويل       | قصير           | طويل       |
| ق ١     | خرسانة عادية بسمك = ٤٠ سم |  | ٠,٧ × ٢,٤ × ٢,٧٥  |  | ٩ / ١٢ Ø م     | ٩ / ١٢ Ø م | -----          | -----      |
| ق ٢     |                           |  | ٠,٧ × ٢,٦٥ × ٣,٢٥ |  | ٥ / ١٦ Ø م     | ٥ / ١٦ Ø م | -----          | -----      |
| ق ٣     |                           |  | ٠,٧ × ٣,٧٥ × ٣,٧٥ |  | ٩ / ١٦ Ø م     | ٩ / ١٦ Ø م | -----          | -----      |
| ق ٤     |                           |  | ٠,٧ × ٣,٣ × ٣,٣   |  | ٧ / ١٦ Ø م     | ٧ / ١٦ Ø م | -----          | -----      |
| ق ٥     |                           |  | ٠,٧ × ٣,٠ × ٣,٠   |  | ٩ / ١٢ Ø م     | ٩ / ١٢ Ø م | -----          | -----      |
| ق ٦     |                           |  | ٠,٩ × ٥,٤٥ × ٦,٢  |  | ٨ / ١٦ Ø م     | ٨ / ١٦ Ø م | ٥ / ١٢ Ø م     | ٨ / ١٦ Ø م |

| نموذج | ابعاد (م) |        | التسليح السفلي |      | التسليح العلوي |      | كانات    |
|-------|-----------|--------|----------------|------|----------------|------|----------|
|       | عرض       | ارتفاع | عدل            | مكسح | عدل            | مكسح |          |
| س ١   | ٠,٣٠      | ٠,٧٠   | ١٦ Ø ٣         | --   | ١٦ Ø ٣         | --   | ١٦ Ø ٦ م |
| س ٢   | ٠,٣٠      | ٠,٧٠   | ١٦ Ø ٤         | --   | ١٦ Ø ٤         | --   | ١٦ Ø ٦ م |

Figure 1.44 . Table foundation

## ***Unit (2) High Rise Building***



### 2.1 INTRODUCTION

#### 2.1.1 High Rise Building Consists of:

- Basement of (2.90) m height
- Ground Floor of (2.90) m height
- 11 Typical Floors each (2.90) m height

#### 2.1.2 Material Properties Used:

- $F_{cu} = 250 \text{ kg/cm}^2$
- $F_{y(\text{main steel})} = 3600 \text{ kg/cm}^2$
- $F_{y(\text{stirrups})} = 2400 \text{ kg/cm}^2$
- Weight of used brick =  $1500 \text{ kg/m}^3$
- Bearing Capacity of Soil =  $10 \text{ kg/cm}^2$

#### 2.1.3 Cover Thickness

- Slabs Cover = 2 cm
- Beams Cover = 2 cm
- Columns Cover = 2.5 cm
- Foundations Cover = 7 cm
- Stairs Cover = 2 cm
- Ramp Cover = 2 cm

#### 2.1.4 Loads Used:

- L.L= According to every Floor
- Cover = 0.15 ton
  - رمل تسوية بسمك 5 سم  $0.05 * 1.5$
  - مونة أسمنتية بسمك 1 سم  $0.01 * 2.1$
  - بلاط سيراميك بسمك 2 سم  $0.02 * 2.1$
  - محارة أسفل البلاط بسمك 2 سم  $0.02 * 2.1$
- Wall = According to every Floor
- D.L = Own weight + Covering Material + Wall Load

### **2.1.5 Design Method:**

- Ultimate limit state design

### **2.1.6 Computer Programs Used in Analysis :**

- (Etabs + Safe + SAP2000 + CSI Column + Excel)

### **2.1.7 Design Code:**

- Egyptian code of practice 2020

## 2.2 DESIGN OF SLABS:

### 2.2.1 Basement Slab: (Flat Slab System)

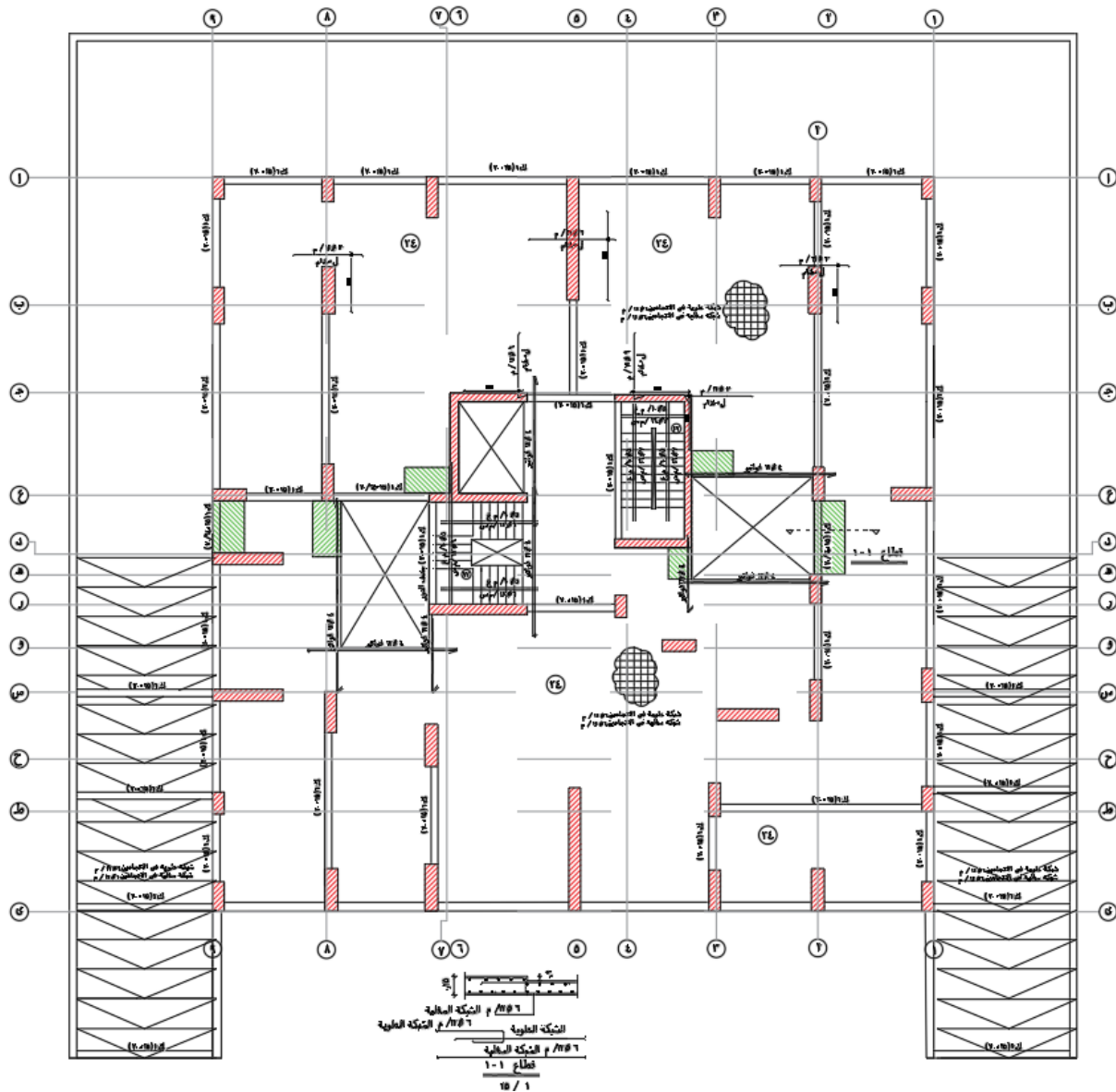


Figure 2.1 Statical System of Basement Roof

- ❖ Slab Thickness = 24 cm
- ❖ Own weight =  $0.24 \times 2.5 = 0.6 \text{ t/m}^2$ .
- ❖ Covering =  $150 \text{ kg/m}^2 = 0.15 \text{ t/m}^2$ .
- ❖ Live load =  $300 \text{ kg/m}^2 = 0.3 \text{ t/m}^2$
- ❖ Wall load =  $300 \text{ kg/m}^2 = 0.3 \text{ t/m}^2$ .

### Solving This flat slab By Using CSI Safe program:

- $D.L = O.W + W_{wall} + \text{Covering material}$   
 $= 0.6 + 0.3 + 0.15 = 1.05 \text{ t/m}^2$
- $L.L = 300 \text{ kg/cm}^2 = 0.3 \text{ t/m}^2$
- $W_u = 1.4 D.L + 1.6 L.L = 1.4 (1.05) + (1.6 * 0.3) = 1.95 \text{ t/m}^2$

### For ultimate design:-

- $As = \left[ \frac{Mu}{F_y * J * d} \right]$
- $M_u = As * F_y * J * d = 6 * \left( \frac{\pi * (1.2)^2}{4} \right) * 3600 * 0.85 * 23 * (10)^{-5}$
- $M(r) = 4.775 \text{ t.m} \Rightarrow \text{Use } 6 \text{ } \phi 12 / \text{m in each Direction}$
- Additional RFT (3  $\phi 12 / \text{m}$ ) and ( 6  $\phi 12 / \text{m}$  ) upper and lower

**In X-Direction: ( Upper and Lower ) :**

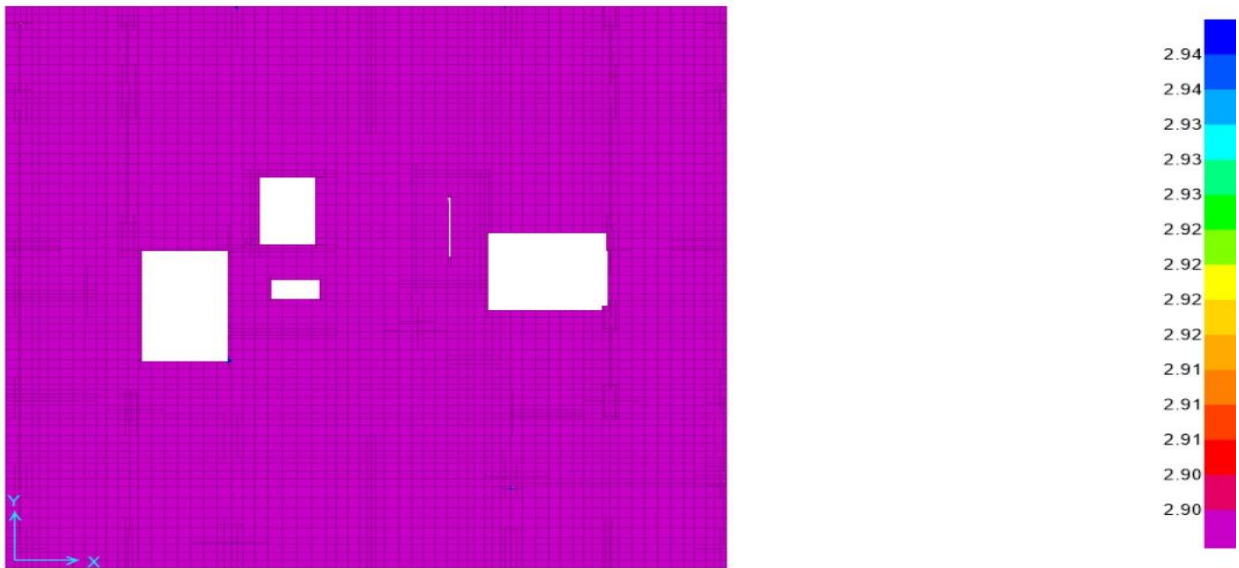


Figure 2.2 Additional Reinforcement in X-Direction ( Upper and Lower )

**In Y-Direction ( Upper and Lower )**

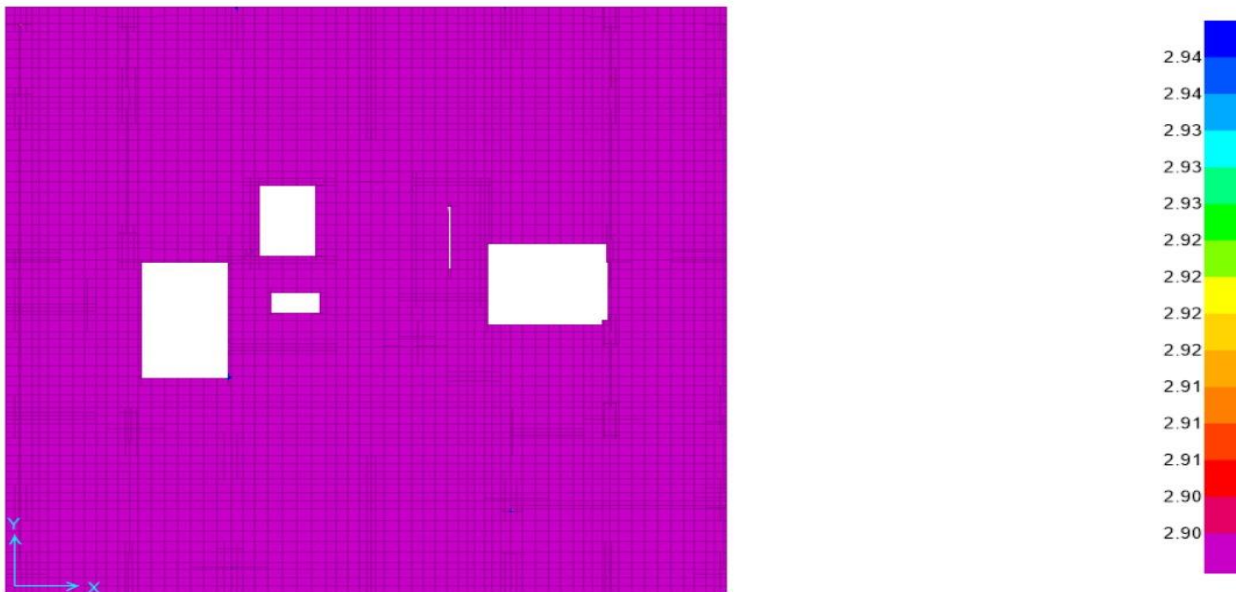


Figure 2.3 Additional Reinforcement in Y-Direction ( Upper and Lower )



### 2.2.1.1 Check for Long Term Deflection:

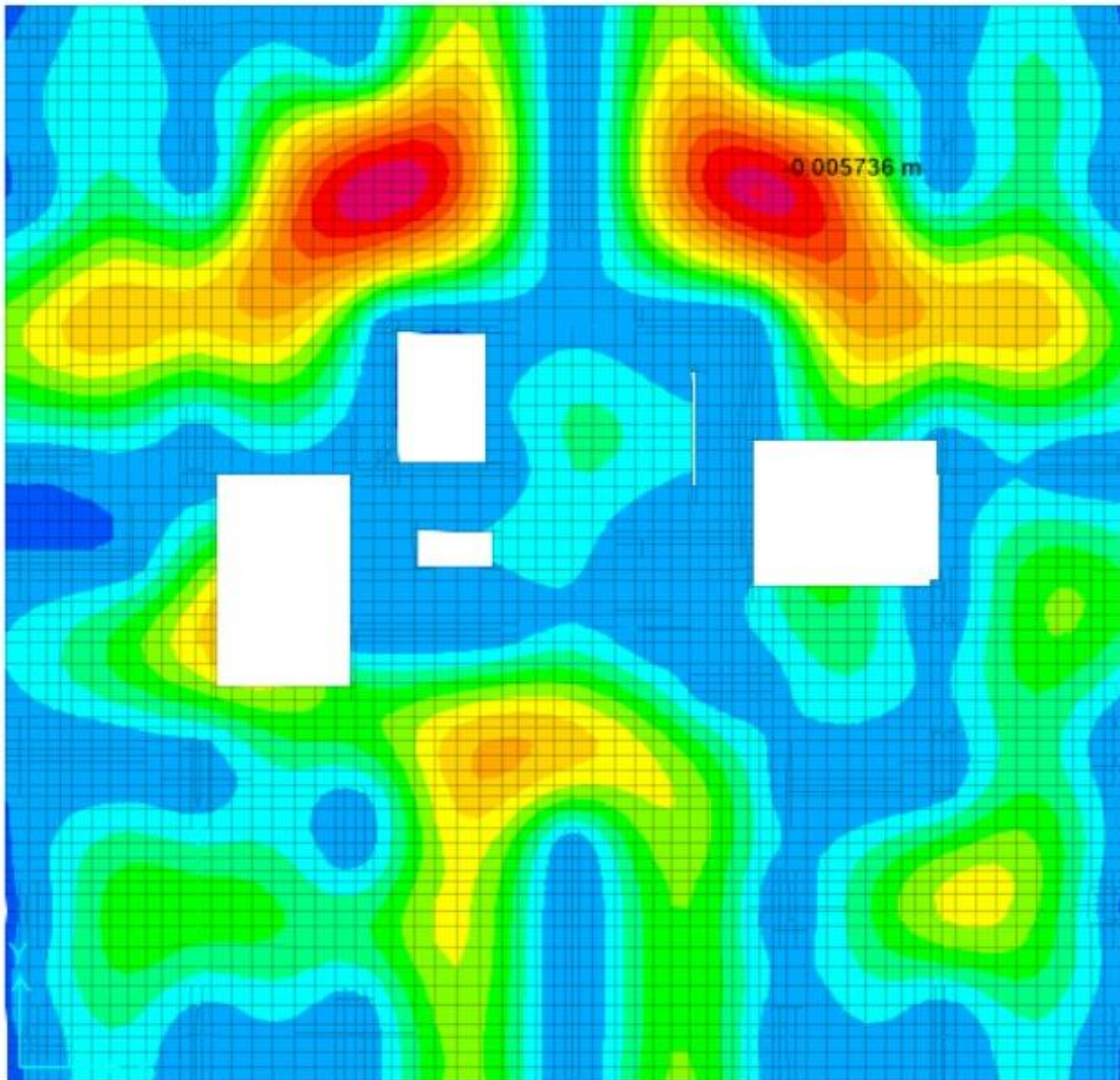


Figure2.4 Long Term Deflection

- From Code Check =  $L/250$
- Span for Check = 4.32 m
- Allowable Deflection = 0.017 m
- Maximum Deflection = - 0.00573

## 2.2.2 Ground Slab & Rebated (Flat Slab System)

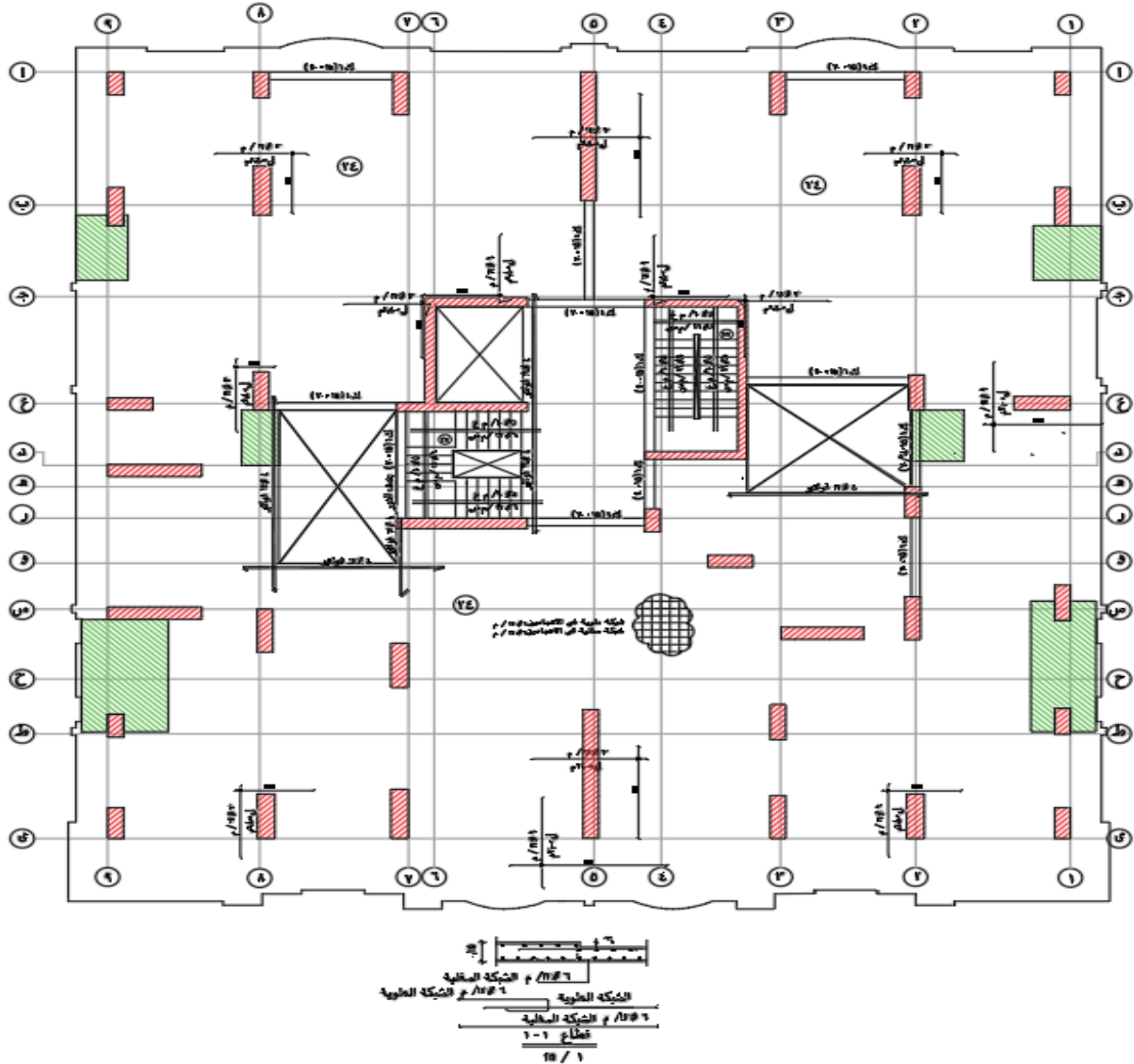


Figure 2.5 Statical System of Ground Roof

- ❖ Slab Thickness = 24 cm
- ❖ Own weight =  $0.24 \times 2.5 = 0.6 \text{ t/m}^2$ .
- ❖ Covering =  $150 \text{ kg/m}^2 = 0.15 \text{ t/m}^2$ .
- ❖ Live load =  $300 \text{ kg/m}^2 = 0.3 \text{ t/m}^2$
- ❖ Wall load =  $300 \text{ kg/m}^2 = 0.3 \text{ t/m}^2$ .

### Solving This flat slab By Using CSI Safe program:

- $D.L = O.W + W_{wall} + \text{Covering material}$   
 $= 0.6 + 0.3 + 0.15 = 1.05 \text{ t/m}^2$
- $L.L = 300 \text{ kg/cm}^2 = 0.3 \text{ t/m}^2$
- $W_u = 1.4 D.L + 1.6 L.L = 1.4 (1.05) + (1.6 * 0.3) = 1.95 \text{ t/m}^2$

### For ultimate design:-

- $A_s = \left[ \frac{Mu}{F_y * j * d} \right]$
- $M_u = A_s * F_y * j * d = 6 * \left( \frac{\pi * (1.2)^2}{4} \right) * 3600 * 0.85 * 23 * (10)^{-5}$
- $M(r) = 4.775 \text{ t.m} \Rightarrow \text{Use } 6 \text{ } \phi 12 / \text{m in each Direction}$
- Additional RFT (3  $\phi 12 / \text{m}$ ) & (6  $\phi 12 / \text{m}$ ) upper and lower

**In X-Direction ( Upper and Lower ) :**

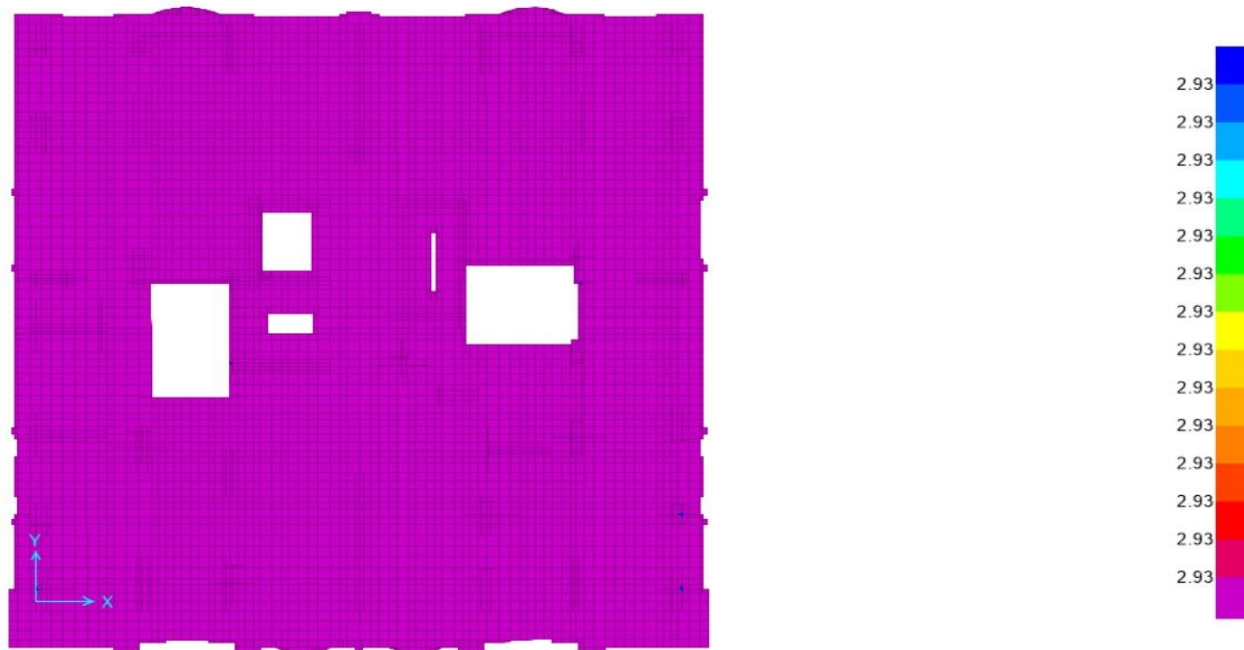


Figure 2.6 Additional Reinforcement in X-Direction ( Upper and Lower )

**In Y-Direction ( Upper and Lower )**

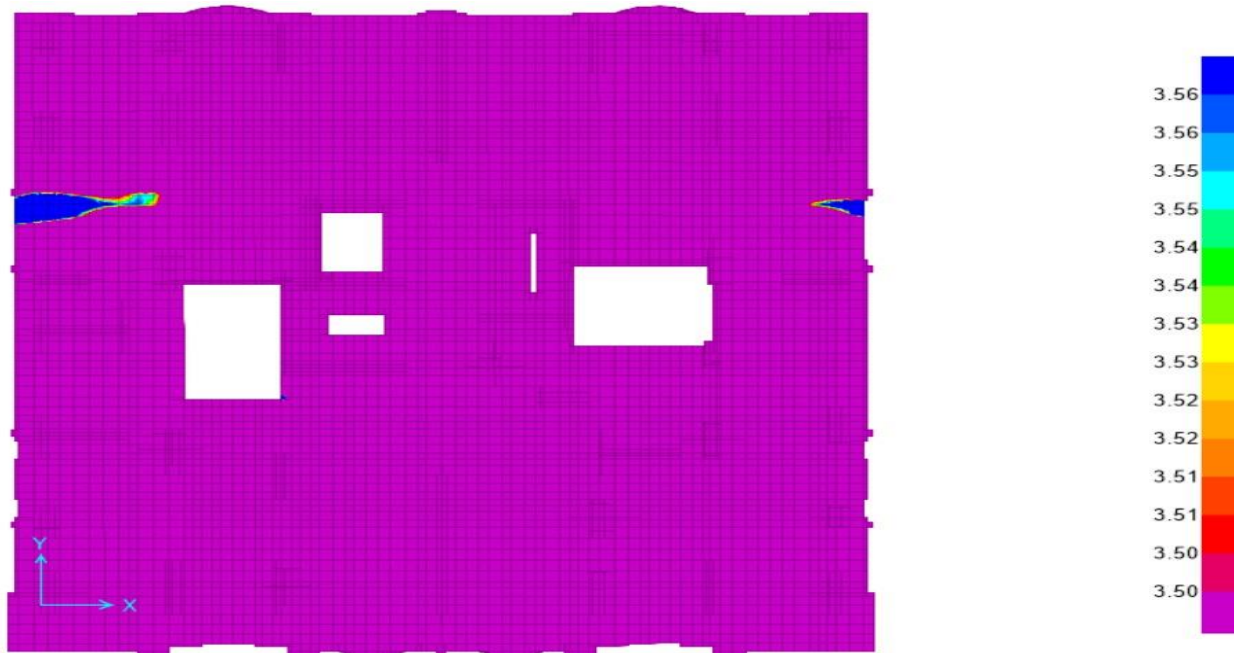


Figure 2.7 Additional Reinforcement in Y-Direction ( Upper and Lower )



### 2.2.2.1 Check for Long Term Deflection:

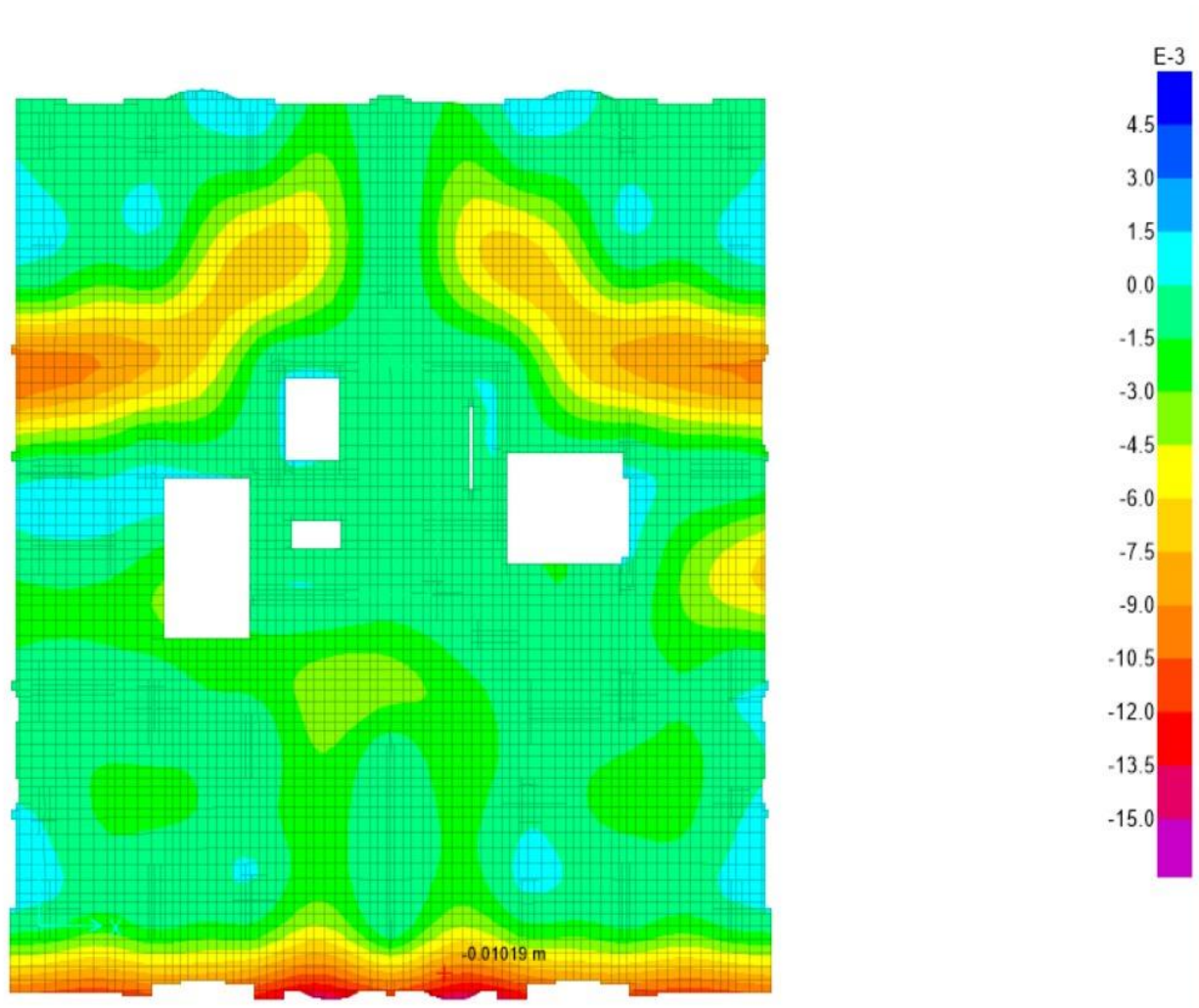


Figure 2.8 Long Term Deflection

- From Code Check =  $L/400$  **for cantiliver**
- From Code Check =  $L/250$
- Span for Check = 2.05 m **for cantiliver**
- Span for Check = 2 m
- Allowable Deflection = 0.01812 m
- Allowable Deflection = 0.005 m **for cantiliver**
- Maximum Deflection = -0.01019 m

**Ok use camper At cantilever 1 cm each 1 m**

## 2.2.3 Check of Punching Shear: (Basement Roof)

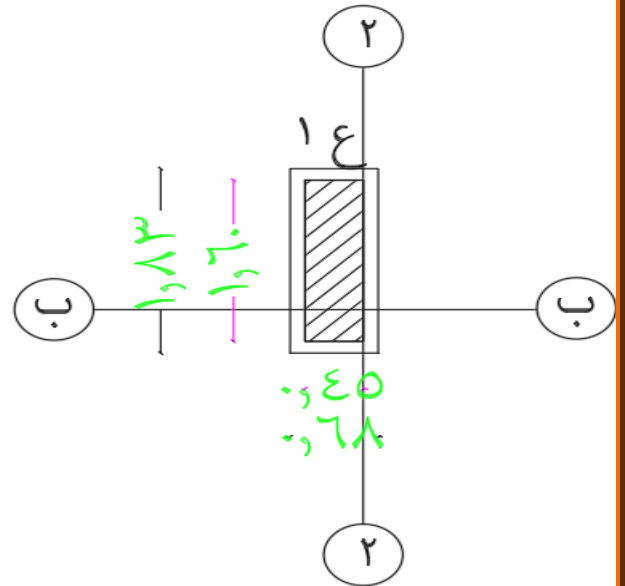
### 2.2.3.1 Interior Column (C1 = 45\*160) on ( 2 - ب ) Axis:

- ❖ Slab Thickness = 24 cm
- ❖ Own weight =  $0.24 * 2.5 = 0.6 \text{ t/m}^2$ .
- ❖ Covering =  $150 \text{ kg/m}^2 = 0.15 \text{ t/m}^2$ .
- ❖ Live load =  $300 \text{ kg/m}^2 = 0.3 \text{ t/m}^2$
- ❖ Wall load =  $300 \text{ kg/m}^2 = 0.3 \text{ t/m}^2$ .

### Solving This flat slab

- $D.L = O.W + W_{\text{wall}} + \text{Covering material}$   
 $= 0.6 + 0.3 + 0.15 = 1.05 \text{ t/m}^2$
- $L.L = 300 \text{ kg/cm}^2 = 0.3 \text{ t/m}^2$
- $W_u = 1.4 D.L + 1.6 L.L = 1.4 (1.05) + (1.6 * 0.3) = 1.95 \text{ t/m}^2 = 19.5 \text{ KN/m}^2$
- 
- $d = t_s - 20 \text{ mm} = 250 - 20 = 230 \text{ mm} = 0.23 \text{ m}$
- $b_o = 2 * (1830 + 680) = 5020 \text{ mm}$
- $Q_{up} = W_u (L_1 * L_2 - A_p) = 1.95 (2.2 * 1.8 - 1.83 * 0.68) = 5.295 \text{ t/m}^2$
- $q_{up} = \frac{Q_{up}}{b_o * d} * \beta = \frac{52.95 * 1000}{5020 * 230} * 1.15 = 0.052 \text{ N/mm}^2$
- $q_{cup} = \text{the least of:-}$ 
  - $1.70 \text{ N/mm}^2$
  - $0.316 \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.316 \sqrt{\frac{30}{1.5}} = 1.41 \text{ N/mm}^2$
  - $0.316 \left( \frac{a}{b} + 0.5 \right) \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.316 \left( \frac{50}{160} + 0.5 \right) \sqrt{\frac{30}{1.5}} = 1.148 \text{ N/mm}^2$
  - $0.8 \left( \frac{\alpha * d}{b_o} + 0.2 \right) \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.8 \left( \frac{4 * 230}{5020} + 0.2 \right) \sqrt{\frac{30}{1.5}} = 1.371 \text{ N/mm}^2$
- $q_{up} = 0.052 \text{ N/mm}^2 \leq q_{cup} = 1.148 \text{ N/mm}^2$

OK safe punching



## 2.4 Design of Stairs Three Flight Stair Axis (ξ – 5)

### 2.4.1 Manual solution

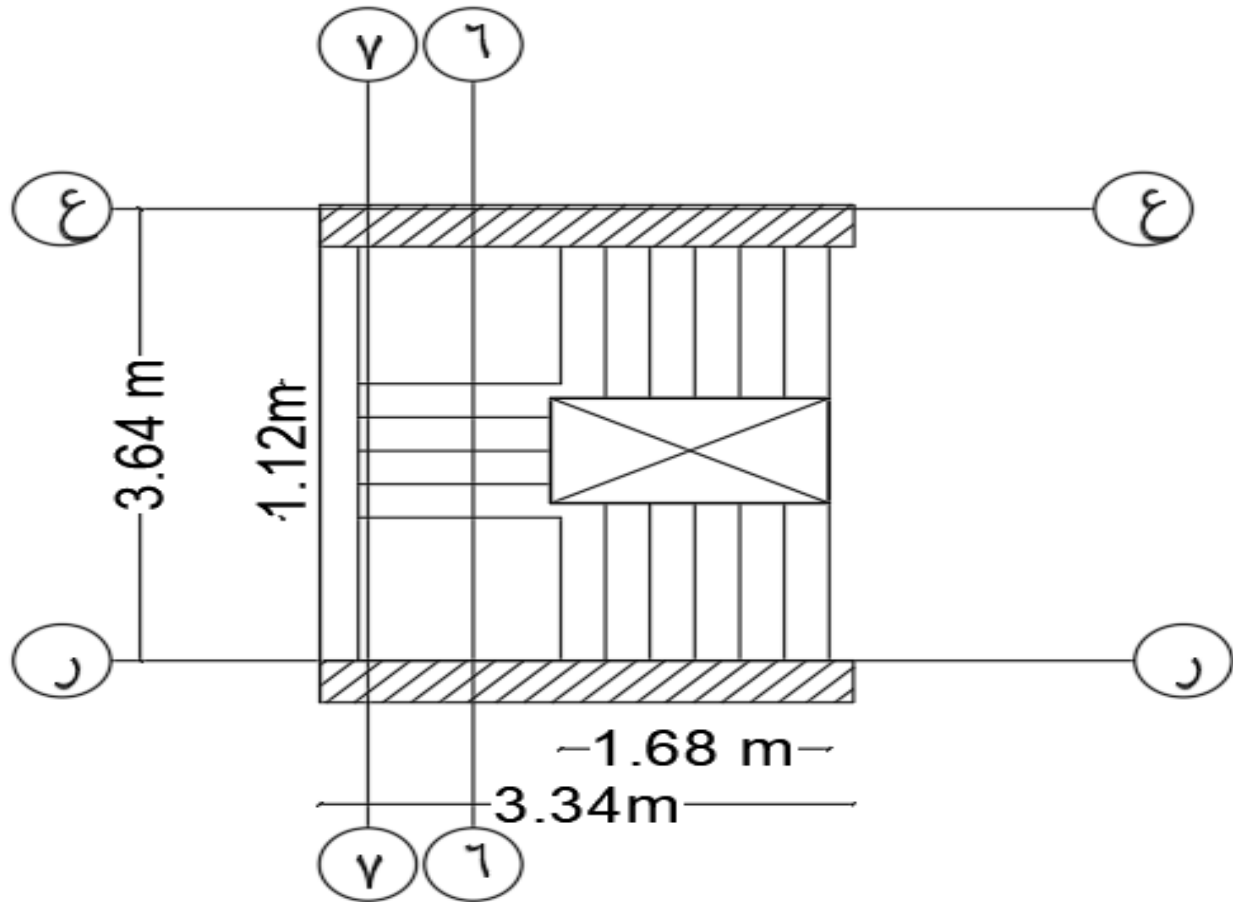


Figure 2.9 Stair Cross Section

#### 1) Dimensions:

- $t_s = \frac{Span}{24:30} = 0.14 \text{ m}$
- $H_{Story} = 2.90 \text{ m}$
- $Rise = 0.15 \text{ m}$
- $Going = 0.30 \text{ m}$
- $\theta = \tan^{-1}\left(\frac{0.15}{0.30}\right) = 26.56^\circ$
- $t^* = \frac{t_s}{\cos \theta} = \frac{14}{\cos(26.56)} = 15.6 \text{ cm}$
- $t_{av} = t^* + \frac{Rise}{2} = 23.1 \text{ cm}$

### 2) Loads:

- $W_{u \text{ flight}} = 1.4 \text{ D.L} + 1.6 \text{ L.L}$   
$$= 1.4 (25 * 0.231 + 1.0) + 1.6 (3)$$
$$= 14.30 \text{ KN/m}^2$$
- $W_{u \text{ landing}} = 1.4 \text{ D.L} + 1.6 \text{ L.L}$   
$$= 1.4 (25 * 0.14 + 1.5) + 1.6 (3)$$
$$= 11.80 \text{ KN/m}^2$$

### 3) For Strips

Shown In The figure

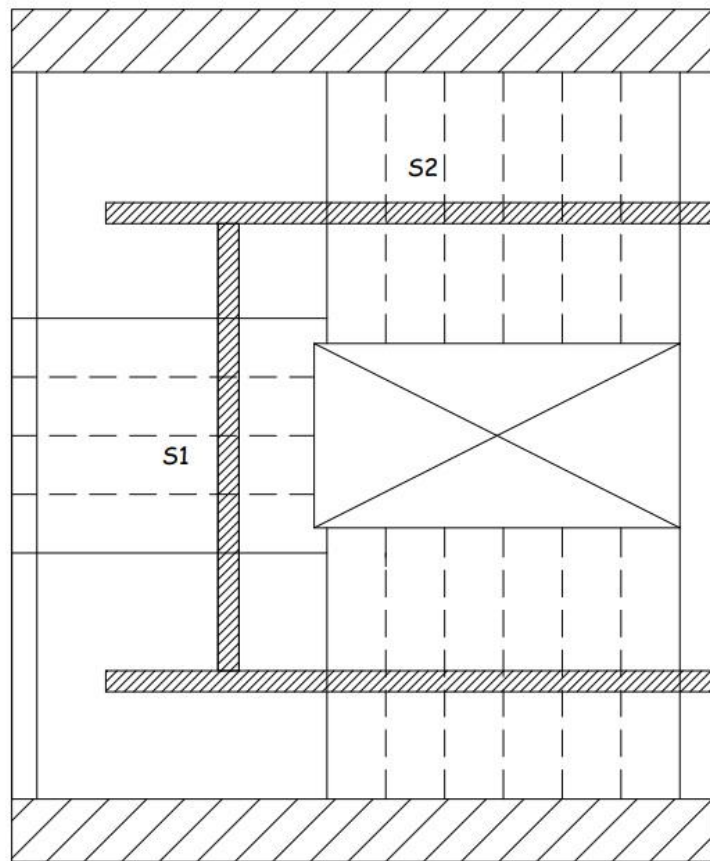


Figure 2.10 Strips Of Stair



## For Strip (F1):

- $R_1 = 8.01 \text{ KN}$
- $M_{u1} = 8.01 \cdot 1.15 - 14.30 \cdot 0.56^2 / 2 = 6.69 \text{ KN.m}$

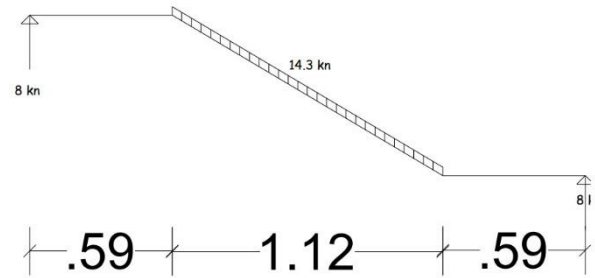


Figure 2.11 Strip (F1) of Stair

## For Strip (F2):

- $R_2 \cdot 3.93 = 11.8 \cdot 1.5 \cdot (.75 + 1.68 + .75) + 10.17 \cdot 1.5 \cdot (.75 + 1.68 + .75) + (14.3 \cdot 1.68 \cdot 1.59)$

$$R_2 = 36.38 \text{ Kn}$$

- $R_2 \cdot 3.93 = (10.17 \cdot 1.5 \cdot .75) + (11.8 \cdot 1.5 \cdot .75) + (14.3 \cdot 1.68 \cdot 1.59)$

$$R_3 = 20.6 \text{ Kn}$$

- $M_{u2} = 36.38 \cdot 1.5 - 10.17 \cdot 1.5 \cdot .75 + 11.8 \cdot 1.5 \cdot .75 = 29.85 \text{ k.m}$
- $M_{u3} = 20.6 \cdot 0.75 = 15.45 \text{ Kn.m}$
- $M_{u4} = \frac{29.85 + 15.45}{2} = 22.65$

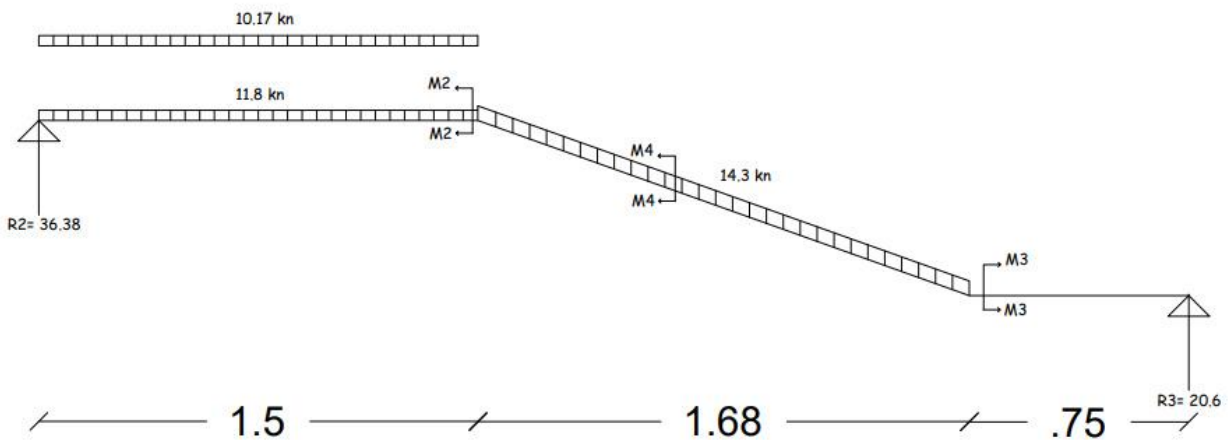


Figure 2.12 Strip (F2) of Stair

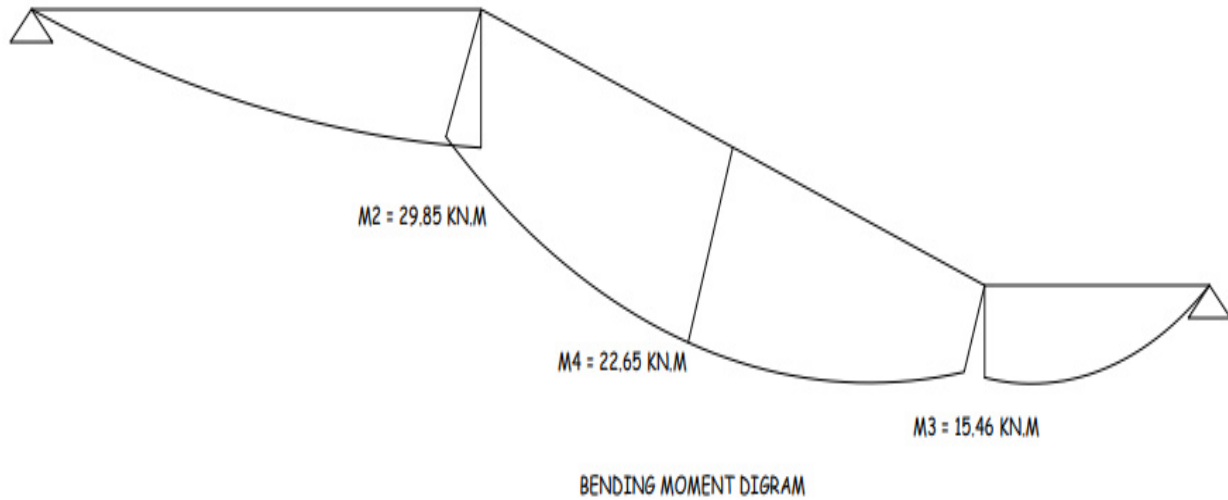


Figure 2.13 BMD

#### 4) Design of Section of Flight :

- $M_{u4} = 22.65 \text{ kN.m}$
- $d = c_1 \sqrt{\frac{MU}{FCU * b}} \Rightarrow 120 = c_1 \sqrt{\frac{22.65 * 10^6}{30 * 1000}} \Rightarrow c_1 = 4.36 \Rightarrow J = .826$
- $A_s = \frac{M_u}{F_y * j * d} = \frac{22.65 * 10^6}{350 * .826 * 120} = 652.88 \text{ mm}^2$
- Use  $A_s \Rightarrow 6 \text{ } \phi 12 / \text{m}$

#### 4) Design of Section of STRIP 1 :

$$A_s = \frac{M_u}{F_y * j * d} = \frac{6.69 * 10^6}{350 * .826 * 120} = 192.83 \text{ mm}^2$$

5  $\phi 12 / \text{m}$

**2.3.2 Using Sap Program**

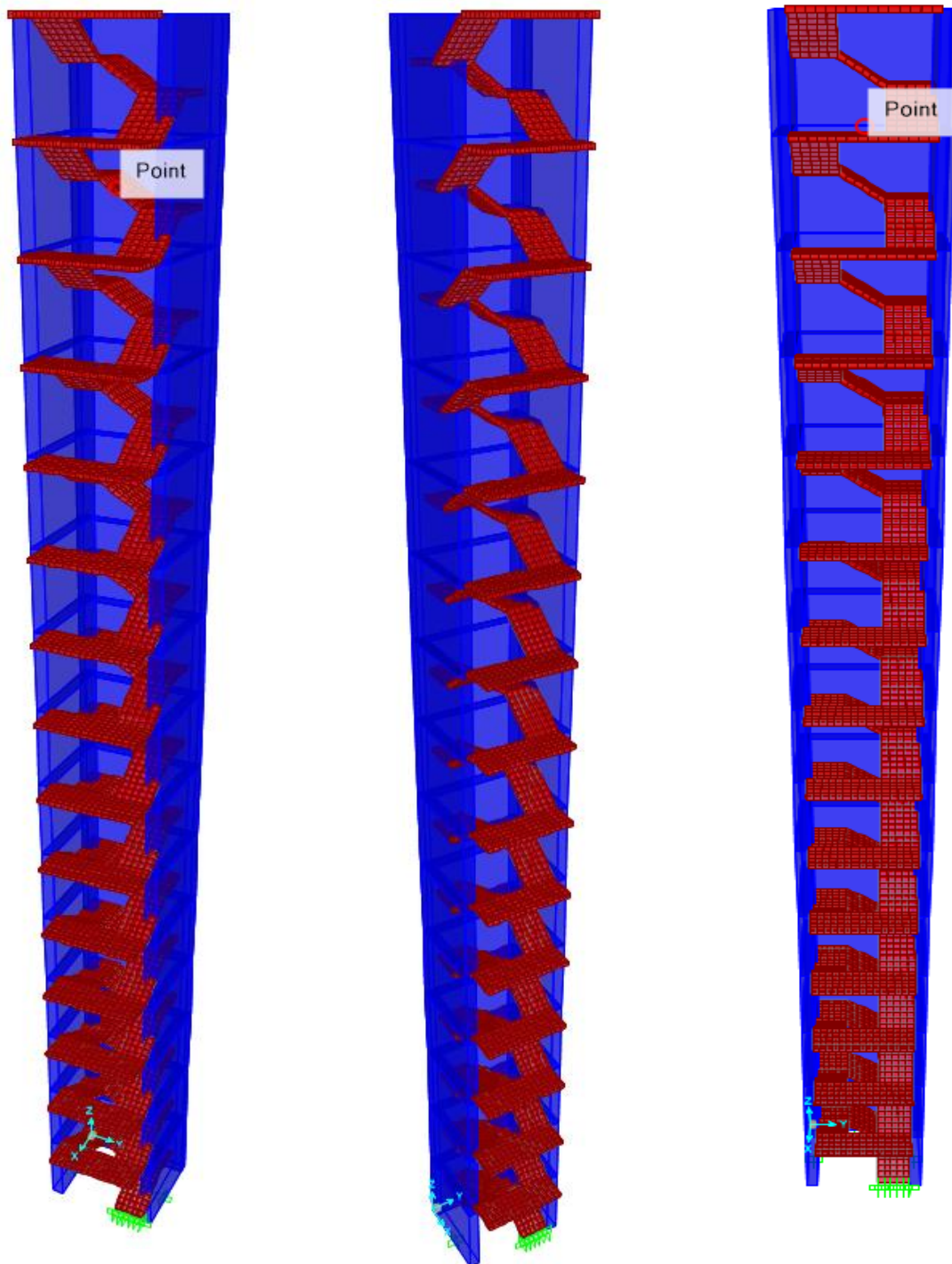


Figure 2.14 Stair 3D

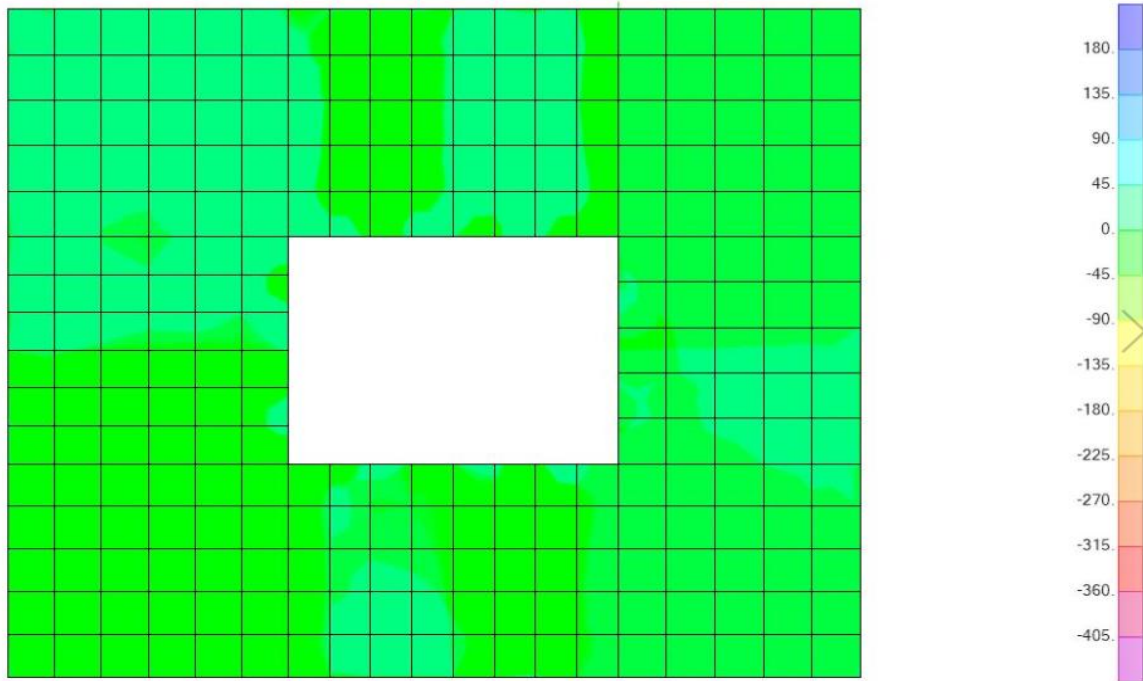


Figure 2.15 Bending Moment In X-Direction

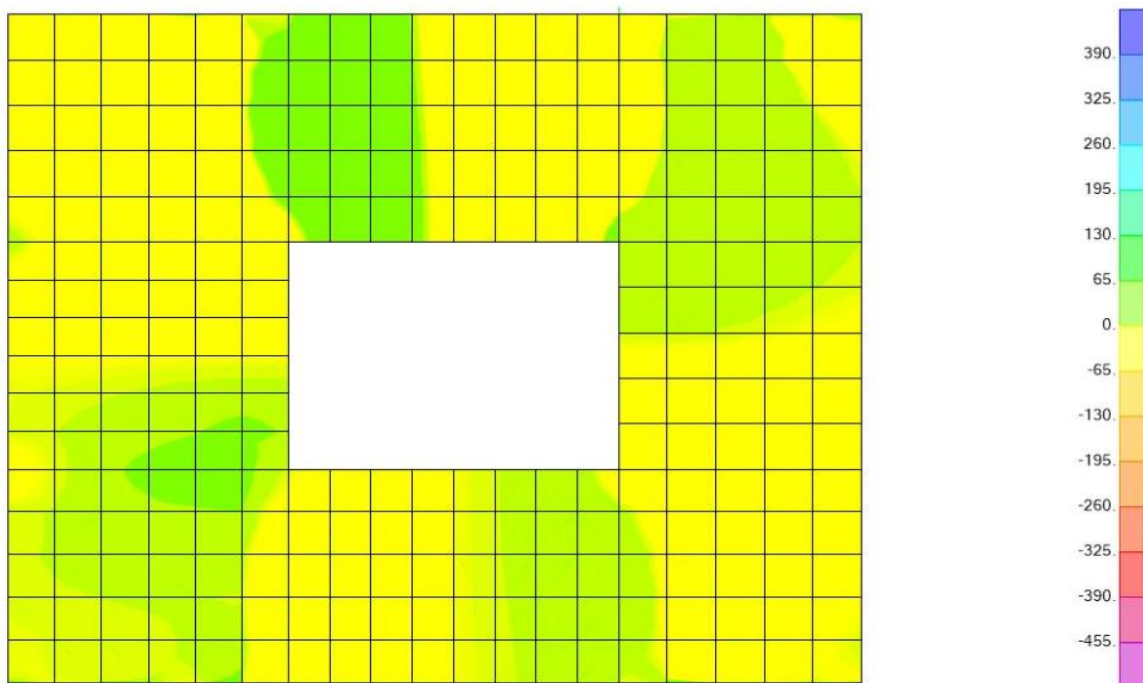
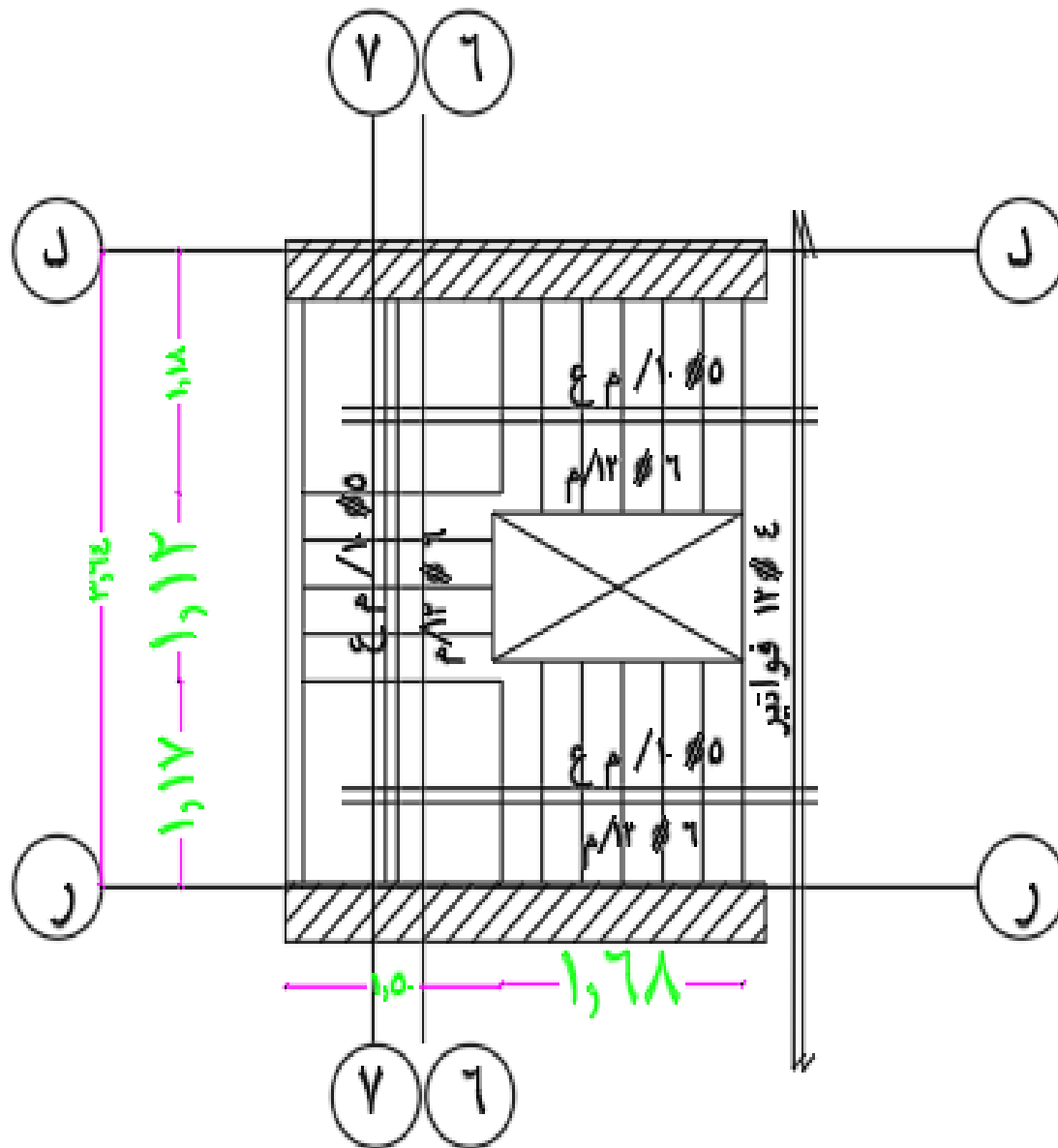


Figure 2.16 Bending Moment In Y-Direction



مسقط انشائي لسلم المتكرر

Figure 2.17 Stair Reinforcement in plan

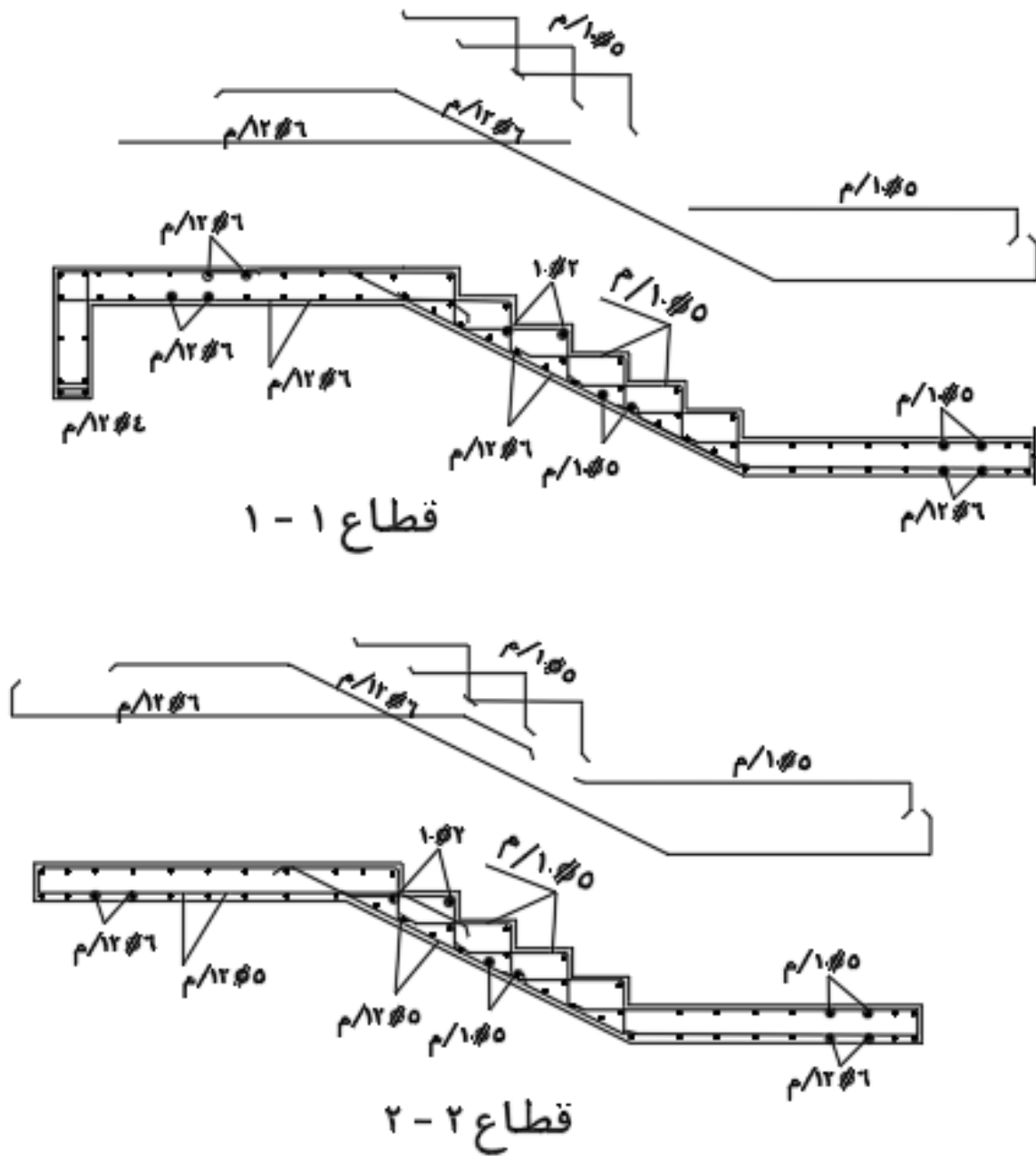


Figure 2.18 Reinforcement in sec1-1 and sec 2-2

## 2.4 Design of Stairs ( Two Flight Stair Axis 3- ξ )

### 2.4.2 Manual solution

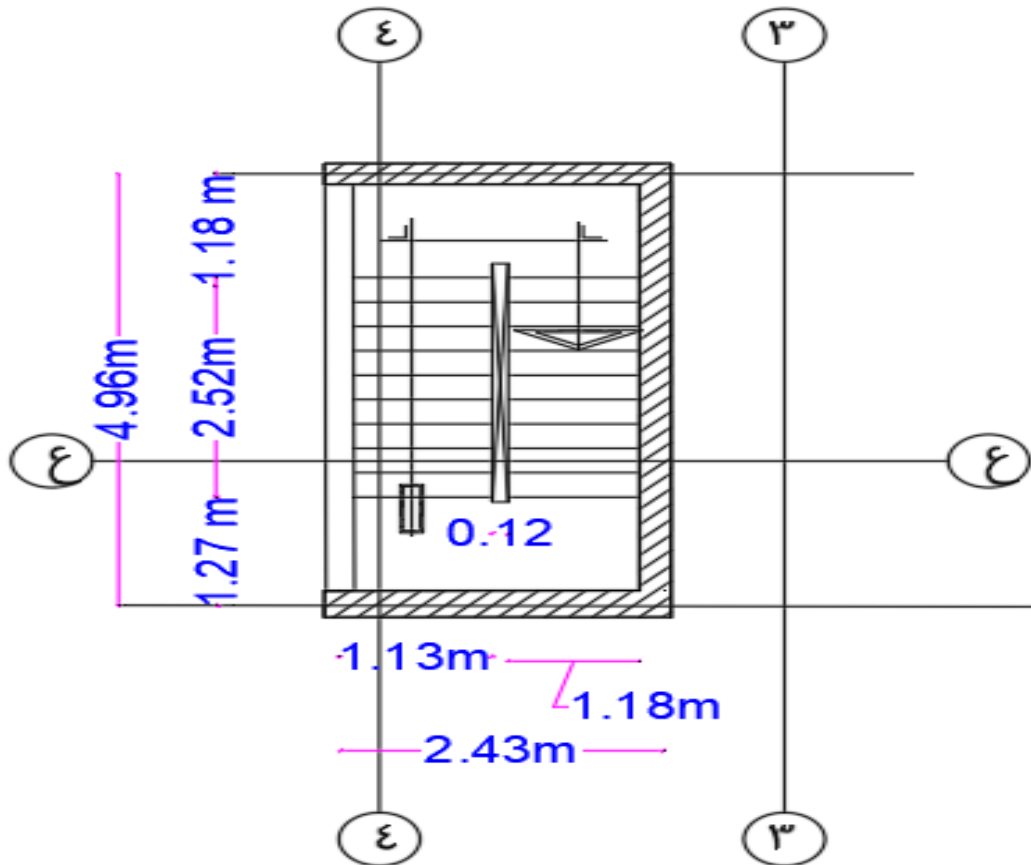


Figure 2.19 Stair Cross Section

#### 1) Dimensions:

- $t_s = \frac{Span}{24:30} = 0.22 \text{ m}$
- $H_{Story} = 2.90 \text{ m}$
- $Rise = 0.15 \text{ m}$
- $Going = 0.30 \text{ m}$
- $\theta = \tan^{-1}\left(\frac{0.15}{0.30}\right) = 26.56^\circ$
- $t^* = \frac{t_s}{\cos \theta} = \frac{16}{\cos(26.56)} = 17.88 \text{ cm}$
- $t_{av} = t^* + \frac{Rise}{2} = 25.38 \text{ cm}$

### 2 ) Loads:

- $W_{su} = 1.4 \text{ D.L} + 1.6 \text{ L.L}$   
 $= 1.4 (25 \times 0.2538 + 1.0) + 1.6(3)$   
 $= 15 \text{ KN/m}^2$

- $W_{u \text{ landing}} = 1.4 \text{ D.L} + 1.6 \text{ L.L}$   
 $= 1.4 (25 \times 0.16 + 1.5) + 1.6 (3)$   
 $= 12.5 \text{ KN/m}^2$

### 3) For Strips

Shown In The figure

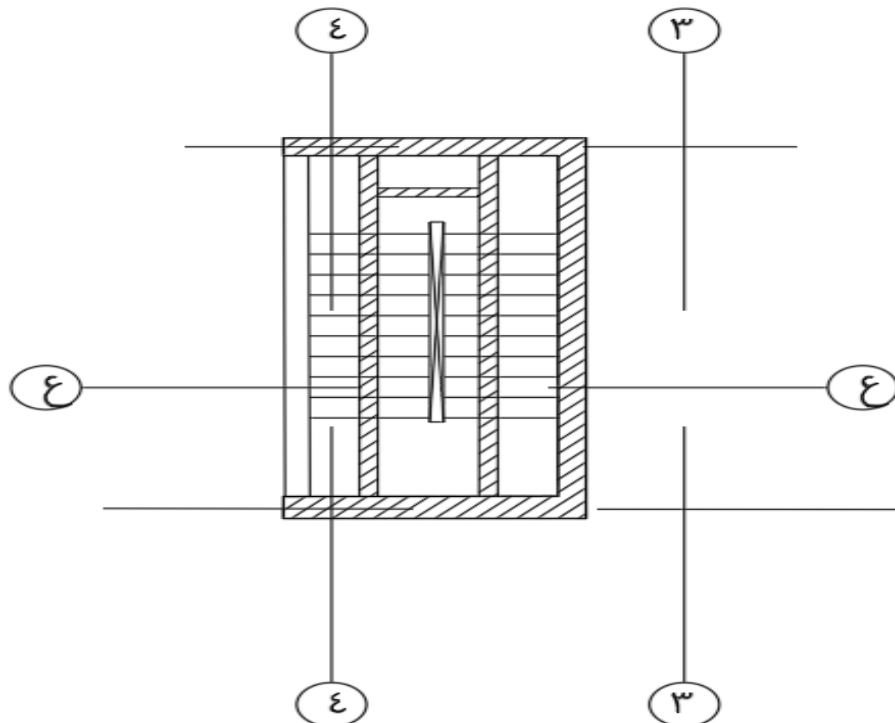


Figure 2.20 Strips Of Stair



## For Strip (F1):

- $R_1 = 12.5 * \frac{1.25}{2} * 2 + 15 * \frac{2.52}{2} = 34.52 \text{ KN}$
- $M_{u1} = 34.52 * 2.52 - 15 * \frac{2.52^2}{2} - 12.5 * 1.25 * 1.885 = 45.28 \text{ KN.m}$

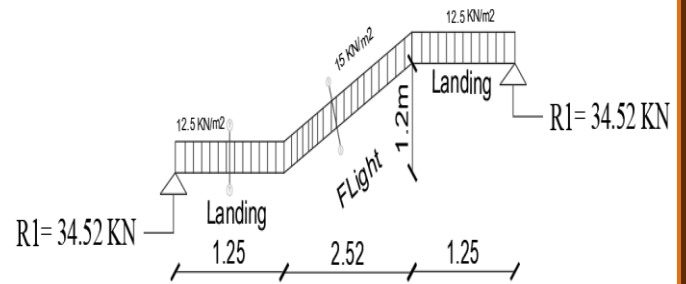


Figure 2.21 Strip (F1) of Stair

## Moment :-

.  $M1 = M3$

$$R_1 * 1.25 - 12.5 * \frac{1.25 * 1.25}{2} = 34.52 * 1.25 - 12.5 * \frac{1.25 * 1.25}{2} = 33.38 \text{ KN.m}$$

.  $M2 = M4$

$$R_1 (1.25 + 2.52) - 12.5 * 1.25 \left( \frac{1.25}{2} + 2.52 \right) - 15 * 2.52 * \frac{2.52}{2} = 34.52 (3.77) - 12.5 * 1.25 \left( \frac{1.25}{2} + 2.52 \right) - 15 * 2.52 * \frac{2.52}{2} = 33.37 \text{ KN.m}$$

.  $Mu1$

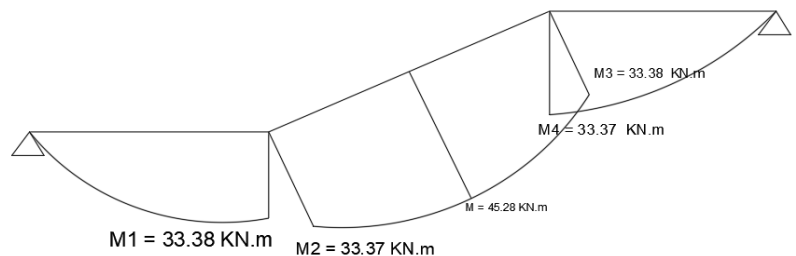
$$M_{u1} = 34.52 * 2.52 - 15 * \frac{2.52^2}{2} - 12.5 * 1.25 * 1.885 = 45.28 \text{ KN.m}$$

## 4) Design of Section of Flight :

- $M_{u1} = 45.28 \text{ KN.m}$
- $d = c_1 \sqrt{\frac{MU}{FCU * b}} \Rightarrow 140 = c_1 \sqrt{\frac{45.28 * 10^6}{30 * 1000}}$
- $c_1 = 3.60 \Rightarrow J = 0.786$

$$A_s = \frac{M_u}{F_{y_s} * j * d} = \frac{45.28 * 10^6}{360 * 0.786 * 140} = 1156.14 \text{ mm}^2$$

- Use  $A_s \Rightarrow 8 \text{ } \phi 16 / \text{m}$



### 2.4.3 Using Sap Program

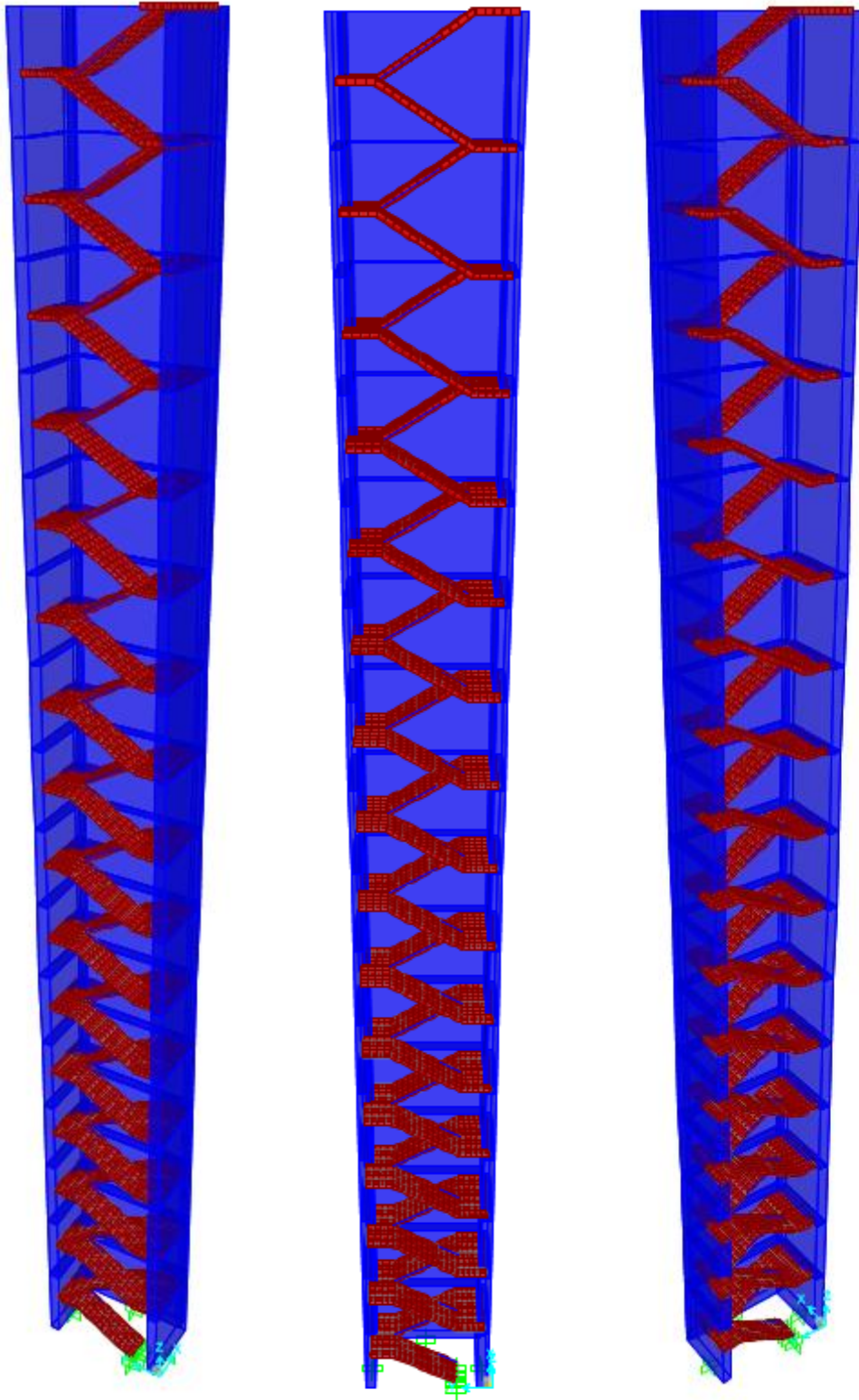


Figure 2.22 Stair 3D

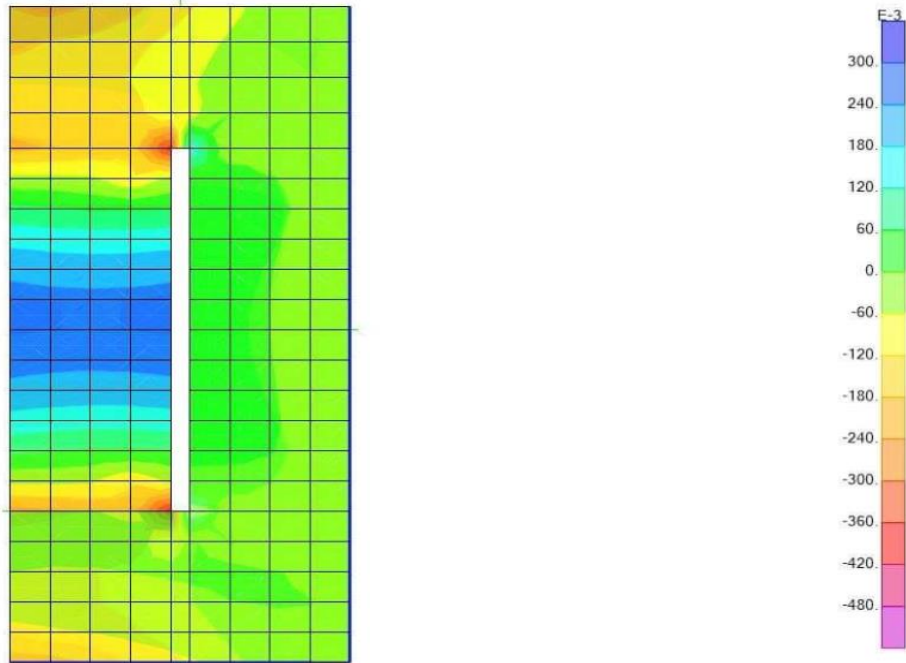


Figure 2.23 Bending Moment In X-Direction

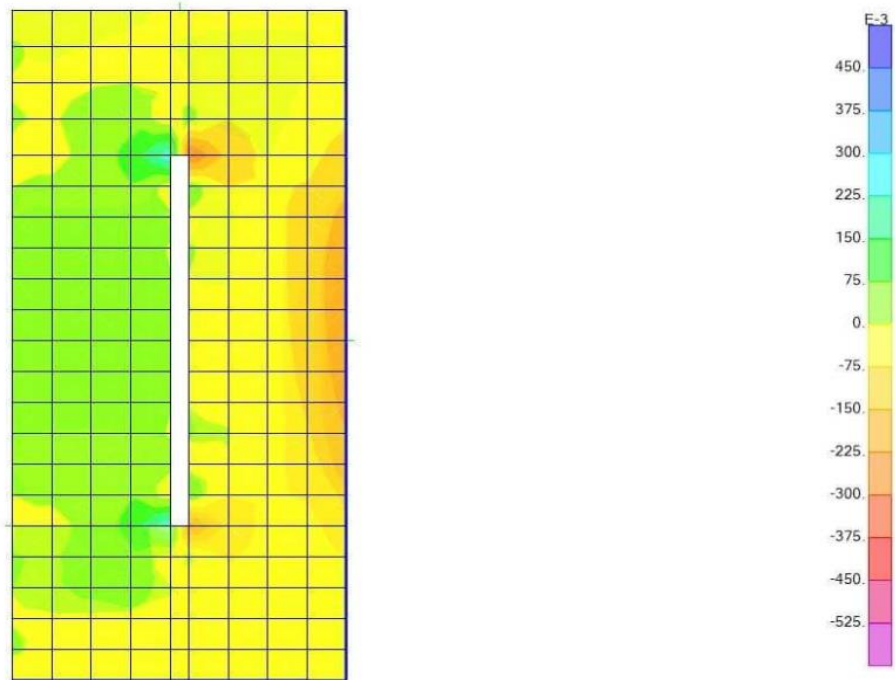


Figure 2.24 Bending Moment In Y-Direction

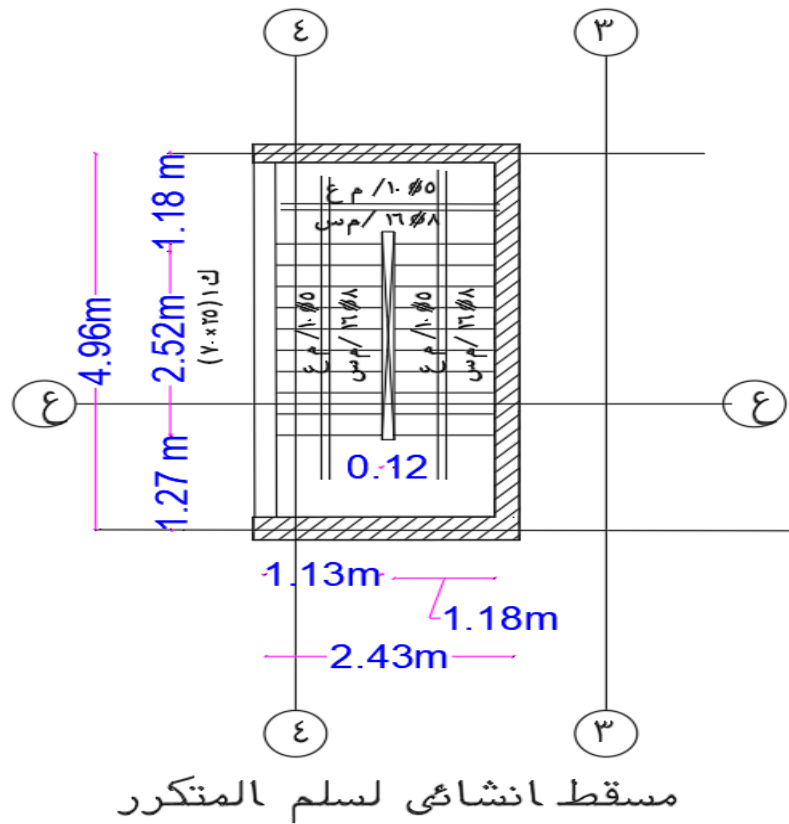


Figure 2.25 Stair Reinforcement in plan

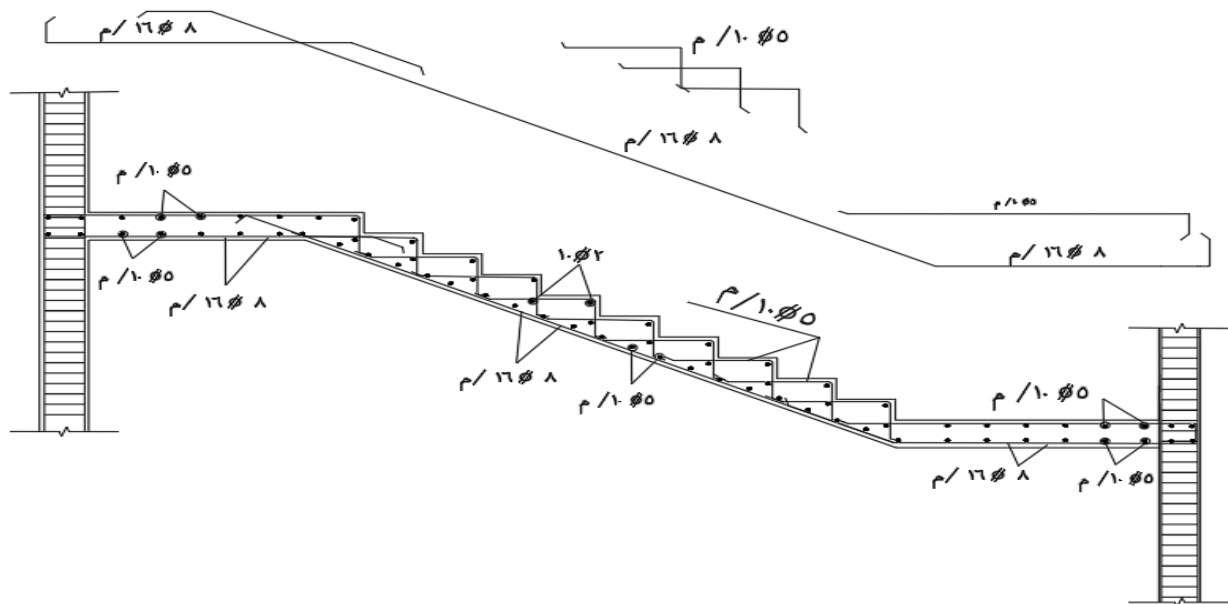


Figure 2.26 Reinforcement details

## 2.5 Design of Beams

### Concrete dimensions

- ❖ Assume  $b = 25 \text{ cm}$
- ❖  $t = \frac{L}{12} = 80 \text{ cm}$
- ❖ **Loads on beams**
  - ❖ Own weight
  - ❖ Load from slab
  - ❖ Load from wall
- ❖  $A_s = \frac{M_u}{F_y * j * d}$  USE 2  $\varnothing 16 / \text{m}$  ( 1 ك ) and 4  $\varnothing 16 / \text{m}$  ( 2 ك )

### 2.5.1 Basement Slab Beams :

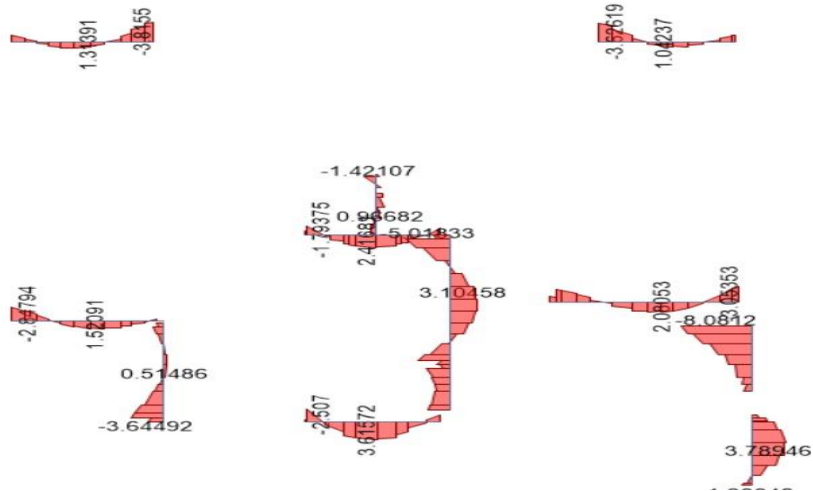


Figure 2.27 Moment Of Basement Slab Beams

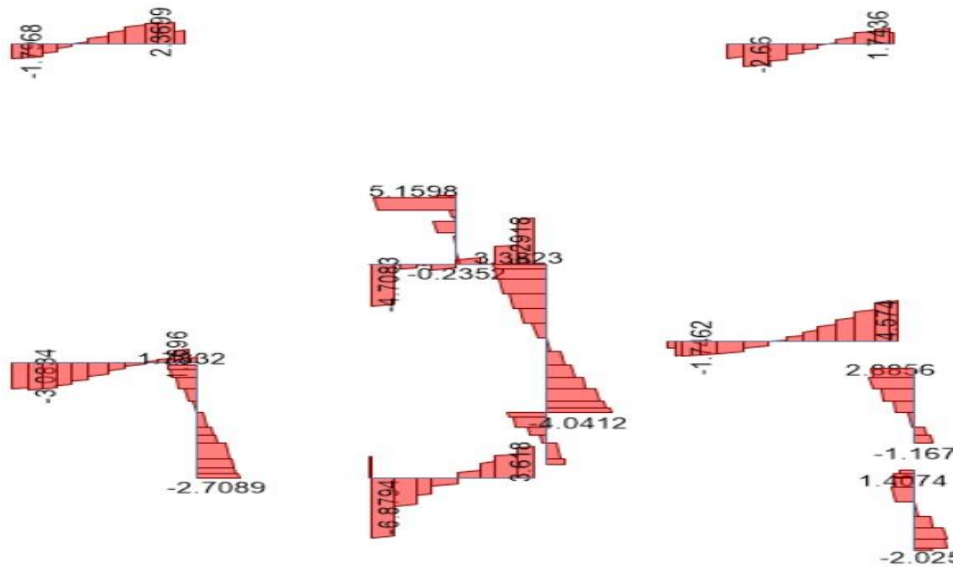


Figure 2.28 Shear Of Basement Slab Beam



## 2.6.1 Design of Column Section (subjected to axial compression force)

❖ For  $\xi 1$  on axis (5-5), (  $\varphi - \varphi$  )

### ➤ Concrete Dimension

- $P_U = 725$  ton (From Etabs Program)
- $P_{U \text{ actual}} = 725 * 1.1 = 797.5$  ton
- $P_{U \text{ actual}} = 0.35( A_c - A_s )F_{cu} + 0.67 A_s F_y \rightarrow \text{assume } A_s = 0.01 A_c$
- $797.5 * 10^3 = 0.35 * ( 0.99 A_c ) * 300 + 0.67 * 3600 * 0.01 A_c$
- $A_c = 6227.06 \text{ cm}^2$
- Assume  $b = 45 \text{ cm}$
- $t = \frac{Ac}{b} = 160 \text{ cm}$
- $A_{\text{cact}} = 45 * 160 = 72 \text{ cm}^2$

❖ Check of Buckling (braced column)

- in short direction
  - $k = 0.80$  (1-2)
- $$\lambda = \frac{k*H_o}{b} = \frac{0.8*2.90}{0.45} = 5.15 < 15 \Rightarrow \text{Short Column}$$

- in Long direction
  - $k = 0.80$  (1-2)
- $$\lambda = \frac{k*H_o}{t} = \frac{0.80*2.90}{1.60} = 1.45 < 15 \Rightarrow \text{Short Column}$$

Neglected buckling in short and long direction

- $A_s = 0.01 * 450 * 1600 = 72 \text{ cm}^2$

Use 36  $\phi 18$

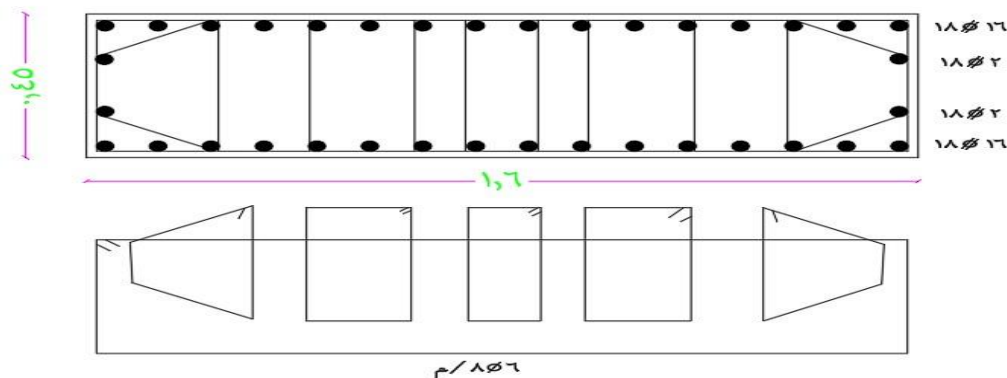


Figure 2.30 Column Cross Section

**2.6.2 Table of Columns**

| Joint | $P_{u \text{ etabs}}$<br>(T) | $P_{u \text{ act}}$ (T) | $\mu = A_s/A_c$ | b (cm) | t (cm) | §     | Sample |
|-------|------------------------------|-------------------------|-----------------|--------|--------|-------|--------|
| 1     | 725.0078                     | 725                     | 1.00%           | 45     | 160    | 36#16 | C1     |
| 2     | 716.2165                     | 716                     | 1.00%           | 45     | 160    | 36#16 | C1     |
| 3     | 690.8475                     | 691                     | 1.00%           | 45     | 160    | 36#16 | C1     |
| 4     | 640.3195                     | 640                     | 1.00%           | 45     | 145    | 26#18 | C2     |
| 5     | 625.9972                     | 626                     | 1.00%           | 45     | 145    | 26#18 | C2     |
| 6     | 625.3069                     | 625                     | 1.00%           | 45     | 145    | 26#18 | C2     |
| 7     | 622.3681                     | 622                     | 1.00%           | 45     | 145    | 26#18 | C2     |
| 8     | 565.0497                     | 565                     | 1.00%           | 40     | 140    | 26#18 | C3     |
| 9     | 550.3094                     | 550                     | 1.00%           | 40     | 140    | 26#18 | C3     |
| 10    | 537.0696                     | 537                     | 1.00%           | 40     | 140    | 26#18 | C3     |
| 11    | 533.5909                     | 534                     | 1.00%           | 40     | 140    | 26#18 | C3     |
| 12    | 532.9364                     | 533                     | 1.00%           | 40     | 140    | 26#18 | C3     |
| 13    | 500.1122                     | 500                     | 1.00%           | 40     | 125    | 20#18 | C4     |
| 14    | 495.8886                     | 498                     | 1.00%           | 40     | 125    | 20#18 | C4     |
| 15    | 487.1236                     | 487                     | 1.00%           | 40     | 125    | 20#18 | C4     |
| 16    | 468.4492                     | 469                     | 1.00%           | 40     | 115    | 20#18 | C5     |
| 17    | 460.033                      | 460                     | 1.00%           | 40     | 115    | 20#18 | C5     |
| 18    | 459.304                      | 459                     | 1.00%           | 40     | 115    | 20#18 | C5     |
| 19    | 440.7449                     | 441                     | 1.00%           | 40     | 115    | 20#18 | C5     |
| 20    | 435.8601                     | 436                     | 1.00%           | 40     | 115    | 20#18 | C5     |
| 21    | 393.9528                     | 394                     | 1.00%           | 40     | 100    | 20#18 | C6     |
| 22    | 393.5869                     | 394                     | 1.00%           | 40     | 100    | 20#18 | C6     |
| 23    | 393.5869                     | 394                     | 1.00%           | 40     | 100    | 20#18 | C6     |
| 24    | 393.5869                     | 394                     | 1.00%           | 40     | 85     | 14#18 | C7     |
| 25    | 393.5869                     | 394                     | 1.00%           | 40     | 85     | 14#18 | C7     |
| 26    | 393.5869                     | 394                     | 1.00%           | 40     | 85     | 14#18 | C7     |
| 27    | 393.5869                     | 394                     | 1.00%           | 40     | 75     | 12#18 | C8     |
| 28    | 393.5869                     | 394                     | 1.00%           | 40     | 75     | 12#18 | C8     |
| 29    | 393.5869                     | 394                     | 1.00%           | 40     | 75     | 12#18 | C8     |
| 30    | 393.5869                     | 394                     | 1.00%           | 40     | 75     | 12#18 | C8     |

Figure 2.31 Table Columns Load And Section (Ultimat)



## 2.7 Design of ShearWall (W<sub>1</sub>=0.45m\*4.20m) on asix (7-7)

Case (1):

- P<sub>u</sub> = 3827.3 KN
- M<sub>ux</sub> = 12.362 kN.m
- M<sub>uy</sub> = 1560 kN.m

$$1. R_b = \frac{P_u}{F_{cu} * b * t} = \frac{3827.3 * 10^3}{30 * 450 * 4200} = 0.067 \approx 0.30$$

$$2. \frac{M_{ux}}{F_{cu} * b * t^2} = \frac{12.362 * 10^6}{30 * 450 * 4200^2} = 0.0001$$

$$3. \frac{M_{uy}}{F_{cu} * t * b^2} = \frac{1560 * 10^6}{30 * 4200 * 450^2} = 0.06$$

$$4. p = 3.8$$

$$5. \mu = 3.8 * 30 * 10^{-4} = 0.0114$$

$$6. A_s = 0.0114 * 450 * 4200 = 21546 \text{ mm}^2$$

case (2):

- P<sub>u</sub> = 150.263 Kn
- M<sub>ux</sub> = 38.504 kn.m
- M<sub>uy</sub> = 1522.1 kn.m

$$7. R_b = \frac{P_u}{F_{cu} * b * t} = \frac{150.263 * 10^3}{30 * 450 * 4200} = 0.3$$

$$8. \frac{M_{ux}}{F_{cu} * b * t^2} = \frac{38.504 * 10^6}{30 * 450 * 4200^2} = 0.00016$$

$$9. \frac{M_{uy}}{F_{cu} * t * b^2} = \frac{1522.1 * 10^6}{30 * 4200 * 450^2} = 0.06$$

$$10. p = 3.8$$

$$11. \mu = 3.8 * 30 * 10^{-4} = 0.0114$$

$$12. A_s = 0.0114 * 450 * 4200 = 21546 \text{ mm}^2$$

USE 88 ϕ18

**Minimum RFT**

$$A_s = 0.01 A_s = 215.46 \text{ mm}^2$$

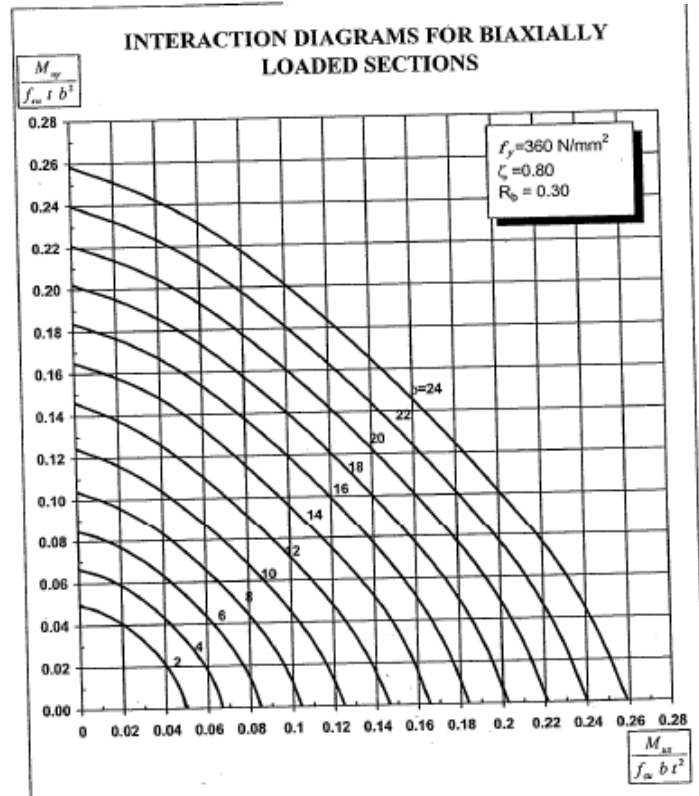
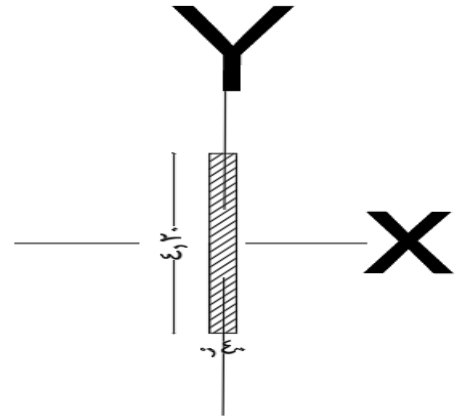


Figure 2.32 Interaction Diagram for Biaxial Loaded Wall

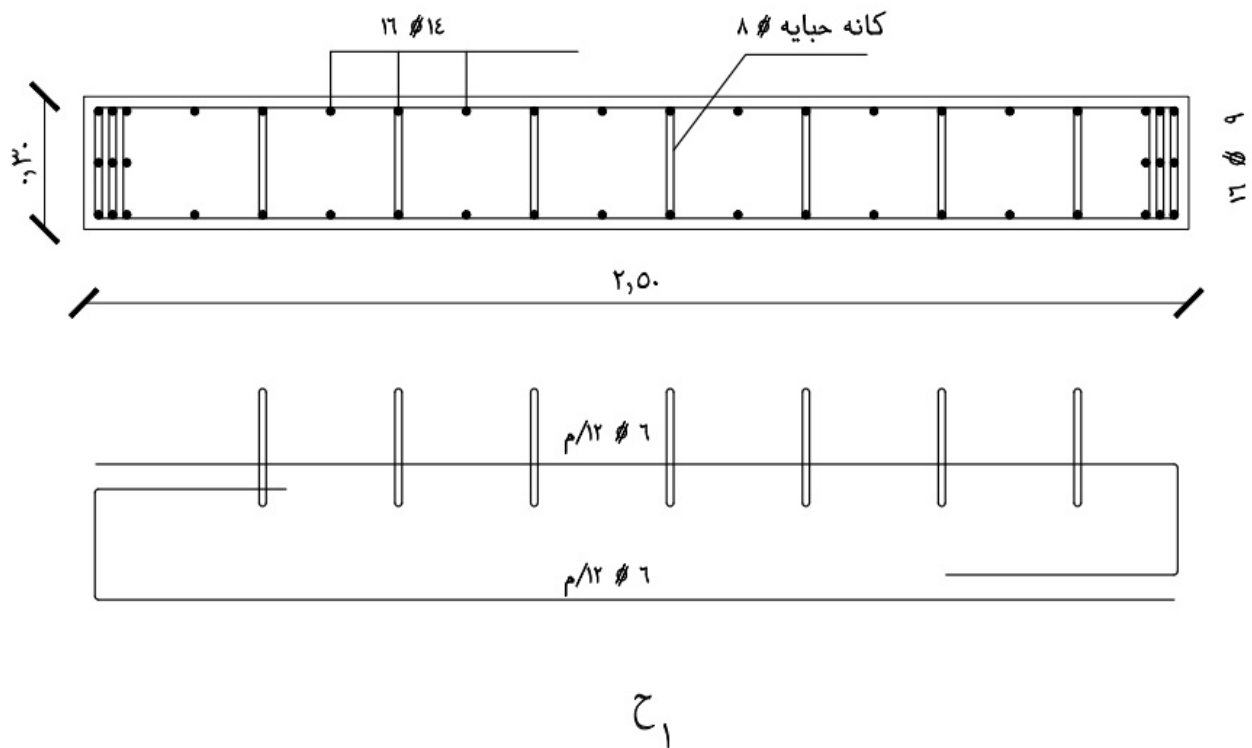


Figure 2.33 W1 Reinforcement

## 2.7.1 Table of Walls:

| Wall Of Section |    |             | Case Of Loading      |                            |                            | As     | Additional steel at each corner |
|-----------------|----|-------------|----------------------|----------------------------|----------------------------|--------|---------------------------------|
|                 |    |             | P <sub>u</sub> (ton) | M <sub>ux</sub><br>(ton.m) | M <sub>uy</sub><br>(ton.m) |        |                                 |
| 1 $\tau$        | W1 | 0.45m*4.20m | 15.0263              | 3.8504                     | 152.21                     | 88 D18 | 12 $\phi$ 18                    |
|                 |    |             | 382.73               | 1.2362                     | 1586                       |        |                                 |
|                 |    |             | 1353.8               | 1.47                       | 1625.88                    |        |                                 |
| 2 $\tau$        | W2 | 0.40m*3.40m | 151.783              | 1.8397                     | 472.56                     | 76 D18 | 12 $\phi$ 18                    |
|                 |    |             | 682.09               | 0.2301                     | 7.2415                     |        |                                 |
|                 |    |             | 493.92               | 1.7306                     | 468.94                     |        |                                 |
| 3 $\tau$        | W3 | 0.40m*2.40m | 90.6534              | 0.7153                     | 344.46                     | 58D18  | 12 $\phi$ 18                    |
|                 |    |             | 309.32               | 1.0469                     | 350.194                    |        |                                 |
|                 |    |             | 395.79               | 0.4618                     | 340.09                     |        |                                 |
| 4 $\tau$        | W4 | 0.40m*2.10m | 969.94               | 1.886                      | 14.8041                    | 44D18  | 12 $\phi$ 18                    |
|                 |    |             | 183.731              | 0.9124                     | 1046.67                    |        |                                 |
|                 |    |             | 671.89               | 1.844                      | 1039.6                     |        |                                 |

Figure 2.34 Table Wall Load And RFT (Ultimate)

### 2.8 Design of Core 1-2 (using ETABS program)

#### Case (1):

- $P_u = 1101.4891 \text{ t}$
- $M_{ux} = 362.9719 \text{ t.m}$
- $M_{uy} = 2894.1329 \text{ t.m}$

#### Case (2):

- $P_u = 339.2703 \text{ t}$
- $M_{ux} = 1786.8875 \text{ t.m}$
- $M_{uy} = 1883.3574 \text{ t.m}$

#### Case (3):

- $P_u = 782.5734 \text{ t}$
- $M_{ux} = 332.4167 \text{ t.m}$
- $M_{uy} = 2853.7354 \text{ t.m}$

Use  $A_{Stotal} = 227 \text{ } \phi 18$

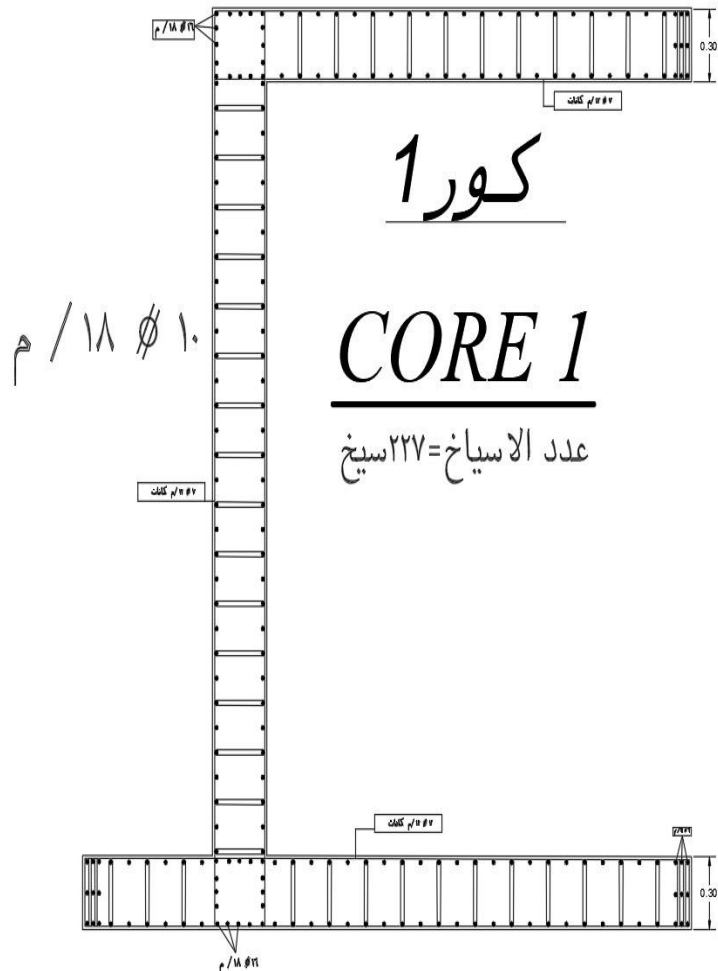


Figure 2.35 Core Cross Section

### 2.9 Design of Core 1-2 (using ETABS Column program)

#### Case (1):

- $P_u = 993.2064 \text{ t}$
- $M_{ux} = 92.4005 \text{ t.m}$
- $M_{uy} = 4464.5441 \text{ t.m}$

#### Case (2):

- $P_u = 1283.4098 \text{ t}$
- $M_{ux} = 69.0354 \text{ t.m}$
- $M_{uy} = 4716.7734 \text{ t.m}$

#### Case (3):

- $P_u = 1869.7428 \text{ t}$
- $M_{ux} = 22.2974 \text{ t.m}$
- $M_{uy} = 221.5333 \text{ t.m}$

Use  $A_{stotal} = 220 \text{ } \phi 18$

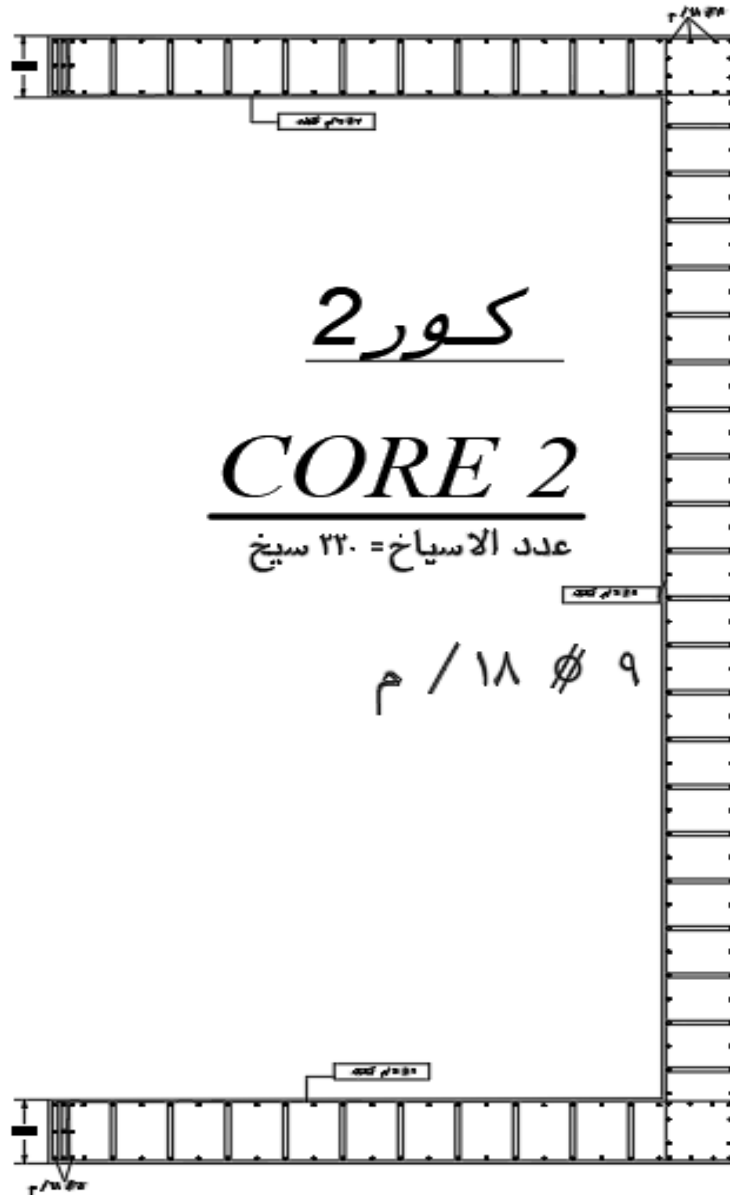


Figure 2.36 Core Cross Section

## 2.10 Effect of Earthquake on tower

### 2.10.1 Manual solution

- Zone (2)  $\Rightarrow a_g = 0.125g = 1.226 \text{ m / sec}^2$
- Type Soil (c)  $\Rightarrow S = 1.5, T_B = 0.1, T_C = 0.25, T_D = 1.20$
- Importance Factor ( $\gamma_I$ ) = 1.0
- Damping Correction Factor ( $\eta$ ) = 1.0
- Response Modification factor (R) = 5

$$F_b = Sd(T_1) * \lambda * \frac{w}{g}$$

1.  $T_1 = C_t H^{3/4} = 0.05 * 40.60^{3/4} = 0.8042 \text{ sec}$
  2.  $T_C < T_1 < T_D \Rightarrow S_d(T_1) = ag * r_I * \frac{2.5 \eta}{R} * [\frac{T_C}{T_1}]$
  3.  $= 1.2226 * 1 * 1.5 * \frac{2.5}{5} * \frac{.25}{0.8042} = .286 > 0.2 \text{ ag} * r_I = .2 * 1.25 * 1 = .25$
  4.  $\lambda = 1 \Rightarrow \text{as } T_1 > 2 T_C$
  5.  $W_i = D.L + \Psi L.L$ 
    - a.  $D.L_{\text{(Basement Slab)}} = O.w + \text{cover} + W_{\text{wall}} = 2.50 * 0.25 + 0.15 + 0.27 = 1.045 \text{ t/m}^2$
    - b.  $W_{\text{(Basement Slab)}} = 1.045 + 0.25 * 0.4 = 1.145 \text{ t/m}^2$
    - c.  $D.L_{\text{(Ground Or Repatd Slab)}} = O.w + \text{cover} + W_{\text{wall}} = 2.50 * 0.25 + 0.15 + 0.27 = 1.045 \text{ t/m}^2$
    - d.  $W_{\text{(Ground Or Repatd Slab)}} = 1.045 + 0.25 * 0.3 = 1.12 \text{ t/m}^2$
  6.  $W_{\text{total}} = 15520 \text{ ton} \Rightarrow \text{FROM ETABS BROGRAM}$
- Note (1.1 For Col. And Shear wall And Beams )**
7.  $F_b = 0.286 * 1 * \frac{15520}{9.81} = 452.46 \text{ ton}$

| FLOOR | W floor | H     | W*H   | FORCE | ton(ULT) | Shear ton      | Base moment | TOTAL M |
|-------|---------|-------|-------|-------|----------|----------------|-------------|---------|
| base  |         | 0     |       |       |          | 558.1531       | 0.0         | 11997.7 |
| 1     | 480     | 3     | 1440  | F1    | 7.903612 | 550.2495       | 23.7        | 10432.8 |
| 2     | 480     | 6     | 2880  | F2    | 15.80722 | 534.4423       | 94.8        | 8891.6  |
| 3     | 480     | 9     | 4320  | F3    | 23.71084 | 510.7314       | 213.4       | 7397.8  |
| 4     | 480     | 12    | 5760  | F4    | 31.61445 | 479.1170       | 379.4       | 5975.1  |
| 5     | 480     | 15    | 7200  | F5    | 39.51806 | 439.5989       | 592.8       | 4647.3  |
| 6     | 480     | 18    | 8640  | F6    | 47.42167 | 392.1772       | 853.6       | 3438.1  |
| 7     | 480     | 21    | 10080 | F7    | 55.32528 | 336.8519       | 1161.8      | 2371.1  |
| 8     | 480     | 24    | 11520 | F8    | 63.2289  | 273.6231       | 1517.5      | 1470.1  |
| 9     | 480     | 27    | 12960 | F9    | 71.13251 | 202.4905       | 1920.6      | 758.7   |
| 10    | 480     | 30    | 14400 | F10   | 79.03612 | 123.4544       | 2371.1      | 260.8   |
| 11    | 480     | 33    | 15840 | F11   | 86.93973 | 36.5147        | 2869.0      | 260.8   |
|       |         |       |       |       |          |                |             |         |
|       |         |       |       |       |          |                |             |         |
|       |         |       |       |       |          |                |             |         |
|       |         |       |       |       |          |                |             |         |
|       |         | TOTAL | 95040 |       | 521.6384 |                |             |         |
|       |         |       |       |       |          | moment at base | 11997.7     |         |

Figure 2.37 Table Over Turning And Torsional Moment

$$M_{\text{overturning}} = 11997.7 \text{ t.m}$$

$$M_{\text{torsional}} = V_i * e_{\text{min}} = 252 * 0.05 * 41.5 = 521.63 \text{ t.m}$$

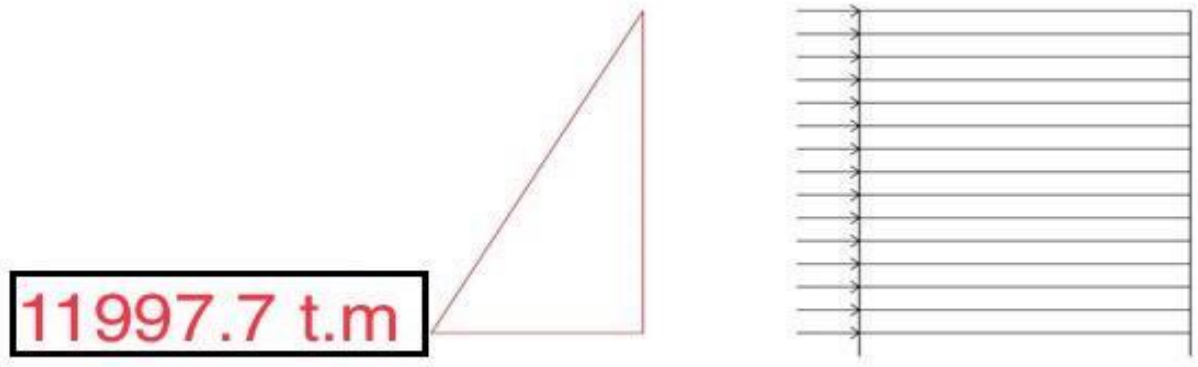


Figure 2.38 Over Turning Moment

### Check of stresses between (Shear wall , and cores) and Foundation

#### Inertia of Walls

##### ▪ W1 (0.45\*4.20)

$$A = 0.45 * 4.20 = 1.89 \text{ m}^2$$

$$I_x = \frac{0.45 * 4.20^3}{12} = 2.78 \text{ m}^4$$

$$I_y = \frac{4.20 * 0.45^3}{12} = 0.032 \text{ m}^4$$

##### ▪ W2 (0.40\*3.40)

$$A = 0.40 * 3.40 = 1.36 \text{ m}^2$$

$$I_x = \frac{3.40 * 0.40^3}{12} = 0.0181 \text{ m}^4$$

$$I_y = \frac{0.40 * 3.40^3}{12} = 1.310 \text{ m}^4$$

##### ▪ W3 (0.40\*2.40)

$$A = 0.40 * 2.40 = 0.96 \text{ m}^2$$

$$I_x = \frac{2.40 * 0.40^3}{12} = 0.0128 \text{ m}^4$$

$$I_y = \frac{0.40 * 2.40^3}{12} = 0.0575 \text{ m}^4$$

##### ▪ W4 (0.40\*2.10)

$$A = 0.40 * 2.10 = 0.84 \text{ m}^2$$

$$I_x = \frac{2.10 * 0.40^3}{12} = 0.0112 \text{ m}^4$$

$$I_Y = \frac{0.40 \cdot 2.10^3}{12} = 0.30 \text{ m}^4$$

## ▪ Core 1

$$A = 2.72 \text{ m}^2$$

$$I_X = \frac{2.63 \cdot 0.30^3}{12} + \frac{0.30 \cdot 3.12^3}{12} + \frac{3.34 \cdot 0.30^3}{12} = 0.77 \text{ m}^4$$

$$I_Y = \frac{0.30 \cdot 2.63^3}{12} + \frac{3.12 \cdot 0.30^3}{12} + \frac{0.30 \cdot 3.34^3}{12} = 1.39 \text{ m}^4$$

## ▪ Core 2

$$A = 2.54 \text{ m}^2$$

$$I_X = 2 \cdot \frac{2.61 \cdot 0.30^3}{12} + \frac{0.30 \cdot 4.66^3}{12} = 2.54 \text{ m}^4$$

$$I_Y = 2 \cdot \frac{0.30 \cdot 2.61^3}{12} + \frac{4.66 \cdot 0.30^3}{12} = 0.049 \text{ m}^4$$

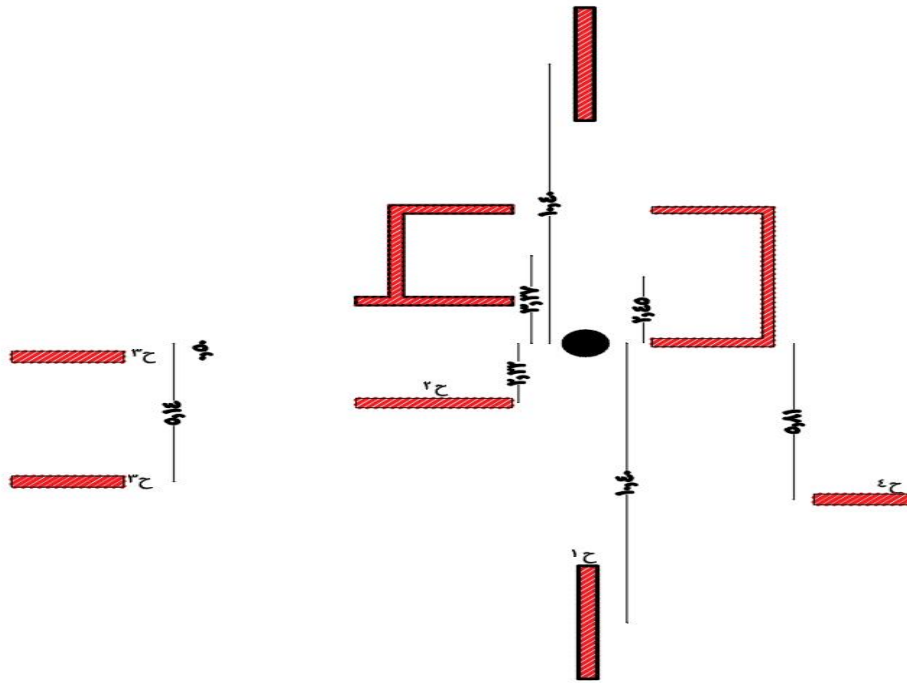


Figure 2.39 Moment of inertia (Ix)

$$I_{X \text{ group}} = (1.89 + 2.78 \cdot (10.40)^2) \cdot 2 + (0.0181 + 1.36 \cdot (2.22)^2) + (0.0128 + 0.96 \cdot (5.14)^2) + (0.0128 + 0.96 \cdot (0.50)^2) + (0.0112 + 0.84 \cdot (5.81)^2) + (0.77 + 2.72 \cdot (3.27)^2) + (2.54 + 2.54 \cdot (2.45)^2) = 522.74 \text{ m}^4$$



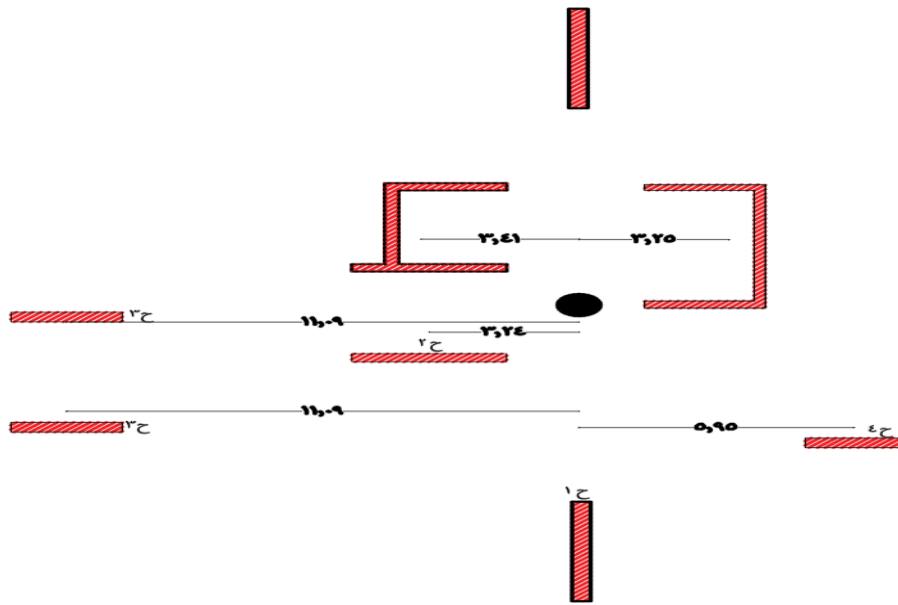


Figure 2.40 Moment of inertia (Iy)

$$I_{Y \text{ group}} = (0.032 + 2.78 * (0)^2) * 2 + (1.310 + 1.36 * (3.24)^2) + (0.0575 + 0.96 * (11.09)^2) * 2 + (0.30 + 0.84 * (5.59)^2) + (1.39 + 2.72 * (3.41)^2) + (0.049 + 2.54 * (3.25)^2) = 338.215 \text{ m}^4$$

$$I_{X \text{ group}} > I_{Y \text{ group}}$$

X Is Critical Direction

$$F_{1,2} = \frac{-N}{A} \pm \frac{M_x * X}{I_y} \leq 1.25 F_c$$

$$A_{\text{wall, Core}} = 13.3 \text{ m}^2 \text{ (From CAD Program)}$$

$$N_{\text{wall}} = 5700.67 \text{ (From ETAP Program)}$$

$$M_x = 11997.7 \text{ t.m}$$

$$Y = 10.10 \text{ m}$$

$$I_{Y \text{ group}} = 338.215 \text{ m}^4$$

$$F_{1,2} = \frac{-5700.67}{13.3} \pm \frac{11997.7 * 10.10}{338.215}$$

$$F_1 = \frac{-5700.67}{13.3} - \frac{11997.7 * 10.10}{338.215} = -78.7 \text{ Kg/Cm}^2 < 1.25 * 105 = 131.25 \text{ Kg/Cm}^2$$

$$F_2 = \frac{-5700.67}{13.3} + \frac{11997.7 * 10.10}{338.215} = -7.03 \text{ Kg/Cm}^2 < 1.25 * 105 = 131.25 \text{ Kg/Cm}^2$$

There is no tension OK Safe

## 2.10.2 Check of Displacement

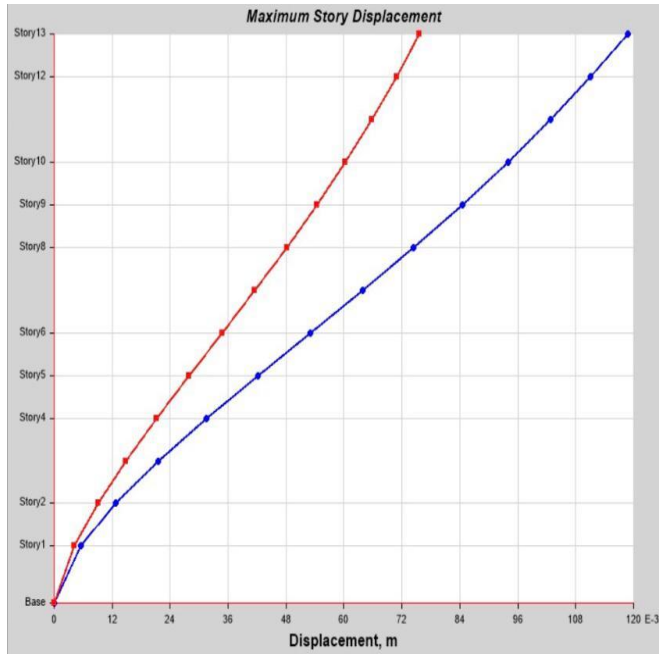


Figure 2.41 Maximum Story Displacement in X direction

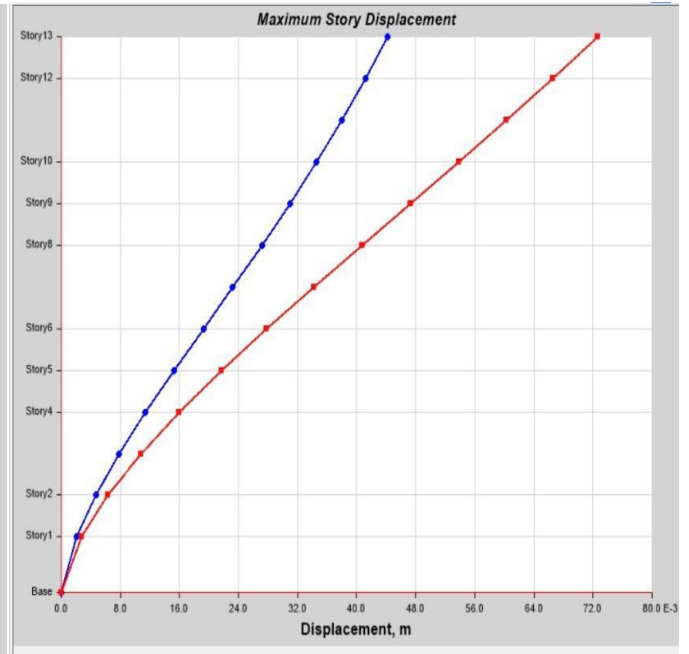


Figure 2.42 Maximum Story Displacement in Y direction

| Story   | H(m) | $d_{sx}$ (m) | $d_{sy}$ (m) |
|---------|------|--------------|--------------|
| Story12 | 40.6 | 0.058263     | 0.036124     |
| Story11 | 37.7 | 0.055066     | 0.033662     |
| Story10 | 34.8 | 0.051606     | 0.031054     |
| Story9  | 31.9 | 0.047794     | 0.028302     |
| Story8  | 29   | 0.043594     | 0.025409     |
| Story7  | 26.1 | 0.039017     | 0.022394     |
| Story6  | 23.2 | 0.034107     | 0.019288     |
| Story5  | 20.3 | 0.028939     | 0.016139     |
| Story4  | 17.4 | 0.023621     | 0.013007     |
| Story3  | 14.5 | 0.018293     | 0.009967     |
| Story2  | 11.6 | 0.013135     | 0.007107     |
| Story1  | 8.7  | 0.008377     | 0.004531     |
| GEROUND | 5.8  | 0.004313     | 0.002364     |
| BASMENT | 2.9  | 0.001319     | 0.000759     |

Figure 2.43 Table Story Response Data

## ***HIGH RISE BUILDING***

Max Displacement at 12<sup>th</sup> Floor

- X-Direction = 0.058263
- Y-Direction = 0.036124

Height of Building = 40.6 m

Allowable Displacement according to ECP =  $\frac{H}{500} = \frac{40.6}{500} = 0.0812$  m Working Method

Ultimate Method =  $0.0812 \times 1.5 = 0.1218$  m

OK , Safe

## 2.10.3 Check of Drift

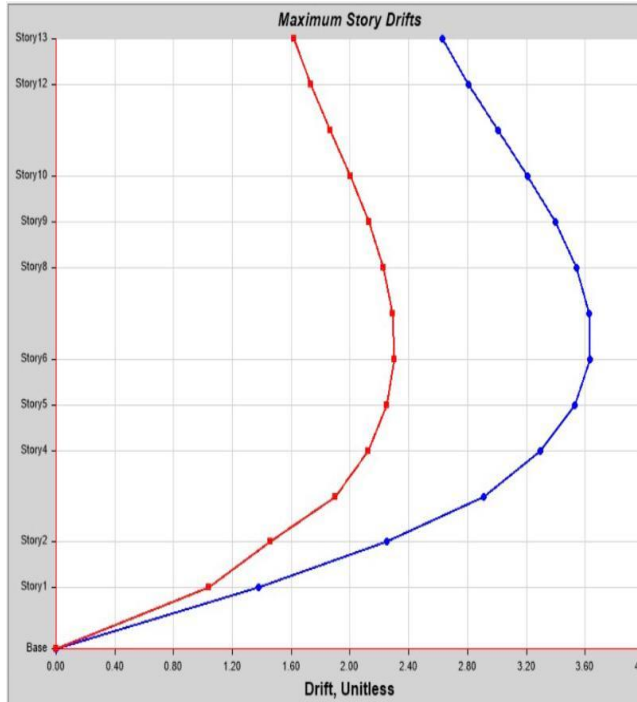


Figure 2.44 Maximum Story Drifts In X direction

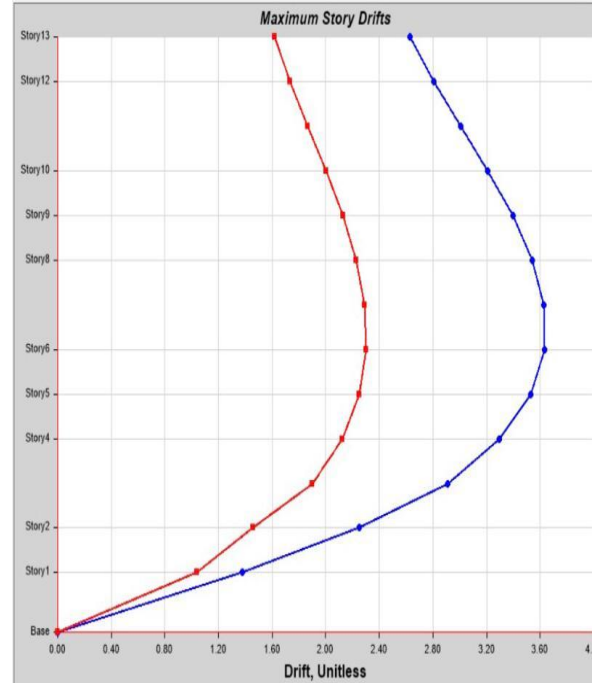


Figure 2.45 Maximum Story Drifts In y direction

| Story   | H(m) | $d_{ry}$ (m) | $d_{rx}$ (m) | $d_{rvx}$ (m) | $d_{rvy}$ (m) |
|---------|------|--------------|--------------|---------------|---------------|
| Story12 | 3    | 0.058263     | 0.03612      | 0.0032        | 0.0025        |
| Story11 | 3    | 0.055066     | 0.03366      | 0.0035        | 0.0026        |
| Story10 | 3    | 0.051606     | 0.03105      | 0.0038        | 0.0028        |
| Story9  | 3    | 0.047794     | 0.0283       | 0.0042        | 0.0029        |
| Story8  | 3    | 0.043594     | 0.02541      | 0.0046        | 0.003         |
| Story7  | 3    | 0.039017     | 0.02239      | 0.0049        | 0.0031        |
| Story6  | 3    | 0.034107     | 0.01929      | 0.0052        | 0.0031        |
| Story5  | 3    | 0.028939     | 0.01614      | 0.0053        | 0.0031        |
| Story4  | 3    | 0.023621     | 0.01301      | 0.0053        | 0.003         |
| Story3  | 3    | 0.018293     | 0.00997      | 0.0052        | 0.0029        |
| Story2  | 3    | 0.013135     | 0.00711      | 0.0048        | 0.0026        |
| Story1  | 3    | 0.008377     | 0.00453      | 0.0041        | 0.0022        |
| GEROUND | 3    | 0.004313     | 0.00236      | 0.003         | 0.0016        |
| BASMENT | 2.65 | 0.001319     | 0.00076      | 0.0013        | 0.0008        |

Figure 2.46 Table Story Response Values

Max drift at X-direction = 0.0053m

Max drift at Y-direction = 0.0031m

Allowable Drift according to ECP =  $\frac{H}{500} = \frac{2.90}{500} = 0.0058$  m Working Method

Ultimate Method =  $0.006 * 1.5 = 0.009$  m

OK , Safe

## 2.10.4 Check of Maximum Distance Between C.M, C.R

| Story   | Diaphragm | Mass X<br>tonf-s <sup>2</sup> /m | Mass Y<br>tonf-s <sup>2</sup> /m | XCM<br>m | YCM<br>m | Cum Mass X<br>tonf-s <sup>2</sup> /m | Cum Mass Y<br>tonf-s <sup>2</sup> /m | XCCM<br>m | YCCM<br>m |
|---------|-----------|----------------------------------|----------------------------------|----------|----------|--------------------------------------|--------------------------------------|-----------|-----------|
| Story1  | D1        | 94.41336                         | 94.41336                         | 12.3997  | 12.4329  | 94.41336                             | 94.41336                             | 12.3997   | 12.4329   |
| Story2  | D2        | 97.23639                         | 97.23639                         | 12.4163  | 11.8639  | 97.23639                             | 97.23639                             | 12.4163   | 11.8639   |
| Story3  | D3        | 97.23651                         | 97.23651                         | 12.4163  | 11.8639  | 97.23651                             | 97.23651                             | 12.4163   | 11.8639   |
| Story4  | D4        | 97.23639                         | 97.23639                         | 12.4163  | 11.8639  | 97.23639                             | 97.23639                             | 12.4163   | 11.8639   |
| Story5  | D5        | 97.23639                         | 97.23639                         | 12.4163  | 11.8639  | 97.23639                             | 97.23639                             | 12.4163   | 11.8639   |
| Story6  | D6        | 97.23639                         | 97.23639                         | 12.4163  | 11.8639  | 97.23639                             | 97.23639                             | 12.4163   | 11.8639   |
| Story7  | D7        | 97.23639                         | 97.23639                         | 12.4163  | 11.8639  | 97.23639                             | 97.23639                             | 12.4163   | 11.8639   |
| Story8  | D8        | 97.23651                         | 97.23651                         | 12.4163  | 11.8639  | 97.23651                             | 97.23651                             | 12.4163   | 11.8639   |
| Story9  | D9        | 97.23639                         | 97.23639                         | 12.4163  | 11.8639  | 97.23639                             | 97.23639                             | 12.4163   | 11.8639   |
| Story10 | D10       | 97.23639                         | 97.23639                         | 12.4163  | 11.8639  | 97.23639                             | 97.23639                             | 12.4163   | 11.8639   |
| Story11 | D11       | 97.23639                         | 97.23639                         | 12.4163  | 11.8639  | 97.23639                             | 97.23639                             | 12.4163   | 11.8639   |
| Story12 | D12       | 97.23639                         | 97.23639                         | 12.4163  | 11.8639  | 97.23639                             | 97.23639                             | 12.4163   | 11.8639   |
| Story13 | D13       | 90.81317                         | 90.81317                         | 12.3938  | 11.8701  | 90.81317                             | 90.81317                             | 12.3938   | 11.8701   |

Figure 2.47 Table Story Response Values

Max Eccentricity

$e_x = 0.22$

$e_y = 0.5$

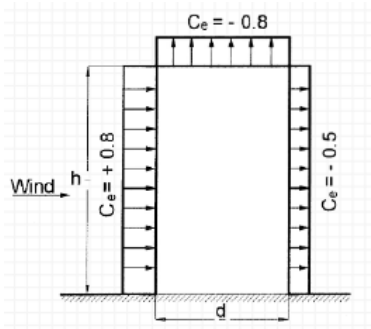
Allowable Difference  $\leq 0.15 b = 0.15 * 26.57 = 3.985$  m  $\Rightarrow$  OK Safe

$\leq 0.15 t = 0.15 * 28.2269 = 4.234$  m  $\Rightarrow$  OK Safe

## 2.11 Effect of Wind on tower

### 2.11.1 Manual solution

- $C_t = 1 \Rightarrow$  معامل طوبوغرافيه الارض
- $C_s = 1 \Rightarrow$  معامل المنشأ
- $\rho = 1.25 \text{ Kg/m}^3 \Rightarrow$  كثافه الهواء
- $V = 30 \text{ m/sec} \Rightarrow$  سرعه الرياح الاساسيه
- $K \Rightarrow$  معامل التعرض ويتغير حسب الارتفاع عن سطح الارض
- $C_e = 0.8 + 0.5 = 1.30 \Rightarrow$  معامل ضغط الرياح الخارجي علي سطح المبني



$$F = q * C_e * K * A_{\text{story}} = P_e * A_{\text{story}}$$

- $q = 0.5 * 10^{-3} * \rho * V^2 * C_t * C_s = 56.25 \text{ Kg/m}^2$
- $P_e = q * C_e * K = 56.25 * 10^{-3} * 1.3 * k = 0.073 \text{ K}$ 
  - $P_{e1} = 0.073 * 1 = 0.073 \text{ ton/m}^2$
  - $P_{e2} = 0.073 * 1.15 = 0.084 \text{ ton/m}^2$
  - $P_{e3} = 0.073 * 1.4 = 0.102 \text{ ton/m}^2$
  - $P_{e4} = 0.073 * 1.6 = 0.117 \text{ ton/m}^2$

$$F_1 = 0.073 * 10 * 27.85 = 20.33 \text{ ton}$$

$$F_2 = 0.084 * 10 * 27.85 = 23.39 \text{ ton}$$

$$F_3 = 0.102 * 10 * 27.85 = 28.41 \text{ ton}$$

$$F_4 = 0.117 * 10 * 27.85 = 34.54 \text{ ton}$$

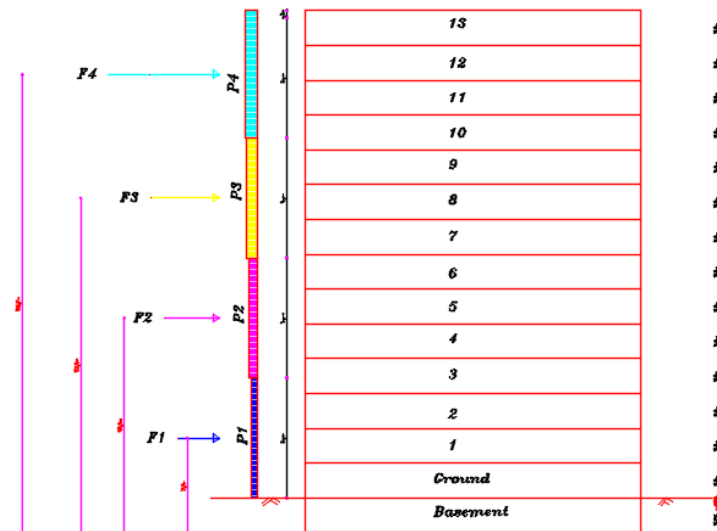


Figure 2.48 Wind Load

### **Check Of Over Turning Moment**

Wt = 13130.9 ton

Resisting Moment =  $13130.9 * \frac{26.5744}{2} = 174444 \text{ ton.m}$

Over Turning Moment = 11997.7 ton.m

Factor Of Safety =  $\frac{\text{Resisting Moment}}{\text{Over Turning Moment}} = \frac{174444}{11997.7} = 14.54 > 1.50$

Ok, Safe

### 2.12 Design of Deep Foundation (Raft on Piles)

Use Thickness of Raft = 120 cm

Pile diameter = 60 cm

Pile capacity = 120 ton

Total building weight (working method) = reactions of column , shear wall and core + O.w R.w  
+ O.w raft

- Reactions of columns ( Etabs program ) =15054 ton
- O.W raft = ( o.w slab+ cover + car weight ) \* area  
$$= (2.5*1.20+ 0.15 + 0.5 ) * 691 = 2522.15 \text{ ton}$$

Total building weight = 15054 + 2522.15 = 17576.15 ton

No. of piles = total building / pile capacity

$$= 17576.15/120 = 146.46 = 147 \text{ piles}$$

**Increase no. of piles by 20% t1o carry lateral loads generated by earthquakes.**

No.of piles = 147 \*1.25= 183.75 ~ 186 piles

Piles spacing (s) = (2→3) D

S= 1.80 m , 2 m on the side and the middle



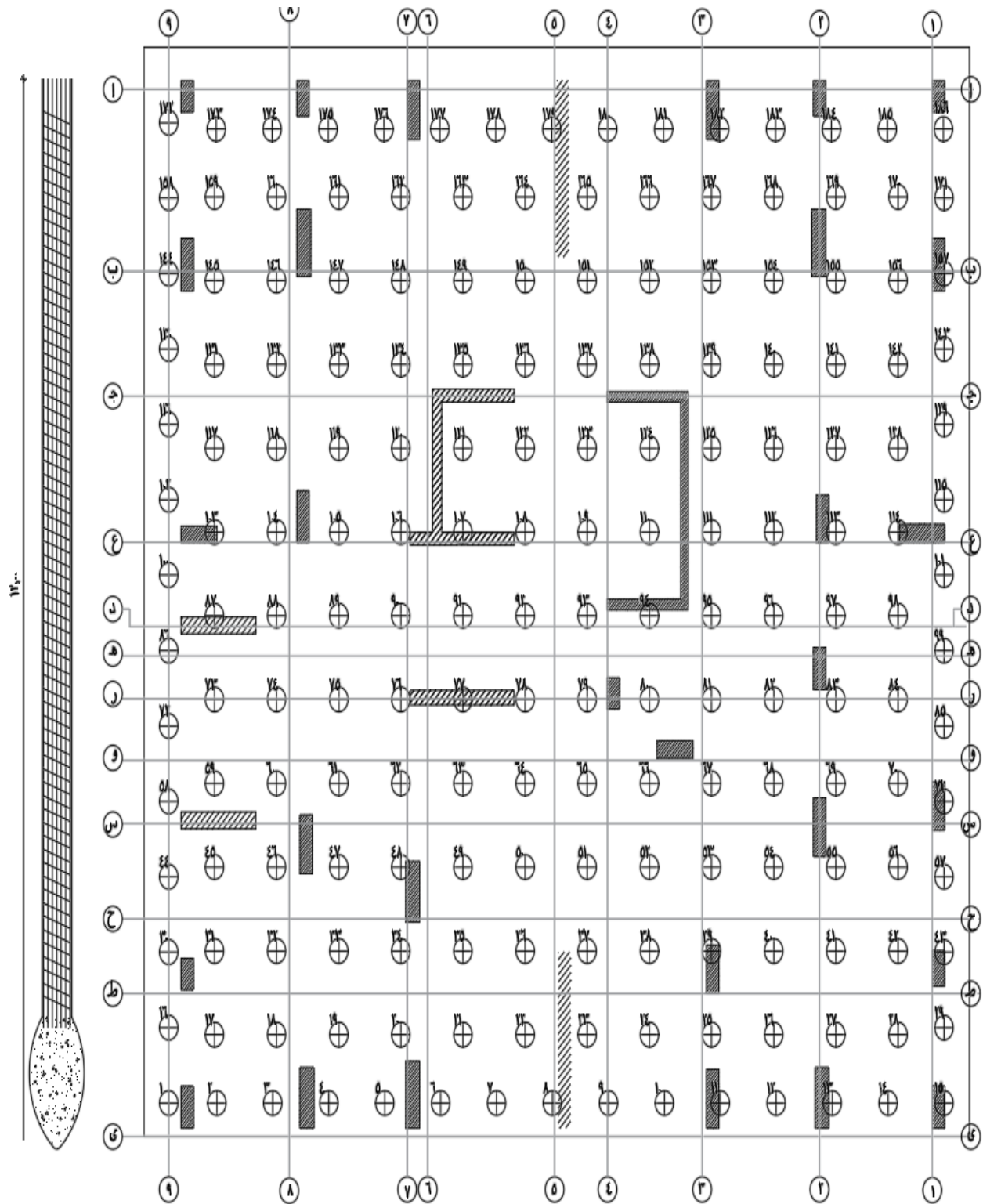


Figure 2.49 Piles arrangement

### 2.12.1 Design of Raft on Piles (Safe program)

$$t_{\text{raft}} = 120 \text{ cm}$$

$$d_{\text{pile}} = 60 \text{ cm}$$

$$P = K \Delta \rightarrow P = 100 \text{ ton}, \Delta = 1 \text{ cm}$$

$$K_{\text{stiffness}} = p/\Delta = 100/0.01 = 10000 \text{ t/m}$$

$$\text{Use } 85\% K_{\text{stiffness}} = 0.85 * 10000 = 8500 \text{ t/m}$$

- For raft

$$A_s = (M_u) / f_y J d$$

$$M_u = A_s * f_y * J * d = 12 * (\pi/4 * 1.8^2) * 3600 * 133 * 0.826 * 10^{-5} = 120.76 \text{ t.m}$$

- Use 8  $\phi$  18 Upper and Lower Reinforcement

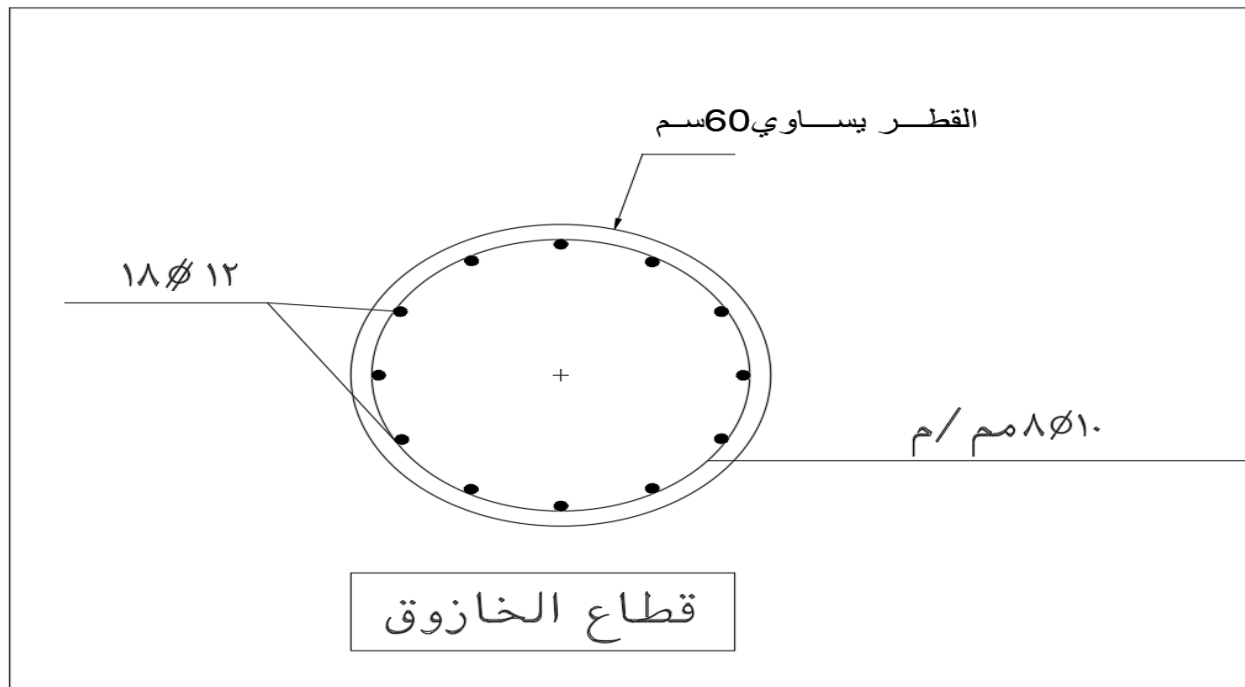
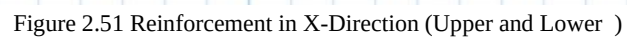


Figure 2.50 Section of Pile



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## 2.12.1.1 Table of piles forces :-

|               |               |               |               |               |               |               |               |               |               |               |               |               |               |              |
|---------------|---------------|---------------|---------------|---------------|---------------|---------------|---------------|---------------|---------------|---------------|---------------|---------------|---------------|--------------|
| 235           | 188           | 189           | 190           | 191           | 192           | 195           | 196           | 197           | 198           | 199           | 202           | 203           | 204           | 205          |
| Fz = 97.7499  | Fz = 101.2592 | Fz = 112.8257 | Fz = 115.1507 | Fz = 124.5207 | Fz = 125.5589 | Fz = 129.7303 | Fz = 144.2637 | Fz = 132.9576 | Fz = 125.4488 | Fz = 127.9738 | Fz = 121.5507 | Fz = 117.0104 | Fz = 106.9327 | Fz = 102.082 |
| 224           | 235           | 267           | 278           | 289           | 300           | 311           | 322           | 333           | 344           | 355           | 366           | 377           | 241           |              |
| Fz = 75.4784  | Fz = 83.8766  | Fz = 94.4006  | Fz = 96.2033  | Fz = 93.9012  | Fz = 97.6125  | Fz = 113.178  | Fz = 114.8306 | Fz = 99.4341  | Fz = 95.7569  | Fz = 98.0693  | Fz = 96.0996  | Fz = 85.2182  | Fz = 79.0157  |              |
| 241           | 254           | 266           | 277           | 288           | 299           | 310           | 321           | 332           | 343           | 354           | 365           | 376           | 240           |              |
| Fz = 89.0177  | Fz = 71.4033  | Fz = 81.4951  | Fz = 80.4899  | Fz = 76.9399  | Fz = 81.6523  | Fz = 91.7105  | Fz = 94.0865  | Fz = 84.3542  | Fz = 79.4546  | Fz = 82.4853  | Fz = 82.2869  | Fz = 72.2487  | Fz = 70.8385  |              |
| 279           | 253           | 265           | 276           | 287           | 298           | 309           | 320           | 331           | 342           | 353           | 364           | 375           | 239           |              |
| Fz = 83.235   | Fz = 58.1261  | Fz = 66.4483  | Fz = 73.2271  | Fz = 82.9023  | Fz = 90.3003  | Fz = 91.2423  | Fz = 92.825   | Fz = 93.5159  | Fz = 85.7459  | Fz = 74.3294  | Fz = 66.7317  | Fz = 58.3956  | Fz = 53.776   |              |
| 278           | 262           | 264           | 275           | 286           | 297           | 308           | 319           | 330           | 341           | 352           | 363           | 374           | 238           |              |
| Fz = 51.2571  | Fz = 62.2472  | Fz = 72.4706  | Fz = 82.1903  | Fz = 96.7929  | Fz = 105.1432 | Fz = 101.8692 | Fz = 102.7569 | Fz = 109.0342 | Fz = 99.3298  | Fz = 81.3236  | Fz = 70.9427  | Fz = 60.9575  | Fz = 49.8175  |              |
| 277           | 290           | 283           | 274           | 285           | 296           | 307           | 318           | 329           | 340           | 351           | 362           | 373           | 235           |              |
| Fz = 67.158   | Fz = 81.931   | Fz = 90.1752  | Fz = 94.5513  | Fz = 107.996  | Fz = 112.5819 | Fz = 109.1602 | Fz = 106.7208 | Fz = 114.5229 | Fz = 107.1921 | Fz = 91.995   | Fz = 85.6519  | Fz = 77.7309  | Fz = 60.1393  |              |
| 275           | 249           | 262           | 273           | 284           | 295           | 306           | 317           | 328           | 339           | 350           | 361           | 372           | 234           |              |
| Fz = 83.8979  | Fz = 90.4274  | Fz = 90.8579  | Fz = 92.4076  | Fz = 101.7463 | Fz = 107.1264 | Fz = 105.7465 | Fz = 109.9787 | Fz = 117.1779 | Fz = 107.8662 | Fz = 94.5523  | Fz = 84.6688  | Fz = 70.7457  | Fz = 63.5253  |              |
| 274           | 248           | 261           | 272           | 283           | 294           | 305           | 316           | 327           | 338           | 349           | 360           | 371           | 233           |              |
| Fz = 85.0304  | Fz = 81.5659  | Fz = 83.8223  | Fz = 86.1821  | Fz = 98.6369  | Fz = 103.5461 | Fz = 100.117  | Fz = 100.4688 | Fz = 104.9503 | Fz = 100.7945 | Fz = 94.0996  | Fz = 85.2646  | Fz = 65.7483  | Fz = 54.9922  |              |
| 273           | 247           | 260           | 271           | 282           | 293           | 304           | 315           | 326           | 337           | 348           | 359           | 370           | 232           |              |
| Fz = 74.3845  | Fz = 80.8507  | Fz = 84.3001  | Fz = 84.2373  | Fz = 85.3265  | Fz = 83.9737  | Fz = 80.7466  | Fz = 83.2291  | Fz = 94.6686  | Fz = 97.0304  | Fz = 94.9492  | Fz = 85.3076  | Fz = 70.1309  | Fz = 54.9805  |              |
| 272           | 246           | 259           | 270           | 281           | 292           | 303           | 314           | 325           | 336           | 347           | 358           | 369           | 231           |              |
| Fz = 73.3897  | Fz = 77.4399  | Fz = 85.1954  | Fz = 85.9254  | Fz = 87.8535  | Fz = 79.0142  | Fz = 75.2104  | Fz = 77.7818  | Fz = 85.629   | Fz = 96.6342  | Fz = 98.361   | Fz = 86.2368  | Fz = 70.6326  | Fz = 65.5077  |              |
| 271           | 245           | 258           | 269           | 280           | 291           | 302           | 313           | 324           | 335           | 346           | 357           | 368           | 229           |              |
| Fz = 69.7883  | Fz = 73.3541  | Fz = 78.6447  | Fz = 84.3439  | Fz = 89.7909  | Fz = 87.7293  | Fz = 93.4184  | Fz = 96.2909  | Fz = 92.9885  | Fz = 98.552   | Fz = 90.8546  | Fz = 80.7827  | Fz = 73.1383  | Fz = 63.4385  |              |
| 270           | 242           | 256           | 268           | 279           | 290           | 301           | 312           | 323           | 334           | 345           | 356           | 367           | 228           |              |
| Fz = 82.9384  | Fz = 88.6383  | Fz = 95.9438  | Fz = 102.9863 | Fz = 107.278  | Fz = 109.551  | Fz = 124.5531 | Fz = 126.4754 | Fz = 113.186  | Fz = 112.1949 | Fz = 105.8477 | Fz = 98.365   | Fz = 91.5112  | Fz = 86.4577  |              |
| 269           | 241           | 255           | 267           | 278           | 289           | 300           | 311           | 322           | 333           | 344           | 355           | 366           | 226           |              |
| Fz = 119.5882 | Fz = 119.442  | Fz = 130.3568 | Fz = 139.3997 | Fz = 143.1236 | Fz = 139.7867 | Fz = 142.3318 | Fz = 160.2094 | Fz = 144.6237 | Fz = 138.0707 | Fz = 142.5207 | Fz = 136.4297 | Fz = 136.8568 | Fz = 123.5426 |              |

Table 2.55 Piles Forces

## 2.12.1.2 Check of punching shear on raft

- for 1 ξ ( 45 \* 160 )

Col load = 725 ton

t<sub>raft</sub> = 120 cm

d = t<sub>raft</sub> - cover = 120 - 7 = 113 cm

$$q_{up} = P_{col} / 2((a+d)+(b+d))*d$$

$$\frac{725 * 10^4}{2 * (2230 + 1530) * 1130} = 0.85$$

$q_{cup}$  = the least of:-

- 1.70 N/mm<sup>2</sup>
- $0.316 \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.316 \sqrt{\frac{30}{1.5}} = 1.41 \text{ N/mm}^2$
- $0.316 \left( \frac{a}{b} + 0.5 \right) \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.316 \left( \frac{400}{1150} + 0.5 \right) \sqrt{\frac{30}{1.5}} = 1.198 \text{ N/mm}^2$
- $0.8 \left( \frac{\alpha * d}{b_0} + 0.2 \right) \sqrt{\frac{f_{cu}}{\gamma_c}} = 0.8 \left( \frac{4 * 1330}{8420} + 0.2 \right) \sqrt{\frac{30}{1.5}} = 2.97 \text{ N/mm}^2$
- $q_{up} = 0.85 \text{ N/mm}^2 \leq q_{cup} = 1.198 \text{ N/mm}^2$

OK . Safe

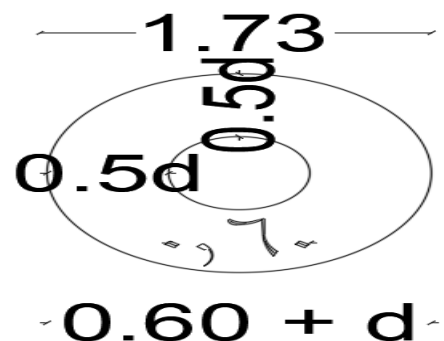
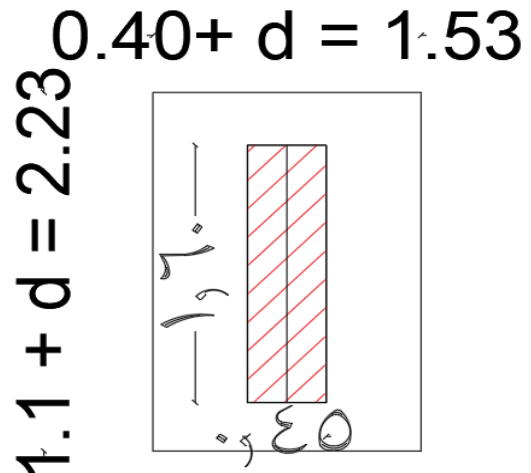
- For pile no. ( 16 )

Pile load = 120 ton

d<sub>pile</sub> = 60 cm

$$q_{up} = Q_{up} / \pi D d = ( 120 * 10^3 ) / ( \pi * 173 * 113 )$$

$$= 1.95 \text{ kg/cm}^2 < 10 \text{ kg/cm}^2 \quad \text{OK . Safe}$$



## 2.13 Design of Retaining wall:

### 1 Loads

- Thickness of wall = 25 cm
- Use Fill sand  $\phi_{30}$
- $\gamma_{soil} = 1.5 \text{ t/m}^3$
- $K_a = \frac{1 - \sin \phi}{1 + \sin \phi} = 0.33$
- $q(\text{sur charge}) = 1 \text{ t/m}^2$
- $e1 = (\Sigma \gamma * h + q) K_a - 2 * C * \sqrt{K_a}$   
 $= (1.5 * 2.9 + 1) * 0.33 = 1.78$
- $F_1 = 0.5 * 1.53 * 2.9 = 2.58 \text{ t/m}^2$

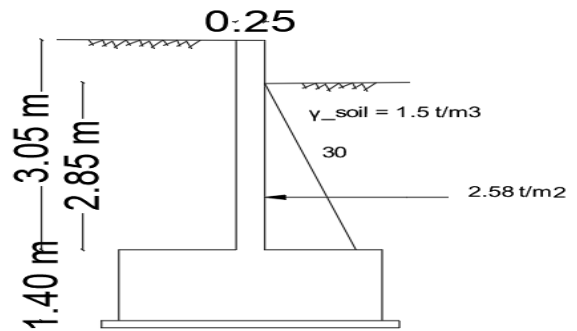


Figure 2.56 Section of Retaining Wall (Critical Section)

### 2.13.1 Design of critical section (as uncracked section)

- $M_{over} = 2.58 * \frac{1}{3} * 2.9 = 2.494 \text{ t.m/m}^2$
- $t = 250 \text{ mm}$ ,  $d = 220 \text{ mm}$
- $d = C_1 \sqrt{\frac{M}{b * F_{CU}}} = 220 = C_1 \sqrt{\frac{2.494 * 10^7 * 1.5}{1000 * 30}} \rightarrow C_1 = 6.275$ ,  $J = .826$
- $A_s = \frac{M * 1.5}{F_y * d * J} = \frac{2.494 * 10^7 * 1.5}{.826 * 350 * 220} = 571.85 \text{ mm}^2$
- Use  $A_{s \min} 6 \text{ } \phi 12 / \text{m}^2$  (Vertical) in both side
- Use  $A_{s \min} 6 \text{ } \phi 12 / \text{m}^2$  (horizontal) in both side

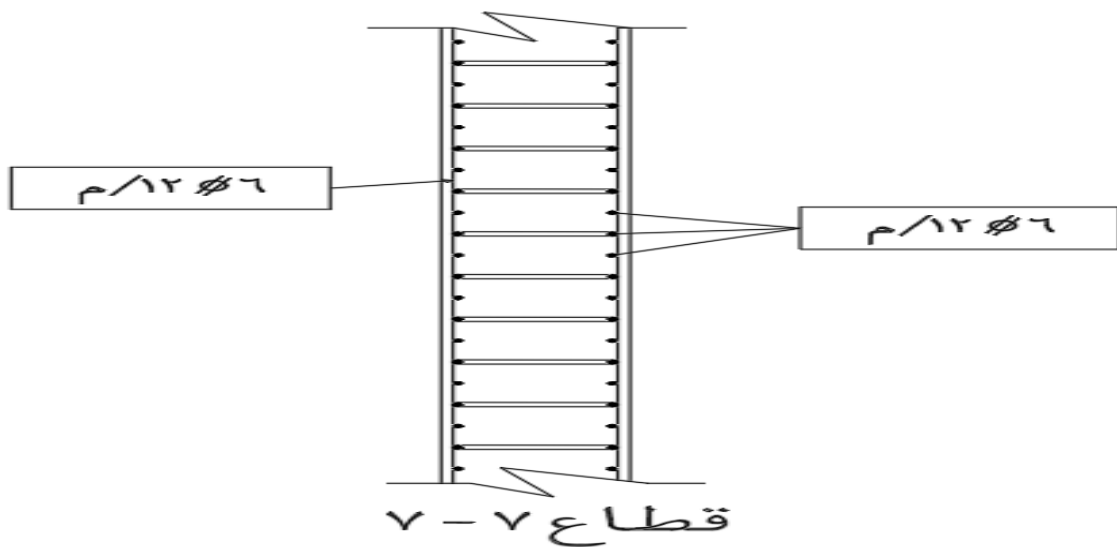
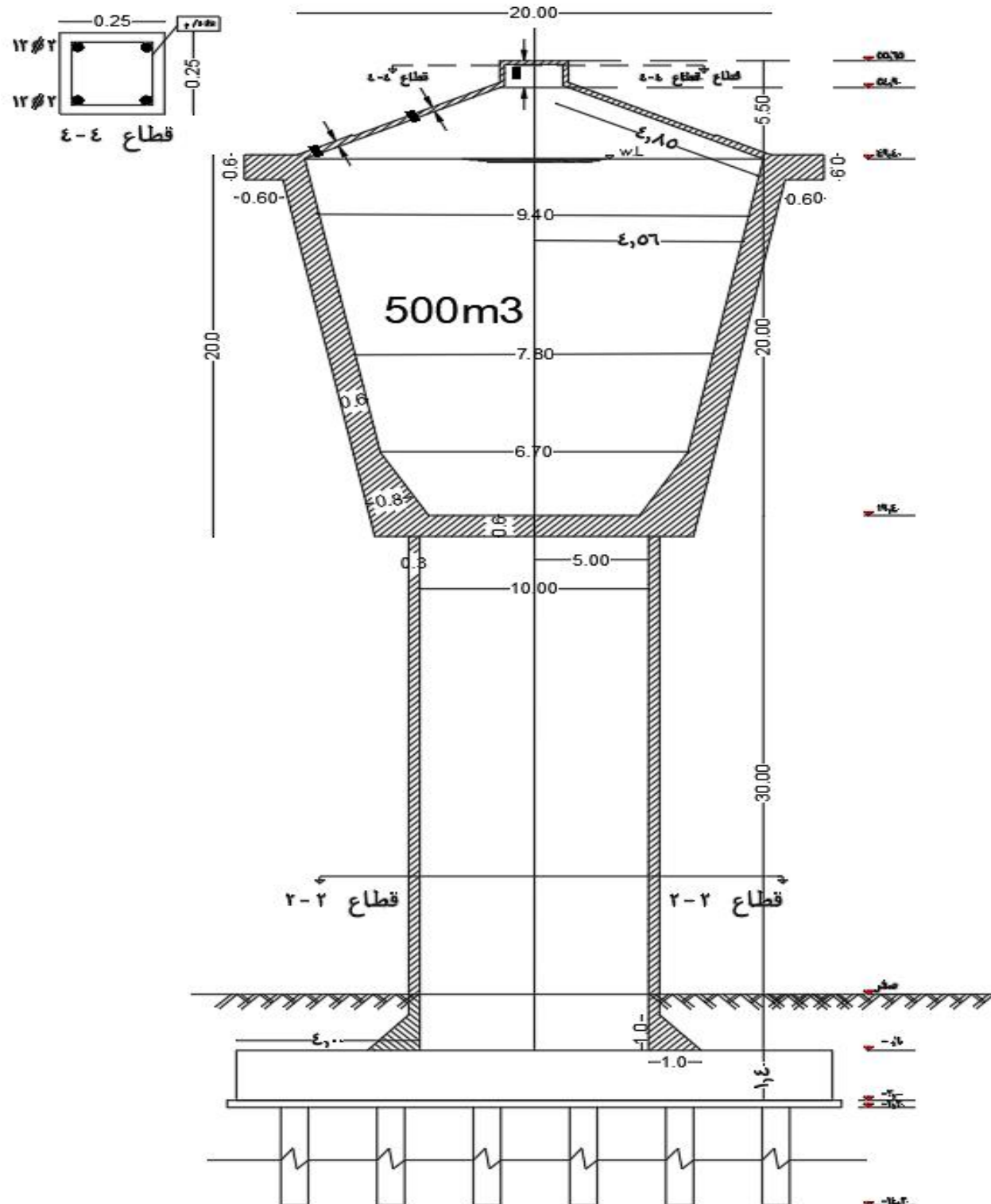


Figure 2.56 RFt of Retaining wall

## Unit (3) Elevated Tank



### **3.1 INTRODUCTION**

#### **3.1.1 Elevated tank consists of:**

- Covering Cone
- Horizontal R-Ring Beam
- Vertical Ring Beam
- posts
- Covering Cone of Tank
- Circular Wall
- Conical Wall
- Shaft
- Tank Supporting Tower
- Foundation

#### **3.1.2 Material Properties Used:**

- $F_{cu}=30 \text{ N/mm}^2$
- $F_y(\text{main steel}) = 360 \text{ N/mm}^2$
- $F_y(\text{stirrups}) = 240 \text{ N/mm}^2$
- Bearing Capacity of Soil =  $10 \text{ t/m}^2$

#### **3.1.3 Cover Thickness $\nless 5 \text{ cm}$**

#### **3.1.4 Loads Used:**

- $L.L = 0.1 \text{ t/m}^2$
- $\text{Cover} = 0.05 \text{ t/m}^2$

#### **3.1.5 Design Method:**

- Ultimate Limit State Design & Working Method Design



## ELEVATED TANK

### 3.2 Dimensions

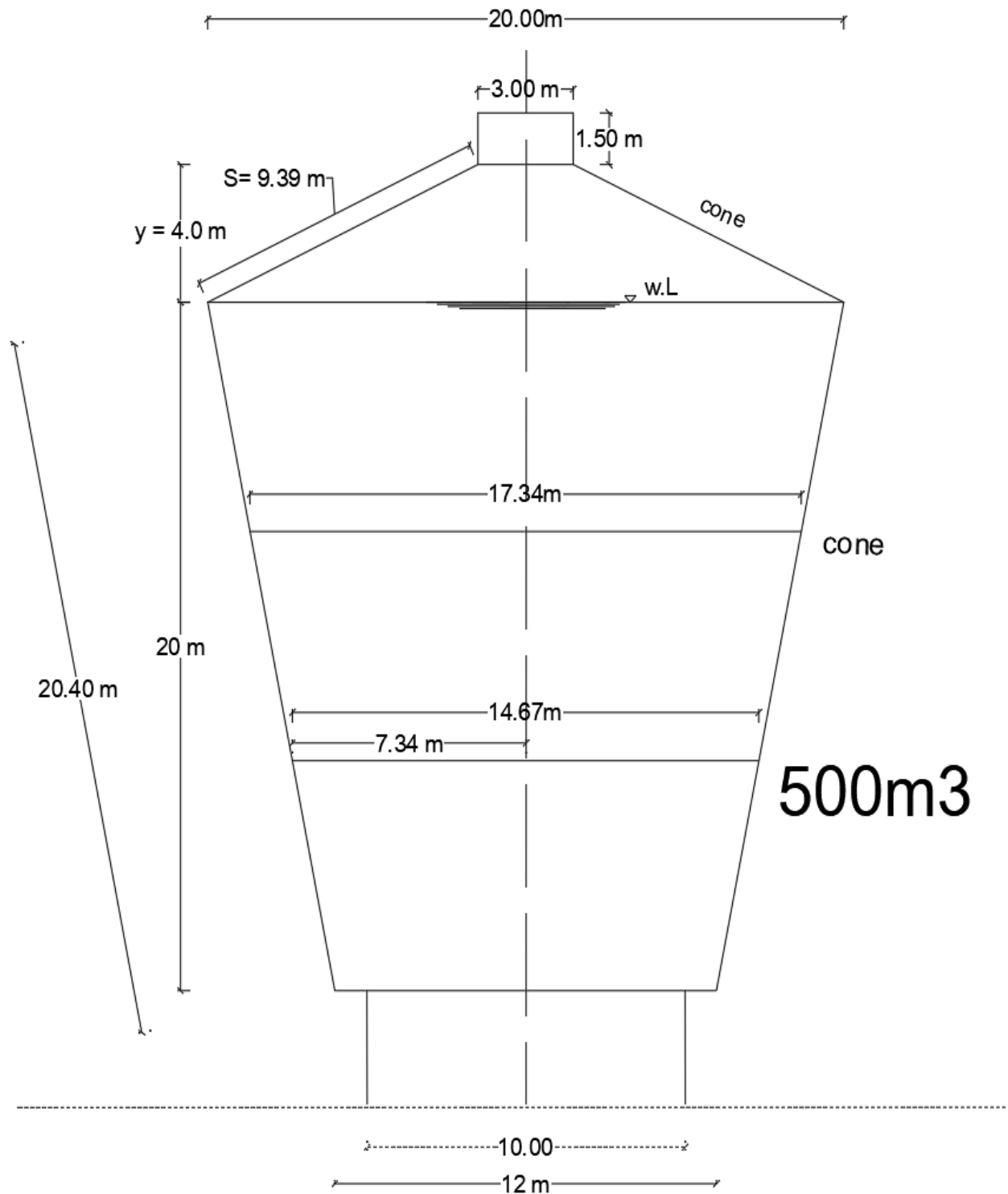


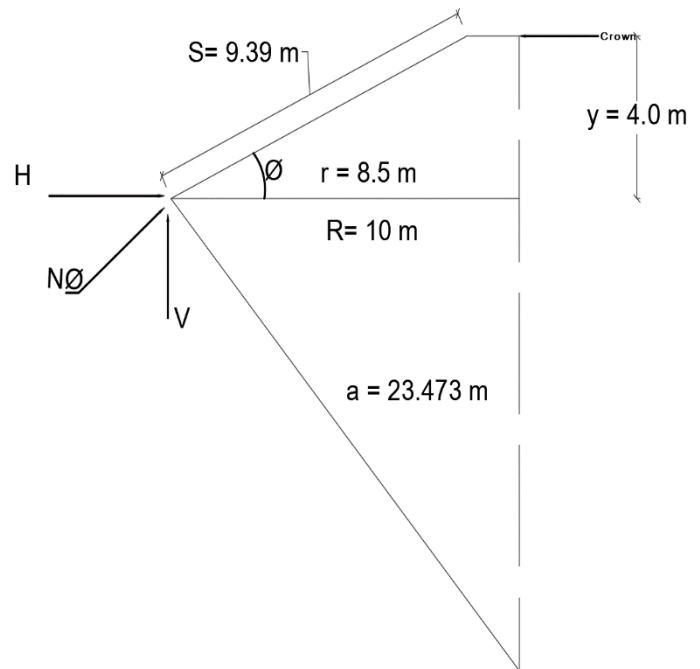
Figure 3.1 Elevated Tank Dimensions

## 3.3 Design

### 3.3.1 Covering cone

#### (1) Geometric Design

- $D = 20.00 \text{ m}$
- $r = \frac{D}{2} = \frac{20}{2} = 10 \text{ m}$
- $y = \frac{r}{2:4} = \frac{10}{2:4} = 4 \text{ m}$
- $S = \sqrt{r^2 + y^2} = \sqrt{8.5^2 + 4^2} = 9.39 \text{ m}$
- $a = \frac{r \cdot S}{y} = \frac{10 \cdot 9.39}{4} = 23.475 \text{ m}$
- $\tan \emptyset = \frac{4}{8.5}$
- $\emptyset = 25.20^\circ$
- Assume That:
  - $T_s = 0.12 \text{ m}$
  - Covering =  $0.05 \text{ t/m}^2$
  - L.L =  $0.1 \text{ t/m}^2$



#### (2) Loads

- $W = D.L + L.L = \text{own weight} + \text{Covering} + L.L$   
 $= (2.5 * 0.12) + 0.05 + 0.1 = 0.45 \text{ t/m}^2$

#### Posts

Height = 1.5 m  
 assume  $b_p = 25 \text{ cm}$   
 $t_p = 25 \text{ cm}$   
 No. of posts = 15

$$V (\text{vertical reaction}) = \frac{W \cdot A}{\pi \cdot D} = \frac{0.45 * \frac{\pi (3)^2}{4}}{\pi (3)} = 0.34 \text{ t/m}$$

$$\text{Load for Every post} = \frac{0.34 * \pi * 3}{15} + (2.5 * 0.25 * 0.25 * 1.5) = 0.44 \text{ ton}$$

$$A_s = \frac{1.4 * 0.44 * 1000}{\frac{3600}{1.15}} = 0.2 \text{ cm}^2$$

USE

4  $\emptyset$  12

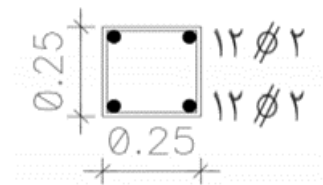


Figure3.2 Post reinforcement

## ELEVATED TANK

### (3) Internal Forces (Meridian Force & Ring Force )

#### At Crown

$$N_{\theta} = N_{\phi} = 0.0 \quad \text{Safe}$$

#### At Footing

- $W_{\phi} = W * A_{\text{surface}}$
- $W_{\phi} = 0.45 * \frac{2\pi r s}{2} = 0.45 * \frac{2\pi * 10 * 9.39}{2} = 132.747 \text{ ton}$

$$V_1(\text{vertical reaction}) = \frac{W_{\phi}}{2\pi r} = \frac{132.747}{2\pi * 10} = 2.112 \text{ t/m}$$

$$N_{\phi} = \frac{V_1}{\sin \phi} = \frac{2.112}{\sin(25.20)} = 4.95 \text{ t/m} \quad (\text{Compression})$$

$$N_{\theta} = P_r * a = W_{\phi} * \cos \phi * a = 0.45 * \cos(25.20) * 23.475 = 9.55 \text{ t/m} \quad (\text{Compression})$$

### (4) Check Of Stress

$$F_c = \frac{N_{\theta}}{b t} = \frac{9.55 * 10^3}{100 * 15} = 6.37 \text{ Kg/cm}^2 < F_{call} = 60 \text{ Kg/cm}^2 \quad \longrightarrow \text{ok safe}$$

### (5) Moment At Edge

$$X = 0.6 \sqrt{a * t} = 0.6 \sqrt{23.474 * 0.15} = 1.120 \text{ m}$$

$$M = \frac{W x^2}{2} = \frac{0.45 * 1.12^2}{2} = 0.28 \text{ t.m} \quad \text{Very small (neglected)}$$

Use 6  $\phi$  10 / m For ring and Meridian direction

## ELEVATED TANK

### 3.3.2 Horizontal R-Ring Beam

$$T = (H_1 + H_2) * r = \left( \frac{V}{\tan \theta} + \frac{V}{\tan \theta} \right)$$

$$= \left( \frac{2.11}{\frac{4}{10}} + \frac{2.11}{\frac{20}{4}} \right) * 10 = 56.97 \text{ ton}$$

Assume  $b = 0.5 \text{ t}$

$$A_c = b * T = 60 \text{ T}$$

$$0.5 \text{ t}^2 = 60 * 56.97$$

$$t = \sqrt{\frac{60 * 41.69}{0.5}} = 82.6 = 85 \text{ cm}$$

Use  $b = 60 \text{ cm}$  ,  $t = 120 \text{ cm}$

$$A_s = \frac{T}{f_s} = \frac{56.97 * 1000}{2000} = 28.48 \text{ cm}^2$$

$$A_{s(\text{modified})} = \frac{A_s}{\beta_{cr}} = \frac{28.48}{0.65} = 43.8 \text{ cm}^2$$

Use 10  $\phi$  25  $A_{s \text{ act}} = 49.08 \text{ cm}^2$

Use 5  $\phi$  25 / one side

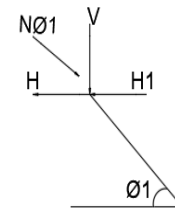
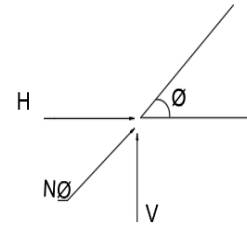
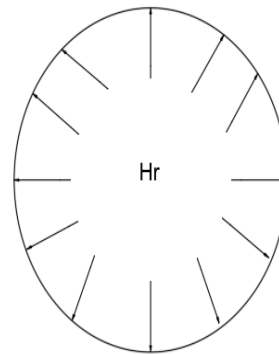


Figure 3.3 Ring Beam

#### (6) Check Of Ft

$$F_t = \frac{T}{A_c + n A_s} = \frac{56.97 * 1000}{(60 * 120) + 10(49.08)} = 7.41 \text{ Kg / cm}^2 < F_{all} = 18 \text{ Kg/cm}^2 , \text{ ok safe}$$

## ELEVATED TANK

### 3.3.3 Desing of conical wall of the tank:-

ASSume  $t = 60 \text{ cm}$

$$W = 0.60 * 2.5 = 1.5 \text{ t/m}$$

$$a1 = \frac{r1.s}{y} = \frac{10*6.8}{6.667} = 10.2 \text{ m}$$

$$a2 = \frac{r2.s}{y} = \frac{8.67*6.8}{6.667} = 8.84 \text{ m}$$

$$a3 = \frac{r3.s}{y} = \frac{7.34*6.8}{6.667} = 7.48 \text{ m}$$

$$a4 = \frac{r4.s}{y} = \frac{6*6.8}{6.667} = 6.11 \text{ m}$$

$$\phi1 = 78.69^\circ$$

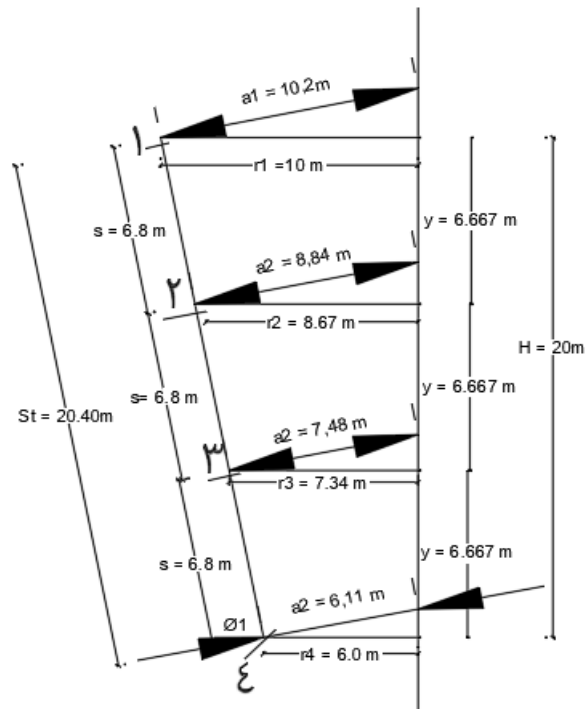


Figure 3.4 Conical wall

At sec ( 1 ) :-  $V1 = 2.11 \text{ t/m}$  = Vertical reaction of the covering cone

$$N\phi1 = (W * \cos \phi1 + \gamma_w h1) * a1$$

$$= (1.50 * \cos (78.69)) * 10.2 = 3 \text{ t/m (tension)}$$

$$N \phi1 = \frac{V1}{\sin \phi1} = \frac{2.11}{\sin(78.69)} = 2.15 \text{ t/m (comp)}$$

At sec ( 2 ) :-

$$N\phi2 = (W * \cos \phi1 + \gamma_w h2) * a2$$

$$= (1.50 * \cos (78.69) + 1 * 6.667) * 8.84 = 61.53 \text{ t/m (tension)}$$

$W \phi2$  = weight of cone above sec + weight of covering cone + weight of water above sec

$$W_{\text{cone}} = W * A_{\text{sur}} = W\pi (r1 + r2) * s = 1.5 * \pi (10 + 8.67) * 6.8 = 598.26 \text{ ton}$$

$$W_{\text{cover}} = V1 * \pi D = 2.11 * \pi * 20 = 132.57 \text{ ton}$$

$$W_{\text{water}} = \gamma_w * V_w = \gamma_w * A1 * 2\pi r_{cg} = 1 * 0.5 * 1.13 * 6.667 (2\pi) (8.67 + \frac{1.13}{3}) = 214.12 \text{ ton}$$

$$W \phi2 = 598.26 + 132.57 + 214.12 = 944.94 \text{ ton}$$

$$V2 = \frac{W \phi2}{\pi r2} = \frac{944.94}{\pi * 8.67} = 34.7 \text{ ton/m}$$

$$N \phi2 = \frac{V2}{\sin \phi1} = \frac{34.7}{\sin(78.69)} = 35.38 \text{ t/m (comp)}$$

## ELEVATED TANK

At sec ( 3 ) :-

$$N\theta 3 = ( W * \cos \theta 1 + \gamma w h 3 ) * a 3 \\ = ( 1.50 * \cos ( 78.69 ) + 1 * 13.334 ) * 7.48 = 101.93 \text{ t/m ( tension )}$$

W Ø3 = weight of cone above sec + weight of covering cone + weight of water above sec

$$W_{\text{cone}} = W * A_{\text{sur}} = W\pi ( r_2 + r_3 ) * s = 1.5 * \pi ( 8.67 + 7.34 ) * 6.8 = 513.02 \text{ ton}$$

$$W_{\text{cover}} = V_1 * \pi D = 2.11 * \pi * 20 = 132.57 \text{ ton}$$

$$W_{\text{water}} = \gamma w * V_w = \gamma w * A_1 * 2 \pi r_{cg} = 1 * 0.5 * 2.66 * 13.334 ( 2 \pi ) ( 7.34 + \frac{2.66}{3} ) = 916.67 \text{ ton}$$

$$W \text{ Ø} 3 = 513.02 + 132.57 + 916.67 = 1562.26 \text{ ton}$$

$$V_3 = \frac{W \text{ Ø} 3}{\pi r^3} = \frac{1562.26}{\pi * 7.34} = 67.74 \text{ ton/m}$$

$$N \text{ Ø} 3 = \frac{V_3}{\sin \theta 1} = \frac{67.74}{\sin ( 78.69 )} = 69.08 \text{ t/m ( comp )}$$

At sec ( 4 ) :-

$$N\theta 4 = ( W * \cos \theta 1 + \gamma w h 4 ) * a 4 \\ = ( 1.50 * \cos ( 78.69 ) + 1 * 20 ) * 6.11 = 123.99 \text{ t/m ( tension )}$$

W Ø3 = weight of cone above sec + weight of covering cone + weight of water above sec

$$W_{\text{cone}} = W * A_{\text{sur}} = W\pi ( r_3 + r_4 ) * s = 1.5 * \pi ( 7.34 + 6 ) * 6.8 = 427.47 \text{ ton}$$

$$W_{\text{cover}} = V_1 * \pi D = 2.11 * \pi * 20 = 132.57 \text{ ton}$$

$$W_{\text{water}} = \gamma w * V_w = \gamma w * A_1 * 2 \pi r_{cg} = 1 * 0.5 * 4 * 20 ( 2 \pi ) ( 6 + \frac{4}{3} ) = 1843.06 \text{ ton}$$

$$W \text{ Ø} 4 = 427.47 + 132.57 + 1843.06 = 2400.1 \text{ ton}$$

$$V_4 = \frac{W \text{ Ø} 4}{\pi r^4} = \frac{2400.1}{\pi * 6} = 127.32 \text{ ton/m}$$

$$N \text{ Ø} 4 = \frac{V_4}{\sin \theta 1} = \frac{127.32}{\sin ( 78.69 )} = 129.841 \text{ t/m ( comp )}$$

### Calculate As

At sec ( 2 )

$$A_s = \frac{N\theta}{f_s} = \frac{61.53 * 1000}{0.9 * 2000} = 34.18 \text{ cm}^2$$

Use 16  $\phi$  18  $A_{s \text{ act}} = 40.7 \text{ cm}^2$

Use 8  $\phi$  18 / one side

## ELEVATED TANK

At sec ( 3 )

$$A_s = \frac{N\theta}{f_s} = \frac{101.93 \times 1000}{0.8 \times 2000} = 63.70 \text{ cm}^2$$

Use 18  $\phi$  22  $A_{s \text{ act}} = 69.11 \text{ cm}^2$

Use 9  $\phi$  22 / one side

At sec ( 4 )

$$A_{s \text{ min}} \geq 0.15\% \cdot b \cdot d \text{ or } 0.6 \cdot b \cdot d / f_y$$

$$A_{s \text{ min}} \geq 0.15\% \cdot 1000 \cdot 750 \text{ or } 0.6 \cdot 1000 \cdot 750 / 360 = 1125 \text{ mm}^2 \text{ or } 1250 \text{ mm}^2$$

$$A_{s \text{ min}} = 1250 \text{ mm}^2$$

$$A_{s \text{ modify}} = 1250 / 0.85 = 1470.58 \text{ mm}^2$$

Use 8  $\phi$  18 / m  $A_{s \text{ act}} = 20.35 \text{ cm}^2$

### 3.3.4 Desing of Floor of the tank:-

#### (1) Loads

$$\text{Assume } t_f = 60 \text{ cm}$$

$$W = D.L + \gamma_w \cdot h = \gamma_c \cdot t_f + \gamma_w \cdot h$$

$$W = (2.5 \cdot 0.6) + 1 \cdot 20 = 21.5 \text{ t/m}^2$$

$$N_f = N\phi 4 \cdot \cos \phi 1 = 129.841 / \cos(78.69) = 662 \text{ t}$$

#### (2) Rdial moment of floor slab $M_r$ at point 5&6

$$M_{r5} = -0.125 W R_4^2 = -0.125 \cdot 21.5 \cdot 6^2 = -96.75 \text{ t.m (tension at inner face)}$$

$$M_{r6} = 0.075 W R_4^2 = 0.075 \cdot 21.5 \cdot 6^2 = 58.05 \text{ t.m (tension at outer face)}$$

#### (3) Tangention moment of floor slab $M_r$ at point 5&6

$$M_{t5} = -0.025 W R_4^2 = 0.025 \cdot 21.5 \cdot 6^2 = -19.35 \text{ t.m (tension at inner face)}$$

$$M_{t6} = 0.075 W R_4^2 = 0.075 \cdot 21.5 \cdot 6^2 = 58.05 \text{ t.m (tension at outer face)}$$

## ELEVATED TANK

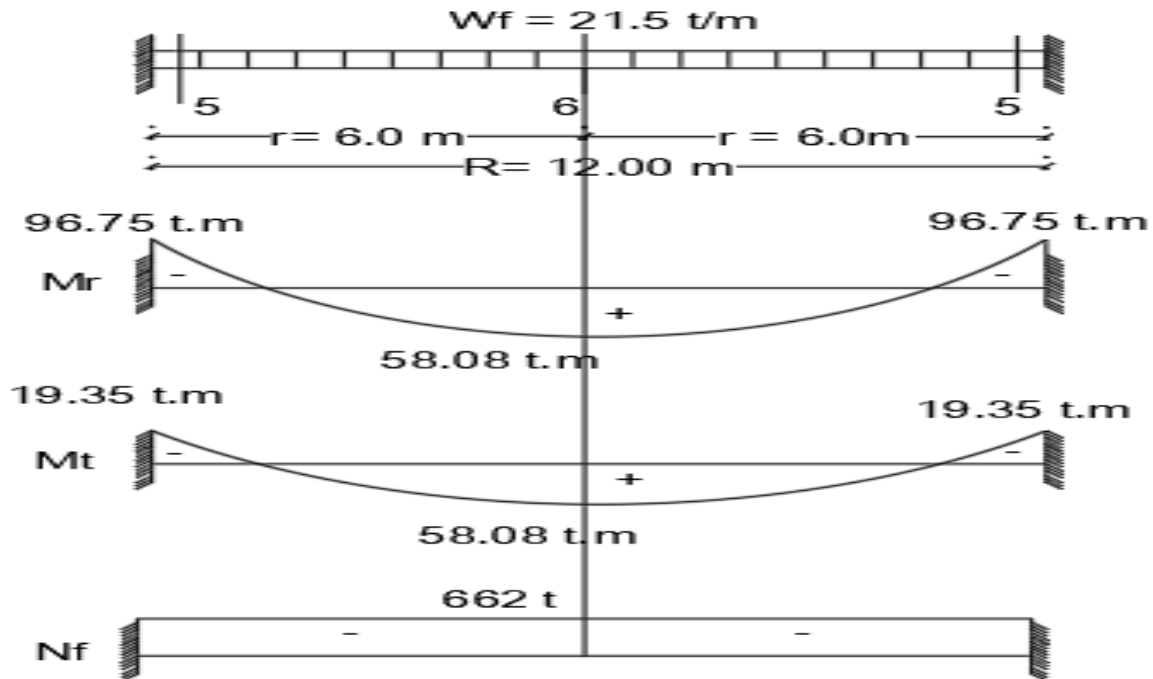


Figure 3.5 Moment in circular floor of Tank

### (4) Design

#### Sec.5: Radial

$$M = 96.75 \text{ t.m} \quad N = -662 \text{ t}$$

$$t = \sqrt{\frac{M}{3b}} - 3\text{cm} = \sqrt{\frac{96.75 \times 10^5}{3 \times 100}} - 3\text{cm} = 179.58 \text{ cm} \quad \text{Use } t = 180 \text{ cm} \quad \rightarrow \text{use hunch}$$

$$\therefore T_f = 80 \text{ cm}$$

#### check of tensile strength (ft)

$$F(t) = F_{ct}(N) + F_{ct}(M) < \frac{F_{ct}}{\eta_1 \eta_2}$$

$$F_t = \frac{6M}{bt^2} - \frac{N}{bt} = \frac{6 \times 96.75 \times 10^5}{100 \times 80^2} - \frac{662 \times 10^3}{100 \times 80} = 7.95 \text{ Kg/cm}^2 < F_{cto} = \frac{F_{ctr}}{\xi} = 17.6 \text{ Kg/cm}^2$$

#### Calculate As

$$\bullet \quad e = \frac{M}{N} = \frac{96.75}{662} = 0.146 < \frac{t}{2} - \text{cover} = \frac{0.80}{2} - 0.05 = 0.35$$



## ELEVATED TANK

- $es = e + \frac{t}{2} - \text{cover} = 0.146 + 0.4 - 0.03 = 0.516 \text{ m} \rightarrow \text{Small eccentricity}$
- $K = \frac{P}{f_c * b * t} = \frac{662 * 10^3}{73 * 100 * 80} = 1.133 \rightarrow \text{this point under curve use minimum reinforcement for steel}$   
360/520
- $A_s \min \geq 0.15\% * b * d \text{ or } 0.6 * b * d / f_y$
- $A_s \min \geq 0.15\% * 1000 * 750 \text{ or } 0.6 * 1000 * 750 / 360 = 1125 \text{ mm}^2 \text{ or } 1250 \text{ mm}^2$   
 $A_s \min = 1250 \text{ mm}^2$
- $A_s \text{ modify} = \frac{1250}{0.85} = 1470.58 \text{ mm}^2$
- Use 8  $\phi$  18 / m  $\rightarrow A_{sact} = 20.35 \text{ cm}^2$

### Sec.6: Radial

$M = 58.08 \text{ t.m}$        $N = -662 \text{ t}$

#### Calculate $A_s$

- $e = \frac{M}{N} = \frac{58.08}{662} = 0.087 < \frac{t}{2} - \text{cover} = \frac{0.80}{2} - 0.05 = 0.35 \rightarrow \text{Small eccentricity}$
- $K = \frac{P}{f_c * b * t} = \frac{662 * 10^3}{73 * 100 * 80} = 1.13 \rightarrow \text{this point under curve use minimum reinforcement for steel 360/520}$
- $A_s \min \geq 0.15\% * b * d \text{ or } 0.6 * b * d / f_y$
- $A_s \min \geq 0.15\% * 1000 * 750 \text{ or } 0.6 * 1000 * 750 / 360 = 1125 \text{ mm}^2 \text{ or } 1250 \text{ mm}^2$   
 $A_s \min = 1250 \text{ mm}^2$
- $A_s \text{ modify} = \frac{1250}{0.85} = 1470.58 \text{ mm}^2$
- Use 8  $\phi$  16 / m  $\rightarrow A_{sact} = 20.35 \text{ cm}^2$

## ELEVATED TANK

### Sec.5: Tangention

Calculate As

- $d = C1 \sqrt{\frac{M}{F_{cu} * b}} \Rightarrow 550 = C1 \sqrt{\frac{19.35 * 1.5 * 10^7}{25 * 1000}} \quad C1 = 5.10 \Rightarrow j = .826$
- $A_s = \frac{M}{F_y * j * d} = \frac{19.35 * 1.5 * 10^7}{360 * 0.826 * 550} = 1774.70 \text{ mm}^2$
- $A_s \text{ modify} = \frac{1774.70}{0.85} = 2087.88 \text{ mm}^2$
- Use 9  $\phi$  18/m  $\Rightarrow A_{sact} = 2290.2 \text{ mm}^2$

### Sec.6: Tangention

M= 58.08 t.m

- $d = C1 \sqrt{\frac{M}{F_{cu} * b}} \Rightarrow 550 = C1 \sqrt{\frac{58.08 * 1.5 * 10^7}{25 * 1000}} \quad C1 = 2.94 \Rightarrow j = 0.75$
- $A_s = \frac{M}{F_y * j * d} = \frac{58 * 1.5 * 10^7}{550 * 0.740 * 360} = 5858 \text{ mm}^2$
- Use 10  $\phi$  20 /m  $\Rightarrow A_{sact} = 3141.5 \text{ mm}^2$

| Section     | M(t.m ) | N(ton )  | Stage | T ( cm ) | A <sub>s</sub>  |
|-------------|---------|----------|-------|----------|-----------------|
| 1           | -----   | -2.15    | 1     | 60       | 8 $\phi$ 16 /m  |
| 2           | -----   | -35.38   | 1     | 60       | 8 $\phi$ 18 /m  |
| 3           | -----   | - 69.08  | 1     | 60       | 8 $\phi$ 18 /m  |
| 4           | 96.75   | -129.841 | 1     | 80       | 8 $\phi$ 18 /m  |
| 5( radial ) | 96.75   | -662     | 1     | 80       | 8 $\phi$ 18 /m  |
| 6( radial ) | 58.08   | -662     | 2     | 80       | 8 $\phi$ 16 /m  |
| 5( tang )   | 19.35   | -----    | 1     | 60       | 9 $\phi$ 18 /m  |
| 6( tang )   | 58.08   | -----    | 2     | 60       | 10 $\phi$ 20 /m |

Figure 3.6 Table RFT Floor

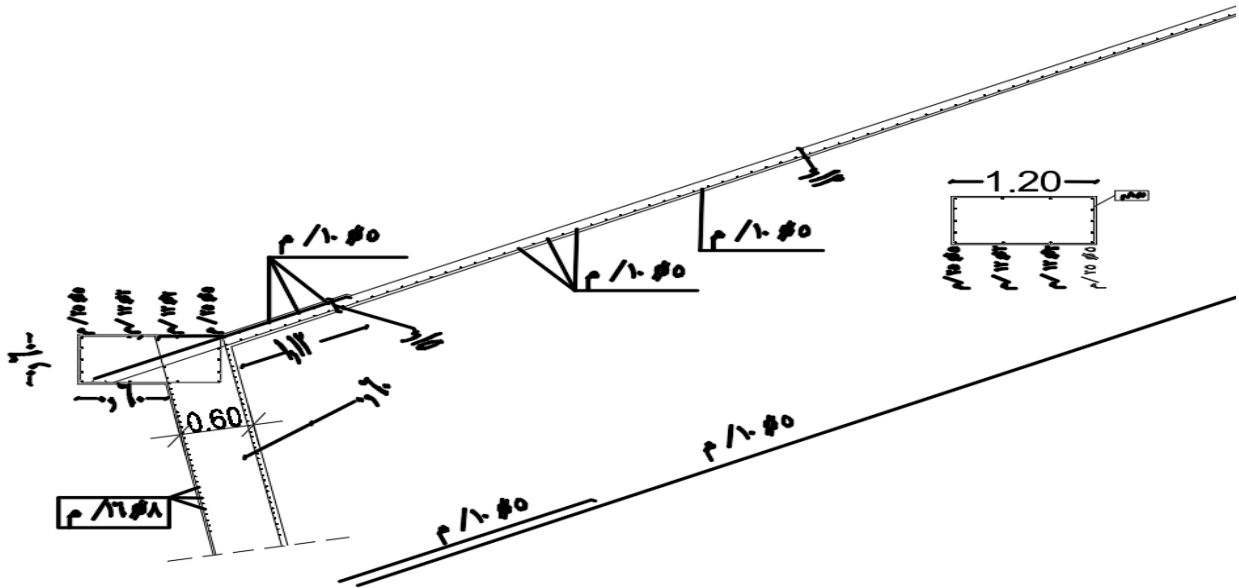


Figure 3.7 RFT details

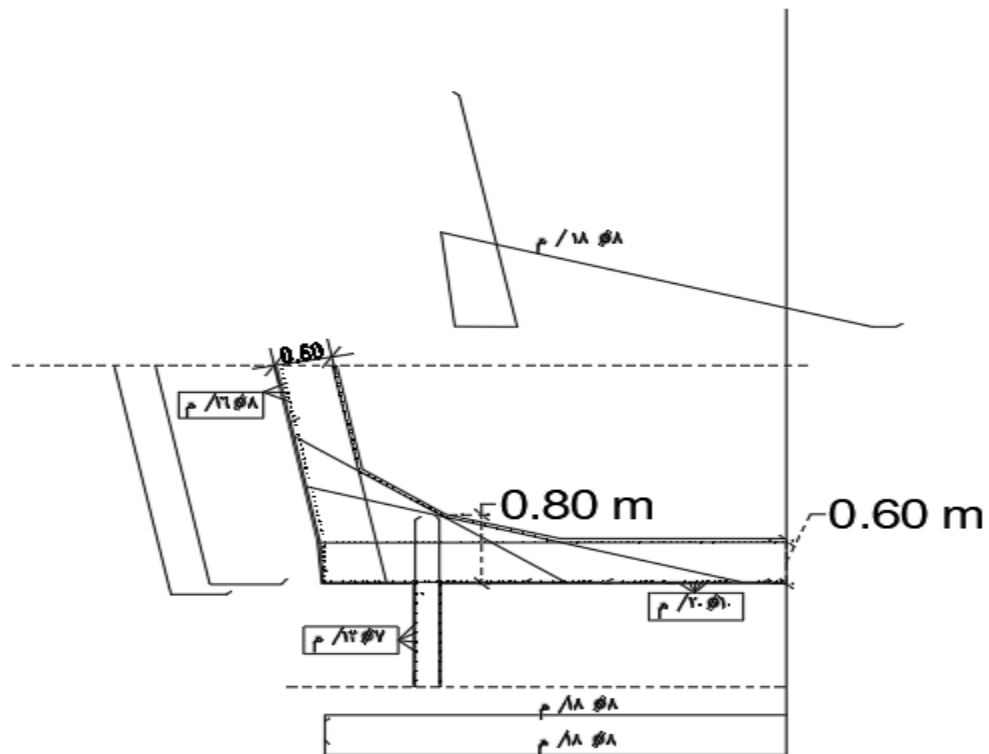


Figure 3.8 Floor RFT details

### 3.3.5 Design of water tower :-

- Assume thickness of shaft = 30 cm

#### Check of stability :-

##### ❖ Vertical load :-

- $W_{\text{cone}} = W * A_{\text{surface}} = 0.45 * 132.74 = 59.73 \text{ ton}$
- $W_{B1} = \alpha_c * t_b * B * \pi d = 2.5 * 0.60 * 1.20 * \pi * 10 = 56.54 \text{ ton}$
- $W_{\text{conical wall}} = W * A = 2.5 * 0.45 * \pi * \frac{9.7^2}{4} = 26.46 \text{ ton}$
- $W_{\text{water}} = \alpha_w * \pi w_{\text{er}} = 1 * 500 = 500 \text{ ton}$
- $W_{\text{shaft}} = 2.5 * 30 * \frac{\pi * (10^2 - 9.40^2)}{4} = 685.65 \text{ t}$

Case of empty tank =  $\sum$  vertical long with Wt of water  
= 828.38 ton  $\approx$  828.4 ton

Case of Full tank =  $\sum$  vertical long with Wt of water = 828.38 + 500 = 1328.38 ton

##### ❖ Horizontal loads: -

##### → earthquake load

$F_b = S_d(T_1) * \lambda * \frac{W}{g} \rightarrow$  ultimate base shear force

$$T_1 = C_t * H^{3/4} = 0.075 * (50)^{3/4} = 1.41 \text{ sec}$$

##### ❖ Seismic zone: -

Second zone ( $a_g = 0.125 \text{ g} = 0.125 * 9.81 = 1.23 \text{ m/sec}^2$ )

##### ❖ Soil type (c): -

$S = 1.50$  ,  $T_b = 0.10$  ,  $T_c = 0.25$  ,  $T_d = 1.20$

$$T_c \leq T_1 \leq T_d$$

$$S_d(T_1) = a_g * Y_I * s * \frac{2.5}{R} \left( \frac{T_c}{T_1} \right) \geq 0.20 a_g * Y_I$$

- $Y_I$  (importance factor) = 1.2
- $R$  (reduction factor) = 9.40

$$S_d(T_1) = 1.23 * 1.2 * 1.5 * \frac{2.5}{9.40} \left( \frac{0.25}{1.41} \right) = 0.1043 \geq 0.20 * 1.23 * 1.2 = 0.3$$

Use: -

- $S_d(T_1) = 0.3$

## ELEVATED TANK

- $\lambda = 1 \rightarrow$  as  $T1 > 2Tc$
- $W_{\text{tank}} = 1328 \text{ ton}$

$$F_b = 0.3 * 1 * \frac{1328}{9.81} = 40.61 \text{ ton}$$

$$M_{\text{Over turning}} = 40.61 * 50 = 2030.5 \text{ m.t}$$

$$M_{\text{stability}} = W_t. (\text{without water}) * R = 828 * 9.40 = 7783.2 \text{ m.t}$$

$$M_{\text{stability}} = W_t. (\text{with water}) * R = 1328 * 9.40 = 12483.2 \text{ m.t}$$

### Check of overturning :

$$\text{Factor of safety} = \frac{M_{\text{stability}}}{M_{\text{Overturning}}}$$

- ❖ F.O.S (with out water) =  $\frac{7783.2}{2030.5} = 3.83 > 1.5$  OK ,safe
- ❖ F.O.S (with water) =  $\frac{12483.2}{2030.5} = 6.147 > 1.5$  OK ,safe

### Check the sliding :

- ❖ Resisting force =  $M * W_{\text{total}}$ 
  - With out water =  $0.3 * 828 = 248.4 \text{ ton}$
  - With water =  $0.3 * 1328 = 398.4 \text{ ton}$

$$\text{Factor of safety} = \frac{\text{Resisting}}{\text{sliding force}}$$

- ❖ F.O.S (with out water) =  $\frac{248.4}{40.61} = 6.11 > 1.5$  OK ,safe
- ❖ F.O.S (with water) =  $\frac{398.4}{40.61} = 9.81 > 1.5$  OK ,safe

### Wind load ( height = 54 m )

$$P_e = q * C_e * K$$

$$q = 0.5 * 10^{-3} * P * V^2 * C_e * C_s$$

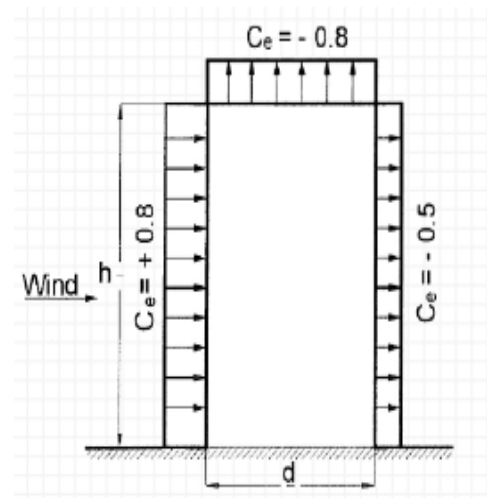
$$P = 1.25 \text{ kg/m}^3 \rightarrow \text{كثافة الهواء}$$

$$V = 30 \text{ m / sec} \rightarrow \text{سرعة الرياح الأساسية}$$

$$C_e = 1 \rightarrow \text{معامل طبوغرافية الأرض}$$

$$C_s = 1 \rightarrow \text{معامل المنشأ}$$

$$\therefore q = 0.5 * 10^{-3} * 1.25 * (30)^2 * 1 * 1 = 0.5625 \text{ KN/m}^2 \rightarrow \text{ضغط الرياح الأساسي}$$



## ELEVATED TANK

$C_e = 0.8 + 0.5 = 1.3 \rightarrow$  معامل ضغط الرياح  $\rightarrow$  Assume Zone (A) as more critical

$$Pe_1 = q * Ce * K_{0-10} = 0.5625 * 1.3 * 1 = 0.73125 \text{ KN /m}^2$$

$$Pe_2 = q * Ce * K_{10-20} = 0.5625 * 1.3 * 1.15 = 0.84 \text{ KN /m}^2$$

$$Pe_3 = q * Ce * K_{20-30} = 0.5625 * 1.3 * 1.4 = 1.024 \text{ KN/m}^2$$

$$Pe_4 = q * Ce * K_{30-50} = 0.5625 * 1.3 * 1.6 = 1.17 \text{ KN /m}^2$$

$$Pe_5 = q * Ce * K_{50-80} = 0.5625 * 1.3 * 1.85 = 1.353 \text{ KN/m}^2$$

$$\underline{\mathbf{F_i} = \mathbf{P_{ei}} * (\mathbf{hs*bs})}$$

$$F5 = 1.353 * 4 * 1.5 = 8.118 \text{ KN}$$

$$F_4 = 1.17 * 20 * 10 = 234 \text{ KN}$$

$$F_3 = 1.024 \cdot 10^6 = 61.44 \text{ kN}$$

$$F_2 = 0.84 \cdot 10 \cdot 6 = 50.4 \text{ KN}$$

$$F1 = 0.73125 * 10 * 6 = 43.875 \text{ KN}$$

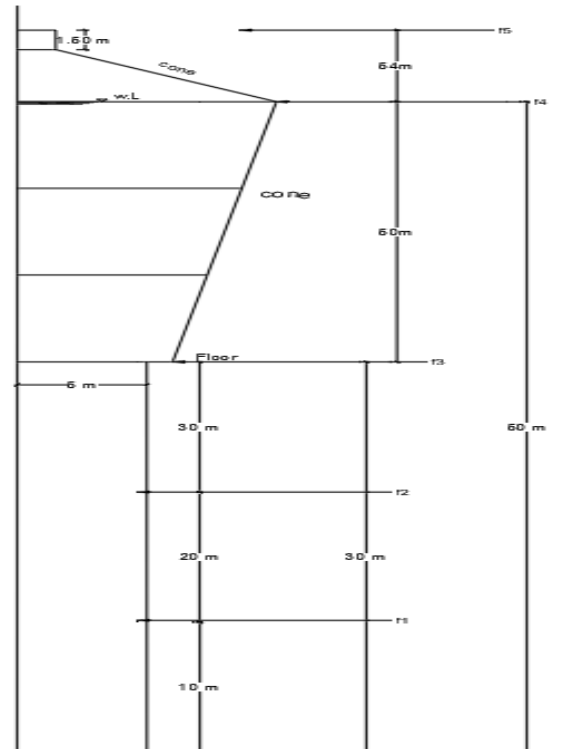


Figure3.9 Wind load

$$\begin{aligned} M_{\text{over turning}} &= \sum f * h \\ &= (43.875 * 10) + (50.4 * 20) + (61.4 * 30) + (234 * 50) + (8.118 * 54) \\ &= 15427 \text{ KN.m} = 1542.7 \text{ ton.m} \end{aligned}$$

$$M_{\text{over turning (wind load)}} < M_{\text{over turning (earthquakes)}} \rightarrow \text{wind load is safe}$$

### 3.3.6 Check of stresses between shaft and foundation :-

$$F_{1,2} = \frac{-N}{A} \pm \frac{M_x}{I_x} * y \leq 1.25 * F_c \text{ all } = 1.25 * 105 = 131.25 \text{ kg/cm}^2$$

$$N = 1328 \text{ ton}$$

$$M_x = 2030.5 \text{ m.t}$$

$$A = \frac{\pi * (10^2 - 9.40^2)}{4} = 9.14 \text{ m}^2$$

$$I_x = \frac{\pi * (10^4 - 9.40^4)}{64} = 107.62 \text{ m}^4$$

$$Y = 5 \text{ m}$$

$$F_{1,2} = \frac{-1328}{9.14} \pm \frac{2030.5}{107.62} * 5$$

$$F_1 = -50.95 \text{ t/m}^2 = -5.095 \text{ kg/cm}^2 < 131.25 \text{ kg/cm}^2$$

$$F_2 = -239.63 \text{ t/m}^2 = -23.963 \text{ kg/cm}^2 < 131.25 \text{ kg/cm}^2$$

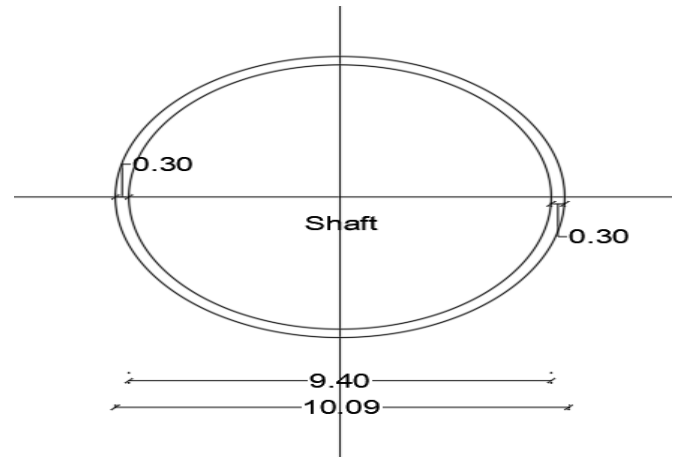


Figure 3.10 Dimensions Shaft

Ok, Safe, no tension

### 3.3.7 Design of Shaft

$$P = 1328 \text{ t}$$

$$M = 2030.5 \text{ m.t}$$

$$r = 5 \text{ m}$$

$$r_s = 4.70 \text{ m}$$

$$r' = r - \text{cover} = 5 - 0.05 = 4.95 \text{ m}$$

$$A_c = \pi * (r^2 - r_s^2) = \pi * (5^2 - 4.70^2) = 9.14 \text{ m}^2$$

$$\frac{P_u}{f_{cu} * A_c} = \frac{1328 * 10^3}{250 * 9.14 * 100^2} = 0.058$$

$$\frac{M_u}{f_{cu} * A_c * r} = \frac{2030.5 * 10^5}{250 * 9.14 * 100^2 * 500} = 0.0177$$

$$\eta = \frac{r'}{r} = \frac{4.95}{5} = 0.99$$

## ELEVATED TANK

$$\frac{rs}{r} = \frac{4.70}{5} = 0.94$$

$$\mu = \rho * f_{cu} * 10^{-4} = 2 * 25 * 10^{-4} = 0.005$$

$$\mu_{min} = 0.008$$

$$A_{s \text{ total}} = \mu_{min} * A_c = 0.008 * 9.14 * (100)^2 = 731.2 \text{ cm}^2$$

$$\text{Use } 250 \text{ } \phi 20 \quad A_s = 785.39 \text{ cm}^2$$

$$A_s / \text{side} = 392.69 \text{ cm}^2 \rightarrow 125 \text{ } \phi 20 \rightarrow \text{use } 7 \text{ } \phi 20 / \text{m} \text{ ( for vertical bars )}$$

$$\text{Use } 7 \text{ } \phi 12 / \text{m} \text{ ( for horizontal bars ) } \dots\dots \text{ stirrups}$$

### 3.3.8 Design of foundation

$$\text{Assume } t_f = 1.4 \text{ m}$$

$$W_t = W_{\text{tank}} + O.W_{\text{raft}} = 1328 + 2.5 * 1.40 * \frac{\pi * (10)^2}{4} = 1602.88 \text{ t}$$

$$\text{NO. of piles} = (1.2:1.4) * \frac{W_t}{\text{pile capacity}} = 1.2 * \frac{1602.88}{100} = 19.2 = 19 \text{ piles}$$

$$F_{1,2} = \frac{-N}{n} \pm \frac{M}{\sum r^2} * r_i$$

$$N = 1328 \text{ t}$$

$$M = 2030.5 \text{ m.t}$$

$$n = \text{use } 25 \text{ pile}$$

$$\sum r^2 = (12 * 5.62^2) + (9 * 4.20^2) + (4 * 1.76^2) = 532.52 \text{ m}^2$$

#### For point (1)

$$F_{1,2} = \frac{-1328}{25} \pm \frac{2030.5}{532.52} * 5.62$$

$$F_1 = -31.69 \text{ t} < 100 \text{ t}$$

$$F_2 = -74 \text{ t} < 100 \text{ t}$$

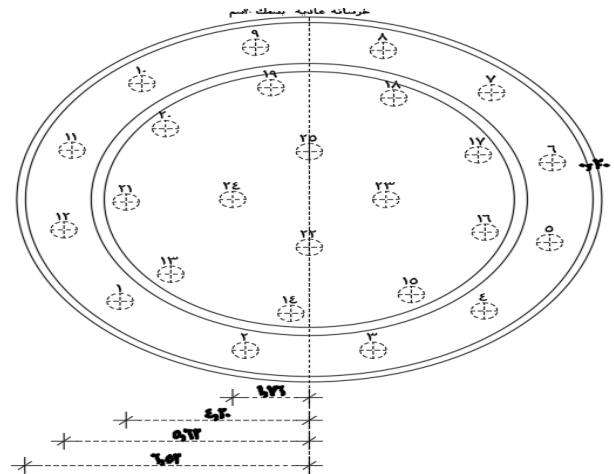


Figure 3.11 Piles



## ELEVATED TANK

### For point (2)

$$F_{1,2} = \frac{-1328}{25} \pm \frac{2030.5}{532.52} * 4.20$$

$$F_1 = -37.105 \text{ t} < 100 \text{ t}$$

$$F_2 = -69.13 \text{ t} < 100 \text{ t}$$

### For point (3)

$$F_{1,2} = \frac{-1328}{25} \pm \frac{2030.5}{532.52} * 1.76$$

$$F_1 = -46.40 \text{ t} < 100 \text{ t}$$

$$F_2 = -59.83 \text{ t} < 100 \text{ t}$$

For Raft:

$$A_s = \frac{M_u}{f_y * j * d}$$

$$M_u = A_s * f_y * j * d = 8 * \frac{\pi * 2^2}{4} * 3600 * 0.826 * 133 * 10^{-5} = 99.39 \text{ t.m}$$

$$A_{s_{act}} = \frac{M_u}{f_y * j * d} = \frac{99.39 * 10000}{360 * 0.826 * 133} = 25.132 \text{ cm}^2$$

$$A_s = \frac{0.15}{100} * b * d = \frac{0.15}{100} * 1000 * 1330 = 1995 = 19.95 \text{ cm}^2$$

Use

8  $\phi$  20 / m` upper reinforcement

8  $\phi$  20 / m` lower reinforcement

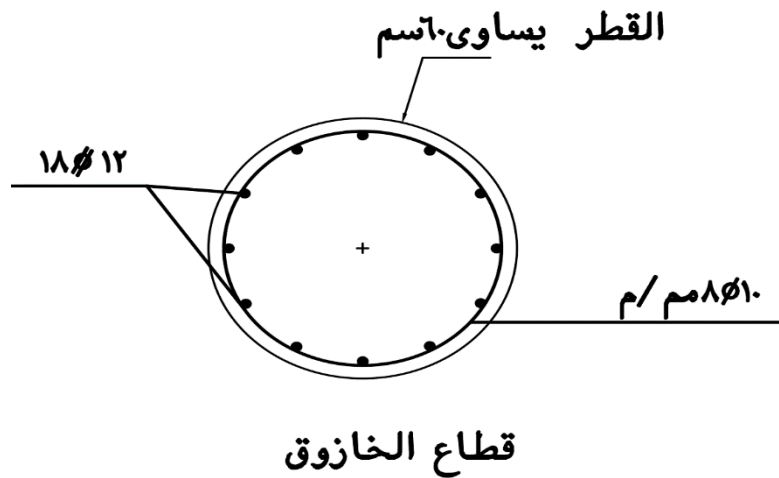


Figure 3.12 sec Pile

## ELEVATED TANK

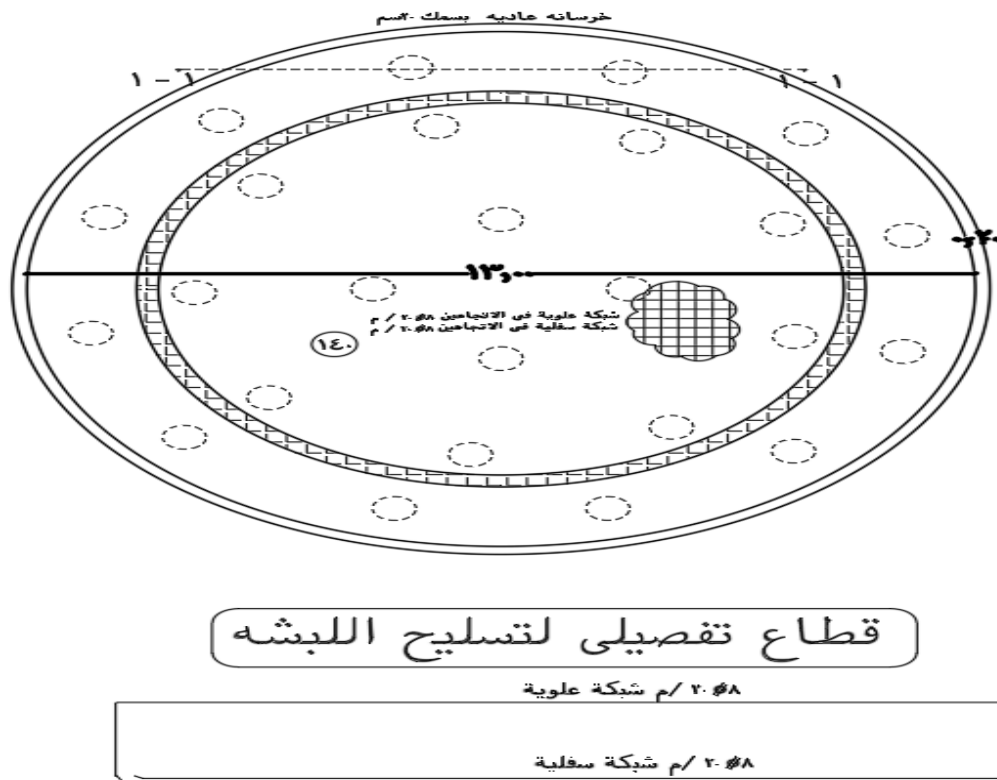


Figure 3.13 Raft RFT details

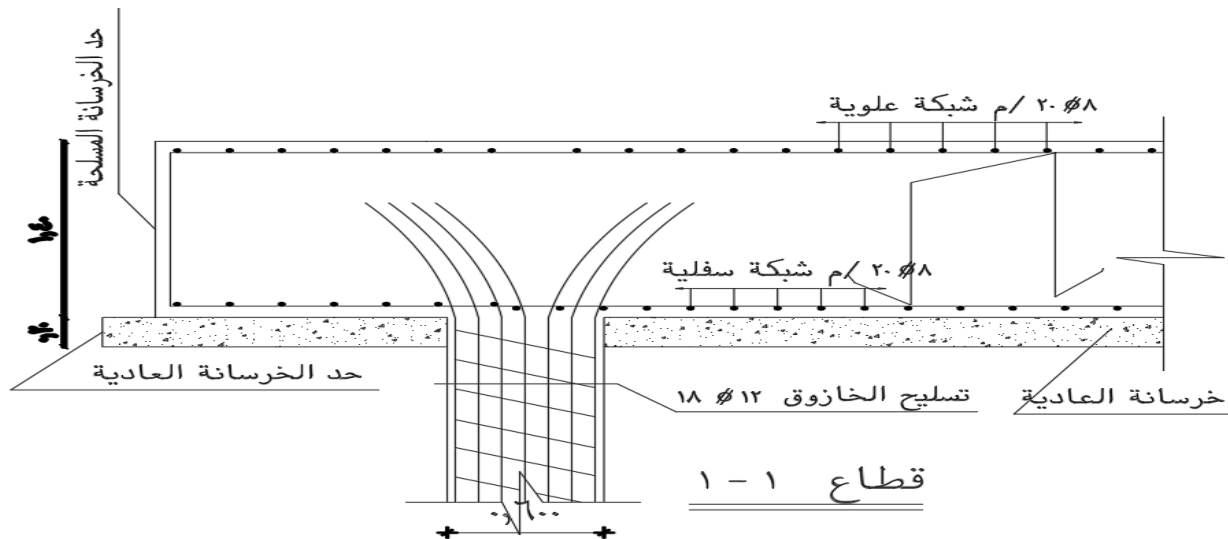
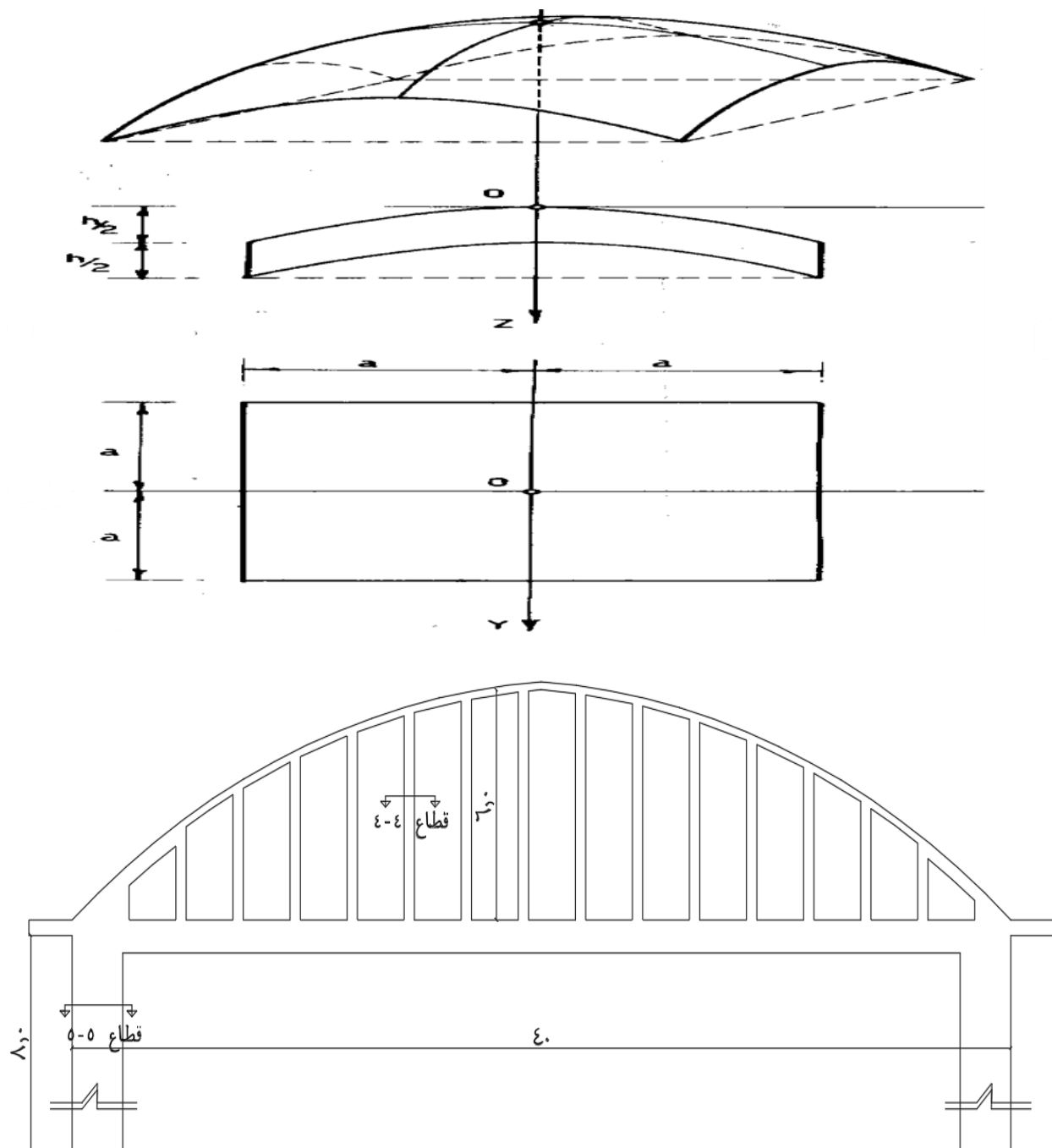


Figure 3.14 Raft RFT details

# Unit ( 4 ) Reinforced Concret Hall parabolic shell



## **4.1 INTRODUCTION**

### **4.1.1 Material Properties Used:**

- $F_{cu}=250 \text{ kg/cm}^2$  ,  $F_c = 60 \text{ kg/cm}^2$
- $F_y(\text{main steel})=3600 \text{ kg/cm}^2$  ,  $F_s = 2000 \text{ kg/cm}^2$
- $F_y(\text{stirrups})=2400 \text{ kg/cm}^2$
- Weight of used brick =  $1400 \text{ kg/m}^3$
- Bearing Capacity of Soil =  $1.0 \text{ kg/m}^2$
- passion ratio=0.2
- $E_c=22000 \text{ mpa}$

### **4.1.2 Cover Thickness**

- Columns Cover = 2.5 cm
- Foundations Cover = 5 cm
- Posts Cover = 2.5 cm
- Hanger Cover = 2.5 cm

### **4.1.3 Loads Used:**

- $L.L= 0.1 \text{ t/m}^2$
- Cover =  $0.05 \text{ t/m}^2$
- $D.L = \text{Own weight} + \text{Covering Material}$

#### 4.2 Concrete dimensions: -

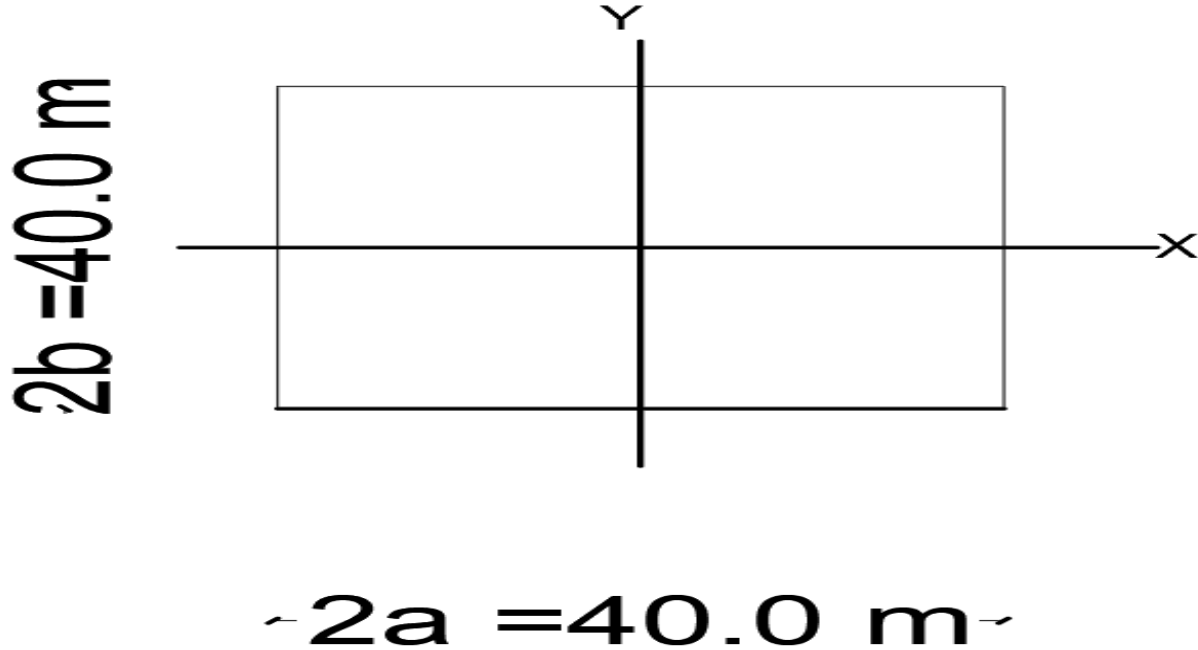


Figure 4.1 Concrete dimensions

$$a = 20 \text{ m}$$

$$L = 2a = 40 \text{ m}$$

$$b = 20 \text{ m}$$

$$L = 2b = 40 \text{ m}$$

$$t_s = 10 \text{ cm}$$

$$h_x = h_y = \frac{Lsh}{6-8} = \frac{40}{7} = 5.7 \approx 6 \text{ m}$$

$$\text{Total Load Surface ( P )} = D.L + L.L$$

$$= \text{Own weight} + \text{Covering Material} + L.L$$

$$= ( 2.5 * 0.1 + 0.05 ) + 0.1 = 0.4 \text{ t / m}^2 = 400 \text{ Kg /m}^2$$

$$r = a = \frac{a^2 + y^2}{2y} = \frac{20^2 + 3^2}{2*3} = 68.2 \text{ m}$$

The constant factor for determining the force is

$$\frac{4Pr}{\pi^3} = \frac{4 \cdot 0.40 \cdot 68.2}{\pi^3} = 3.52 \text{ t/m}^2$$

the maximum compressive Force at the middle of the shell

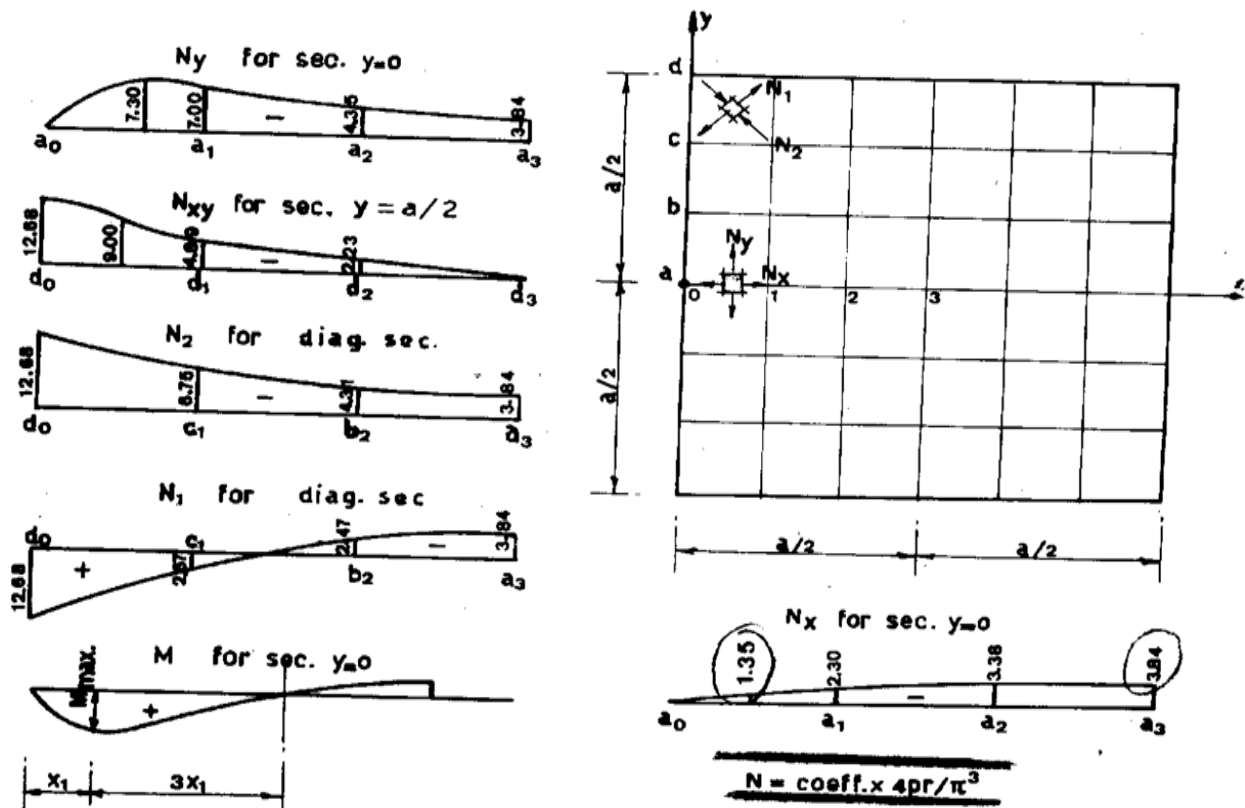
$$N_x = N_y = -3.84 \cdot 3.52 = -13.516 \text{ t/m}^2$$

the maximum compressive Force at the zone adjacent of the edge

$$N_y = -7.30 \cdot 3.52 = -25.696 \text{ t/m}^2$$

the maximum principal compressive and principal tensile force and the share Force at the corner of the shell

$$N_2 = N_1 = N_{xy} = 12.68 \cdot 3.52 = 44.63 \text{ t/m}^2$$



Internal Forces in Shallow Convex Shells with Square Plan

Figure 4.2 Internal forces in shallow convex shells with square plan

### 4.3 Check of Stresses

- **Maximum compressive stress at corner**

$$F_c = \frac{N_x}{bt} = \frac{13.516 \times 1000}{10 \times 100} = 13.516 \text{ kg/cm}^2 < F_{c \text{ all}} = 60 \text{ kg/cm}^2, \text{ Ok}$$

- **Maximum compressive stress at center**

$$F_c = \frac{N_{x0}}{bt} = \frac{25.696 \times 1000}{10 \times 100} = 25.696 \text{ kg/cm}^2 < F_{c \text{ all}} = 60 \text{ kg/cm}^2, \text{ Ok}$$

- **Principle compressive stress at corner**

$$F_c = \frac{N_{xy}}{bt} = \frac{44.63 \times 1000}{10 \times 100} = 44.63 \text{ kg/cm}^2 < F_{c \text{ all}} = 60 \text{ kg/cm}^2, \text{ Ok}$$

### 4.4 Moment :-

$$t_s = 10 \text{ cm}$$

$$X = 0.60 \sqrt{rt} = 0.60 (68.2 \times 0.10)^{0.50} = 1.57 \text{ m}$$

$$\therefore M = 0.094 \times r \times t \times p = 0.094 \times 68.2 \times 0.10 \times 0.40 = 0.256 = 0.3 \text{ t.m/m}$$

The reinforcement is generally one mesh 6  $\Phi$  10 in each direction

the compressive Force  $N_x$  in this section can be determined by linear interpolation thus

$$N_x = \text{Coeff} \times \frac{4Pr}{\pi^3} \times X \times 12 / 40 = -1.35 \times 3.52 \times 1.57 \times 12 / 40 = -2.238 \text{ t/m}$$

Due to the high principal compressive and tensile stresses at the corners, it is recommended to increase the thickness of the shell gradually from 10 to 16 cms along a length of about 0.15 measured along the diagonal

The maximum compressive stresses :-

$$\sigma_c = -16700 / 100 \times 10 = -16.7 \text{ Kg /cm}^2$$

$$\sigma_c = -31800 / 100 \times 10 = -31.8 \text{ Kg /cm}^2$$

$$\sigma_c = -55200 / 100 \times 16 = -34.8 \text{ Kg /cm}^2$$

The diagonal tension reinforcement on both Sides at the corner can be calculated as follows

$$A_{s1} = \frac{55+44}{2\sigma_s} = \frac{55+44}{2 \cdot 1.4} = 35 \text{ cm}^2 / \text{m}$$

$$A_{s2} = \frac{44+34}{2\sigma_s} = \frac{44+34}{2 \cdot 1.4} = 28 \text{ cm}^2 / \text{m}$$

$$A_{s3} = \frac{34+26}{2\sigma_s} = \frac{34+26}{2 \cdot 1.4} = 22 \text{ cm}^2 / \text{m}$$

$$A_{s4} = \frac{26+20}{2\sigma_s} = \frac{26+20}{2 \cdot 1.4} = 16 \text{ cm}^2 / \text{m}$$

$$A_{s5} = \frac{20+11.6}{2\sigma_s} = \frac{20+11.6}{2 \cdot 1.4} = 11.2 \text{ cm}^2 / \text{m}$$

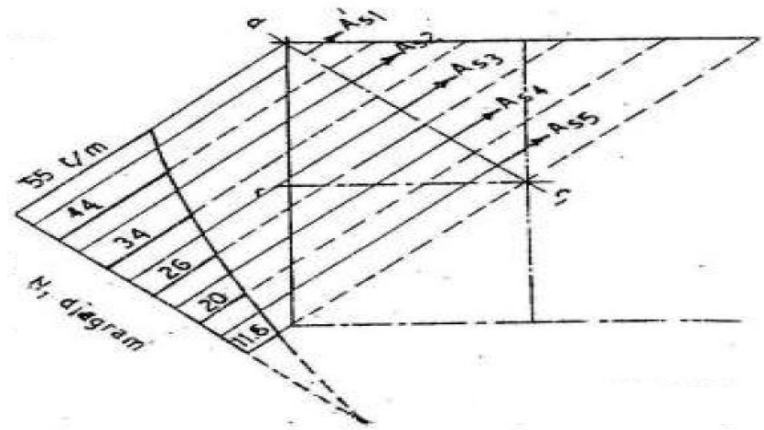


Figure 4.3 The diagonal tension reinforcement

## 4.5 Concrete Dimensions of System :

Parabolic Shell Thickness (ts) = 10 cm

1) Tie

- assume b = 40 cm
- t = 70 cm

2) Hanger

- assume b = 25 cm
- t = 25 cm

3) Main Girder

- assume b = 40 cm
- $t = \frac{\text{span}}{30-35} = \frac{400}{30-35} = 140 \text{ cm}$

### 4.5.1 Design of Girder

Weq = 1.4 o.w Girder +  $\sum v$  + W tie + W Hanger

- Own Weight Of Girder =  $1.4 \cdot 2.5 \cdot 0.4 \cdot 1.40 = 1.96 \text{ t/m}$
- Load From Hanger = W Hanger =  $1.4 (2.5 \cdot 0.25 \cdot 0.40 \cdot 6) = 2.1 \text{ t}$



$$W_{tie} = 1.4 ( 2.5 * 0.4 * 0.7 * 3.33 ) = 3.26 \text{ ton}$$

$$W_{tie} + W_{Hanger} = 3.26 + 2.1 = 5.36 \text{ ton}$$

Load From Slab

$$S = N_{xy} * L = 44.63 * 20.62 = 373.14 \text{ ton}$$

$$H = S \cos \theta = 373.14 * \frac{20}{20.62} = 361.92 \text{ ton}$$

$$V = S \sin \theta = 373.14 * \frac{5}{20.62} = 90.5 \text{ ton}$$

$$W_u = \frac{5.36 * 9}{40} + 1.96 + \frac{90.5}{40} = 7.69 \text{ t/m}$$

$$M_u = 0.05 * \frac{wL^2}{8} = 0.05 * \frac{7.69 * 40^2}{8} = 76.91 \text{ t.m}$$

$$N_u = 0.95 * \frac{wL^2}{8F} = 243.516 \text{ t ( comp )}$$

$$e = \frac{M_u}{N_u} = \frac{76.91}{238.45} = 0.315 < \frac{t}{2} \quad \text{small eccentricity}$$

$$\frac{N_u}{F_{cu} * b * t} = \frac{243.516 * 10^3}{250 * 40 * 140} = 0.17$$

$$\frac{M_u}{F_{cu} * b * t^2} = \frac{76.91 * 10^5}{250 * 40 * 140^2} = 0.039$$

**From chart**

$$\rho = 1$$

$$\mu = \rho * F_{cu} * 10^{-4} = 1 * 25 * 10^{-4} = 25 * 10^{-4}$$

$$A_s = A_s' = \mu * b * t = 25 * 10^{-4} * 400 * 1400 = 1400 \text{ mm}^2$$

$$A_{s\text{total}} = 2800 \text{ mm}^2$$

$$A_{s\text{min}} = \frac{0.8}{100} * b * t$$

$$= \frac{0.8}{100} * 400 * 1400 = 4480 \text{ mm}^2$$

$$A_s = A_s' = \frac{4480}{2} = 2240 \text{ mm}^2$$

$$\text{use } A_{s\text{min}} = 2240 \text{ mm}^2$$

$$\text{use } 24 \text{ } \phi 12$$

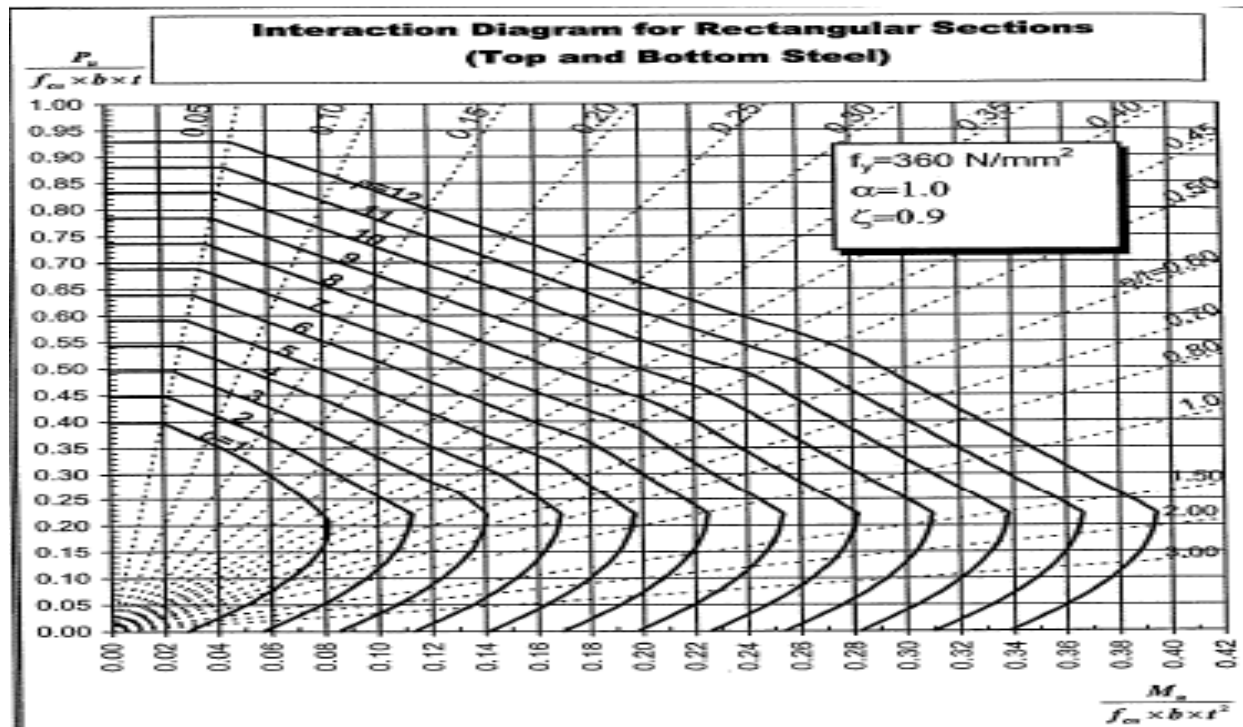


Figure 4.4 Interaction Diagram

#### 4.5.2 Design of Tie

$$P = R_x = \frac{0.95 WL^2}{8h} = 243.51 \text{ ton}$$

$$A_s = \frac{P}{\frac{f_y}{\gamma_s}} = \frac{243.51 \times 10^4}{\frac{3600}{1.15}} = 777.9 \text{ mm}^2 = 7.77 \text{ cm}^2$$

use 4  $\phi$  18

#### 4.5.3 Design of Hanger :-

Load From Hanger = 1.4 (O.W Of Hanger + O.W Of Tie)

$$= 1.4 (2.5 * 0.4 * 0.25 * 5 + 2.5 * 0.4 * 0.8 * 4) = 2.198 \text{ ton}$$

$$A_s = \frac{2.198 \times 10^4}{\frac{3600}{1.15}} = 7.021 \text{ mm}^2$$

use 4  $\phi$  16

#### 4.5.4 Design of column (40\*110): -

$$P = \frac{W_{eq} * L}{2} = \frac{7.69 * 40}{2} = 153.8 \text{ t}$$

$$P_{u \text{ actual}} = 0.35 (A_c) F_{cu} + 0.67 A_s F_y$$

$$\text{assume } A_s = 0.01 A_c$$

$$154 * 10^4 = 0.35 * 0.72 * 25 + 0.67 * 360 * A_s$$

$$A_s = 63.82 \text{ cm}^2$$

USE 24  $\phi$  20

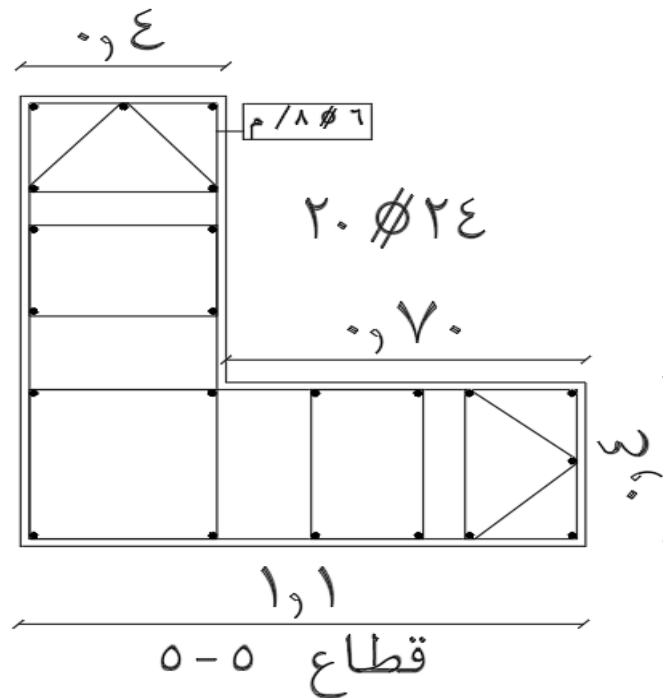
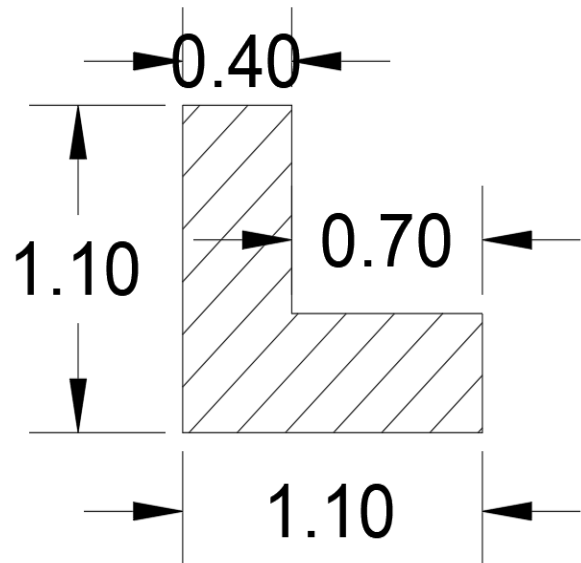


Figure 4.5 RFT details

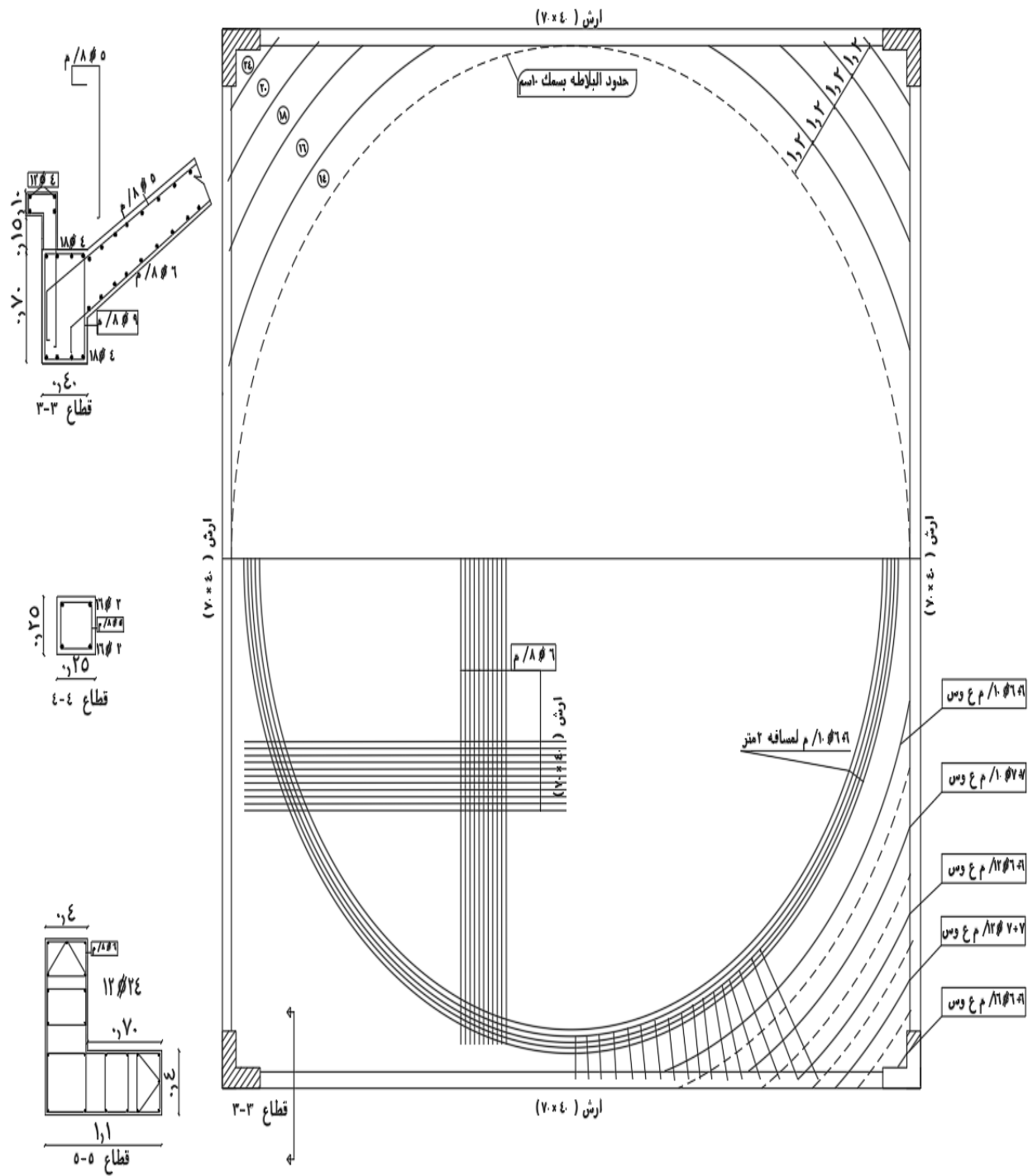


Figure 4.6 RFT details

#### 4.5.5 Design of Foundation : -

Column Working Load = 153.7 ton

Column Dimension a = 40 cm

b = 110 cm

Plain Concrete Depth = 0.4 m

Plain Concrete Extension = 0.40 m

##### 1 Concrete Dimension :

- $B.C = \frac{153.7}{A_{pc}}$
- $10 = \frac{106.7}{A_{pc}}$
- $A_{pc} = 15.31 \text{ m}^2 = L_{pc} * B_{pc}$   
 $= (1.10 + 2c) * (0.40 + 2c)$

C = 1.58 m

- Use take  $L_{pc} = 4.15 \text{ m}$
- $B_{pc} = 4.15 \text{ m}$

∴  $L_{RC} = 3.75 \text{ m}$

$B_{RC} = 3.75 \text{ m}$

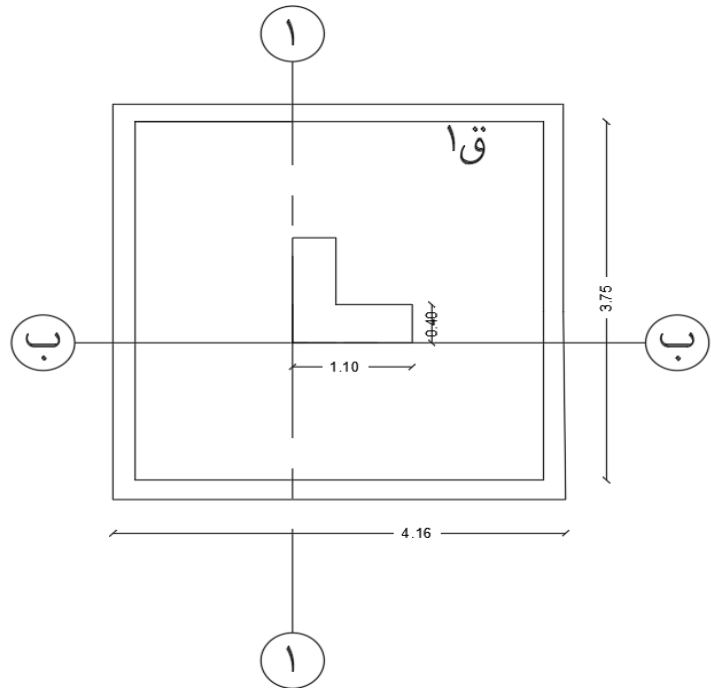


Figure 4.7 Section of footing (elevation)

##### 2 Check of Stress:

- $q_{act} = \frac{153.7}{A_{pc}}$
- $q_{act} = \frac{153.7}{4.15 * 4.15} = 8.92 \text{ t/m}^2 < B.C = 10 \text{ t/m}^2, \text{ Ok}$

$$q_{act} = \frac{P_U}{A_{RC}} = \frac{153.7}{3.75 * 3.75} = 109.3 \text{ KN}$$

$$Z = \frac{3.75 - 1.10}{2} = 0.63$$

$$M_U = \frac{q_{act} * Z^2}{2} = \frac{109.3 * 1^2}{2} = 54.65 \text{ KN.m}$$

$$d = c1 \sqrt{\frac{M_U}{F_{CU} * b}} = C1 \sqrt{\frac{54.65 * 10^6}{35 * 1000}} \Rightarrow d \approx 500 \Rightarrow t = 600$$

$$Q_p = P_{co} - q_{act} * (a + d) * (b + d)$$

- $Q_p = 1537 - 109.3 * (0.4 + 0.6) * (1.10 + 0.6) = 1351.19 \text{ KN}$

- $q_p = \frac{Q_p}{2((a+d)+(a+d))*d}$
- $q_p = \frac{1351.19 * 10^3}{2((400 + 600) + (1100 + 600)) * 600} = 0.417 \text{ N/mm}^2$

$$q_{p(allow)} = 1 -$$

$$1.7$$

$$2- 0.316 * (0.5 + \frac{a}{b}) * \sqrt{\frac{F_{CU}}{\gamma_c}} = 0.316 * (0.5 + \frac{0.4}{1.10}) * \sqrt{\frac{25}{1.5}} = 1.11 \text{ N/mm}^2$$

$$3- 0.316 * \sqrt{\frac{F_{CU}}{\gamma_c}} = 0.316 * \sqrt{\frac{25}{1.5}} = 1.29 \text{ N/mm}^2$$

$$4- 0.8 * (\frac{\alpha * d}{2(a+d+b+d)} + 0.2) * \sqrt{\frac{F_{CU}}{\gamma_c}} =$$

$$= 0.8 * (\frac{4 * 600}{2(400 + 600 + 1100 + 600)} + 0.2) * \sqrt{\frac{25}{1.5}} = 2.1 \text{ N/mm}^2$$

$$q_p < q_{p(allow)} \Rightarrow \text{OK, safe}$$

### 3 Check of Shear

$$Q_{sh1} = q_{act} * c = 109.3 * 1.58 = 172.69 \text{ ton}$$

$$q_{sh} = \frac{Q_{sh}}{B * d} = \frac{172.69 * 10^3}{3750 * 600} = 0.1 \text{ KN / mm}^2$$

$$q_{sh(allow)} = 0.16 * \sqrt{\frac{F_{CU}}{\gamma_c}} = 0.16 * \sqrt{\frac{25}{1.5}} = 0.65 \text{ KN / mm}^2$$

$$q_{sh} < q_{sh(allow)} \text{ OK .Safe}$$

$$A_s = \frac{M_U}{F_Y * J * d} = \frac{54.65 * 10^6}{360 * 0.826 * 600} = 306.30 \text{ mm}^2$$

$$A_s = 9 \phi 12 / \text{m}$$

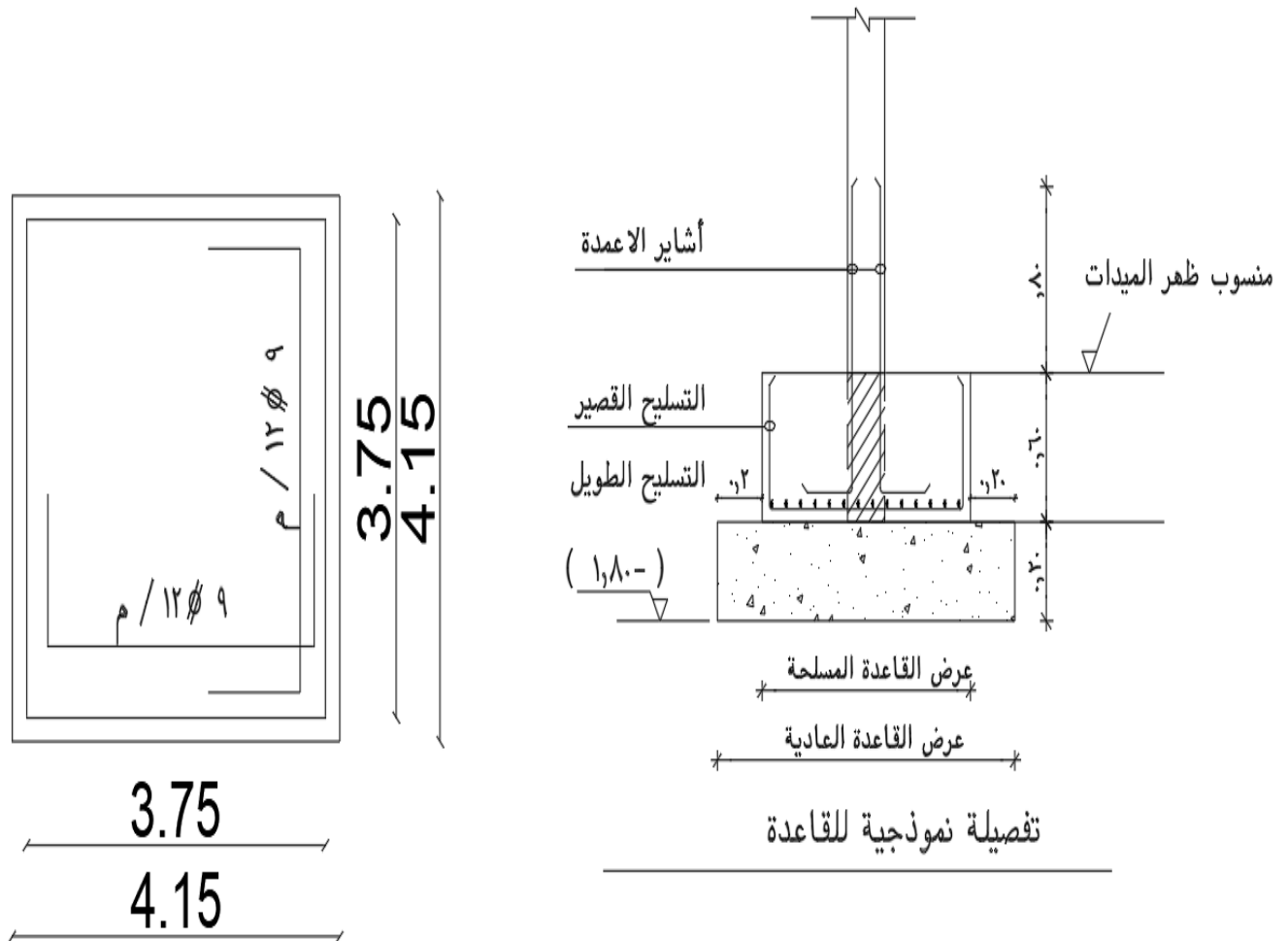
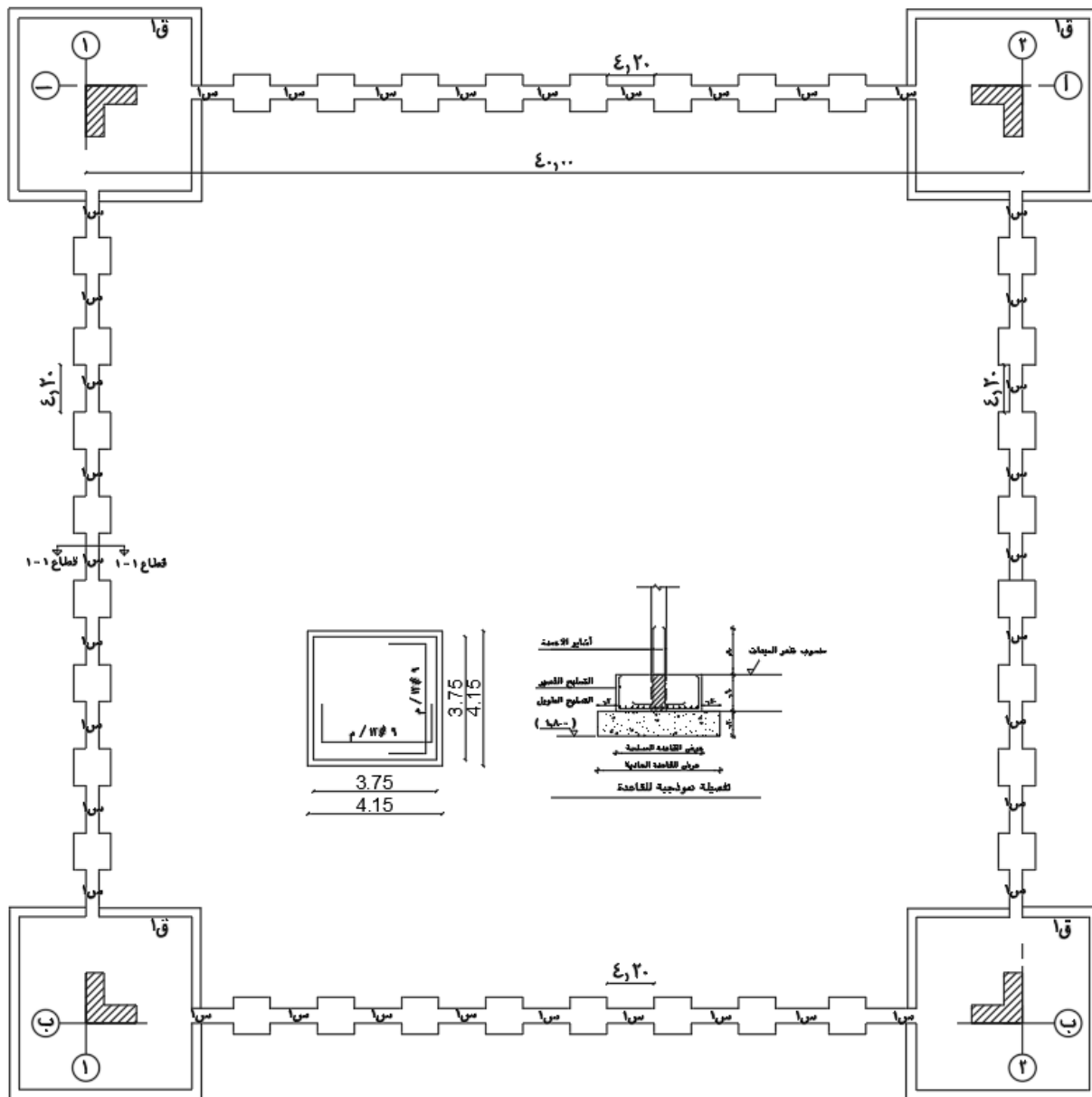


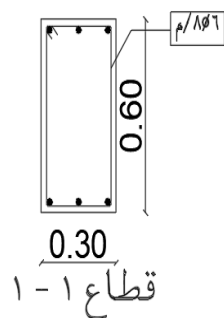
Figure 4.8 RFT details

Figure 4.9 RFT details foundation





## جدول القواعد على اجهاد لا يقل عن ١,٠٠ كجم/سم<sup>٢</sup>



| النموذج | الابعاد |                 | التسليح السفلي |              | التسليح العلوي |      |
|---------|---------|-----------------|----------------|--------------|----------------|------|
|         | عادية   | مسلحة           | قصير           | طويل         | قصير           | طويل |
| ق ١     | ٠,٢     | ٠,٦x٣,٧٥x٣,٧٥   | ٩ / ١٢ Ø ٩ م   | ٩ / ١٢ Ø ٩ م | ---            | ---  |
| ق ٢     | صفر     | ٠,٤ x ٠,٨ x ٠,٨ | ٥ / ١٢ Ø ٥ م   | ٥ / ١٢ Ø ٥ م | ---            | ---  |

| نموذج | ابعاد (م) |        | التسليح السفلي |      | التسليح العلوي |      | كانات      |
|-------|-----------|--------|----------------|------|----------------|------|------------|
|       | عرض       | ارتفاع | عدل            | مكسح | عدل            | مكسح |            |
| س ١   | ٠,٣٠      | ٠,٦٠   | ١٦ Ø ٣         | —    | ١٦ Ø ٣         | --   | ١٠/٨ Ø ٦ م |

Figure 4.10 Table RFT

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